

TECHNICAL MEMORANDUM

X-344

BASIC PRESSURE MEASUREMENTS ON A 0.0667-SCALE MODEL
OF THE NORTH AMERICAN X-15 RESEARCH AIRPLANE
AT TRANSONIC SPEEDS

By Robert S. Osborne and Virginia C. Stafford

Langley Research Center
Langley Field, Va.

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SUMMARY

In order to determine local loadings and interference effects at transonic speeds, basic pressure measurements have been made on a 0.0667-scale model of an interim configuration of the North American X-15 research airplane in the Langley 8-foot transonic pressure tunnel at Mach numbers from 0.60 to 1.18, angles of attack from -20° to 20° , and angles of sideslip from -5° to 5° . The results are presented in tabular form without analysis.

INTRODUCTION

As part of the wind-tunnel program required for the development of the North American X-15 research airplane, a 0.0667-scale pressure model of an interim configuration of the X-15 has been tested in the Langley 8-foot transonic pressure tunnel in order to determine local pressures on the fuselage-fairing-canopy area and local pressures and interference effects on the horizontal and vertical tail surfaces at transonic speeds. Tests of this model at Mach numbers from 2.30 to 4.65 are reported in reference 1. Tests of force models of the configuration are reported in references 2 to 5.

The X-15 is a rocket-powered research airplane designed for hypersonic speeds at very high altitudes. It employs a 5-percent-thick low-aspect-ratio trapezoidal wing mounted in the midposition on a fuselage consisting of a body of revolution with large side fairings. The 45° sweptback horizontal tail is all-movable for pitch control and may be deflected differentially for roll control. The outboard panels of the

upper and lower vertical-tail surfaces may be deflected for directional control; the inboard panels are fixed and support the speed brakes.

The model was tested at Mach numbers from 0.60 to 1.18, angles of attack from -20° to 20° , and angles of sideslip from -5° to 5° . Local pressure coefficients were obtained for the model with the speed brakes open and closed and with vertical-tail deflections from -7.5° to 7.5° . The results are presented in tabular form without analysis.

SYMBOLS

b	exposed span
C_p	pressure coefficient, $\frac{p_l - p}{q}$
c	local chord measured parallel to fuselage center line
\bar{c}	mean aerodynamic chord
l	fuselage length
M	free-stream Mach number
p	free-stream static pressure
p_l	local static pressure
q	free-stream dynamic pressure
R	Reynolds number
x	distance from nose of fuselage or from leading edge of airfoil surface measured parallel to fuselage center line
y	spanwise distance measured from juncture of airfoil surface and fuselage or fuselage side fairings
α	angle of attack of fuselage center line
β	angle of sideslip
δ_e	deflection of horizontal tail, positive when trailing edge is down

δ_v deflection of upper and lower vertical tail surfaces, positive when trailing edge is to the left

ϕ angular location of fuselage orifices, measured counterclockwise from bottom of fuselage when facing forward

Orifice locations:

C canopy

F fuselage side fairings

J fuselage-canopy juncture

APPARATUS AND TESTS

Model

The 0.0667-scale pressure model of the North American X-15 research airplane used in this investigation was supplied by the contractor. It was constructed of stainless steel. Photographs of the model are presented in figure 1, and dimensional details are shown in figure 2 and table I.

The model closely approximated configuration 3 of the X-15. Distinguishing features of this configuration as compared to earlier configurations include an increased fuselage diameter, shortened fuselage side fairings, increased leading-edge radii on wing and horizontal tail, a larger vertical tail having 10° full-wedge airfoil sections and with the total exposed area distributed 55 percent above the fuselage and 45 percent below, and reduced speed-brake area. Unlike configuration 3, however, the wing of the configuration tested had not been moved forward 3.6 inches (full scale), the horizontal tail had not been moved rearward 5.4 inches (full scale), and there were no landing skids. The contractor's code designation of the model tested was B₂W₂X₄H₃VU₅JU₂VL₇JL₂.

The horizontal tail was fixed at a deflection angle of 0° for these tests, whereas the speed brakes and the outer panels of the upper and lower vertical tail surfaces could be deflected about hinge lines having 0° sweepback.

Tunnel and Model Support

The tests were conducted in the Langley 8-foot transonic pressure tunnel which is a single-return, rectangular, slotted-throat wind tunnel

having controls that allow for the independent variation of Mach number, stagnation pressure, temperature, and humidity.

The model was attached to a sting support which had a diameter approximately equal to that of the base of the fuselage. The sting support was cylindrical for 4.5 base diameters downstream of the model base and at its downstream end was attached to the larger tunnel support sting which could be rotated by means of an arc-shaped support strut to obtain changes in angle of attack. Variations in angle of sideslip were obtained by insertion of properly angled couplings in the model support system.

Measurements and Accuracy

Local static pressures were measured by using a total of 205 flush-mounted orifices distributed over the right half of the model. Measurements on the right wing were limited to one row of orifices on the upper and lower surfaces and on the right horizontal tail to two rows of orifices on the upper and lower surfaces. One orifice was located in the base of the upper vertical tail. Orifice locations are diagrammed and tabulated in figure 3. Pressure coefficients were determined from these measurements and are believed to be accurate within ± 0.005 .

The angle of attack was set to within $\pm 0.1^\circ$ and the angle of sideslip to within $\pm 0.2^\circ$ by means of a pendulum-type attitude indicator and calibrations of sting deflection with respect to model load. Model loads were determined from wind-tunnel tests of a nearly identical force model of the X-15 airplane. Control deflection inaccuracies are unknown but are believed to be small.

The average test Mach number was determined to within ± 0.003 from a calibration with respect to the pressure in the chamber surrounding the slotted test section.

Tests

The complete model was tested with the horizontal tail undeflected and the speed brakes closed for vertical-tail deflections of -7.5° , 0° , and 7.5° . The model was also tested with the horizontal and vertical tails undeflected and the speed brakes opened 35° .

The model was tested at Mach numbers from 0.60 to 1.18, angles of attack from -20° to 20° , and angles of sideslip of -5° , 0° , and 5° . The tests were conducted at a tunnel stagnation pressure of approximately 1 atmosphere, the average test Reynolds number based on the wing mean

aerodynamic chord varying from approximately 2.2×10^6 to 2.8×10^6 over the Mach number range. (See fig. 4.) For all tests, boundary-layer transition strips consisting of No. 120 carborundum grains were installed along the 10-percent-chord lines of the wing and tail surfaces and at 10 percent of the body length.

Corrections

Tunnel boundary interference at subsonic speeds is minimized by the slotted test section, and no corrections for this interference have been applied. With respect to supersonic boundary-reflected disturbances, the effects of reflected compression waves on the pressure distribution over the model are negligible at a Mach number of 1.03, and the effects of reflected expansion waves were minimized by testing the model a few inches off the tunnel center line. (See ref. 6.) At a Mach number of 1.18 the reflected disturbances passed well downstream of the base of the model.

SUMMARY OF RESULTS

The pressure coefficients for each surface of a 0.0667-scale model of the X-15 airplane are tabulated as a function of configuration, Mach number, angle of attack, and angle of sideslip and are presented in tables II to VI as follows:

Surface	Table
Fuselage	II
Right wing	III
Right horizontal tail	IV
Vertical tails and speed brakes	V
Base of upper vertical tail	VI

Langley Research Center,
National Aeronautics and Space Administration,
Langley Field, Va., July 13, 1960.

REFERENCES

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6. Wright, Ray H., Ritchie, Virgil S., and Pearson, Albin O.: Characteristics of the Langley 8-Foot Transonic Tunnel With Slotted Test Section. NACA Rep. 1389, 1958. (Supersedes NACA RM L51H10 by Wright and Ritchie and RM L51K14 by Ritchie and Pearson.)

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TABLE I.- DIMENSIONS OF THE 0.0667-SCALE PRESSURE MODEL OF
THE NORTH AMERICAN X-15 RESEARCH AIRPLANE

Wing:

Airfoil section	Modified NACA 66-005
Total area, sq in.	128.150
Exposed area, sq in.	67.260
Total span, in.	17.898
Exposed span, in.	12.028
Total aspect ratio	2.50
Leading-edge sweepback, deg	36.75
Quarter-chord-line sweepback, deg	25.64
Trailing-edge sweepforward, deg	17.75
Root chord at center line, in.	11.932
Exposed root chord, in.	8.801
Tip chord, in.	2.386
Total taper ratio	0.20
Mean aerodynamic chord based on total area, in.	8.220
Longitudinal distance from fuselage nose to $\bar{c}/4$, in.	23.411
Dihedral, deg	0
Incidence, deg	0

Horizontal tail (in plane of surface):

Airfoil section	Modified NACA 66-005
Total area, sq in.	73.850
Exposed area, sq in.	32.710
Total span, in.	14.980
Exposed span, in.	9.068
Exposed aspect ratio	2.51
Leading-edge sweepback, deg	50.58
Quarter-chord-line sweepback, deg	45.00
Trailing-edge sweepback, deg	19.28
Root chord at center line, in.	8.175
Exposed root chord, in.	5.616
Tip chord, in.	1.686
Exposed taper ratio	0.30
Mean aerodynamic chord based on exposed area, in.	4.002
Hinge line, percent exposed \bar{c}	37
Longitudinal distance from total wing $\bar{c}/4$ to exposed tail $\bar{c}/4$, in.	11.450
Dihedral, deg	-15.00

Upper vertical tail (exposed panel):

Airfoil section	10° wedge
Area, sq in.	26.075
Span, in.	3.668
Aspect ratio	0.52
Leading-edge sweepback, deg	30
Trailing-edge sweepback, deg	0

TABLE I.- DIMENSIONS OF THE 0.0667-SCALE PRESSURE MODEL OF
THE NORTH AMERICAN X-15 RESEARCH AIRPLANE - Concluded

Root chord, in.	8.171
Tip chord, in.	6.053
Taper ratio	0.74
Mean aerodynamic chord, in.	7.153
Longitudinal distance from total wing $\bar{c}/4$ to exposed panel $\bar{c}/4$, in. . . .	9.630
Movable portion -	
Area, sq in.	16.945
Span, in.	2.501
Root chord, in.	7.495
Tip chord, in.	6.053
Hinge line, percent exposed panel \bar{c}	29
Speed brake (one side) -	
Area, sq in.	3.533
Chord, in.	2.560
Average span, in.	1.380
Lower vertical tail (exposed panel):	
Airfoil section	10° wedge
Area, sq in.	22.385
Span, in.	3.068
Aspect ratio	0.42
Leading-edge sweepback, deg	30
Trailing-edge sweepback, deg	0
Root chord, in.	8.171
Tip chord, in.	6.420
Taper ratio	0.79
Mean aerodynamic chord, in.	7.322
Longitudinal distance from total wing $\bar{c}/4$ to exposed panel $\bar{c}/4$, in. . . .	9.516
Movable portion -	
Area, sq in.	13.220
Span, in.	1.901
Root chord, in.	7.495
Tip chord, in.	6.420
Hinge line, percent exposed panel \bar{c}	30
Speed brake (one side) -	
Area, sq in.	3.533
Chord, in.	2.560
Average span, in.	1.380
Fuselage:	
Length, in.	39.353
Maximum depth, in.	3.733
Maximum width with side fairings, in.	5.867
Fineness ratio without side fairings	10.54
Base diameter, in.	3.193

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE

(a) $M = 0.60$ Pressure coefficients C_p on -

Model orifice number	$\frac{x}{l}$	β , deg	Fuselage								Model orifice number	$\frac{x}{l}$	β , deg	Fuselage - Concluded							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$											$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
1	.026	0				.152	.234	.327	.431	.544	55	.503	F				.004	-.058	-.307	-.678	-.687
2	.026	90				.116	.082	.006			56	.503	135				-.008	-.052	-.124	-.200	-.351
3	.026	180				.151	.077	.010			57	.588	0				-.038	.041	.123	.208	.306
4	.084	0				.028	.100	.188	.286	.396	58	.588	180				-.030	-.104	.212	-.267	-.357
5	.084	45				-.001	.026	-.005			59	.604	15				-.036	.033	.105	.182	.269
6	.084	90				.014	-.026	-.077			60	.604	30				-.030	.047	.121	.193	.272
7	.084	135				-.003	-.031	-.071			61	.604	50				-.035	.055	.145	.228	.319
8	.084	180				.041	-.022	-.074			62	.604	F				-.042	.068	.174	.271	.366
9	.134	J				.084	.036	.003			63	.604	F				-.038	-.140	-.214	-.263	-.325
10	.156	J				.003	-.030	-.043	-.036		64	.604	135				-.030	-.113	-.229	-.366	-.469
12	.173	30				-.077	-.061	-.050	-.027		65	.690	0				-.010	.038	.088	.140	.214
13	.173	60				-.083	-.097	-.143	-.212		66	.690	180				-.017	-.050	-.117	-.238	-.302
14	.173	90				-.084	-.099	-.157	-.247		67	.706	15				-.005	.031	.072	.113	.172
15	.173	135				-.084	-.086	-.098	-.125		68	.706	30				.001	.036	.071	.108	.158
16	.173	J				-.057	-.073	-.071	-.058		69	.706	50				.006	.046	.084	.122	.171
17	.173	C				-.026	-.063	-.083	-.084		70	.706	F				.014	.053	.092	.128	.172
18	.190	J				-.109	-.106	-.096	-.078		71	.706	F				-.023	-.051	-.099	-.167	-.230
19	.190	C				-.123	-.148	-.162	-.164		72	.706	135				-.012	-.046	-.097	-.203	-.251
20	.197	C				-.317	-.369	-.403	-.409		73	.753	0				.080	.110	.146	.183	.236
21	.207	C				-.234	-.251	-.267	-.271	-.221	74	.753	180				.067	.039	.010	-.097	-.172
23	.216	C				-.254	-.278	-.271	-.245	-.219	75	.770	15				-.086	-.062	-.029	.017	.060
24	.226	23				-.088	-.076	-.058	-.022	.023	76	.770	30				-.031	-.011	.015	.048	.073
25	.226	45				-.084	-.089	-.115	-.147	-.191	77	.770	50				-.008	.019	.046	.078	.106
26	.226	68				-.091	-.114	-.175	-.267	-.396	78	.770	F				.017	.039	.063	.081	.087
27	.226	90				-.092	-.112	-.175	-.279	-.426	79	.770	F				.046	.048	.038	.018	-.024
28	.226	135				-.167	-.183	-.205	-.243	-.322	80	.770	F				-.006	-.020	-.042	-.086	-.146
29	.226	C				-.251	-.254	-.258	-.263	-.273	81	.770	135				-.040	-.048	-.060	-.094	-.136
30	.263	C				-.108	-.124	-.118	-.094	-.102	82	.820	15				-.063	-.033	.003	.053	.105
31	.283	0				-.094	-.062	-.009	.063	.153	83	.820	38				-.049	-.014	.022	.067	.110
32	.299	30				-.075	-.075	-.074	-.063	-.043	84	.820	F				-.040	-.004	.040	.092	.153
33	.299	60				-.060	-.033	-.026	-.030	-.047	85	.820	F				-.027	.012	.052	.102	.154
34	.299	F				-.066	-.036	-.048	-.104	-.204	86	.820	F								
35	.299	F				-.044	-.083	-.203	-.398	-.691	87	.820	F				.021	.060	.109	.162	.188
36	.299	F				-.069	-.143	-.262	-.422	-.625	88	.820	F				-.045	-.077	-.127	-.201	-.221
37	.299	135				-.106	-.077	-.095	-.150	-.216	89	.820	F				-.042	-.068	-.113	-.164	-.189
38	.319	C				-.026	-.055	-.092	-.118	-.181	90	.820	F				-.050	-.073	-.104	-.140	-.189
39	.350	20				-.042	-.030	-.013	.021	.073	91	.820	F				-.059	-.080	-.106	-.140	-.187
40	.350	40				-.031	-.024	-.024	-.020	-.016	92	.820	142				-.013	-.100			
41	.350	60				-.031	-.005	.023	.058	.096	94	.897	30				-.120	-.092	-.057	-.005	.055
42	.350	F				-.035	-.014	.023	.056	.106	95	.897	50				-.124	-.097	-.057	.002	.071
43	.350	F				-.036	-.088	-.245	-.502	-.889	96	.897	F				-.124	-.091	-.048	.016	.086
44	.350	F				-.031	-.075	-.150	-.287	-.486	97	.897	F				-.133	-.159	-.190	-.236	-.296
45	.350	135				-.009	-.026	-.063	-.104	-.126	98	.897	130				-.126	-.141	-.167	-.209	-.281
46	.367	0				-.045	-.014	.033	.103	.192	99	.897	150				-.132	-.146	-.157	-.198	-.274
47	.367	180				-.018	-.044	-.060	-.085	-.135	102	.984	0				-.259	-.241	-.241	-.223	-.193
48	.486	0				-.028	.020	.085	.165	.259	104	.984	60				-.190	-.179	-.186	-.184	-.189
49	.486	180				-.027	-.069	-.142	-.178	-.175	105	.984	90				-.184	-.176	-.183	-.188	-.213
50	.503	15				-.022	.022	.073	.138	.219	106	.984	135				-.164	-.161	-.150	-.153	-.166
51	.503	30				-.015	.020	.058	.101	.159	107	.984	180				-.309	-.317	-.297	-.344	-.394
52	.503	50				-.010	.032	.078	.130	.188											
53	.503	F				.008	.033	.053	.074	.098											
54	.503	F				.009	-.085	-.316	-.578	-.953											

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(a) M = 0.60 - Continued

Pressure coefficients C_p on -

Model orifice number	$\frac{x}{l}$	β , deg	Fuselage								Model orifice number	$\frac{x}{l}$	β , deg	Fuselage - Concluded										
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$			
			$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$											$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$										
1	.026	0	-.069	.010	.110	.163	.243	.335	.441	.553	55	.503	F	.091	.048	.030	.012	-.059	-.317	-.684	-.686			
2	.026	90	-.309	.002	.106	.125	.086	.007	-.119	-.309	56	.503	135	.148	.064	.023	.000	-.047	-.118	-.198	-.361			
3	.026	180	.554	.345	.208	.158	.078	.012	-.064	.57	.588	0	0	-.216	-.064	-.019	.059	.138	.221	.320	.320			
4	.084	0	-.131	-.077	-.005	.037	.109	.195	.295	.406	58	.588	180	.291	.123	.024	-.019	-.096	-.203	-.256	-.352			
5	.045	84	-.195	-.054	.005	.008	.030	-.002	-.017	-.059	59	.604	15	-.365	-.194	-.060	-.018	.053	.124	.198	.283			
6	.084	90	-.398	-.085	-.010	.024	-.025	-.083	-.203	-.391	60	.604	30	-.456	-.175	-.058	-.011	.064	.136	.206	.285			
7	.084	135	-.079	-.010	.007	.004	-.027	-.070	-.122	-.199	61	.604	50	-.769	-.212	-.065	-.015	.076	.162	.245	.330			
8	.084	180	.410	.211	.091	.046	-.021	-.074	-.081	-.122	62	.604	F	-1.156	-.236	-.093	-.020	.088	.192	.286	.379			
9	.134	J	.393	.221	.120	.090	.036	.009	-.016	-.050	63	.604	F	.354	.176	.042	-.026	-.139	-.202	-.265	-.361			
10	.156	J	.219	.086	.027	.011	-.022	-.037	-.032	.004	64	.604	135	.285	.136	.029	-.017	-.108	-.218	-.363	-.705			
12	.173	30	-.170	-.102	-.079	-.065	-.054	-.042	-.019	.004	65	.690	0	-.307	-.109	-.007	.018	.065	.112	.162	.235			
13	.173	60	-.280	-.137	-.082	-.073	-.093	-.143	-.213	-.321	66	.690	180	.180	.071	.030	.006	-.033	-.098	-.224	-.296			
14	.173	90	-.449	-.179	-.088	-.073	.096	-.156	-.255	-.407	67	.706	15	-.250	-.096	.005	.024	.060	.096	.136	.193			
15	.173	135	-.220	-.130	-.084	-.073	.082	-.094	-.122	-.166	68	.706	30	-.195	-.059	.008	.030	.062	.097	.131	.180			
16	.173	J			-.043	-.050	-.068	-.060	-.053	-.040	69	.706	50	-.163	-.060	.016	.035	.069	.108	.145	.193			
17	.173	J	.174	.046	-.008	-.018	-.061	-.082	-.082	-.096	70	.706	F	-.208	-.059	.022	.042	.076	.117	.150	.192			
18	.190	J	-.040	-.102	-.105	-.104	-.101	-.089	-.074	-.126	71	.706	F	.122	.049	.028	.002	-.051	-.079	-.157	-.311			
19	.190	J	.054	-.047	-.097	-.112	-.147	-.157	-.163	-.145	72	.706	135	.122	.047	.025	.013	-.025	-.077	-.191	-.243			
20	.197	C	.069	-.154	-.267	-.307	-.367	-.396	-.408	-.402	73	.753	0	-.172	.018	.097	.114	.143	.179	.212	.264			
21	.207	C	-.072	-.161	-.208	-.224	-.249	-.265	-.265	-.221	74	.753	180	.183	.117	.113	.097	.067	.044	-.080	-.162			
23	.216	C	.029	-.140	-.225	-.247	-.273	-.263	-.241	-.216	75	.770	15	-.006	-.062	-.038	-.034	-.018	.014	.063	.095			
24	.226	23	-.137	-.094	-.089	-.080	-.070	-.050	-.012	.029	76	.770	30	-.054	-.046	.003	.017	.030	.056	.086	.109			
25	.226	45	-.192	-.097	-.074	-.072	-.087	-.113	-.148	-.190	77	.770	50	-.058	-.027	.023	.037	.060	.089	.116	.139			
26	.226	68	-.289	-.138	-.085	-.083	-.110	-.173	-.267	-.402	78	.770	F	-.014	-.001	.057	.060	.077	.101	.115	.116			
27	.226	90	-.428	-.172	-.092	-.086	-.108	-.175	-.283	-.436	79	.770	F	-.127	.046	.083	.083	.080	.073	.047	-.005			
28	.226	135	-.141	-.140	-.156	-.157	-.179	-.202	-.245	-.326	80	.770	F	.006	.011	.042	.032	.017	-.005	-.057	-.137			
29	.226	C	-.175	-.215	-.236	-.241	-.249	-.253	-.259	-.270	81	.770	135	.009	-.020	.012	.001	-.010	-.022	-.063	-.118			
30	.263	C	.075	-.049	-.090	-.100	-.119	-.109	-.088	-.098	82	.820	15	-.086	-.089	.015	.025	.046	.081	.127	.171			
31	.283	C	-.212	-.111	-.099	-.084	-.051	-.001	.072	.161	83	.820	38	-.081	-.071	.017	.038	.065	.100	.138	.174			
32	.299	30	-.272	-.084	-.074	-.069	-.071	-.067	-.057	-.042	84	.820	F	-.060	-.048	.029	.045	.076	.117	.164	.215			
33	.299	60	-.232	-.138	-.074	-.052	-.029	-.022	-.027	-.048	85	.820	F	-.050	-.047	.040	.054	.083	.123	.168	.210			
34	.299	F	-.482	-.235	-.094	-.058	-.027	-.044	-.108	-.211	86	.820	F	-.019	-.015									
35	.299	F	-.657	-.197	-.052	-.037	-.084	-.207	-.415	-.711	87	.820	F	-.002	-.004	.067	.092	.128	.174	.222	.233			
36	.299	F	-.241	-.050	-.035	-.062	-.148	-.272	-.441	-.634	88	.820	F	-.009	-.020	.024	.010	-.036	-.072	-.151	-.185			
37	.299	135			-.028	-.041	-.065	-.105	-.165	-.233	89	.820	F	-.039	-.019	.028	.016	-.021	-.050	-.108	-.152			
38	.319	C	.140	.021	-.011	-.020	-.051	-.087	-.117	-.183	90	.820	F	-.029	-.030	.027	.014	-.019	-.036	-.076	-.149			
39	.350	20	-.064	-.039	-.036	-.032	-.023	-.005	.029	.080	91	.820	F	-.018	-.043	.016	.011	-.021	-.029	-.071	-.141			
40	.350	40	-.162	-.063	-.031	-.021	.016	-.016	-.015	-.011	92	.820	142	-.035	-.064	.004	-.006	-.027	-.030	-.054	-.121			
41	.350	60	-.124	-.094	-.038	-.022	.004	.032	.065	.105	93	.820	165	-.011	-.074									
42	.350	F	-.431	-.141	-.053	-.025	-.006	-.018	-.053	-.105	94	.897	30	-.174	-.121	.188	.231	.271	.301	.331	.359			
43	.350	F	-.823	-.244	-.046	-.028	-.090	-.256	-.533	-.916	95	.897	50	-.184	-.139	.149	.153	.180	.209	.246	.285			
44	.350	F	-.099	-.015	-.007	-.025	-.073	-.147	-.295	-.495	96	.897	F	-.180	-.147	.078	.094	.113	.147	.186	.233			
45	.350	135	.004	.009	.010	-.002	-.022	-.060	-.098	-.126	97	.897	F	-.065	-.113	.038	.024	-.025	-.007	-.046	-.122			
46	.367	0	-.172	-.075	-.048	-.037	-.005	.042	.113	.201	98	.897	130	-.070	-.113	.112	.129	.075	.131	.100	.004			
47	.367	180	.215	.068	.007	-.012	-.042	-.053	-.079	-.137	99	.897	150	-.072	-.118	.225	.206	.187	.272	.244	.164			
48	.486	0	-.172	-.129	-.038	-.014	.034	.098	.179	.275	102	.984	0	-.350	-.242	-.422	-.421	-.410	-.390	-.339	-.311			
49	.486	180	.255	.088	.008	-.018	-.060	-.136	-.172	-.168	103	.984	30	-.255	-.193									
50	.503	15	-.211	-.101	-.033	-.008	.034	.086	.152	.231	104	.984	60	-.190	-.158	-.343	-.348	-.357	-.321	-.326	-.344			
51	.503	30	-.378	-.087	-.026	-.003	.032	.069	.113	.170	105	.984	90	-.195	-.168	-.405	-.297	-.323	-.302	-.324	-.310			
52	.503	50	-.473	-.109	-.023	.003	.044	.090	.141	.200	106	.984	135	-.154	-.150	-.456	-.449	-.439	-.421	-.488	-.535			
53	.503	F	-.545	-.325	-.012	.020	.044	.064	.081	.106	107	.984	180	-.254	-.305	-.453	-.451	-.450	-.465	-.525	-.551			
54	.503	F	-.743	-.291	-.010	.020	-.095	-.333	-.607	-.995														

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(a) $M = 0.60$ - ContinuedPressure coefficients C_p on -

Model orifice number	$\frac{x}{t}$	β , deg	Fuselage								Model orifice number	$\frac{x}{t}$	β , deg	Fuselage - Concluded							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 3^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 3^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
1	.026	0	-.049	.043		.175	.262	.354	.461	.569	54	.503	F	-.810	-.307		.016	-.093	-.348	-.610	-.876
2	.026	90	-.213	.084		.191	.158	.085	-.045	-.232	55	.503	F	.164	.091		.003	-.073	-.248	-.850	-.752
3	.026	180	.569	.358		.178	.113	.045	-.006	-.051	56	.503	135	.231	.110		-.004	-.060	-.117	-.171	-.259
4	.084	0	-.089	-.041		.049	.125	.227	.326	.431	57	.588	0	-.377	-.175		-.014	.065	.147	.232	.332
5	.084	45	-.307	-.080		.054	.095	.114	.135	.126	58	.588	180	.324	.148		-.014	-.088	-.173	-.259	-.390
6	.084	90	-.331	-.045		.063	.033	-.034	-.149	-.312	59	.604	15	-.412	-.194		-.018	.062	.141	.221	.317
7	.084	135	.099	.103		.053	-.006	-.069	-.163	-.272	60	.604	30	-.427	-.196		-.012	.070	.157	.227	.325
8	.084	180	.436	.238		.065	.012	-.033	-.067	-.088	61	.604	50	-.494	-.223		-.024	.071	.162	.247	.345
9	.134	J	.413	.235		.105	.052	.027	.001	-.014	62	.604	F	-1.133	-.271		-.038	.078	.181	.278	.382
10	.156	J	.351	.212		.132	.086	.063	.038	.011	63	.604	F	.373	.174		-.040	-.161	-.254	-.301	-1.201
12	.173	30	-.189	-.116		-.049	-.007	.034	.083	.125	64	.604	135	.322	.153		-.024	-.114	-.218	-.276	-.513
13	.173	60	-.362	-.151		-.039	-.030	-.050	-.097	-.173	65	.690	0	-.189	-.073		.013	.060	.107	.156	.232
14	.173	90	-.417	-.141		-.035	-.055	-.109	-.214	-.359	66	.690	180	.203	.095		.003	-.033	-.089	-.191	-.198
15	.173	135	-.055	-.013		-.013	-.038	-.068	-.107	-.147	67	.706	15	-.301	-.062		.024	.065	.103	.144	.211
16	.173	J	.213	.106		.051	.023	.012	-.006	-.035	68	.706	30	-.206	-.049		.031	.071	.109	.143	.204
17	.173	C	.311	.169		.075	.042	.021	.006	.003	69	.706	50	-.294	-.050		.036	.073	.106	.136	.192
18	.190	J	.073	.002		-.024	-.040	-.045	-.071	-.114	70	.706	F	-.373	-.047		.045	.081	.109	.127	.183
19	.190	C	.161	.027		-.050	-.085	-.107	-.115	-.122	71	.706	F	.126	.066		.007	-.025	-.094	-.259	-.415
20	.190	C	.102	-.134		-.304	-.360	-.392	-.416	-.409	72	.706	135	.149	.076		.011	-.022	-.084	-.230	-.323
21	.207	C	.015	-.111		-.187	-.222	-.246	-.252	-.250	73	.753	0	-.007	.055		.094	.125	.158	.189	.245
22	.216	C	.061	-.126		-.236	-.259	-.261	-.247	-.234	74	.753	180	.197	.137		.087	.071	.033	-.044	-.041
24	.226	23	-.184	-.109		-.060	-.025	.019	.073	.135	75	.770	15	-.145	.067		.119	.150	.172	.193	.237
25	.226	45	-.215	-.121		-.051	-.031	-.030	-.033	-.049	76	.770	30	-.197	.030		.087	.113	.134	.152	.189
26	.226	68	-.386	-.163		-.065	-.069	-.108	-.177	-.279	77	.770	50	-.153	.025		.071	.091	.111	.122	.157
27	.226	90	-.425	-.160		-.069	-.088	-.152	-.254	-.398	78	.770	F	-.149	.035		.072	.089	.103	.094	.109
28	.226	135	-.056	-.098		-.154	-.194	-.230	-.271	-.324	79	.770	F	-.825	-.023		.100	.095	.079	-.028	-.086
29	.226	C	-.060	-.137		-.241	-.278	-.305	-.324	-.325	80	.770	F	.005	.060		.058	.042	.002	-.118	-.238
30	.263	C	.122	-.006		-.070	-.078	-.078	-.068	-.062	81	.770	135	.075	.072		.060	.039	.004	-.115	-.223
31	.283	0	-.072	-.090		-.070	-.037	.014	.086	.176	82	.820	15	-.091	.011		.040	.063	.087	.125	.174
32	.299	30	-.219	-.111		-.063	-.033	-.006	.032	.077	83	.820	38	-.186	.003		.038	.063	.093	.129	.175
33	.299	60	-.285	-.188		-.064	-.019	.009	.027	.033	84	.820	F	-.109	.009		.034	.063	.097	.149	.206
34	.299	F	-.669	-.297		-.071	-.017	-.013	-.051	-.122	85	.820	F	-.121	.000		.027	.059	.097	.146	.197
35	.299	F	-.659	-.187		-.026	-.066	-.185	-.382	-.659	86	.820	F	-.114	.017		.042	.080	.119	.168	.209
36	.299	F	-.163	-.021		-.074	-.178	-.335	-.540	-.772	87	.820	F	-.111	.031		.054	.103	.153	.242	.288
37	.299	135	.037	.013		-.058	-.105	-.158	-.202	-.252	88	.820	F	.043	.024		.021	-.027	-.091	-.247	-.297
38	.319	C	.183	.055		-.010	-.028	-.042	-.052	-.056	89	.820	F	-.001	.026		.020	-.011	-.060	-.151	-.216
39	.350	20	-.273	-.060		-.022	.008	.046	.100	.167	90	.820	F	.011	.022		.023	-.006	-.045	-.125	-.214
40	.350	40	-.123	-.083		-.024	.008	.032	.063	.093	91	.820	F	.035	.022		.026	.003	-.033	-.117	-.210
41	.350	60	-.110	-.118		-.036	.008	.049	.098	.152	92	.820	142				.029	.009	-.027	-.115	-.302
42	.350	F	-.484	-.201		-.041	.003	.013	.005	-.022	93	.820	165	.085	.040						
43	.350	F	-.857	-.229		-.021	-.081	-.239	-.501	-.862	94	.897	30	-.229	-.107		-.083	-.062	-.029	.025	.080
44	.350	F	-.019	.019		-.038	-.108	-.203	-.324	-.523	95	.897	50	-.217	-.108		-.089	-.066	-.027	.034	.094
45	.350	135	.098	.050		-.012	-.044	-.078	-.093	-.103	96	.897	F	-.243	-.116		-.102	-.072	-.030	.039	.105
46	.367	0	-.045	-.042		-.024	.007	.057	.127	.215	97	.897	F	-.847	-.849		-.085	-.110	-.143	-.217	-.316
47	.367	180	.245	.092		.013	.006	.003	.015	.019	98	.897	130	-.017	-.078		-.076	-.098	-.127	-.202	-.340
48	.486	0	-.175	-.077		-.006	.044	.110	.194	.291	99	.897	150	-.023	-.082		-.076	-.095	-.143	-.237	-.256
49	.486	180	.287	.113		-.005	-.045	-.079	-.107	-.179	102	.984	0	-.337	-.256		-.216	-.210	-.220	-.209	-.176
50	.503	15	-.207	-.113		-.003	.053	.121	.200	.292	103	.984	30	-.287	-.208		-.169	-.167	-.177	-.166	-.144
51	.503	30	-.198	-.112		-.001	.057	.116	.183	.258	104	.984	60	-.190	-.171		-.161	-.165	-.176	-.186	-.195
52	.503	50	-.278	-.114		-.002	.056	.116	.181	.255	105	.984	90	-.196	-.169		-.161	-.163	-.176	-.201	-.234
53	.503	F	-.708	-.257		.011	.060	.098	.135	.175	106	.984	135	-.112	-.117		-.131	-.135	-.140	-.154	-.206
											107	.984	180	-.232	-.275		-.263	-.281	-.304	-.301	-.372

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(a) $M = 0.60$ - ContinuedPressure coefficients C_p on -

Model orifice number	$\frac{x}{l}$	ϕ , deg	Fuselage								Model orifice number	$\frac{x}{l}$	ϕ , deg	Fuselage - Concluded							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$																		
1	.026	0			.128	.170	.256	.353	.456	.565	55	.503	F			.039	.009	-.064	-.236	-.842	-.755
2	.026	90			.179	.191	.168	.087	-.043	-.229	56	.503	135			.034	.000	-.052	-.111	-.161	-.253
3	.026	180			.229	.181	.115	.052	-.006	-.047	57	.588	0			-.066	-.020	.059	.144	.231	.328
4	.084	0			.025	.046	.120	.226	.324	.429	58	.588	180			.037	-.010	-.082	-.171	-.254	-.387
5	.084	45			.015	.052	.096	.113	.140	.128	59	.604	15			-.068	-.022	.058	.141	.222	.315
6	.084	90			.045	.062	.041	-.031	-.148	-.310	60	.604	30			-.071	-.019	.067	.154	.232	.324
7	.084	135			.076	.055	.004	-.073	-.162	-.270	61	.604	50			-.085	-.027	.069	.162	.248	.343
8	.084	180			.110	.066	.017	-.033	-.067	-.088	62	.604	F			-.117	-.043	.074	.183	.278	.380
9	.134	J			.138	.105	.060	.025	.000	-.015	63	.604	F			.037	-.035	-.148	-.250	-.272	-1.168
10	.156	J			.153	.130	.093	.062	.039	.012	64	.604	135			.034	-.020	-.106	-.215	-.270	-.504
12	.173	30			-.074	-.043	.003	.047	.090	.136	65	.690	0			-.014	.011	.058	.107	.156	.231
13	.173	60			-.070	-.041	-.025	-.049	-.091	-.164	66	.690	180			.031	.009	-.024	-.087	-.191	-.198
14	.173	90			-.056	-.037	-.047	-.107	-.208	-.352	67	.706	15			.003	.024	.066	.107	.146	.213
15	.173	135			-.006	-.010	-.031	-.069	-.106	-.148	68	.706	30			.010	.034	.075	.113	.146	.205
16	.173	J			.067	.053	.030	.014	-.003	-.035	69	.706	50			.020	.041	.079	.113	.138	.195
17	.173	C			.103	.079	.053	.020	.008	.013	70	.706	F			.028	.049	.087	.116	.131	.184
18	.190	J			-.019	-.023	-.034	-.043	-.068	-.113	71	.706	F			.041	.020	-.011	-.083	-.257	-.424
19	.190	C			-.035	-.045	-.080	-.109	-.112	-.118	72	.706	135			.043	.022	-.007	-.073	-.223	-.327
20	.197	C			-.260	-.301	-.351	-.397	-.409	-.408	73	.753	0			.077	.092	.125	.158	.192	.244
21	.207	C			-.161	-.188	-.219	-.244	-.250	-.247	74	.753	180			.104	.093	.078	.041	-.036	-.044
23	.216	C			-.211	-.237	-.254	-.258	-.247	-.233	75	.770	15			.130	.151	.184	.207	.226	.268
24	.226	23			-.082	-.061	-.020	.019	.074	.134	76	.770	30			.098	.116	.142	.164	.176	.213
25	.226	45			-.075	-.051	-.028	-.023	-.028	-.041	77	.770	50			.081	.094	.115	.134	.142	.174
26	.226	68			-.084	-.067	-.062	-.106	-.171	-.276	78	.770	F			.082	.092	.109	.123	.108	.123
27	.226	90			-.080	-.069	-.084	-.148	-.248	-.394	79	.770	F			.118	.118	.117	.094	-.018	-.081
28	.226	135			-.134	-.154	-.187	-.234	-.270	-.324	80	.770	F			.089	.082	.072	.029	-.098	-.229
29	.226	C			-.227	-.244	-.276	-.305	-.320	-.324	81	.770	135			.102	.095	.080	.037	-.088	-.205
30	.263	C			-.059	-.072	-.079	-.080	-.066	-.067	82	.820	15			.087	.100	.124	.147	.177	.218
31	.283	0			-.085	-.072	-.038	.015	.088	.175	83	.820	38			.069	.081	.104	.130	.158	.198
32	.299	30			-.084	-.064	-.034	-.003	.034	.076	84	.820	F			.057	.069	.098	.130	.174	.227
33	.299	60			-.103	-.067	-.018	.012	.027	.032	85	.820	F			.042	.057	.087	.126	.167	.214
34	.299	F			-.134	-.077	-.015	-.012	-.042	-.118	86	.820	F			.050	.067	.104	.140	.186	.220
35	.299	F			-.048	-.025	-.063	-.183	-.375	-.652	87	.820	F			.058	.077	.123	.172	.261	.299
36	.299	F			-.032	-.071	-.172	-.331	-.535	-.770	88	.820	F			.064	.049	.008	-.056	-.225	-.281
37	.299	135			-.034	-.056	-.104	-.157	-.201	-.252	89	.820	F			.065	.054	.025	-.021	-.123	-.194
38	.319	C			.001	-.007	-.022	-.038	-.047	-.059	90	.820	F			.062	.057	.036	-.008	-.095	-.193
39	.350	20			-.038	-.025	.007	.047	.101	.167	91	.820	F			.070	.065	.052	.014	-.080	-.181
40	.350	40			-.045	-.026	.007	.034	.064	.094	92	.820	142			.090	.085	.073	.033	-.061	-.249
41	.350	60			-.061	-.037	.007	.049	.099	.152	94	.897	30			-.074	-.066	-.050	-.023	.026	.070
42	.350	F			-.086	-.044	.002	.015	.008	-.018	95	.897	50			-.077	-.070	-.051	-.018	.041	.092
43	.350	F			-.050	-.022	-.070	-.231	-.492	-.854	96	.897	F			-.091	-.080	-.055	-.016	.050	.106
44	.350	F			-.008	-.036	-.103	-.200	-.319	-.523	97	.897	F			-.056	-.059	-.076	-.118	-.193	-.300
45	.350	135			.008	-.010	-.042	-.080	-.095	-.104	98	.897	130			-.051	-.054	-.061	-.100	-.180	-.341
46	.367	0			.037	-.027	.005	.055	.126	.213	99	.897	150			-.056	-.051	-.062	-.113	-.225	-.253
47	.367	180			.030	.017	.012	.006	.014	.014	102	.984	0			-.292	-.289	-.292	-.278	-.255	-.229
48	.486	0			-.034	-.009	.041	.109	.192	.288	103	.984	30			-.224	-.221	-.220	-.220	-.201	-.181
49	.486	180			.022	-.002	-.040	-.078	-.101	-.181	104	.984	60			-.168	-.164	-.164	-.178	-.182	-.201
50	.503	15			-.041	-.007	.051	.122	.201	.291	105	.984	90			-.136	-.160	-.160	-.175	-.195	-.233
51	.503	30			-.039	-.004	.054	.117	.184	.257	106	.984	135			-.329	-.138	-.138	-.142	-.167	-.221
52	.503	50			-.042	-.006	.053	.116	.183	.254	107	.984	180			-.010	-.326	-.322	-.341	-.393	-.488
53	.503	F			-.032	.009	.058	.099	.137	.178											
54	.503	F			-.027	.019	-.080	-.340	-.603	-.867											

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(a) $M = 0.60$ - ContinuedPressure coefficients C_p on -

Model orifice number	$\frac{x}{l}$	β , deg	Fuselage								Model orifice number	$\frac{x}{l}$	β , deg	Fuselage - Concluded											
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = -7.5^\circ; \text{brakes closed}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = -7.5^\circ; \text{brakes closed}$											
1	.026	0			.132	.171	.258	.355	.456	.565	55	.503	F			.040	.007	-.066	-.245	-.832	-.758				
2	.026	90			.175	.184	.159	.083	-.045	-.229	56	.503	135			.033	-.001	-.057	-.108	-.165	-.259				
3	.026	180			.233	.179	.114	.050	-.003	-.047	57	.588	0			-.064	-.019	.061	.147	.230	.330				
4	.084	0			.026	.047	.120	.228	.324	.429	58	.588	180			.039	-.009	-.084	-.168	-.250	-.378				
5	.084	45			.017	.050	.094	.115	.136	.121	59	.604	15			-.066	-.023	.058	.141	.220	.315				
6	.084	90			.045	.058	.034	-.033	-.147	-.309	60	.604	30			-.070	-.020	.066	.152	.229	.321				
7	.084	135			.076	.053	-.001	-.066	-.157	-.262	61	.604	50			-.086	-.030	.067	.161	.245	.342				
8	.084	180			.110	.066	.014	-.029	-.061	-.087	62	.604	F			-.120	-.047	.071	.182	.275	.379				
9	.134	J			.143	.107	.058	.025	.002	-.015	63	.604	F			.036	-.036	-.156	-.250	-.279	-1.194				
10	.156	J			.148	.124	.085	.061	.038	.009	64	.604	135			.033	-.022	-.111	-.214	-.272	-.489				
12	.173	30			-.075	-.053	-.010	.034	.079	.125	65	.690	0			-.010	.012	.059	.109	.157	.231				
13	.173	60			-.064	-.043	-.029	-.050	-.094	-.171	66	.690	180			.034	.008	-.028	-.084	-.184	-.197				
14	.173	90			-.053	-.038	-.052	-.106	-.207	-.355	67	.706	15			.000	.020	.061	.104	.140	.208				
15	.173	135			-.005	-.016	-.033	-.065	-.106	-.144	68	.706	30			.003	.025	.066	.105	.136	.197				
16	.173	J			.063	.048	.024	.011	-.005	-.033	69	.706	50			.009	.027	.066	.101	.128	.184				
17	.173	C			.102	.071	.040	.020	.006	.001	70	.706	F			.013	.034	.071	.103	.120	.173				
18	.190	J			-.021	-.029	-.040	-.045	-.065	-.116	71	.706	F			.023	.000	-.032	-.098	-.248	-.397				
19	.190	C			-.036	-.053	-.083	-.107	-.111	-.121	72	.706	135			.031	.006	-.025	-.084	-.220	-.319				
20	.197	C			-.254	-.301	-.354	-.389	-.409	-.406	73	.753	0			.081	.095	.128	.163	.192	.246				
21	.207	C			-.161	-.188	-.221	-.245	-.248	-.244	74	.753	180			.109	.093	.076	.041	-.035	-.040				
23	.216	C			-.208	-.233	-.255	-.255	-.242	-.229	75	.770	15			.061	.080	.113	.140	.158	.202				
24	.226	23			-.078	-.062	-.024	.021	.072	.133	76	.770	30			.038	.055	.082	.106	.119	.157				
25	.226	45			-.070	-.052	-.028	-.026	-.031	-.046	77	.770	50			.031	.042	.064	.088	.097	.132				
26	.226	68			-.081	-.065	-.063	-.106	-.171	-.276	78	.770	F			.040	.049	.066	.083	.076	.091				
27	.226	90			-.078	-.067	-.087	-.144	-.246	-.393	79	.770	F			.082	.079	.075	.060	-.028	-.087				
28	.226	135			-.130	-.155	-.187	-.223	-.264	-.318	80	.770	F			.038	.034	.018	-.019	-.125	-.241				
29	.226	C			-.222	-.242	-.272	-.299	-.314	-.316	81	.770	135			.041	.030	.010	-.025	-.135	-.245				
30	.263	C			-.053	-.068	-.079	-.078	-.067	-.061	82	.820	15			-.047	-.033	-.004	.028	.064	.118				
31	.283	0			-.083	-.071	-.037	.017	.087	.175	83	.820	38			-.031	-.016	.015	.050	.088	.137				
32	.299	30			-.077	-.064	-.034	-.007	.035	.076	84	.820	F			-.027	-.012	.022	.062	.112	.172				
33	.299	60			-.098	-.066	-.019	.013	.029	.030	85	.820	F			-.032	-.015	.024	.068	.114	.170				
34	.299	F			-.130	-.074	-.017	-.008	-.042	-.121	86	.820	F			-.011	.007	.050	.093	.141	.185				
35	.299	F			-.046	-.031	-.065	-.180	-.372	-.645	87	.820	F			.006	.024	.072	.128	.213	.261				
36	.299	F			-.031	-.070	-.174	-.324	-.524	-.760	88	.820	F			.012	-.007	-.053	-.115	-.268	-.303				
37	.299	135			-.030	-.054	-.099	-.150	-.195	-.252	89	.820	F			.007	-.009	-.041	-.084	-.171	-.232				
38	.319	C			.008	-.005	-.025	-.035	-.045	-.054	90	.820	F			-.003	-.017	-.039	-.079	-.151	-.237				
39	.350	20			-.035	-.024	.006	.048	.100	.165	91	.820	F			-.012	-.021	-.040	-.078	-.156	-.244				
40	.350	40			-.043	-.025	.006	.034	.061	.092	92	.820	142			-.016	-.029	-.047	-.087	-.170	-.348				
41	.350	60			-.059	-.036	.006	.050	.097	.151	94	.897	30			-.121	-.111	-.083	-.043	.011	.069				
42	.350	F			-.082	-.043	.001	.016	.006	-.020	95	.897	50			-.132	-.120	-.090	-.045	.018	.080				
43	.350	F			-.046	-.023	-.071	-.229	-.488	-.852	96	.897	F			-.142	-.127	-.093	-.041	.027	.094				
44	.350	F			-.004	-.035	-.102	-.194	-.314	-.514	97	.897	F			-.097	-.108	-.132	-.175	-.237	-.329				
45	.350	135			.012	-.009	-.044	-.075	-.089	-.101	98	.897	130			-.092	-.100	-.119	-.160	-.227	-.348				
46	.367	0			-.035	-.025	.006	.058	.127	.213	99	.897	150			-.098	-.105	-.123	-.171	-.253	-.281				
47	.367	180			.032	.016	.008	.009	.017	.021	102	.984	0			-.252	-.249	-.218	-.192	-.211	-.193				
48	.486	0			-.031	-.008	.043	.112	.191	.290	103	.984	30			-.187	-.185	-.171	-.157	-.150	-.118				
49	.486	180			.026	-.002	-.041	-.073	-.099	-.178	104	.984	.60			-.168	-.166	-.167	-.173	-.182	-.192				
50	.503	15			-.036	-.005	.051	.121	.198	.289	105	.984	90			-.164	-.164	-.167	-.177	-.197	-.234				
51	.503	30			-.036	-.003	.054	.117	.179	.255	106	.984	135			-.128	-.133	-.130	-.141	-.150	-.215				
52	.503	50			-.038	-.005	.053	.116	.178	.252	107	.984	180			-.271	-.276	-.282	-.304	-.345	-.427				
53	.503	F			-.032	.009	.058	.098	.131	.172															
54	.503	F			-.024	.016	-.087	-.339	-.606	-.875															

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(a) M = 0.60 - Continued

Pressure coefficients C_p on -

Model orifice number	$\frac{x}{t}$	β , deg	Fuselage								Model orifice number	$\frac{x}{t}$	β , deg	Fuselage - Concluded							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 7.5^\circ; \text{brakes closed.}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 7.5^\circ; \text{brakes closed.}$							
1	.026	0			.134	.172	.259	.353	.457	.568	55	.503	F			.048	.017	-.059	-.228	-.819	-.742
2	.026	90			.178	.194	.161	.088	-.038	-.223	56	.503	135			.044	.007	-.050	-.104	-.158	-.250
3	.026	180			.236	.185	.113	.054	-.003	-.045	57	.588	0			-.052	-.008	.071	.154	.237	.336
4	.084	0			.030	.049	.122	.225	.322	.431	58	.588	180			.050	.002	-.073	-.154	-.239	-.377
5	.084	45			.022	.055	.092	.120	.138	.128	59	.604	15			-.053	-.010	.068	.150	.228	.324
6	.084	90			.050	.062	.038	-.027	-.140	-.304	60	.604	30			-.055	-.007	.077	.162	.238	.330
7	.084	135			.080	.059	-.001	-.065	-.153	-.261	61	.604	50			-.070	-.015	.077	.171	.253	.351
8	.084	180			.115	.070	.015	-.030	-.064	-.088	62	.604	F			-.103	-.032	.083	.191	.283	.387
9	.134	J			.145	.106	.055	.028	.004	-.012	63	.604	F			.048	-.019	-.136	-.239	-.248	-1.143
10	.156	J			.153	.132	.091	.066	.040	.013	64	.604	135			.047	-.009	-.097	-.200	-.258	-.502
12	.173	30			-.062	-.040	-.004	.041	.087	.135	65	.690	0			.011	.033	.077	.127	.170	.247
13	.173	60			-.063	-.041	-.030	-.047	-.091	-.165	66	.690	180			.052	.027	-.011	-.068	-.177	-.188
14	.173	90			-.048	-.034	-.050	-.104	-.202	-.346	67	.706	15			.027	.048	.085	.128	.160	.228
15	.173	135			.000	-.006	-.035	-.064	-.103	-.139	68	.706	30			.031	.053	.091	.132	.159	.219
16	.173	J			.069	.056	.026	.017	.000	-.028	69	.706	50			.039	.057	.094	.132	.151	.207
17	.173	C			.112	.087	.044	.023	.010	.015	70	.706	F			.045	.064	.100	.134	.142	.196
18	.190	J			-.015	-.022	-.041	-.041	-.064	-.108	71	.706	F			.055	.034	-.003	-.067	-.244	-.410
19	.190	C			-.023	-.043	-.086	-.105	-.110	-.116	72	.706	135			.060	.039	.002	-.058	-.209	-.316
20	.197	C			-.255	-.295	-.351	-.388	-.406	-.403	73	.753	0			.112	.124	.156	.190	.217	.269
21	.207	C			-.157	-.183	-.220	-.241	-.247	-.245	74	.753	180			.136	.120	.101	.065	-.019	-.023
23	.216	C			-.204	-.233	-.252	-.252	-.244	-.230	75	.770	15			.138	.156	.186	.212	.224	.265
24	.226	23			-.075	-.058	-.022	.019	.073	.137	76	.770	30			.112	.127	.152	.176	.182	.218
25	.226	45			-.066	-.049	-.030	-.020	-.026	-.037	77	.770	50			.100	.111	.130	.153	.154	.188
26	.226	68			-.078	-.063	-.064	-.103	-.169	-.274	78	.770	F			.102	.111	.125	.142	.125	.138
27	.226	90			-.076	-.064	-.085	-.142	-.242	-.391	79	.770	F			.136	.135	.128	.112	-.005	-.070
28	.226	135			-.126	-.148	-.186	-.224	-.266	-.314	80	.770	F			.102	.095	.079	.044	-.088	-.222
29	.226	C			-.214	-.238	-.275	-.301	-.316	-.317	81	.770	135			.110	.102	.078	.042	-.085	-.207
30	.263	C			-.051	-.066	-.076	-.073	-.064	-.059	82	.820	15			.115	.126	.148	.175	.200	.245
31	.283	0			-.078	-.067	-.035	.018	.088	.179	83	.820	38			.109	.121	.144	.172	.195	.235
32	.299	30			-.076	-.061	-.034	-.003	.035	.079	84	.820	F			.104	.115	.141	.173	.213	.264
33	.299	60			-.094	-.064	-.015	.015	.032	.037	85	.820	F			.090	.104	.134	.171	.207	.253
34	.299	F			-.124	-.074	-.020	-.007	-.037	-.116	86	.820	F			.097	.113	.147	.185	.224	.257
35	.299	F			-.039	-.026	-.060	-.171	-.360	-.643	87	.820	F			.105	.122	.165	.217	.296	.325
36	.299	F			-.027	-.065	-.165	-.311	-.510	-.755	88	.820	F			.094	.078	.026	-.039	-.224	-.291
37	.299	135			-.024	-.051	-.100	-.149	-.195	-.247	89	.820	F			.097	.079	.048	.004	-.078	-.179
38	.319	C			.012	-.001	-.024	-.035	-.044	-.054	90	.820	F			.097	.087	.061	.022	-.073	-.170
39	.350	20			-.031	-.019	.008	.048	.101	.169	91	.820	F			.101	.095	.072	.039	-.059	-.163
40	.350	40			-.038	-.022	.009	.037	.065	.098	92	.820	142			.112	.106	.085	.048	-.053	-.255
41	.350	60			-.054	-.032	.009	.053	.098	.155	94	.897	30			.280	.293	.317	.340	.368	.396
42	.350	F			-.077	-.038	.005	.019	.010	-.015	95	.897	50			.207	.212	.223	.247	.284	.320
43	.350	F			-.041	-.018	-.066	-.210	-.470	-.837	96	.897	F			.123	.130	.147	.177	.216	.260
44	.350	F			.000	-.031	-.101	-.192	-.310	-.510	97	.897	F			.085	.083	.063	.034	-.064	-.181
45	.350	135			.017	-.003	-.044	-.071	-.087	-.096	98	.897	130			.174	.172	.159	.136	.040	-.128
46	.367	0			-.030	-.021	.009	.059	.127	.218	99	.897	150			.291	.281	.272	.241	.132	.106
47	.367	180			.037	.024	.009	.018	.024	.024	102	.984	0			-.412	-.413	-.413	-.399	-.341	-.294
48	.486	0			-.025	-.001	.046	.114	.194	.295	103	.984	30			-.427	-.426	-.422	-.405	-.362	-.329
49	.486	180			.033	.005	-.035	-.068	-.093	-.172	104	.984	60			-.319	-.317	-.320	-.318	-.308	-.361
50	.503	15			-.029	.002	.057	.127	.202	.298	105	.984	90			-.215	-.272	-.262	-.289	-.289	-.181
51	.503	30			-.029	.004	.060	.122	.185	.263	106	.984	135			-.444	-.444	-.444	-.452	-.516	-.560
52	.503	50			-.030	.002	.058	.121	.185	.261	107	.984	180			-.459	-.459	-.476	-.485	-.528	-.612
53	.503	F			-.023	.016	.063	.104	.139	.182											
54	.503	F			-.017	.022	-.070	-.309	-.573	-.849											

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(a) $M = 0.60$ - ContinuedPressure coefficients C_p on -

Model orifice number	$\frac{x}{t}$	ϕ , deg	Fuselage								Model orifice number	$\frac{x}{t}$	ϕ , deg	Fuselage - Concluded							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$																		
1	.026	0	-.069	.007		.154	.249	.355	.454	.571	55	.503	F	.240	.141		-.003	-.110	-.236	-.949	-.858
2	.026	90	-.114	.169		.268	.229	.136	-.008	-.205	56	.503	135	.300	.151		-.018	-.103	-.145	-.175	-.248
3	.026	180	.547	.338		.129	.064	.005	-.028	-.061	57	.588	0	-.354	-.222		-.037	.048	.134	.220	.326
4	.084	0	-.132	-.067		.028	.112	.210	.307	.425	58	.588	180	.301	.131		-.047	-.129	-.225	-.266	-.365
5	.084	45	-.360	-.110		.100	.173	.217	.249	.260	59	.604	15	-.369	-.211		-.037	.054	.146	.229	.339
6	.084	90	-.259	.026		.128	.100	.034	-.113	-.291	60	.604	30	-.369	-.215		-.562	-.281	-.034	.109	.259
7	.084	125	.254	.205		.064	-.017	-.124	-.245	-.374	61	.604	50	-.446	-.252		-.017	.088	.184	.270	.379
8	.084	180	.410	.209		.031	-.030	-.061	-.095	-.125	62	.604	F	.784	-.342		-.027	.097	.203	.297	.413
9	.134	J	.396	.220		.065	.028	-.003	-.024	-.046	63	.604	F	.388	.190		-.064	-.185	-.327	-.359	-1.068
10	.156	J	.468	.336		.218	.175	.137	.088	.011	64	.604	135	.347	.164		-.048	-.148	-.250	-.286	-.457
12	.173	30	-.257	-.168		-.044	.043	.111	.180	.240	65	.690	0	-.329	-.117		-.015	.038	.091	.143	.227
13	.173	60	-.447	-.176							66	.690	180	.184	.074		-.029	-.068	-.134	-.251	-.306
14	.173	90	-.372	-.096		.002	-.004	-.069	-.182	-.337	67	.706	15	-.344	-.084		.004	.057	.103	.148	.230
15	.173	135	.104	.099		.036	-.008	-.068	-.134	-.181	68	.706	30	.381	-.073		.030	.079	.121	.159	.237
16	.173	J	.340	.223		.133	.101	.069	.023	-.034	69	.706	50	-.471	-.056		.054	.098	.127	.150	.224
17	.173	J	.434	.292		.156	.115	.083	.067	.072	70	.706	F	-.591	-.066		.071	.109	.127	.137	.204
18	.190	J	.200	.108		.045	.024	-.012	-.066	-.135	71	.706	F	.133	.084		.033	-.012	-.114	-.405	-.578
19	.190	C	.269	.129		-.008	-.047	-.076	-.083	-.066	72	.706	135	.168	.097		.023	-.019	-.094	-.301	-.406
20	.197	C	.062	-.155		-.355	-.396	-.426	-.426	-.404	73	.753	0	-.176	.008		.062	.097	.137	.174	.236
21	.207	C	.114	-.034		-.165	-.213	-.228	-.226	-.216	74	.753	180	.184	.118		.052	.022	-.003	-.114	-.182
23	.216	C	.033	-.136		-.263	-.290	-.276	-.238	-.218	75	.770	15	-.252	.111		.249	.274	.293	.306	.363
24	.226	23	-.215	-.157		-.063	-.005	.068	.145	.227	76	.770	30	-.283	.088		.169	.197	.217	.228	.281
25	.226	45	-.330	-.180		-.060	.012	.052	.071	.081	77	.770	50	-.324	.059		.125	.151	.165	.164	.210
26	.226	68	-.465	-.191		-.030	-.023	-.041	-.094	-.186	78	.770	F	-.353	.056		.113	.133	.138	.100	.133
27	.226	90	-.405	-.140		-.030	-.060	-.124	-.232	-.387	79	.770	F	-.390	.100		.155	.136	.084	-.146	-.199
28	.226	135	.030	-.046		-.150	-.212	-.261	-.307	-.355	80	.770	F	.003	.104		.114	.080	.021	-.216	-.365
29	.226	C	.042	-.132		-.295	-.356	-.392	-.410	-.398	81	.770	135	.138	.146		.133	.095	.035	-.177	-.328
30	.263	C	.092	-.041		-.115	-.128	-.120	-.109	-.119	82	.820	15	-.184	.098						
31	.283	0	-.203	-.114		-.095	-.052	.011	.080	.181	83	.820	38	-.223	.080		.121	.142	.170	.200	.252
32	.299	30	-.227	-.168		-.076	-.023	.041	.104	.174	84	.820	F	-.204	.055		.093	.122	.160	.208	.270
33	.299	60	-.414	-.245		-.061	.000	.054	.090	.111	85	.820	F	-.236	.040		.072	.106	.153	.208	.260
34	.299	F	-.814	-.376		-.072	.005	.034	.015	-.048	86	.820	F	-.251	.051		.076	.118	.170	.230	.272
35	.299	F	-.645	-.176		-.003	-.055	-.176	-.378	-.662	87	.820	F	-.283	.064		.088	.131	.202	.315	.375
36	.299	F	-.078	.021		-.113	-.237	-.416	-.650	-.926	88	.820	F	.100	.072		.088	.025	-.075	-.316	-.522
37	.299	135	.117	.041		-.093	-.172	-.238	-.293	-.302	89	.820	F	.056	.072		.080	.030	-.028	-.179	-.308
38	.319	C	.148	.023		-.040	-.071	-.100	-.137	-.202	90	.820	F	.076	.076		.090	.048	-.002	-.146	-.284
39	.350	20	-.202	-.093		-.049	.011	.076	.144	.235	91	.820	F	.105	.091		.108	.069	.016	-.130	-.272
40	.350	40	-.131	-.137		-.035	.024	.079	.128	.186	92	.820	142				.143	.115	.054	-.105	-.254
41	.350	60	-.249	-.156		-.037	.022	.085	.140	.211	93	.820	165	.190	.155		.170	.136	.084	-.065	-.219
42	.350	F	-.537	-.264		-.039	.026	.060	.063	.053	94	.897	30	-.271	-.073		-.056	-.028	.016	.067	.121
43	.350	F	-.862	-.235		-.011	-.073	-.240	-.508	-.882	95	.897	50	-.271	-.080		-.058	-.032	.014	.075	.134
44	.350	F	.058	.057		-.056	-.162	-.272	-.393	-.593	96	.897	F	-.305	-.097		-.073	-.040	.014	.082	.146
45	.350	135	.180	.086		-.039	-.102	-.140	-.145	-.117	97	.897	F	.057	-.036		-.035	-.070	-.117	-.212	-.340
46	.367	0	-.142	-.078		-.053	-.015	.045	.113	.211	98	.897	130	.042	-.029		-.017	-.044	-.090	-.185	-.314
47	.367	180	.227	.075		-.025	-.057	-.063	-.100	-.165	99	.897	150	.035	-.028		-.023	-.050	-.082	-.164	-.296
48	.486	0	.170	-.135		-.035	.022	.098	.179	.288	102	.984	0	-.364	-.248		-.271	-.258	-.244	-.226	-.193
49	.486	180	.265	.094		-.036	-.082	-.152	-.176	-.177	103	.984	30	-.300	-.205						
50	.503	15	-.160	-.129		-.387	-.081	.104	.200	.318	104	.984	60	-.218	-.180		-.164	-.172	-.176	-.189	-.209
51	.503	30	-.177	-.148		-.013	.068	.157	.239	.332	105	.984	90	-.239	-.180		-.164	-.176	-.185	-.225	-.264
52	.503	50	-.221	-.150		.000	.079	.161	.236	.322	106	.984	135	-.111	-.114		-.134	-.149	-.159	-.191	-.230
53	.503	F	-.943	-.217		.021	.090	.150	.201	.252	107	.984	180	-.252	-.306		-.319	-.328	-.316	-.372	-.427
54	.503	F	-.826	-.339		.013	-.109	-.372	-.627	-.893											

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(a) $M = 0.60$ - ConcludedPressure coefficients C_p on -

Model orifice number	$\frac{x}{l}$	β , deg	Fuselage								Model orifice number	$\frac{x}{l}$	β , deg	Fuselage - Concluded							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$											$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
1	.026	0			.101	.150	.245	.342	.452	.564	55	.503	F			.058	.010	-.089	-.214	-.908	-.850
2	.026	90			.270	.275	.237	.151	.010	-.189	56	.503	135			.045	-.005	-.085	-.136	-.160	-.228
3	.026	180			.197	.144	.071	.010	-.021	-.059	57	.588	0			-.067	-.029	.054	.134	.226	.327
4	.084	0			-.013	.027	.109	.200	.307	.419	58	.588	180			.017	-.030	-.104	-.204	-.248	-.348
5	.084	45			.069	.096	.169	.217	.253	.259	59	.604	15			-.073	-.029	.061	.146	.236	.339
6	.084	90			.128	.138	.107	.037	-.095	-.276	60	.604	30			-.605	-.540	-.459	-.299	-.164	-.103
7	.084	135			.134	.086	-.007	-.111	-.227	-.365	61	.604	50			-.070	-.012	.093	.183	.277	.379
8	.084	180			.091	.045	-.026	-.058	-.086	-.125	62	.604	F			-.095	-.022	.101	.200	.305	.413
9	.134	J			.111	.077	.033	.004	-.016	-.044	63	.604	F			.045	-.030	-.157	-.304	-.309	-.978
10	.156	J			.259	.231	.182	.141	.100	.013	64	.604	135			.040	-.023	-.121	-.225	-.267	-.421
12	.173	30			-.073	-.032	.042	.108	.174	.241	65	.690	0			-.011	.006	.056	.103	.161	.237
14	.173	90			.011	.011	.003	-.059	-.165	-.324	66	.690	180			.019	-.008	-.041	-.105	-.230	-.294
15	.173	135			.074	.051	-.001	-.060	-.122	-.179	67	.706	15			.009	.030	.077	.122	.167	.242
16	.173	J			.168	.144	.106	.074	.035	-.029	68	.706	30			.030	.052	.099	.138	.177	.248
17	.173	C			.205	.170	.121	.087	.075	-.071	69	.706	50			.057	.077	.118	.145	.167	.233
18	.190	J			.069	.056	.028	-.005	-.053	-.131	70	.706	F			.074	.094	.129	.145	.155	.215
19	.190	C			.042	.008	-.041	-.070	-.077	-.071	71	.706	F			.082	.057	.018	-.075	-.380	-.560
20	.197	C			-.293	-.338	-.390	-.419	-.418	-.404	72	.706	135			.075	.048	.010	-.058	-.277	-.393
21	.207	C			-.117	-.154	-.201	-.223	-.220	-.214	73	.753	0			.082	.095	.125	.162	.204	.255
23	.216	C			-.232	-.258	-.281	-.274	-.236	-.216	74	.753	180			.099	.083	.053	.032	-.084	-.163
24	.226	23			-.097	-.066	-.002	.064	.136	.223	75	.770	15			.276	.286	.309	.323	.338	.385
25	.226	45			-.072	-.034	.019	.051	.076	.083	76	.770	30			.197	.210	.233	.251	.260	.304
26	.226	68			-.063	-.031	-.012	-.033	-.082	-.175	77	.770	50			.155	.168	.189	.203	.197	.235
27	.226	90			-.038	-.027	-.050	-.114	-.217	-.373	78	.770	F			.145	.153	.171	.174	.134	.158
28	.226	135			-.106	-.140	-.197	-.252	-.297	-.349	79	.770	F			.186	.184	.176	.124	-.126	-.194
29	.226	C			-.241	-.286	-.343	-.384	-.404	-.395	80	.770	F			.160	.150	.124	.064	-.187	-.345
30	.263	C			-.099	-.109	-.119	-.118	-.106	-.114	81	.770	135			.186	.171	.142	.080	-.141	-.307
31	.283	0			-.100	-.089	-.050	.005	.082	.177	83	.820	38			.198	.205	.220	.240	.267	.306
32	.299	30			-.106	-.076	-.018	.040	.106	.172	84	.820	F			.164	.175	.197	.227	.273	.322
33	.299	60			-.112	-.066	.002	.054	.094	.114	85	.820	F			.140	.152	.179	.215	.269	.309
34	.299	F			-.135	-.067	.013	.036	.024	-.039	86	.820	F			.137	.151	.185	.226	.284	.317
35	.299	F			-.012	.004	-.039	-.159	-.355	-.634	87	.820	F			.146	.156	.195	.254	.366	.408
36	.299	F			-.021	-.079	-.217	-.394	-.627	-.896	88	.820	F			.154	.141	.088	-.018	-.269	-.497
37	.299	135			-.037	-.082	-.156	-.226	-.283	-.296	89	.820	F			.152	.137	.097	.033	-.138	-.280
38	.319	C			-.019	-.029	-.059	-.094	-.126	-.191	90	.820	F			.163	.152	.122	.061	-.099	-.248
39	.350	20			-.059	-.040	.013	.072	.150	.232	91	.820	F			.179	.173	.146	.086	-.081	-.235
40	.350	40			-.065	-.034	.025	.076	.134	.186	92	.820	142			.217	.210	.192	.130	-.053	-.213
41	.350	60			-.068	-.034	.023	.081	.146	.211	93	.820	165			.246	.240	.216	.158	-.012	-.172
42	.350	F			-.091	-.039	.027	.060	.073	.062	94	.897	30			.351	.353	.364	.382	.411	.434
43	.350	F			-.012	.007	-.053	-.216	-.482	-.843	95	.897	50			.259	.261	.269	.285	.324	.356
44	.350	F			.003	-.044	-.144	-.256	-.376	-.571	96	.897	F			.170	.173	.185	.211	.258	.294
45	.350	135			.009	-.029	-.087	-.131	-.139	-.113	97	.897	F			.136	.134	.114	.054	-.093	-.247
46	.367	0			-.054	-.046	-.014	.042	.117	.210	98	.897	130			.234	.237	.224	.177	.019	-.138
47	.367	180			.002	-.018	-.044	-.059	-.090	-.155	99	.897	150			.350	.349	.331	.280	.171	.032
48	.486	0			-.046	-.028	.025	.095	.185	.284	102	.984	0			-.431	-.439	-.430	-.411	-.362	-.328
49	.486	180			.003	-.024	-.067	-.139	-.164	-.167	104	.984	60			-.343	-.346	-.327	-.323	-.325	-.347
50	.503	15			-.614	-.550	-.474	-.319	-.177	-.109	105	.984	90			-.332	-.371	-.367	-.361	-.345	-.352
51	.503	30			-.054	-.011	.072	.154	.242	.335	106	.984	135			-.464	-.468	-.465	-.466	-.527	-.616
52	.503	50			-.038	.006	.082	.158	.242	.325	107	.984	180			-.471	-.479	-.462	-.454	-.523	-.575
53	.503	F			-.027	.024	.094	.151	.210	.257											
54	.503	F			.008	.038	-.080	-.345	-.598	-.861											

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(b) $M = 0.90$ Pressure coefficients C_p on -

Model orifice number	$\frac{x}{l}$	β , deg	Fuselage							Model orifice number	$\frac{x}{l}$	β , deg	Fuselage - Concluded								
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$				$\alpha = 20^\circ$	$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$										$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$								
1	.026	0				.203	.295	.398	.510	.625	55	.503	F				.014	-.028	-.164	-.317	-.452
2	.026	90				.178	.130	.054	-.064	-.237	56	.503	135				.001	-.022	-.023	-.074	-.385
3	.026	180				.200	.117	.073	.037	.004	57	.588	Q				-.043	.081	.183	.288	.390
4	.084	0				.054	.138	.235	.343	.463	58	.588	180				-.043	-.238	-.363	-.438	-.559
5	.084	45				.029	.052	.029	.027	.016	59	.604	15				-.053	.067	.161	.259	.354
6	.084	90				.033	-.008	-.065	-.175	-.366	60	.604	30				-.044	.084	.178	.273	.359
7	.084	135				.014	-.020	-.058	-.120	-.178	61	.604	50				-.050	.092	.203	.308	.404
8	.084	180				.061	.001	-.042	-.066	-.102	62	.604	F				-.054	.107	.233	.350	.452
9	.134	J				.101	.047	.018	.001	-.045	63	.604	F				-.062	-.336	-.543	-.842	-1.191
10	.156	J				.044	.004	.001	.016	.070	64	.604	135				-.050	-.295	-.485	-.705	-.909
12	.173	30				-.096	-.072	-.041	-.005	.041	65	.690	0				-.008	.046	.086	.164	.263
13	.173	60				-.103	-.113	-.152	-.211	-.290	66	.690	180				-.023	-.048	-.194	-.418	-.680
14	.173	90				-.099	-.111	-.161	-.260	-.378	67	.706	15				.011	.047	.065	.127	.216
15	.173	135				-.081	-.078	-.077	-.104	-.135	68	.706	30				.015	.053	.067	.128	.211
16	.173	J				-.033	-.050	-.039	-.012	-.016	69	.706	50				.020	.062	.078	.141	.227
17	.173	J				.005	-.034	-.037	-.032	-.025	70	.706	F				.031	.070	.087	.152	.236
18	.190	J				-.111	-.106	-.081	-.047	-.063	71	.706	F				-.011	-.022	-.200	-.642	-.834
19	.190	C				-.117	-.139	-.135	-.132	-.094	72	.706	135				-.003	-.014	-.158	-.428	-.618
20	.197	C				-.316	-.367	-.386	-.394	-.383	73	.753	0				.142	.171	.161	.180	.253
21	.207	C				-.207	-.215	-.217	-.206	-.135	74	.753	180				.126	.115	.010	-.134	-.292
23	.216	C				-.479	-.525	-.538	-.536	-.532	75	.770	15				-.030	-.016	-.017	.002	.054
24	.226	23				-.135	-.118	-.082		.046	76	.770	30				.001	.010	-.019	-.023	.038
25	.226	45				-.141	-.152	-.159	-.170	-.177	77	.770	50				.021	.041	.003	-.018	.046
26	.226	68				-.174	-.198	-.241	-.307	-.404	78	.770	F				.047	.065	.012	-.042	.006
27	.226	90				-.212	-.227	-.262	-.330	-.415	79	.770	F				.082	.092	.012	-.088	-.118
28	.226	135				-.502	-.486	-.455	-.455	-.511	80	.770	F				.033	.031	-.047	-.192	-.407
29	.226	C				-.571	-.606	-.695	-.704	-.738	81	.770	135				-.001	.013	-.037	-.135	-.209
30	.263	C				-.085	-.075	-.048	-.029	-.040	82	.820	15				-.050	-.019	-.004	.007	.055
31	.283	0				-.121	-.085	-.027	.057	.163	83	.820	38				-.037	-.001	.014	.016	.055
32	.299	30				-.102	-.099	-.090	-.064	-.024	84	.820	F				-.027	.011	.033	.044	.093
33	.299	60				-.080	-.050	-.032	-.015	.011	85	.820	F				-.012	.024	.044	.048	.088
34	.299	F				-.081	-.048	-.060	-.097	-.119	87	.820	F				.045	.069	.084	.090	.135
35	.299	F				-.061	-.113	-.240	-.460	-.693	88	.820	F				-.024	-.041	-.062	-.153	-.353
36	.299	F				-.084	-.176	-.303	-.454	-.601	89	.820	F				-.029	-.041	-.063	-.163	-.309
37	.299	135				-.056	-.082	-.119	-.161	-.183	90	.820	F				-.037	-.057	-.088	-.217	-.376
38	.319	C				-.025	-.052	-.071	-.074	-.128	91	.820	F				-.053	-.069	-.101	-.233	-.387
39	.350	20				-.048	-.038	-.010	.038	.104	92	.820	142				-.222	-.223	-.225	-.229	-.235
40	.350	40				-.036	-.029	-.019	.000	.030	94	.897	30				-.148	-.106	-.045	.011	.067
41	.350	60				-.038	-.008	.031	.080	.142	95	.897	50				-.155	-.110	-.043	.022	.081
42	.350	F				-.040	-.016	-.016	-.027	-.032	96	.897	F				-.155	-.106	-.036	.034	.098
43	.350	F				-.043	-.111	-.282	-.564	-.888	97	.897	F				-.172	-.207	-.222	-.261	-.336
44	.350	F				-.034	-.078	-.147	-.278	-.606	98	.897	130				-.159	-.181	-.189	-.235	-.308
45	.350	135				-.008	-.020	-.049	-.056	-.077	99	.897	150				-.158	-.173	-.179	-.227	-.297
46	.367	0				-.052	-.018	.040	.122	.222	102	.984	0				-.279	-.274	-.268	-.259	-.256
47	.367	180				-.015	-.040	-.039	-.032	-.036	104	.984	60				-.249	-.239	-.240	-.247	-.287
48	.486	0				-.021	.036	.115	.211	.317	105	.984	90				-.246	-.242	-.247	-.263	-.321
49	.486	180				-.020	-.043	-.050	-.040	-.060	106	.984	135				-.229	-.234	-.217	-.245	-.302
50	.503	15				-.010	.041	.110	.192	.284	107	.984	180				-.354	-.363	-.333	-.384	-.485
51	.503	30				-.003	.038	.091	.155	.229											
52	.503	50				.002	.053	.116	.185	.265											
53	.503	F				.019	.054	.098	.146	.212											
54	.503	F				.015	-.061	-.174	-.370	-.404											

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(b) M = 0.90 - Continued

Pressure coefficients C_p on -

Model orifice number	$\frac{x}{l}$	β , deg	Fuselage								Model orifice number	$\frac{x}{l}$	β , deg	Fuselage - Concluded							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx -5^\circ$; $\delta_0 = 0^\circ$; $\delta_v = 0^\circ$; brakes open											$\beta \approx -5^\circ$; $\delta_0 = 0^\circ$; $\delta_v = 0^\circ$; brakes open							
1	.026	0	-.002	.065	.161	.211	.302	.402	.513	.633	55	.503	F	.199	.091	.037	.021	-.025	-.169	-.313	-.448
2	.026	90	-.250	.045	.153	.180	.127	.048	-.070	-.248	56	.503	135	.222	.095	.031	.010	-.018	-.019	-.075	-.423
3	.026	180	.643	.411	.251	.198	.114	.070	.034	.001	57	.588	0	-.498	-.358	-.106	-.019	.101	.195	.296	.401
4	.084	0	-.115	-.046	.017	.062	.145	.239	.347	.468	58	.588	180	.385	.182	.039	-.033	-.238	-.339	-.402	-.550
5	.084	45	-.180	-.041	.022	.025	.052	.031	.026	.013	59	.604	15	-.581	-.400	-.110	-.024	.089	.176	.268	.365
6	.084	90	-.384	-.078	.018	.034	-.006	-.069	-.177	-.378	60	.604	30	-.680	-.414	-.108	-.016	.104	.192	.281	.371
7	.084	135	-.003	.014	.025	.016	-.020	-.060	-.117	-.180	61	.604	50	-.856	-.442	-.119	-.021	.114	.216	.318	.416
8	.084	180	.486	.254	.106	.058	.002	-.044	-.068	-.103	62	.604	F	-1.084	-.551	-.150	-.025	.130	.249	.361	.464
9	.134	J	.465	.263	.140	.099	.046	.018	-.001	-.046	63	.604	F	.451	.236	.051	-.050	-.327	-.505	-.788	-1.185
10	.156	J	.303	.139	.067	.044	.005	.001	.019	.073	64	.604	135	.383	.194	.040	-.039	-.289	-.440	-.659	-.901
12	.173	30	-.226	-.146	-.106	-.091	-.069	-.039	-.002	.045	65	.690	0	-.691	-.200	.006	.029	.079	.110	.184	.283
13	.173	60	-.307	-.187	-.112	-.100	-.112	-.152	-.212	-.298	66	.690	180	.235	.061	.033	.008	-.022	-.180	-.381	-.631
14	.173	90	-.540	-.240	-.111	-.095	-.109	-.162	-.258	-.383	67	.706	15	-.607	-.153	.035	.049	.082	.093	.150	.239
15	.173	135	-.194	-.147	-.090	-.078	-.075	-.077	-.104	-.137	68	.706	30	-.579	-.131	.036	.053	.085	.095	.149	.231
16	.173	J			-.022	-.033	-.049	-.037	-.010	.015	69	.706	50	-.633	-.148	.045	.058	.094	.105	.162	.247
17	.173	C	.246	.081	.021	.006	-.032	-.037	-.031	-.023	70	.706	F	.743	.168	.052	.067	.101	.113	.172	.255
18	.190	J	-.010	-.109	-.110	-.107	-.101	-.079	-.045	-.064	71	.706	F	.185	.035	.042	.021	-.009	-.200	-.553	-.566
19	.190	C	.089	-.050	-.100	-.115	-.138	-.137	-.131	-.090	72	.706	135	.178	.036	.043	.029	.002	-.171	-.390	-.443
20	.197	C	.091	-.149	-.272	-.314	-.365	-.386	-.391	-.383	73	.753	0	-.295	.015	.166	.184	.210	.198	.216	.288
21	.207	C	-.051	-.173	-.197	-.204	-.214	-.214	-.202	-.132	74	.753	180	.199	.130	.080	.163	.148	.030	-.155	-.463
23	.216	C	-.017	-.292	-.438	-.479	-.525	-.538	-.534	-.533	75	.770	15	.042	-.014	.016	.017	.023	.020	.030	.086
24	.226	23	-.102	-.111	-.133	-.131	-.112	-.076	-.021	.048	76	.770	30	-.028	-.028	.052	.059	.059	.032	.024	.079
25	.226	45	-.195	-.123	-.128	-.134	-.146	-.154	-.169	-.178	77	.770	50	-.126	-.034	.070	.079	.092	.059	.037	.091
26	.226	68	-.218	-.159	-.155	-.168	-.197	-.239	-.307	-.412	78	.770	F	-.303	-.008	.106	.102	.112	.065	.014	.050
27	.226	90	-.376	-.195	-.198	-.206	-.224	-.261	-.329	-.422	79	.770	F	-.356	-.081	.133	.130	.131	.055	-.068	-.125
28	.226	135	-.205	-.432	-.492	-.496	-.481	-.452	-.455	-.516	80	.770	F	-.120	-.055	.084	.081	.069	-.012	-.190	-.446
29	.226	C	-.351	-.518	-.588	-.560	-.608	-.676	-.680	-.731	81	.770	135	-.054	-.071	.050	.049	.049	-.011	-.141	-.208
30	.263	C	.085	-.058	-.084	-.082	-.069	-.044	-.028	-.047	82	.820	15	-.321	-.095	.060	.067	.082	.094	.103	.148
31	.283	0	-.198	-.133	-.127	-.113	-.078	-.020	.062	.171	83	.820	38	-.278	-.064	.062	.083	.105	.117	.118	.153
32	.299	30	-.257	-.096	-.096	-.097	-.095	-.085	-.061	-.021	84	.820	F	-.182	-.028	.074	.091	.118	.135	.148	.194
33	.299	60	-.204	-.155	-.093	-.074	-.045	-.024	-.012	.015	85	.820	F	-.226	-.012	.089	.101	.128	.143	.150	.186
34	.299	F	-.427	-.265	-.110	-.073	-.042	-.052	-.094	-.118	86	.820	F	-.235	.042						
35	.299	F	-.670	-.220	-.063	-.055	-.112	-.238	-.459	-.702	87	.820	F	-.184	.074	.126	.143	.165	.181	.195	.238
36	.299	F	-.165	-.055	-.046	-.080	-.171	-.297	-.443	-.607	88	.820	F	-.134	-.067	.067	.059	.031	.003	-.144	-.423
37	.299	135			-.034	-.049	-.073	-.104	-.148	-.179	89	.820	F	-.180	-.071	.073	.060	.035	.005	-.134	-.317
38	.319	C	.180	.026	-.012	-.021	-.049	-.068	-.076	-.133	90	.820	F	-.177	-.069	.066	.057	.024	-.012	-.168	-.353
39	.350	20	-.037	-.022	-.043	-.042	-.033	-.006	.043	.111	91	.820	F	-.162	-.078	.051	.046	.020	-.020	-.172	-.369
40	.350	40	-.104	-.058	-.035	-.029	-.023	-.015	.003	.034	92	.820	142	-.156	-.087	.038	.026	.004	-.029	-.187	-.437
41	.350	60	-.113	-.087	-.043	-.029	.000	.037	.085	.149	93	.820	165	-.114	-.091						
42	.350	F	-.436	-.143	-.056	-.031	-.010	-.012	-.025	-.029	94	.897	30	-.208	-.129	.239	.309	.375	.407	.430	.456
43	.350	F	-.970	-.273	-.049	-.035	-.105	-.273	-.541	-.902	95	.897	50	-.199	-.145	.231	.240	.285	.314	.343	.378
44	.350	F	-.031	-.015	-.011	-.030	-.075	-.144	-.279	-.671	96	.897	F	-.199	-.153	.168	.193	.226	.257	.289	.326
45	.350	135	.056	.015	.008	-.003	-.016	-.045	-.053	-.080	97	.897	F	-.110	-.148	.129	.110	.060	.085	-.006	-.167
46	.367	0	-.096	-.053	-.044	-.011	.047	.128	.230	.98	98	.897	130	-.111	-.139	.205	.220	.171	.232	.135	-.006
47	.367	180	.257	.077	.008	-.011	-.035	-.033	-.031	-.039	99	.897	150	-.116	-.143	.312	.271	.275	.351	.273	.191
48	.486	0	-.023	-.068	-.025	-.007	.049	.126	.221	.329	102	.984	0	-.413	-.285	-.495	-.499	-.492	-.462	-.444	-.437
49	.486	180	.317	.115	.011	-.013	-.038	-.047	-.035	-.057	103	.984	30	-.252	-.252						
50	.503	15	-.049	-.007	-.019	.003	.054	.119	.199	.292	104	.984	60	-.239	-.219	-.589	-.594	-.604	-.544	-.554	-.587
51	.503	30	-.219	.004	-.012	.010	.049	.100	.162	.237	105	.984	90	-.234	-.234	-.617	-.599	-.621	-.603	-.628	-.654
52	.503	50	-.328	-.012	-.004	.016	.066	.123	.192	.273	106	.984	135	-.248	-.232	-.634	-.604	-.594	-.558	-.626	-.665
53	.503	F	-.359	-.163	.013	.031	.065	.106	.152	.220	107	.984	180	-.333	-.352	-.505	-.519	-.522	-.540	-.585	-.582
54	.503	F	-.295	-.135	.008	.027	-.057	-.166	-.383	-.402											

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(b) $M = 0.90$ - ContinuedPressure coefficients C_p on -

Model orifice number	$\frac{x}{l}$	β , deg	Fuselage								Model orifice number	$\frac{x}{l}$	β , deg	Fuselage - Concluded							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
1	.026	0	.021	.087		.217	.315	.412	.522	55	.503	F	.257	.138		.010	-.021	-.201	-.393		
2	.026	90	-.131	.129		.232	.201	.131	.018	56	.503	135	.299	.150		.003	-.023	.000	-.007		
3	.026	180	.648	.426		.216	.150	.090	.050	57	.588	0	-.487	-.355		-.029	.103	.203	.307		
4	.084	0	-.072	-.026		.080	.175	.266	.371	58	.588	180	.415	.214		.031	-.222	-.356	-.433		
5	.084	45	-.203	-.041		.073	.121	.167	.190	59	.604	15	-.654	-.378		-.046	.089	.188	.292		
6	.084	90	-.277	-.029		.078	.052	-.015	-.117	60	.604	30	-.773	-.371		-.043	.100	.203	.306		
7	.084	135	.175	.137		.066	.006	-.065	-.140	61	.604	50	-.907	-.389		-.053	.098	.208	.319		
8	.084	180	.501	.284		.077	.024	-.020	-.047	62	.604	F	-1.052	-.458		-.068	.104	.229	.349		
9	.134	J	.474	.278		.115	.066	.033	.017	63	.604	F	.467	.243		-.076	-.348	-.493	-.788		
10	.156	J	.423	.271		.166	.124	.103	.089	64	.604	135	.418	.214		-.059	-.296	-.397	-.519		
12	.173	30	-.232	-.161		-.058	-.001	.045	.103	65	.690	0	-.499	-.162		.001	.061	.097	.175		
13	.173	60	-.414	-.199		-.058	-.038	-.050	-.079	66	.690	180	.255	.081		-.018	-.028	-.153	-.422		
14	.173	90	-.468	-.189		-.044	-.052	-.104	-.196	67	.706	15	-.429	-.119		.027	.072	.087	.149		
15	.173	135	-.014	-.001		.005	-.009	-.031	-.050	68	.706	30	-.588	-.339		.035	.079	.091	.151		
16	.173	J	.275	.151		.085	.064	.053	.049	69	.706	50	-.598	-.131		.042	.082	.089	.148		
17	.173	C	.373	.221		.126	.093	.077	.082	70	.706	F	-.699	-.131		.052	.091	.091	.149		
18	.190	J	.109	.021		-.010	-.014	-.016	-.029	71	.706	F	.191	.041		.010	.008	-.156	-.504		
19	.190	C	.196	.052		-.027	-.050	-.058	-.045	72	.706	135	.202	.053		.009	.003	-.141	-.413		
20	.197	C	.115	-.122		-.299	-.345	-.367	-.379	73	.753	0	-.253	.044		.132	.161	.148	.160		
21	.207	C	.029	-.094		-.140	-.151	-.153	-.131	74	.753	180	.191	.116		.122	.124	.037	-.102		
23	.216	C	.001	-.269		-.454	-.496	-.526	-.537	75	.770	15	-.054	.091		.174	.207	.189	.179		
24	.226	23	-.199	-.127		-.105	-.058	-.005	.067	76	.770	30	-.315	.032		.139	.165	.137	.113		
25	.226	45	-.193	-.154		-.107	-.075	-.056	-.036	77	.770	50	-.415	.016		.119	.137	.101	.055		
26	.226	68	-.337	-.198		-.144	-.134	-.149	-.188	78	.770	F	-.450	.014		.119	.133	.081	-.009		
27	.226	90	-.410	-.209		-.183	-.176	-.201	-.264	79	.770	F	-.011	-.011		.150	.151	.069	-.190		
28	.226	135	-.068	-.315		-.448	-.427	-.409	-.401	80	.770	F	-.071	.026		.106	.106	.009	-.271		
29	.226	C	-.180	-.437		-.660	-.793	-.792	-.778	81	.770	135	.045	.056		.108	.104	.016	-.182		
30	.263	C	.136	.003		-.052	-.041	-.027	-.008	82	.820	15	-.252	.035		.076	.098	.096	.091		
31	.283	0	-.075	-.104		-.102	-.061	-.004	.078	83	.820	38	-.355	.032		.078	.103	.102	.098		
32	.299	30	-.231	-.128		-.098	-.060	-.021	.027	84	.820	F	-.226	.043		.076	.104	.110	.121		
33	.299	60	-.238	-.213		-.096	-.039	-.002	.034	85	.820	F	-.336	.037		.070	.101	.113	.120		
34	.299	F	-.697	-.369		-.100	-.038	-.023	-.037	86	.820	F	-.330	.067		.089	.123	.133	.152		
35	.299	F	-.611	-.222		-.056	-.100	-.209	-.389	87	.820	F	-.306	.084		.105	.141	.158	.229		
36	.299	F	-.088	-.028		-.106	-.220	-.372	-.564	88	.820	F	-.061	.007		.061	.030	-.022	-.262		
37	.299	135	.086	.015		-.080	-.122	-.153	-.177	89	.820	F	-.128	.008		.060	.041	-.001	-.194		
38	.319	C	.218	.066		-.013	-.025	-.028	-.022	90	.820	F	-.129	.006		.058	.041	-.006	-.196		
39	.350	20	-.204	-.054		-.036	.000	.045	.110	91	.820	F	-.107	.009		.061	.050	-.004	-.180		
40	.350	40	-.099	-.072		-.038	.000	.034	.077	92	.820	142				.066	.057	.001	-.173		
41	.350	60	-.036	-.107		-.049	.001	.052	.114	93	.820	165	-.011	.031							
42	.350	F	-.646	-.200		-.054	-.004	.017	.030	94	.897	30	-.245	-.121		-.120	-.079	-.019	.046		
43	.350	F	-.910	-.293		-.036	-.095	-.242	-.462	95	.897	50	-.208	-.113		-.128	-.082	-.016	.060		
44	.350	F	.043	.023		-.048	-.115	-.188	-.314	96	.897	F	-.245	-.119		-.140	-.088	-.015	.065		
45	.350	135	.147	.062		-.016	-.041	-.049	-.048	97	.897	F	-.011	-.012		-.115	-.147	-.182	-.238		
46	.367	0	.001	-.017		-.038	.000	.057	.137	98	.897	130	-.040	-.099		-.101	-.125	-.157	-.225		
47	.367	180	.280	.102		.004	.007	.025	.047	99	.897	150	-.052	-.102		-.099	-.125	-.170	-.255		
48	.486	0	-.026	.016		-.005	.056	.135	.230	102	.984	0	-.380	-.279		-.237	-.234	-.230	-.226		
49	.486	180	.343	.146		-.006	-.017	.007	.017	103	.984	30	-.341	-.256		-.219	-.213	-.210	-.209		
50	.503	15	-.077	-.016		.002	.075	.155	.247	104	.984	60	-.254	-.223		-.213	-.214	-.224	-.239		
51	.503	30	-.063	-.006		.006	.077	.150	.230	105	.984	90	-.264	-.229		-.221	-.226	-.238	-.265		
52	.503	50	-.158	.014		.005	.077	.151	.232	106	.984	135	-.201	-.177		-.208	-.212	-.221	-.273		
53	.503	F	-.344	-.208		.017	.080	.139	.199	107	.984	180	-.299	-.285		-.278	-.283	-.319	-.360		
54	.503	F	-.436	-.186		.016	-.051	-.175	-.300												

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(b) $M = 0.90$ - ContinuedPressure coefficients C_p on -

Model orifice number	$\frac{x}{l}$	ϕ , deg	Fuselage								Model orifice number	$\frac{x}{l}$	ϕ , deg	Fuselage - Concluded							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
1	.026	0			.180	.219	.314	.414	.522	.637	55	.503	F			.047	.013	-.018	-.198	-.392	-.291
2	.026	90			.227	.236	.206	.134	.019	-.143	56	.503	135			.042	.007	-.021	.004	-.002	-.194
3	.026	180			.274	.219	.152	.092	.051	.021	57	.588	0			-.125	-.028	.100	.203	.306	.411
4	.084	0			.047	.083	.174	.268	.371	.487	58	.588	180			.050	.025	-.220	-.353	-.432	-.448
5	.084	45			.054	.078	.124	.164	.194	.205	59	.604	15			-.144	-.043	.088	.190	.293	.397
6	.084	90			.072	.081	.056	-.011	-.120	-.248	60	.604	30			-.148	-.040	.099	.204	.307	.409
7	.084	135			.099	.071	.007	-.063	-.147	-.225	61	.604	50			-.169	-.050	.098	.211	.320	.428
8	.084	180			.127	.081	.024	-.019	-.048	-.068	62	.604	F			-.205	-.066	.106	.231	.350	.464
9	.134	J			.157	.118	.063	.035	.017	.009	63	.604	F			.034	-.071	-.344	-.487	-.802	-.932
10	.156	J			.198	.173	.130	.106	.089	.073	64	.604	135			.036	-.053	-.291	-.387	-.520	-1.014
12	.173	30			-.087	-.056	.000	.053	.109	.175	65	.690	0			-.012	.006	.062	.100	.176	.277
13	.173	60			-.085	-.055	-.032	-.044	-.076	-.127	66	.690	180			.018	-.008	-.023	-.156	-.421	-.583
14	.173	90			-.061	-.041	-.050	-.102	-.194	-.335	67	.706	15			.022	.035	.076	.092	.151	.248
15	.173	135			.016	.011	-.006	-.027	-.051	-.081	68	.706	30			.043	.053	.089	.094	.150	.243
16	.173	J			.104	.094	.068	.057	.050	.029	69	.706	50			.053	.063	.098	.097	.152	.244
17	.173	C			.157	.132	.099	.083	.083	.107	70	.706	F			.041	.025	.020	-.148	-.489	-.833
18	.190	J			.000	-.001	-.010	-.013	-.028	-.080	71	.706	F			.041	.023	.014	-.131	-.403	-.921
19	.190	C			.002	-.020	-.046	-.053	-.044	-.017	72	.706	135			.130	.140	.168	.156	.168	.248
20	.197	C			-.256	-.297	-.347	-.374	-.382	-.372	73	.753	0			.146	.135	.135	.054	-.082	-.044
21	.207	C			-.121	-.131	-.146	-.147	-.128	-.096	74	.753	180			.193	.211	.241	.222	.212	.275
23	.216	C			-.416	-.456	-.505	-.526	-.540	-.534	75	.770	15			.161	.175	.195	.169	.144	.205
24	.226	23			-.121	-.104	-.059	.003	.069	.153	76	.770	30			.144	.150	.164	.129	.082	.143
25	.226	45			-.120	-.104	-.071	-.051	-.034	-.018	77	.770	50			.144	.147	.157	.105	.013	.059
26	.226	68			-.154	-.143	-.133	-.148	-.187	-.253	78	.770	F			.169	.177	.174	.084	-.194	-.173
27	.226	90			-.192	-.179	-.174	-.200	-.263	-.352	79	.770	F			.142	.140	.134	.034	-.268	-.511
28	.226	135			-.456	-.445	-.423	-.407	-.397	-.427	80	.770	F			.154	.148	.141	.049	-.159	-.386
29	.226	C			-.559	-.666	-.802	-.808	-.788	-.789	81	.770	135			.146	.155	.172	.165	.153	.198
30	.263	C			-.045	-.045	-.037	-.024	-.007	.004	82	.820	15			.128	.136	.154	.149	.135	.173
31	.283	0			-.110	-.098	-.061	-.003	.076	.182	83	.820	38			.119	.128	.149	.153	.156	.201
32	.299	30			-.109	-.094	-.058	-.020	.028	.098	84	.820	F			.107	.116	.141	.148	.149	.183
33	.299	60			-.131	-.094	-.039	-.002	.034	.079	85	.820	F			.118	.129	.156	.165	.176	.197
34	.299	F			-.152	-.099	-.037	-.023	-.038	-.043	86	.820	F			.128	.139	.172	.191	.262	.256
35	.299	F			-.067	-.053	-.099	-.210	-.388	-.597	87	.820	F			.112	.102	.068	.007	-.283	-.327
36	.299	F			-.058	-.102	-.220	-.379	-.564	-.908	88	.820	F			.109	.104	.082	.033	-.182	-.268
37	.299	135			-.049	-.077	-.123	-.154	-.180	-.246	89	.820	F			.113	.108	.090	.037	-.175	-.356
38	.319	C			.003	-.010	-.022	-.024	-.022	-.011	90	.820	F			.120	.118	.108	.048	-.147	-.345
39	.350	20			-.044	-.033	.000	.047	.111	.195	91	.820	142			.140	.136	.127	.071	-.120	-.351
40	.350	40			-.052	-.035	.000	.036	.078	.133	92	.820	30			-.095	-.086	-.060	-.012	.041	.087
41	.350	60			-.068	-.046	.001	.052	.114	.189	94	.897	30			-.100	-.091	-.060	-.002	.061	.113
42	.350	F			-.092	-.052	-.003	.019	.030	.043	95	.897	50			-.113	-.100	-.065	.002	.072	.127
43	.350	F			-.053	-.031	-.092	-.243	-.457	-.769	96	.897	F			-.071	-.076	-.101	-.142	-.217	-.357
44	.350	F			-.016	-.043	-.114	-.189	-.316	-.497	97	.897	F			-.057	-.061	-.079	-.117	-.209	-.325
45	.350	135			.008	-.012	-.041	-.047	-.048	-.057	98	.897	130			-.061	-.062	-.081	-.133	-.242	-.308
46	.367	0			-.043	-.034	.000	.057	.136	.238	99	.897	150			-.304	-.315	-.320	-.294	-.278	-.273
47	.367	180			.020	.009	.008	.025	.047	.044	102	.984	0			-.263	-.272	-.278	-.257	-.244	-.248
48	.486	0			-.019	-.003	.056	.136	.230	.338	103	.984	30			-.230	-.226	-.235	-.241	-.251	-.302
49	.486	180			.025	.000	-.017	.008	.015	-.014	104	.984	60			-.227	-.228	-.239	-.249	-.276	-.336
50	.503	15			-.021	.006	.075	.157	.247	.349	105	.984	90			-.225	-.227	-.232	-.235	-.286	-.318
51	.503	30			-.019	.008	.077	.151	.231	.321	106	.984	135			-.363	-.364	-.356	-.380	-.423	-.519
52	.503	50			-.015	.007	.077	.153	.232	.321	107	.984	180								
53	.503	F			-.003	.020	.080	.142	.201	.272											
54	.503	F			-.001	.021	-.051	-.175	-.299	-.497											

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(b) M = 0.90 - Continued

Pressure coefficients C_p on -

Model orifice number	$\frac{x}{l}$	β , deg	Fuselage								Model orifice number	$\frac{x}{l}$	β , deg	Fuselage - Concluded							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = -7.5^\circ; \text{brakes closed}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = -7.5^\circ; \text{brakes closed}$							
1	.026	0			.180	.219	.315	.416	.521	.635	55	.503	F			.043	.014	-.021	-.193	-.392	-.291
2	.026	90			.222	.232	.203	.135	.016	-.148	56	.503	135			.040	.005	-.021	.006	-.006	-.208
3	.026	180			.274	.222	.151	.098	.052	.022	57	.588	0			-.125	-.030	.101	.207	.307	.410
4	.084	0			.047	.084	.175	.270	.370	.484	58	.588	180			.050	-.024	-.218	-.342	-.426	-.451
5	.084	45			.055	.072	.119	.165	.183	.194	59	.604	15			-.145	-.045	.087	.193	.291	.396
6	.084	90			.065	.078	.051	-.012	-.119	-.255	60	.604	30			-.149	-.042	.098	.206	.305	.407
7	.084	135			.094	.069	.007	-.058	-.140	-.221	61	.604	50			-.170	-.054	.096	.214	.318	.426
8	.084	180			.127	.082	.025	-.016	-.047	-.068	62	.604	F			-.207	-.070	.103	.234	.348	.462
9	.134	J			.157	.119	.066	.038	.017	.009	63	.604	F			.033	-.069	-.342	-.517	-.749	-1.015
10	.156	J			.189	.166	.123	.103	.087	.071	64	.604	135			.035	-.052	-.290	-.405	-.521	-1.041
12	.173	30			-.098	-.067	-.010	.044	.093	.159	65	.690	0			-.012	.005	.061	.103	.175	.277
13	.173	60			-.087	-.059	-.038	-.048	-.085	-.140	66	.690	180			.015	-.011	-.027	-.151	-.421	-.580
14	.173	90			-.067	-.043	-.053	-.102	-.198	-.342	67	.706	15			.019	.030	.071	.092	.149	.245
15	.173	135			.007	.007	-.010	-.027	-.051	-.084	68	.706	30			.022	.035	.075	.094	.150	.242
16	.173	J			.098	.084	.060	.055	.047	.027	69	.706	50			.032	.041	.077	.091	.145	.237
17	.173	C			.138	.114	.090	.079	.076	.094	70	.706	F			.038	.049	.084	.092	.148	.238
18	.190	J			-.005	-.008	-.015	-.014	-.032	-.081	71	.706	F			.023	.036	.068	.081	.145	.237
19	.190	C			-.014	-.026	-.052	-.057	-.049	-.024	72	.706	135			.029	.009	-.001	-.132	-.429	-.897
20	.197	C			-.255	-.296	-.347	-.371	-.381	-.372	73	.753	0			.132	.141	.168	.160	.169	.248
21	.207	C			-.121	-.132	-.151	-.147	-.133	-.096	74	.753	180			.143	.134	.132	.054	-.089	-.036
23	.216	C			-.413	-.453	-.505	-.522	-.539	-.533	75	.770	15			.132	.148	.179	.169	.158	.215
24	.226	23			-.120	-.106	-.063	-.001	.063	.145	76	.770	30			.104	.117	.138	.116	.090	.145
25	.226	45			-.120	-.107	-.079	-.055	-.042	-.027	77	.770	50			.090	.095	.109	.078	.030	.086
26	.226	68			-.157	-.147	-.138	-.153	-.193	-.261	78	.770	F			.096	.099	.107	.061	-.030	.003
27	.226	90			-.193	-.185	-.180	-.200	-.268	-.359	79	.770	F			.125	.130	.129	.058	-.172	-.189
28	.226	135			-.459	-.449	-.426	-.405	-.402	-.433	80	.770	F			.085	.085	.082	-.006	-.254	-.651
29	.226	C			-.556	-.656	-.800	-.803	-.789	-.792	81	.770	135			.090	.082	.077	.000	-.180	-.309
30	.263	C			-.041	-.043	-.035	-.020	-.007	.004	82	.820	15			-.034	-.021	.000	.003	-.008	.049
31	.283	0			-.111	-.100	-.061	.000	.075	.182	83	.820	38			-.010	.004	.029	.036	.027	.077
32	.299	30			-.108	-.094	-.061	-.019	.025	.093	84	.820	F			.002	.015	.041	.054	.059	.114
33	.299	60			-.128	-.095	-.041	.002	.030	.074	85	.820	F			.003	.016	.046	.063	.066	.109
34	.299	F			-.153	-.099	-.040	-.022	-.043	-.050	86	.820	F			.030	.043	.075	.091	.103	.133
35	.299	F			-.069	-.054	-.103	-.206	-.391	-.603	87	.820	F			.052	.062	.094	.112	.157	.169
36	.299	F			-.056	-.099	-.219	-.365	-.561	-.877	88	.820	F			.032	.021	-.013	-.044	-.246	-.256
37	.299	135			-.049	-.076	-.120	-.146	-.177	-.244	89	.820	F			.022	.014	-.003	-.033	-.208	-.230
38	.319	C			.002	-.009	-.023	-.020	-.022	-.012	90	.820	F			.013	.005	-.013	-.054	-.228	-.368
39	.350	20			-.043	-.034	-.001	.049	.107	.190	91	.820	F			.002	-.005	-.019	-.065	-.241	-.369
40	.350	40			-.051	-.035	-.001	.037	.073	.127	92	.820	142			-.014	-.018	-.033	-.088	-.279	-.464
41	.350	60			-.066	-.046	.001	.055	.112	.186	94	.897	30			-.173	-.156	-.111	-.041	.022	.076
42	.350	F			-.090	-.052	-.004	.021	.027	.038	95	.897	50			-.183	-.166	-.118	-.042	.030	.090
43	.350	F			-.056	-.034	-.093	-.239	-.463	-.776	96	.897	F			-.196	-.176	-.122	-.038	.040	.103
44	.350	F			-.014	-.043	-.113	-.180	-.311	-.491	97	.897	F			-.142	-.158	-.194	-.218	-.268	-.398
45	.350	135			.007	-.011	-.040	-.044	-.048	-.057	98	.897	130			-.128	-.143	-.175	-.199	-.259	-.352
46	.367	0			-.042	-.035	.000	.062	.136	.237	99	.897	150			-.129	-.141	-.175	-.207	-.278	-.326
47	.367	180			.021	.009	.008	.030	.048	.045	102	.984	0			-.300	-.306	-.294	-.256	-.283	-.282
48	.486	0			-.019	-.003	.056	.140	.229	.337	103	.984	30			-.251	-.251	-.244	-.221	-.222	-.230
49	.486	180			.025	-.001	-.016	.012	.016	-.013	104	.984	60			-.219	-.218	-.224	-.228	-.247	-.287
50	.503	15			-.020	.005	.074	.159	.244	.346	105	.984	90			-.220	-.222	-.228	-.233	-.262	-.324
51	.503	30			-.018	.008	.076	.153	.227	.315	106	.984	135			-.203	-.207	-.205	-.207	-.230	-.322
52	.503	50			-.014	.007	.076	.155	.228	.318	107	.984	180			-.293	-.344	-.345	-.359	-.388	-.477
53	.503	F			-.003	.019	.079	.143	.196	.267											
54	.503	F			-.004	.020	-.051	-.170	-.302	-.491											

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(b) $M = 0.90$ - ContinuedPressure coefficients C_p on -

Model orifice number	$\frac{x}{t}$	β , deg	Fuselage								Model orifice number	$\frac{x}{t}$	β , deg	Fuselage - Concluded							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx 0^\circ; \delta_E = 0^\circ; \delta_V = 7.5^\circ; \text{brakes closed}$											$\beta \approx 0^\circ; \delta_E = 0^\circ; \delta_V = 7.5^\circ; \text{brakes closed}$							
1	.026	0			.185	.226	.319	.417	.525	.639	55	.503	F			.056	.024	-.022	-.197	-.376	-.288
2	.026	90			.227	.240	.200	.133	.020	-.146	56	.503	135			.051	.017	-.015	.007	.004	-.192
3	.026	180			.276	.224	.147	.092	.055	.023	57	.588	0			-.104	-.007	.116	.214	.316	.416
4	.084	0			.051	.089	.180	.270	.376	.487	58	.588	180			.063	-.011	-.229	-.334	-.401	-.436
5	.084	45			.061	.082	.123	.165	.192	.201	59	.604	15			-.116	-.017	.108	.202	.303	.404
6	.084	90			.071	.085	.046	-.014	-.117	-.251	60	.604	30			-.121	-.015	.117	.215	.316	.415
7	.084	135			.099	.072	-.003	-.063	-.141	-.225	61	.604	50			-.144	-.026	.116	.222	.330	.434
8	.084	180			.132	.085	.019	-.020	-.044	-.067	62	.604	F			-.182	-.041	.125	.243	.360	.469
9	.134	J			.161	.122	.064	.038	.022	.010	63	.604	F			.048	-.067	-.365	-.466	-.755	-.949
10	.156	J			.197	.175	.124	.106	.092	.072	64	.604	135			.051	-.039	-.303	-.362	-.509	-1.015
12	.173	30			-.083	-.045	.008	.053	.116	.170	65	.690	0			.021	.041	.091	.123	.194	.291
13	.173	60			-.081	-.050	-.036	-.044	-.075	-.132	66	.690	180			.047	.023	-.008	-.135	-.362	-.497
14	.173	90			-.059	-.037	-.059	-.102	-.194	-.338	67	.706	15			.054	.069	.104	.117	.171	.263
15	.173	135			.015	.016	-.011	-.026	-.046	-.081	68	.706	30			.059	.075	.111	.120	.172	.262
16	.173	J			.105	.093	.062	.057	.053	.027	69	.706	50			.070	.082	.114	.118	.168	.256
17	.173	C			.162	.139	.095	.083	.087	.104	70	.706	F			.076	.091	.121	.119	.169	.256
18	.190	J			.001	.001	-.013	-.013	-.025	-.079	71	.706	F			.063	.050	.027	-.140	-.451	-.775
19	.190	C			.004	-.016	-.050	-.056	-.041	-.015	72	.706	135			.066	.050	.026	-.128	-.379	-.835
20	.197	C			-.252	-.294	-.348	-.373	-.376	-.372	73	.753	0			.169	.181	.202	.191	.199	.275
21	.207	C			-.118	-.128	-.149	-.146	-.124	-.094	74	.753	180			.177	.167	.153	.066	-.109	-.175
23	.216	C			-.408	-.450	-.505	-.523	-.534	-.529	75	.770	15			.206	.224	.249	.233	.221	.281
24	.226	23			-.114	-.097	-.056	.001	.074	.154	76	.770	30			.180	.194	.208	.184	.160	.217
25	.226	45			-.111	-.090	-.068	-.049	-.027	-.021	77	.770	50			.167	.173	.183	.150	.105	.164
26	.226	68			-.147	-.136	-.133	-.144	-.184	-.253	78	.770	F			.169	.174	.179	.129	.040	.092
27	.226	90			-.183	-.170	-.178	-.200	-.259	-.353	79	.770	F			.192	.201	.188	.103	-.182	-.118
28	.226	135			-.452	-.439	-.421	-.404	-.393	-.427	80	.770	F			.160	.159	.142	.041	-.262	-.468
29	.226	C			-.545	-.654	-.798	-.790	-.774	-.778	81	.770	135			.166	.161	.141	.048	-.179	-.387
30	.263	C			-.039	-.039	-.037	-.022	-.003	.006	82	.820	15			.180	.190	.203	.203	.201	.244
31	.283	0			-.103	-.089	-.056	.003	.084	.185	83	.820	38			.176	.187	.201	.202	.198	.236
32	.299	30			-.101	-.084	-.053	-.016	.036	.100	84	.820	F			.174	.184	.201	.210	.221	.265
33	.299	60			-.122	-.084	-.034	.004	.040	.083	85	.820	F			.164	.175	.197	.209	.219	.252
34	.299	F			-.144	-.083	-.033	-.020	-.030	-.043	86	.820	F			.174	.187	.213	.226	.246	.265
35	.299	F			-.060	-.046	-.110	-.208	-.395	-.596	87	.820	F			.185	.196	.228	.253	.334	.336
36	.299	F			-.052	-.103	-.250	-.376	-.572	-.908	88	.820	F			.158	.148	.097	.025	-.325	-.370
37	.299	135			-.043	-.069	-.122	-.148	-.169	-.241	89	.820	F			.154	.149	.115	.054	-.165	-.269
38	.319	C			.010	-.004	-.023	-.022	-.014	-.011	90	.820	F			.157	.154	.120	.060	-.158	-.340
39	.350	20			-.037	-.023	.007	.051	.117	.196	91	.820	F			.158	.159	.134	.066	-.137	-.333
40	.350	40			-.044	-.026	.004	.040	.083	.134	92	.820	142			.167	.165	.142	.073	-.133	-.341
41	.350	60			-.060	-.037	.006	.057	.120	.190	94	.897	30			.377	.394	.426	.456	.479	.500
42	.350	F			-.083	-.043	.001	.024	.035	.044	95	.897	50			.310	.320	.335	.361	.396	.424
43	.350	F			-.045	-.025	-.118	-.246	-.478	-.772	96	.897	F			.232	.243	.261	.295	.329	.360
44	.350	F			-.007	-.036	-.116	-.181	-.307	-.491	97	.897	F			.182	.180	.147	.093	-.021	-.232
45	.350	135			.013	-.005	-.038	-.042	-.038	-.052	98	.897	130			.275	.270	.251	.203	.094	-.086
46	.367	0			-.035	-.025	.005	.064	.144	.242	99	.897	150			.387	.380	.365	.311	.196	.113
47	.367	180			.029	.018	.012	.030	.055	.049	102	.984	0			-.502	-.492	-.489	-.462	-.446	-.445
48	.486	0			-.009	.011	.066	.144	.239	.343	103	.984	30			-.545	-.541	-.543	-.509	-.448	-.433
49	.486	180			.035	.009	-.011	.014	.026	-.009	104	.984	60			-.582	-.572	-.586	-.570	-.510	-.565
50	.503	15			-.011	.018	.085	.165	.255	.353	105	.984	90			-.596	-.602	-.626	-.625	-.616	-.664
51	.503	30			-.008	.021	.086	.159	.238	.323	106	.984	135			-.624	.631	-.626	-.601	-.601	-.662
52	.503	50			-.006	.022	.089	.160	.240	.325	107	.984	180			-.484	.487	-.521	-.528	-.542	-.612
53	.503	F			.006	.032	.088	.149	.207	.274											
54	.503	F			.008	.026	-.071	-.173	-.305	-.493											

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(b) $M = 0.90$ - ContinuedPressure coefficients C_p on -

Model orifice number	$\frac{x}{l}$	ϕ , deg	Fuselage								Model orifice number	$\frac{x}{l}$	ϕ , deg	Fuselage - Concluded							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$											$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
1	.026	0	-.005	.067		.210	.315	.421	.537	.658	55	.503	F	.325	.190		-.003	-.027	-.287	-.355	-.320
2	.026	90	-.017	.223		.319	.274	.191	.058	-.118	56	.503	135	.369	.194		-.062	-.035	-.001	.017	-.443
3	.026	180	.630	.409		.181	.104	.068	.033	-.006	57	.588	0	-.494	-.368		-.032	.093	.198	.306	.412
4	.084	0	-.117	-.048		.057	.155	.258	.371	.496	58	.588	180	.391	.194		-.104	-.278	-.382	-.444	-.630
5	.084	45	-.354	-.080		.134	.209	.264	.304	.332	59	.604	15	-.620	-.393		-.048	.088	.201	.315	.427
6	.084	90	-.205	.052		.182	.119	.070	-.077	-.241	60	.604	30	-.740	-.385		.062	.176	.284	.369	.452
7	.084	135	.324	.245		.092	-.012	-.115	-.240	-.369	61	.604	50	-1.004	-.407		-.034	.122	.242	.358	.470
8	.084	180	.475	.258		.058	-.008	-.048	-.076	-.114	62	.604	F	-.942	-.428		-.042	.132	.263	.389	.505
9	.134	J	.461	.267		.089	.039	.016	-.002	-.033	63	.604	F	.484	.251		-.087	-.412	-.541	-.849	-.959
10	.156	J	.540	.392		.266	.210	.165	.109	.053	64	.604	135	.440	.223		-.033	-.354	-.406	-.583	-1.147
12	.173	30	-.273	-.208		-.034	.053	.133	.213	.285	65	.690	0	-.629	-.233		-.007	.049	.094	.182	.284
13	.173	60	-.519	-.217							66	.690	180	.239	.066		.045	-.059	-.255	-.430	-.780
14	.173	90	-.394	-.118		.031	.008	-.055	-.165	-.330	67	.706	15	-.574	-.163		.028	.074	.096	.174	.276
15	.173	135	.155	.131		.084	.035	-.008	-.048	-.079	68	.706	30	-.600	-.163		.049	.095	.111	.189	.285
16	.173	J	.405	.276		.186	.150	.119	.071	.022	69	.706	50	-.673	-.180		.074	.113	.116	.187	.281
17	.173	C	.499	.352		.225	.185	.171	.178	.186	70	.706	F	-.774	-.200		.091	.125	.117	.187	.278
18	.190	J	.241	.139		.087	.065	.029	-.025	-.089	71	.706	F	.208	.055		.033	.036	-.204	-.547	-.841
19	.190	C	.319	.168		.056	.023	.019	.037	.071	72	.706	135	.225	.070		.125	.024	-.180	-.383	-1.009
20	.197	C	.074	-.155		-.355	-.396	-.400	-.392	-.379	73	.753	0	-.277	.004		.140	.164	.153	.186	.263
21	.207	C	.136	.000		-.066	-.084	-.076	-.046	-.040	74	.753	180	.203	.123		.217	.105	-.014	-.132	-.192
23	.216	C	-.008	-.273		-.501	-.541	-.539	-.529	-.533	75	.770	15	-.325	.103		.332	.348	.320	.325	.387
24	.226	23	-.207	-.172		-.100	-.023	.062	.155	.251	76	.770	30	-.463	.078		.245	.263	.226	.223	.288
25	.226	45	-.275	-.210		-.071	-.008	.044	.084	.115	77	.770	50	-.536	.033		.195	.210	.153	.131	.191
26	.226	68	-.431	-.234		-.091	-.061	-.057	-.086	-.147	78	.770	F	-.531	.007		.177	.189	.097	.031	.081
27	.226	90	-.439	-.198		-.111	-.110	-.139	-.212	-.326	79	.770	F	-.507	-.033		.179	.196	-.017	-.292	-.342
28	.226	135	.018	-.149		-.350	-.348	-.340	-.343	-.366	80	.770	F	-.043	.051		.202	.151	-.029	-.385	-.503
29	.226	C	-.008	-.289		-.759	-.810	-.810	-.874	-.815	81	.770	135	.123	.129		.147	.166	.012	-.252	-.679
30	.263	C	.101	-.042		-.104	-.071	-.056	-.039	-.072	82	.820	15	-.281	.126						
31	.283	0	-.197	-.128		-.119	-.073	-.005	.085	.202	83	.820	38	-.389	.096		.188	.197	.194	.196	.244
32	.299	30	-.216	-.179		-.090	-.038	.034	.113	.203	84	.820	F	-.354	.083		.161	.179	.194	.208	.263
33	.299	60	-.389	-.282		-.083	-.015	.045	.100	.167	85	.820	F	-.407	.066		.142	.165	.188	.206	.256
34	.299	F	-1.068	-.442		-.026	-.009	.022	.024	.044	86	.820	F	-.415	.072		.151	.177	.213	.238	.282
35	.299	F	-.535	-.197		-.129	-.088	-.200	-.400	-.528	87	.820	F	-.449	.075		.163	.190	.266	.341	.395
36	.299	F	.006	.016		-.118	-.294	-.508	-.978	-1.223	88	.820	F	.011	.075		.143	.092	-.105	-.507	-.611
37	.299	135	.165	.041		-.034	-.194	-.235	-.223	-.342	89	.820	F	-.031	.058		.151	.099	-.014	-.272	-.360
38	.319	C	.184	.032		-.008	-.063	-.086	-.105	-.151	90	.820	F	-.023	.069		.169	.111	.007	-.254	-.357
39	.350	20	-.186	-.078		-.047	.012	.084	.171	.274	91	.820	F	.011	.087		.204	.136	.030	-.218	-.412
40	.350	40	-.070	-.113		-.041	.023	.086	.154	.232	92	.820	142		.236		.236	.183	.064	-.161	-.538
41	.350	60	-.094	-.151		-.042	.021	.092	.170	.257	93	.820	165	.135	.158		-.045	.200	.108	-.098	-.229
42	.350	F	-.758	-.267		-.046	.025	.066	.097	.125	94	.897	30	-.306	-.094		-.079	-.038	.042	.100	.155
43	.350	F	-.801	-.257		-.064	.091	.257	-.442	-.730	95	.897	50	-.297	-.095		-.083	-.043	.045	.112	.172
44	.350	F	.124	.066		-.038	-.170	-.255	-.414	-.839	96	.897	F	-.355	-.106		-.096	-.052	.048	.120	.180
45	.350	135	.229	.101		-.025	-.087	-.088	-.054	-.090	97	.897	F	.058	-.017		-.024	-.088	-.144	-.262	-.467
46	.367	9	-.089	-.063		-.057	-.016	.051	.141	.251	98	.897	130	.045	-.014		-.029	-.057	-.116	-.227	-.440
47	.367	180	.264	.088		-.025	-.046	-.046	-.051	-.074	99	.897	150	.032	-.020		-.299	-.067	-.104	-.205	-.373
48	.486	0	-.023	-.065		-.022	.043	.132	.235	.348	102	.984	0	-.413	-.283		-.042	-.296	-.276	-.272	-.264
49	.486	180	.322	.126		.038	-.045	-.059	-.037	-.120	103	.984	30	-.367	-.258						
50	.503	15	-.002	-.028		.010	.093	.192	.294	.405	104	.984	60	-.318	-.245		-.223	-.228	-.238	-.280	-.328
51	.503	30	-.021	-.032		.004	.105	.202	.301	.405	105	.984	90	-.319	-.245		-.232	-.238	-.259	-.318	-.390
52	.503	50	-.008	-.008		.021	.114	.209	.302	.398	106	.984	135	-.244	-.201		-.371	-.242	-.250	-.303	-.417
53	.503	F	-.352	-.147		.038	.125	.204	.277	.353	107	.984	180	-.335	-.349		-.010	-.373	-.348	-.403	-.538
54	.503	F	-.612	-.178		.012	-.053	-.186	-.301	-.514											

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(b) M = 0.90 - Concluded

Pressure coefficients C_p on -

Model orifice number	$\frac{x}{l}$	β , deg	Fuselage								Model orifice number	$\frac{x}{l}$	β , deg	Fuselage - Concluded							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$											$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
1	.026	0			.153	.207	.311	.418	.530	.651	55	.503	F	.068	.024	-.018	-.222	-.383	-.307		
2	.026	90			.322	.322	.281	.200	.070	-.103	56	.503	135	.055	.009	-.028	.003	.023	-.369		
3	.026	180			.245	.188	.112	.072	.036	.000	57	.588	0	-.107	-.021	.101	.204	.307	.414		
4	.084	0			.014	.057	.153	.255	.366	.488	58	.588	180	.031	-.039	-.252	-.353	-.412	-.603		
5	.084	45			.096	.133	.208	.263	.304	.334	59	.604	15	-.127	-.033	.098	.207	.316	.427		
6	.084	90			.158	.185	.126	.077	-.067	-.225	60	.604	30	.147	.208	.275	.326	.364	.368		
7	.084	135			.155	.101	-.002	-.104	-.230	-.355	61	.604	50	-.143	-.022	.129	.247	.360	.470		
8	.084	180			.115	.065	-.002	-.042	-.073	-.109	62	.604	F	-.177	-.032	.138	.267	.388	.505		
9	.134	J			.135	.095	.046	.021	.001	-.030	63	.604	F	.039	-.072	-.382	-.414	-.850	-.881		
10	.156	J			.305	.271	.216	.173	.114	.060	64	.604	135	.037	-.057	-.329	-.374	-.520	-1.087		
12	.173	30			-.088	-.035	.053	.131	.207	.291	65	.690	0	.006	.025	.077	.115	.193	.293		
14	.173	90			.018	.034	.015	-.044	-.155	-.309	66	.690	180	.026	.003	-.027	-.179	-.397	-.712		
15	.173	135			.111	.088	.042	-.002	-.045	-.074	67	.706	15	.042	.060	.104	.122	.190	.288		
16	.173	J			.215	.191	.156	.125	.077	.029	68	.706	30	.063	.081	.126	.137	.202	.299		
17	.173	C			.263	.229	.189	.174	.178	.192	69	.706	50	.093	.108	.144	.141	.201	.291		
18	.190	J			.104	.091	.071	.037	-.020	-.080	70	.706	F	.110	.122	.156	.141	.198	.288		
19	.190	C			.088	.060	.028	.022	.036	.073	71	.706	F	.095	.079	.066	-.180	-.466	-.727		
20	.197	C			-.307	-.348	-.390	-.396	-.392	-.377	72	.706	135	.088	.069	.054	-.158	-.360	-.822		
21	.207	C			-.049	-.065	-.080	-.074	-.047	-.031	73	.753	0	.161	.178	.202	.194	.217	.291		
23	.216	C			-.451	-.494	-.536	-.537	-.527	-.525	74	.753	180	.180	.164	.144	.023	-.159	-.360		
24	.226	23			-.131	-.096	-.022	.062	.153	.252	75	.770	15	.365	.373	.388	.364	.367	.422		
25	.226	45			-.110	-.069	-.005	.045	.087	.119	76	.770	30	.281	.291	.308	.277	.269	.327		
26	.226	68			-.121	-.087	-.055	-.052	-.079	-.135	77	.770	50	.235	.243	.257	.207	.182	.238		
27	.226	90			-.126	-.105	-.103	-.131	-.204	-.310	78	.770	F	.221	.226	.237	.153	.086	.132		
28	.226	135			-.350	-.347	-.344	-.337	-.340	-.358	79	.770	F	.259	.259	.243	.034	-.265	-.268		
29	.226	C			-.674	-.735	-.789	-.788	-.864	-.808	80	.770	F	.230	.225	.199	.019	-.339	-.504		
30	.263	C			-.090	-.085	-.067	-.052	-.039	-.057	81	.770	135	.257	.248	.215	.057	-.237	-.646		
31	.283	0			-.125	-.113	-.070	-.005	.083	.197	83	.820	38	.279	.281	.291	.290	.290	.331		
32	.299	30			-.135	-.100	-.035	.034	.110	.203	84	.820	F	.249	.255	.273	.288	.302	.347		
33	.299	60			-.142	-.087	-.013	.047	.100	.169	85	.820	F	.227	.235	.257	.282	.297	.339		
34	.299	F			-.159	-.084	-.005	.028	.027	.047	86	.820	F	.227	.237	.264	.302	.322	.358		
35	.299	F			-.032	-.020	-.075	-.189	-.387	-.513	87	.820	F	.237	.243	.273	.352	.414	.451		
36	.299	F			-.051	-.116	-.274	-.473	-.944	-1.205	88	.820	F	.233	.225	.172	-.054	-.456	-.689		
37	.299	135			-.064	-.108	-.182	-.227	-.221	-.313	89	.820	F	.229	.222	.183	.052	-.228	-.389		
38	.319	C			-.020	-.029	-.053	-.081	-.099	-.147	90	.820	F	.240	.236	.200	.076	-.203	-.328		
39	.350	20			-.063	-.041	.015	.086	.170	.269	91	.820	F	.257	.255	.226	.105	-.171	-.367		
40	.350	40			-.071	-.037	.026	.089	.155	.230	92	.820	142	.289	.292	.275	.137	-.123	-.463		
41	.350	60			-.075	-.037	.024	.093	.170	.255	93	.820	165	.321	.321	.290	.183	-.057	-.225		
42	.350	F			-.100	-.042	.028	.070	.103	.128	94	.897	30	.464	.469	.482	.513	.530	.553		
43	.350	F			-.018	-.001	-.075	-.236	-.447	-.711	95	.897	50	.373	.377	.386	.414	.441	.470		
44	.350	F			-.006	-.053	-.154	-.244	-.402	-.759	96	.897	F	.288	.292	.305	.345	.375	.404		
45	.350	135			.005	-.029	-.076	-.083	-.051	-.077	97	.897	F	.249	.250	.212	.109	-.105	-.357		
46	.367	0			-.056	-.050	-.012	.054	.139	.247	98	.897	130	.349	.354	.327	.219	.032	-.307		
47	.367	180			-.001	-.017	-.037	-.037	-.048	-.070	99	.897	150	.468	.465	.424	.345	.213	-.113		
48	.486	0			-.030	-.015	.049	.136	.234	.346	102	.984	0	-.498	-.499	-.496	-.477	-.459	-.432		
49	.486	180			.007	-.014	-.035	-.045	-.031	-.099	104	.984	60	-.591	-.587	-.574	-.533	-.502	-.525		
50	.503	15			.144	.183	.233	.278	.314	.320	105	.984	90	-.613	-.615	-.623	-.630	-.644	-.652		
51	.503	30			-.031	.011	.107	.205	.303	.402	106	.984	135	-.608	-.602	-.601	-.613	-.642	-.639		
52	.503	50			-.009	.029	.119	.212	.303	.396	107	.984	180	-.524	-.521	-.518	-.545	-.586	-.611		
53	.503	F			.007	.043	.129	.208	.279	.353											
54	.503	F			.033	.048	-.035	-.171	-.295	-.524											

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(c) $M = 0.95$ Pressure coefficients C_p on -

Model orifice number	$\frac{x}{l}$	ϕ , deg	Fuselage							Model orifice number	$\frac{x}{l}$	ϕ , deg	Fuselage - Concluded								
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$	
			$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$										$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$								
1	.026	0				.225	.306	.410	.527	.655	55	.503	F				.027	.009	-.104	-.249	-.309
2	.026	90				.201	.169	.087	-.027	-.199	56	.503	135				.015	.013	.028	-.053	-.410
3	.026	180				.223	.147	.103	.065	.043	57	.588	0				-.036	.084	.201	.317	.428
4	.084	0				.068	.144	.241	.358	.487	58	.588	180				-.031	-.179	-.297	-.378	-.494
5	.045	84				.047	.065	.051	.047	.049	59	.604	15				-.056	.066	.179	.288	.393
6	.090	84				.047	.013	-.039	-.145	-.338	60	.604	30				-.047	.081	.196	.304	.399
7	.084	135				.031	-.001	-.038	-.096	-.158	61	.604	50				-.053	.088	.218	.338	.443
8	.084	180				.077	.013	-.020	-.052	-.085	62	.604	F				-.054	.102	.249	.380	.491
9	.134	J				.119	.065	.037	.020	-.025	63	.604	135				-.053	-.256	-.457	-.858	-1.018
10	.156	J				.071	.034	.024	.046	.106	64	.604	0				-.046	-.226	-.399	-.602	-.795
12	.173	30				-.089	-.066	-.037	.005	.068	65	.690	0				-.011	.025	.093	.191	.299
13	.173	60				-.093	-.098	-.132	-.193	-.277	66	.690	180				-.034	-.305	-.509	-.618	-.795
14	.173	90				-.084	-.091	-.137	-.230	-.348	67	.706	15				.023	.037	.074	.154	.250
15	.173	135				-.058	-.054	-.052	-.073	-.098	68	.706	30				.027	.043	.079	.156	.250
16	.173	J				-.004	-.022	-.014	.017	.053	69	.706	50				.031	.051	.090	.172	.266
17	.173	C				.034	.000	-.008	.004	.018	70	.706	F				.044	.057	.100	.186	.279
18	.190	J				-.086	-.083	-.061	-.018	-.027	71	.706	F				-.001	-.113	-.487	-.567	-1.046
19	.190	C				-.091	-.114	-.110	-.098	-.051	72	.706	135				.006	-.106	-.536	-.674	-.858
20	.197	C				-.272	-.318	-.340	-.343	-.325	73	.753	0				.173	.190	.186	.196	.275
21	.207	C				-.167	-.172	-.167	-.151	-.072	74	.753	180				.158	.145	-.038	-.290	-.610
23	.216	C				-.426	-.464	-.476	-.473	-.464	75	.770	15				.012	.024	.023	.053	.101
24	.226	23				-.156	-.124	-.086	-.021	.065	76	.770	30				.034	.037	-.001	.017	.068
25	.226	45				-.157	-.149	-.151	-.155	-.148	77	.770	50				.050	.060	-.002	-.003	.057
26	.226	68				-.191	-.188	-.219	-.288	-.379	78	.770	F				.076	.086	.000	-.058	.001
27	.226	90				-.223	-.203	-.226	-.288	-.363	79	.770	F				.112	.130	.020	-.230	-.387
28	.226	135				-.475	-.439	-.400	-.399	-.457	80	.770	F				.063	.080	.010	-.502	-.472
29	.226	C				-.616	-.663	-.736	-.804	-.853	81	.770	135				.034	.064	.037	-.434	-.566
30	.263	C				-.325	-.318	-.175	-.085	-.083	82	.820	15				-.017	.010	-.008	-.130	.045
31	.283	0				-.207	-.186	-.089	.038	.173	83	.820	38				-.008	.024	.008	-.073	.048
32	.299	30				-.083	-.105	-.131	-.079	-.005	84	.820	F				-.001	.031	.019	-.042	.089
33	.299	60				-.060	-.046	-.070	-.034	.042	85	.820	F				.015	.044	.028	-.031	.088
34	.299	F				-.054	-.033	-.097	-.113	-.074	86	.820	F								
35	.299	F				-.029	-.077	-.144	-.534	-.634	87	.820	F				.072	.085	.069	.032	.125
36	.299	F				-.046	-.123	-.222	-.425	-.822	88	.820	F				.003	.015	-.022	-.159	-.129
37	.299	135				-.015	-.033	-.063	-.115	-.135	89	.820	F				-.003	.009	-.030	-.203	-.167
38	.319	C				.004	-.016	-.030	-.042	-.097	90	.820	F				-.013	-.013	-.046	-.173	-.223
39	.350	20				-.042	-.034	-.009	.037	.124	91	.820	F				-.028	-.024	-.050	-.134	-.115
40	.350	40				-.029	-.021	-.011	.006	.046	92	.820	142				-.012	-.010	-.011	-.013	-.143
41	.350	60				-.029	-.003	.037	.096	.157	94	.897	30				-.129	-.096	-.045	-.001	.065
42	.350	F				-.030	-.004	.000	.003	-.013	95	.897	50				-.035	-.100	-.041	.013	.081
43	.350	F				-.028	-.074	-.222	-.533	-.649	96	.897	F				-.134	-.097	-.035	.027	.098
44	.350	F				-.019	-.051	-.111	-.229	-.840	97	.897	F				-.147	-.162	-.170	-.224	-.355
45	.350	135				.007	.004	-.021	-.036	-.055	98	.897	130				-.131	-.138	-.141	-.190	-.328
46	.367	0				-.048	-.018	.041	.125	.237	99	.897	150				-.132	-.136	-.134	-.186	-.299
47	.367	180				-.014	-.028	-.010	.018	.010	102	.984	0				-.326	-.327	-.305	-.282	-.272
48	.486	0				-.012	.037	.125	.232	.348	104	.984	60				-.331	-.324	-.301	-.280	-.359
49	.486	180				-.013	-.020	.023	.014	-.011	105	.984	90				-.340	-.336	-.316	-.308	-.408
50	.503	15				.003	.048	.124	.215	.316	106	.984	135				-.313	-.324	-.273	-.268	-.353
51	.503	30				.009	.045	.108	.179	.264	107	.984	180				-.393	-.397	-.361	-.368	-.483
52	.503	50				.014	.061	.132	.211	.299											
53	.503	F				.030	.064	.119	.178	.251											
54	.503	F				.028	-.012	-.082	-.259	-.287											

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(c) $M = 0.95$ - ContinuedPressure coefficients C_p on -

Model orifice number	$\frac{x}{l}$	ϕ , deg	Fuselage								Model orifice number	$\frac{x}{l}$	ϕ , deg	Fuselage - Concluded							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$											$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
1	.026	0	.035	.093	.181	.228	.317	.418	.530	.658	55	.503	F	.237	.115	.050	.035	.012	-.111	-.243	-.311
2	.026	90	-.224	.077	.176	.203	.159	.083	-.032	-.217	56	.503	135	.257	.115	.045	.021	.022	.027	-.057	-.443
3	.026	180	.668	.424	.274	.223	.146	.101	.064	.038	57	.588	0	-.477	-.320	-.107	-.019	.112	.217	.325	.436
4	.084	0	-.098	-.028	.036	.072	.155	.251	.361	.493	58	.588	180	.425	.202	.053	-.021	-.187	-.302	-.370	-.505
5	.084	45	-.169	-.020	.040	.046	.070	.054	.047	.042	59	.604	15	-.533	-.359	-.130	-.032	.095	.195	.298	.400
6	.084	90	-.353	-.052	.036	.050	.021	-.043	-.148	-.357	60	.604	30	-.614	-.364	-.126	-.024	.112	.212	.311	.407
7	.084	135	.027	.033	.041	.032	.000	-.039	-.100	-.169	61	.604	50	-.768	-.401	-.132	-.030	.120	.236	.347	.451
8	.084	180	.510	.263	.124	.076	.020	-.022	-.051	-.089	62	.604	F	-.966	-.461	-.159	-.032	.134	.266	.390	.498
9	.134	J	.490	.276	.158	.119	.066	.038	.019	.032	63	.604	F	.492	.253	.062	-.039	-.270	-.458	-.866	-1.017
10	.156	J	.334	.161	.093	.072	.037	.030	.047	.104	64	.604	135	.424	.216	.053	-.032	-.237	-.410	-.609	-.801
12	.173	30	-.266	-.150	-.102	-.085	-.058	-.029	.011	.064	65	.690	0	-.828	-.532	.029	.037	.080	.116	.206	.314
13	.173	60	-.343	-.186	-.102	-.090	-.094	-.131	-.191	-.288	66	.690	180	.278	.073	.039	.017	-.083	-.394	-.501	-.759
14	.173	90	-.539	-.227	-.100	-.083	-.086	-.137	-.234	-.360	67	.706	15	-.771	-.539	.064	.066	.090	.099	.172	.269
15	.173	135	-.165	-.124	-.068	-.057	-.048	-.048	-.074	-.107	68	.706	30	-.811	-.521	.066	.070	.095	.103	.173	.265
16	.173	J		.007	-.004	-.018	-.009	.018	.047	.069	69	.706	50	-.865	-.528	.075	.074	.104	.115	.188	.282
17	.173	C	.275	.103	.050	.035	.003	-.004	.003	.017	70	.706	F	-.971	-.506	.083	.084	.111	.124	.200	.293
18	.190	J	.016	-.089	-.085	-.083	-.076	-.053	-.017	-.032	71	.706	F	.235	.053	.055	.041	.020	-.394	-.742	-1.071
19	.190	C	.114	-.033	-.074	-.089	-.109	-.106	-.096	-.054	72	.706	135	.225	.053	.056	.050	.037	-.402	-.533	-.850
20	.197	C	.112	-.122	-.234	-.270	-.315	-.338	-.342	-.330	73	.753	0	-.727	-.013	.200	.214	.236	.210	.230	.307
21	.207	C	-.029	-.149	-.160	-.165	-.167	-.163	-.150	-.075	74	.753	180	.237	.165	.211	.194	.192	.099	-.171	-.627
23	.216	C	-.024	-.285	-.394	-.428	-.463	-.476	-.472	-.469	75	.770	15	-.666	-.012	.058	.053	.064	.046	.063	.120
24	.226	23	-.059	-.216	-.170	-.153	-.119	-.078	-.016	.068	76	.770	30	-.654	.017	.091	.088	.091	.034	.029	.095
25	.226	45	-.156	-.243	-.172	-.155	-.143	-.146	-.151	-.157	77	.770	50	-.469	.020	.106	.107	.122	.049	.015	.094
26	.226	68	-.183	-.325	-.213	-.190	-.184	-.218	-.288	-.391	78	.770	F	-.447	.031	.141	.130	.144	.052	-.036	.034
27	.226	90	-.469	-.401	-.253	-.222	-.203	-.227	-.289	-.372	79	.770	F	-.491	-.057	.170	.158	.175	.049	-.133	-.172
28	.226	135	-.409	-.550	-.496	-.476	-.433	-.396	-.402	-.467	80	.770	F	-.096	-.043	.118	.112	.120	.027	-.107	-.241
29	.226	C	-.325	-.490	-.578	-.618	-.664	-.738	-.808	-.854	81	.770	135	.007	-.037	.081	.080	.104	.049	-.028	-.146
30	.263	C	.084	-.036	-.154	-.206	-.238	-.110	-.065	-.090	82	.820	15	-.099	-.059	.097	.097	.112	.107	.100	.152
31	.283	0	-.209	-.127	-.113	-.111	-.166	-.068	.046	.180	83	.820	38	-.066	-.019	.099	.112	.135	.129	.109	.160
32	.299	30	-.246	-.088	-.085	-.090	-.098	-.112	-.074	-.006	84	.820	F	-.019	.007	.112	.119	.144	.145	.134	.200
33	.299	60	-.193	-.138	-.077	-.060	-.036	-.044	-.027	.043	85	.820	F	-.100	.018	.126	.129	.154	.154	.133	.196
34	.299	F	-.438	-.239	-.089	-.054	-.026	-.057	-.108	-.076	86	.820	F	-.213	.063	.175	.098	.171	.186	.189	.232
35	.299	F	-.624	-.183							87	.820	F	-.175	.098	.168	.171	.186	.189	.167	.232
36	.299	F	-.117	-.020	-.011	-.044	-.125	-.241	-.399	-.870	88	.820	F	-.183	-.145	.095	.092	.089	.041	-.095	-.111
37	.299	135			.002	-.011	-.028	-.069	-.114	-.152	89	.820	F	-.189	-.152	.103	.092	.086	.038	-.136	-.145
38	.319	C	.203	.035	.014	.002	.014	-.025	-.044	-.105	90	.820	F	-.216	-.169	.099	.088	.072	.014	-.169	-.258
39	.350	20	-.006	-.012	-.035	-.038	.025	.001	.044	.125	91	.820	F	-.238	-.198	.082	.076	.064	.007	-.164	-.216
40	.350	40	-.106	-.044	-.027	-.024	-.013	-.003	.013	.050	92	.820	142	-.294	-.235	.071	.056	.050	-.004	-.165	-.186
41	.350	60	-.069	.066	-.035	-.025	.007	.047	.101	.163	93	.820	165	-.222	-.256						
42	.350	F	-.553	-.127	-.047	-.025	.004	.006	.003	-.010	94	.897	30	-.184	-.103	.263	.325	.404	.434	.448	.479
43	.350	F	-.820	-.245	-.040	-.025	-.077	-.233	-.553	-.636	95	.897	50	-.187	-.123	.258	.268	.345	.344	.363	.401
44	.350	F	-.003	-.003	.002	-.017	-.048	-.114	-.232	-.871	96	.897	F	-.204	-.133	.202	.225	.262	.290	.314	.349
45	.350	135	.080	.026	.021	.008	.008	-.022	-.031	-.061	97	.897	F	-.215	-.120	.169	.148	.109	.133	.036	-.122
46	.367	0	-.051	-.055	-.046	-.041	-.006	.051	.132	.244	98	.897	130	-.205	-.110	.241	.256	.213	.273	.173	.041
47	.367	180	.278	.079	.011	-.010	-.022	-.005	.019	.008	99	.897	150	-.199	-.114	.345	.307	.309	.383	.318	.202
48	.486	0	-.001	-.021	-.006	-.001	.057	.139	.241	.356	102	.984	0	-.400	-.298	-.552	-.562	-.546	-.505	-.499	-.478
49	.486	180	.346	.126	.023	-.003	-.014	-.019	.013	-.016	103	.984	30	-.357	-.279						
50	.503	15	.002	.042	.003	.013	.066	.136	.222	.322	104	.984	60	-.289	-.278	-.699	-.710	-.710	-.652	-.655	-.715
51	.503	30	-.154	.053	.010	.019	.063	.120	.187	.267	105	.984	90	-.293	-.291	-.720	-.713	-.721	-.677	-.695	-.741
52	.503	50	-.283	.040	.017	.024	.079	.144	.218	.304	106	.984	135	-.305	-.308	-.649	-.620	-.712	-.598	-.647	-.714
53	.503	F	-.266	-.054	.034	.040	.080	.130	.184	.255	107	.984	180	-.367	-.383	-.589	-.595	-.598	-.570	-.595	-.657
54	.503	F	-.219	-.074	.025	.036	-.008	-.088	-.279	-.286											

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(c) $M = 0.95$ - ContinuedPressure coefficients C_p on -

Model orifice number	$\frac{x}{l}$	ϕ , deg	Fuselage								Model orifice number	$\frac{x}{l}$	ϕ , deg	Fuselage - Concluded							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
1	.026	0	.054	.114		.243	.329	.431	.543	.661	55	.503	F	.295	.159		.025	.016	-.173	-.244	-.227
2	.026	90	-.100	.160		.255	.229	.159	.048	-.124	56	.503	135	.331	.167		.020	.014	.053	.044	-.422
3	.026	180	.671	.441		.242	.179	.117	.078	.051	57	.588	0	-.391	-.293		-.017	.113	.228	.340	.447
4	.084	0	-.057	-.006		.104	.187	.280	.388	.509	58	.588	180	.452	.235		-.017	-.171	-.282	-.359	-.391
5	.084	45	-.185	-.017		.090	.135	.181	.205	.222	59	.604	15	-.596	-.363		-.042	.094	.212	.325	.433
6	.084	90	-.258	.000		.094	.072	.006	-.094	-.241	60	.604	30	-.812	-.370		-.039	.103	.226	.338	.443
7	.084	135	.202	.152		.085	.025	-.045	-.125	-.217	61	.604	50	-.875	-.387		-.048	.102	.232	.351	.463
8	.084	180	.523	.297		.096	.045	-.005	-.032	-.056	62	.604	F	-1.037	-.438		-.059	.110	.251	.382	.496
9	.134	J	.497	.291		.133	.081	.051	.031	.018	63	.604	F	.504	.251		-.068	-.279	-.491	-.805	-.980
10	.156	J	.451	.291		.193	.153	.127	.116	.099	64	.604	135	.456	.231		-.051	-.237	-.389	-.587	-1.014
12	.173	30	-.260	-.150		-.078	-.048	.013	.081	.171	65	.690	0	-.671	-.490		-.006	.035	.103	.202	.310
13	.173	60	-.435	-.189		-.046	-.023	-.035	-.065	-.125	66	.690	180	.295	.087		-.039	-.335	-.488	-.573	-.739
14	.173	90	-.455	.159		-.025	-.031	-.083	-.179	.325	67	.706	15	-.731	-.499		.039	.055	.090	.174	.276
15	.173	135	.016	.023		.034	.021	.002	-.019	-.058	68	.706	30	-.567	-.246		.046	.063	.095	.178	.277
16	.173	J	.302	.175		.034	.092	.082	.080	.056	69	.706	50	-.897	-.491		.056	.066	.093	.176	.273
17	.173	J	.400	.243		.160	.127	.110	.114	.135	70	.706	F	-.998	-.492		.068	.073	.097	.182	.278
18	.190	J	.135	.045		.024	.016	.013	.004	-.051	71	.706	F	.237	.047		.021	-.138	-.464	-.739	-1.054
19	.190	C	.219	.074		.008	-.018	-.024	-.010	.021	72	.706	135	.245	.058		.016	-.162	-.501	-.471	-.964
20	.197	C	.135	-.096		-.249	-.290	-.312	-.322	-.315	73	.753	0	-.655	.060		.161	.174	.154	.167	.256
21	.207	C	.054	-.056		-.082	-.094	-.093	-.071	-.034	74	.753	180	.222	.128		.154	.169	.063	-.508	-.753
23	.216	C	-.005	-.250		-.395	-.431	-.459	-.469	-.461	75	.770	15	-.440	.134		.206	.228	.191	.181	.252
24	.226	23	-.149	-.226		-.124	-.069	-.006	.071	.163	76	.770	30	-.623	.062		.173	.186	.135	.109	.180
25	.226	45	-.152	-.249		-.123	-.082	-.052	-.028	-.008	77	.770	50	-.551	.056		.152	.157	.095	.042	.113
26	.226	68	-.213	-.337		-.156	-.132	-.138	-.172	-.237	78	.770	F	-.485	.064		.155	.155	.074	-.045	.021
27	.226	90	-.582	-.362		-.178	-.157	-.176	-.234	-.328	79	.770	F	-.013	-.013		.185	.174	.077	-.262	-.455
28	.226	135	-.258	-.449		-.399	-.370	-.352	-.341	-.388	80	.770	F	-.082	.040		.143	.143	.058	-.288	-.478
29	.226	C	-.279	-.457		-.683	-.806	-.873	-.878	-.845	81	.770	135	.071	.068		.145	.148	.060	-.331	-.779
30	.263	C	.143	.054		-.259	-.272	-.157	-.055	-.022	82	.820	15	.037	.124		.111	.123	.093	.025	.109
31	.283	0	-.080	-.098		-.117	-.104	-.038	.067	.191	83	.820	38	-.052	.112		.114	.129	.102	.033	.110
32	.299	30	-.275	-.122		-.086	-.091	-.065	.016	.109	84	.820	F	.006	.115		.113	.131	.112	.059	.146
33	.299	60	-.255	-.205		-.082	-.057	-.061	.017	.099	85	.820	F	-.239	.111		.107	.129	.112	.059	.141
34	.299	F	-.839	-.321		-.081	-.042	-.105	-.054	-.013	86	.820	F	-.333	.141		.126	.151	.135	.096	.185
35	.299	F	-.552	-.184		-.033	-.075	-.369	-.442	-.542	87	.820	F	-.210	.164		.142	.170	.155	.157	.337
36	.299	F	-.046	-.005		-.078	-.173	-.354	-.876	-1.094	88	.820	F	-.130	.005		.099	.083	.021	-.188	-.342
37	.299	135	.114	.033		-.041	-.078	-.092	-.167	-.319	89	.820	F	-.150	.015		.094	.093	.039	-.169	-.221
38	.319	C	.238	.076		.018	.015	.022	.042	.026	90	.820	F	-.161	.020		.094	.091	.047	-.173	-.292
39	.350	20	-.189	-.038		-.030	.003	.049	.109	.204	91	.820	F	-.133	.018		.097	.100	.059	-.099	-.262
40	.350	40	-.093	-.056		-.029	.005	.042	.077	.142	92	.820	142				.101	.105	.076	-.083	-.085
41	.350	60	-.039	-.092		-.039	.008	.062	.120	.197	93	.820	165	-.044	.037						
42	.350	F	-.811	-.187		-.044	.004	.032	.044	.050	94	.897	30	-.272	-.082		-.095	-.065	-.019	.035	.099
43	.350	F	-.845	-.223		-.022	-.071	-.220	-.388	-.760	95	.897	50	-.250	-.079		-.102	-.069	-.013	.051	.119
44	.350	F	.066	.031		-.032	-.091	-.153	-.288	-.867	96	.897	F	-.277	-.086		-.112	-.076	-.013	.056	.129
45	.350	135	.168	.072		.005	-.016	-.023	-.023	-.152	97	.897	F	-.013	-.012		-.092	-.106	-.142	-.255	-.448
46	.367	0	.021	-.004		-.031	.003	.063	.144	.252	98	.897	130	-.163	-.093		-.078	-.089	-.114	-.222	-.355
47	.367	180	.300	.105		.019	.035	.063	.084	.070	99	.897	150	-.172	-.097		-.077	-.090	-.125	-.186	-.330
48	.486	0	.022	.060		.006	.062	.150	.254	.366	102	.984	0	-.364	-.293		-.275	-.267	-.255	-.248	-.242
49	.486	180	.371	.155		.007	.014	.055	.082	.096	103	.984	30	-.361	-.282		-.271	-.266	-.253	-.241	-.251
50	.503	15	-.071	.036		.018	.085	.174	.272	.378	104	.984	60	-.299	-.274		-.281	-.284	-.275	-.278	-.386
51	.503	30	-.098	.044		.021	.088	.169	.256	.348	105	.984	90	-.300	-.278		-.296	-.301	-.295	-.307	-.416
52	.503	50	-.177	.063		.020	.088	.171	.259	.352	106	.984	135	-.309	-.228		-.271	-.278	-.262	-.287	-.362
53	.503	F	-.264	.136		.031	.093	.163	.231	.306	107	.984	180	-.321	-.313		-.319	-.323	-.333	-.362	-.441
54	.503	F	-.300	-.105		.030	-.014	-.111	-.188	-.345											

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(c) $M = 0.95$ - ContinuedPressure coefficients C_p on -

Model orifice number	$\frac{x}{l}$	β , deg	Fuselage							Model orifice number	$\frac{x}{l}$	β , deg	Fuselage - Concluded								
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$	
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$										$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$								
1	.026	0				.202	.243	.327	.427	.542	.661	55	.503	F		.061	.031	.019	-.165	-.243	-.214
2	.026	90				.251	.259	.235	.163	.052	-.114	56	.503	135	.056	.024	.017	.052	.050	-.405	
3	.026	180				.297	.248	.182	.118	.079	.055	57	.588	0	-.109	-.023	.107	.223	.339	-.445	
4	.084	0				.064	.103	.183	.277	.388	.508	58	.588	180	.067	-.004	-.162	-.281	-.359	-.385	
5	.084	45				.071	.093	.138	.177	.215	.231	59	.604	15	-.138	-.048	.089	.208	.326	.433	
6	.084	90				.089	.100	.077	.008	-.095	-.230	60	.604	30	-.140	-.044	.100	.222	.340	.445	
7	.084	135				.116	.091	.029	-.046	-.129	-.211	61	.604	50	-.156	-.052	.098	.228	.353	.462	
8	.084	180				.147	.102	.045	-.004	-.033	-.055	62	.604	F	-.185	-.066	.104	.248	.382	.498	
9	.134	J				.175	.138	.086	.049	.031	.022	63	.604	F	.049	-.051	-.268	-.491	-.804	-.962	
10	.156	J				.225	.201	.159	.129	.117	.102	64	.604	135	.049	-.037	-.226	-.382	-.582	-1.003	
12	.173	30				-.081	-.050	.005	.058	.122	.192	65	.690	0	-.050	.002	.039	.102	.204	.311	
13	.173	60				-.071	-.043	-.019	-.031	-.059	-.113	66	.690	180	.001	-.022	-.323	-.483	-.559	-.730	
14	.173	90				-.045	-.022	-.027	-.080	-.175	-.307	67	.706	15	.038	.048	.062	.092	.179	.280	
15	.173	135				.042	.041	.025	.001	-.017	-.053	68	.706	30	.045	.058	.072	.096	.182	.281	
16	.173	J				.136	.125	.100	.084	.082	.060	69	.706	50	.057	.069	.077	.095	.180	.278	
17	.173	J				.189	.167	.135	.113	.119	.142	70	.706	F	.063	.081	.086	.099	.185	.282	
18	.190	C				.033	.029	.020	.015	.003	-.043	71	.706	F	.045	.041	-.065	-.404	-.735	-1.012	
19	.190	C				.035	.016	-.011	-.024	-.008	.026	72	.706	135	.044	.035	-.081	-.464	-.553	-.951	
20	.197	C				-.211	-.247	-.290	-.321	-.325	-.312	73	.753	0	.168	.170	.184	.162	.181	.267	
21	.207	C				-.066	-.077	-.089	-.092	-.068	-.032	74	.753	180	.173	.168	.182	.083	-.495	-.690	
23	.216	C				-.364	-.393	-.437	-.461	-.468	-.458	75	.770	15	.227	.241	.262	.225	.212	.284	
24	.226	23				-.159	-.122	-.069	-.008	.075	.166	76	.770	30	.198	.206	.217	.169	.140	.211	
25	.226	45				-.165	-.124	-.079	-.051	-.021	.003	77	.770	50	.183	.184	.188	.126	.070	.140	
26	.226	68				-.200	-.155	-.126	-.135	-.166	-.226	78	.770	F	.184	.181	.182	.100	-.017	.044	
27	.226	90				-.219	-.173	-.152	-.173	-.224	-.322	79	.770	F	.199	.211	.199	.096	-.239	-.415	
28	.226	135				-.416	-.396	-.369	-.354	-.340	-.380	80	.770	F	.173	.173	.173	.083	-.235	-.476	
29	.226	C				-.608	-.676	-.808	-.877	-.877	-.839	81	.770	135	.185	.184	.184	.093	-.242	-.646	
30	.263	C				-.100	-.232	-.270	-.151	-.054	-.018	82	.820	15	.186	.188	.199	.167	.102	.175	
31	.283	0				-.090	-.093	-.101	-.037	.069	.193	83	.820	38	.168	.172	.182	.152	.083	.149	
32	.299	30				-.103	-.085	-.083	-.063	.020	.114	84	.820	F	.161	.163	.177	.157	.106	.181	
33	.299	60				-.117	-.082	-.053	-.064	.020	.102	85	.820	F	.150	.153	.170	.152	.099	.166	
34	.299	F				-.138	-.083	-.038	-.104	-.050	-.007	86	.820	F	.162	.166	.186	.171	.127	.202	
35	.299	F				-.044	-.029	-.068	-.363	-.435	-.530	87	.820	F	.176	.176	.202	.194	.188	.345	
36	.299	F				-.030	-.066	-.167	-.345	-.486	-1.058	88	.820	F	.145	.140	.122	.048	-.154	-.317	
37	.299	135				-.016	-.034	-.075	-.094	-.169	-.314	89	.820	F	.140	.140	.133	.070	-.144	-.199	
38	.319	C				.027	.020	.015	.018	.041	.033	90	.820	F	.144	.143	.140	.089	-.143	-.280	
39	.350	20				-.037	-.027	.003	.049	.112	.208	91	.820	F	.152	.154	.155	.109	-.062	-.272	
40	.350	40				-.044	-.029	.005	.042	.080	.144	92	.820	142	.171	.171	.173	.142	-.032	-.063	
41	.350	60				-.059	-.039	.006	.061	.120	.198	94	.897	30	-.063	-.059	-.041	-.009	.030	.086	
42	.350	F				-.083	-.045	.005	.032	.046	.054	95	.897	50	-.066	-.062	-.042	.002	.054	.115	
43	.350	F				-.044	-.018	-.063	-.214	-.380	-.735	96	.897	F	-.079	-.074	-.048	.006	.063	.129	
44	.350	F				.000	-.025	-.085	-.153	-.293	-.845	97	.897	F	-.048	-.049	-.063	-.100	-.224	-.401	
45	.350	135				.021	.007	-.016	-.023	-.021	-.147	98	.897	130	-.034	-.036	-.041	-.073	-.205	-.331	
46	.367	0				-.034	-.030	.002	.061	.144	.252	99	.897	150	-.039	-.035	-.045	-.083	-.171	-.323	
47	.367	180				.032	.026	.034	.060	.085	.076	102	.984	0	-.352	-.365	-.377	-.334	-.300	-.327	
48	.486	0				.004	.008	.060	.148	.254	.365	103	.984	30	-.326	-.337	-.353	-.321	-.288	-.337	
49	.486	180				.036	.014	.017	.054	.082	.098	104	.984	60	-.302	-.301	-.308	-.295	-.290	-.381	
50	.503	15				.006	.018	.082	.172	.274	.378	105	.984	90	-.307	-.309	-.319	-.309	-.320	-.422	
51	.503	30				.008	.021	.085	.169	.259	.350	106	.984	135	-.346	-.349	-.369	-.320	-.324	-.344	
52	.503	50				.011	.020	.086	.170	.261	.353	107	.984	180	-.423	-.422	-.416	-.409	-.432	-.525	
53	.503	F				.023	.032	.090	.162	.233	.308										
54	.503	F				.016	.035	-.007	-.108	-.189	-.347										

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(c) $M = 0.95$ - ContinuedPressure coefficients C_p on -

Model orifice number	$\frac{x}{l}$	β , deg	Fuselage								Model orifice number	$\frac{x}{l}$	β , deg	Fuselage - Concluded							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = -7.5^\circ; \text{brakes closed}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = -7.5^\circ; \text{brakes closed}$							
1	.026	0			.203	.242	.326	.431	.546	.659	55	.503	F			.058	.026	.011	-.172	-.234	-.224
2	.026	90			.244	.253	.222	.156	.047	-.123	56	.503	135			.053	.020	.011	.051	.044	-.404
3	.026	180			.295	.244	.176	.116	.082	.053	57	.588	0			-.106	-.023	.109	.228	.343	.445
4	.084	0			.064	.101	.183	.280	.390	.506	58	.588	180			.067	-.009	-.173	-.280	-.353	-.384
5	.084	45			.073	.089	.129	.180	.206	.218	59	.604	15			-.136	-.048	.090	.212	.329	.430
6	.084	90			.083	.093	.065	.002	-.095	-.241	60	.604	30			-.137	-.044	.100	.226	.342	.440
7	.084	135			.114	.084	.016	-.047	-.124	-.213	61	.604	50			-.153	-.055	.098	.232	.355	.460
8	.084	180			.148	.099	.038	-.005	-.057	-.241	62	.604	F			-.183	-.067	.105	.252	.385	.495
9	.134	0			.176	.135	.079	.049	.032	.019	63	.604	F			.048	-.058	-.284	-.492	-.795	-.962
10	.156	J			.216	.191	.142	.125	.116	.097	64	.604	135			.047	-.045	-.239	-.384	-.591	-.998
12	.173	30			-.088	-.062	-.011	.046	.109	.175	65	.690	0			-.070	-.002	.039	.108	.208	.309
13	.173	60			-.073	-.046	-.028	-.036	-.067	-.127	66	.690	180			-.007	-.029	-.335	-.485	-.561	-.715
14	.173	90			-.049	-.029	-.038	-.086	-.180	-.318	67	.706	15			.029	.041	.057	.096	.181	.276
15	.173	135			.035	.031	.014	-.002	-.018	-.059	68	.706	30			.031	.047	.063	.100	.183	.276
16	.173	J			.127	.114	.084	.079	.081	.058	69	.706	50			.039	.053	.064	.097	.181	.272
17	.173	C			.171	.145	.118	.107	.113	.130	70	.706	F			.038	.064	.071	.101	.187	.276
18	.190	J			.026	.021	.009	.011	.003	-.049	71	.706	F			.023	.018	-.119	-.548	-.455	-1.073
19	.190	C			.018	.006	-.021	-.029	-.011	.018	72	.706	135			.023	.014	-.131	-.546	-.600	-.959
20	.197	C			-.211	-.248	-.297	-.319	-.322	-.314	73	.753	0			.169	.167	.179	.174	.184	.266
21	.207	C			-.068	-.081	-.100	-.096	-.036	-.069	74	.753	180			.168	.162	.172	.058	-.431	-.713
23	.216	C			-.363	-.396	-.444	-.461	-.465	-.462	75	.770	15			.172	.181	.203	.189	.188	.242
24	.226	23			-.161	-.125	-.074	-.010	.070	.159	76	.770	30			.145	.150	.160	.134	.116	.168
25	.226	45			-.164	-.123	-.087	-.058	-.030	-.010	77	.770	50			.133	.129	.132	.091	.045	.099
26	.226	68			-.200	-.158	-.138	-.142	-.174	-.239	78	.770	F			.139	.133	.131	.077	-.041	.012
27	.226	90			-.221	-.180	-.165	-.177	-.232	-.330	79	.770	F			.157	.164	.158	.092	-.271	-.491
28	.226	135			-.419	-.403	-.378	-.356	-.342	-.388	80	.770	F			.121	.120	.122	.053	-.357	-.541
29	.226	C			-.600	-.674	-.799	-.864	-.867	-.837	81	.770	135			.126	.120	.120	.045	-.528	-.667
30	.263	C			-.211	-.256	-.235	-.157	-.066	-.016	82	.820	15			.005	.005	.015	-.007	-.301	-.003
31	.283	0			-.087	-.110	-.102	-.035	.068	.190	83	.820	38			.029	.032	.046	.033	-.197	.037
32	.299	30			-.098	-.084	-.082	-.065	.015	.105	84	.820	F			.043	.044	.061	.053	-.093	.083
33	.299	60			-.113	-.082	-.053	-.062	.016	.096	85	.820	F			.045	.046	.066	.063	-.063	.090
34	.299	F			-.131	-.080	-.039	-.107	-.055	-.016	86	.820	F			.075	.073	.096	.095	.012	.149
35	.299	F			-.043	-.033	-.080	-.371	-.443	-.544	87	.820	F			.100	.094	.116	.111	.097	.314
36	.299	F			-.026	-.070	-.181	-.325	-.894	-1.086	88	.820	F			.061	.051	.038	.011	-.239	-.336
37	.299	135			-.013	-.036	-.078	-.092	-.172	-.309	89	.820	F			.047	.042	.044	.024	-.206	-.244
38	.319	C			.030	.018	.007	.019	.044	.026	90	.820	F			.041	.035	.032	.018	-.215	-.306
39	.350	20			-.036	-.030	-.001	.049	.107	.201	91	.820	F			.028	.023	.026	.016	-.157	-.271
40	.350	40			-.042	-.031	.001	.041	.076	.137	92	.820	142			.012	.009	.008	.011	-.143	-.128
41	.350	60			-.058	-.041	.004	.062	.118	.194	94	.897	30			-.136	-.126	-.094	-.046	.012	.077
42	.350	F			-.081	-.046	.001	.031	.040	.047	95	.897	50			-.144	-.136	-.102	-.047	.023	.093
43	.350	F			-.044	-.022	-.076	-.224	-.373	-.758	96	.897	F			-.154	-.143	-.105	-.044	.031	.106
44	.350	F			-.001	-.029	-.091	-.152	-.295	-.864	97	.897	F			-.114	-.126	-.149	-.170	-.293	-.479
45	.350	135			.023	.006	-.019	-.022	-.020	-.163	98	.897	130			-.100	-.114	-.132	-.149	-.233	-.354
46	.367	0			-.035	.032	.000	.064	.145	.250	99	.897	150			-.101	-.111	-.131	-.156	-.193	-.308
47	.367	180			.032	.022	.027	.062	.086	.073	102	.984	0			-.348	-.357	-.343	-.337	-.318	-.334
48	.486	0			.003	.005	.059	.152	.257	.364	103	.984	30			-.313	-.319	-.310	-.306	-.303	-.330
49	.486	180			.036	.009	.011	.057	.085	.096	104	.984	60			-.302	-.305	-.305	-.294	-.290	-.364
50	.503	15			.006	.015	.081	.174	.275	.375	105	.984	90			-.307	-.312	-.313	-.310	-.313	-.404
51	.503	30			.009	.019	.083	.169	.257	.344	106	.984	135			-.280	-.281	-.277	-.268	-.271	-.354
52	.503	45			.012	.018	.082	.171	.260	.348	107	.984	180			-.391	-.391	-.380	-.384	-.414	-.498
53	.503	F			.022	.029	.088	.162	.231	.302											
54	.503	F			.017	.031	-.017	-.112	-.185	-.341											

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(c) $M = 0.95$ - ContinuedPressure coefficients C_p on -

Model orifice number	$\frac{x}{l}$	β , deg	Fuselage								Model orifice number	$\frac{x}{l}$	β , deg	Fuselage - Concluded							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 7.5^\circ; \text{brakes closed}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 7.5^\circ; \text{brakes closed}$							
1	.026	0			.204	.243	.329	.429	.540	.660	55	.503	F			.068	.037	.022	-.151	-.263	-.215
2	.026	90			.249	.258	.229	.164	.057	-.116	56	.503	135			.063	.030	.021	.054	.052	-.395
3	.026	180			.297	.248	.180	.120	.082	.054	57	.588	0			-.100	-.011	.122	.232	.341	.448
4	.084	0			.065	.104	.186	.278	.385	.507	58	.588	180			.081	.007	-.162	-.275	-.353	-.388
5	.084	45			.075	.090	.134	.179	.210	.225	59	.604	15			-.130	-.033	.107	.216	.327	.436
6	.084	90			.087	.097	.073	.010	-.088	-.231	60	.604	30			-.131	-.029	.117	.230	.339	.446
7	.084	135			.114	.089	.028	-.040	-.117	-.206	61	.604	50			-.149	-.039	.116	.236	.353	.465
8	.084	180			.147	.102	.046	.000	-.030	-.053	62	.604	F			-.179	-.055	.122	.256	.382	.501
9	.134	J			.177	.140	.086	.053	.033	.022	63	.604	F			.061	-.039	-.266	-.460	-.654	-.961
10	.156	J			.220	.199	.153	.129	.117	.099	64	.604	135			.062	-.024	-.225	-.367	-.514	-1.004
12	.173	30			-.081	-.052	.003	.053	.114	.189	65	.690	0			.034	.046	.087	.121	.210	.318
13	.173	60			-.071	-.042	-.022	-.031	-.060	-.116	66	.690	180			.046	.027	-.077	-.385	-.481	-.718
14	.173	90			-.046	-.025	-.030	-.076	-.165	-.305	67	.706	15			.081	.084	.109	.113	.185	.289
15	.173	135			.039	.037	.022	.002	.017	-.052	68	.706	30			.087	.091	.114	.117	.190	.288
16	.173	J			.132	.121	.093	.083	.083	.059	69	.706	50			.098	.100	.119	.115	.187	.285
17	.173	C			.186	.166	.130	.113	.118	.140	70	.706	F			.105	.109	.127	.119	.192	.288
18	.190	J			.028	.027	.016	.014	.005	-.046	71	.706	F			.076	.069	.038	-.304	-.582	-.732
19	.190	C			.031	.013	-.014	-.025	-.009	.024	72	.706	135			.077	.066	.041	-.319	-.490	-.809
20	.197	C			-.210	-.247	-.292	-.317	-.324	-.312	73	.753	0			.203	.206	.224	.189	.199	.284
21	.207	C			-.069	-.078	-.093	-.092	-.071	-.035	74	.753	180			.204	.197	.207	.130	-.326	-.453
23	.216	C			-.361	-.392	-.437	-.459	-.469	-.461	75	.770	15			.239	.251	.275	.237	.224	.286
24	.226	23			-.157	-.122	-.069	-.010	.072	.165	76	.770	30			.215	.222	.237	.186	.158	.218
25	.226	45			-.163	-.127	-.084	-.053	-.031	.000	77	.770	50			.203	.204	.212	.148	.098	.153
26	.226	68			-.198	-.157	-.128	-.134	-.166	-.230	78	.770	F			.206	.204	.209	.123	.023	.057
27	.226	90			-.215	-.177	-.156	-.173	-.224	-.325	79	.770	F			.225	.231	.227	.107	-.114	-.394
28	.226	135			-.418	-.400	-.373	-.352	-.340	-.383	80	.770	F			.193	.193	.191	.087	-.144	-.464
29	.226	C			-.601	-.674	-.800	-.869	-.880	-.839	81	.770	135			.198	.193	.193	.104	-.104	-.407
30	.263	C			-.078	-.158	-.181	-.113	-.037	-.006	82	.820	15			.216	.221	.231	.212	.189	.234
31	.283	0			-.087	-.092	-.092	-.032	.068	.195	83	.820	38			.212	.218	.231	.215	.190	.229
32	.299	30			-.099	-.084	-.072	-.058	.017	.113	84	.820	F			.211	.216	.231	.222	.210	.261
33	.299	60			-.114	-.081	-.043	-.052	.018	.103	85	.820	F			.203	.207	.226	.221	.207	.253
34	.299	F			-.141	-.084	-.035	-.089	-.054	-.011	86	.820	F			.214	.219	.241	.237	.226	.285
35	.299	F			-.045	-.029	-.074	-.335	-.428	-.532	87	.820	F			.228	.229	.254	.260	.261	.412
36	.299	F			-.036	-.065	-.170	-.224	-.742	-1.051	88	.820	F			.191	.184	.158	.077	-.090	-.305
37	.299	135			-.013	-.034	-.070	-.088	-.152	-.297	89	.820	F			.185	.184	.171	.097	-.086	-.188
38	.319	C			.029	.023	.013	.020	.039	.030	90	.820	F			.191	.188	.173	.099	-.100	-.292
39	.350	20			-.034	-.024	.007	.052	.111	.209	91	.820	F			.193	.194	.183	.106	-.077	-.249
40	.350	40			-.039	-.024	.009	.045	.082	.146	92	.820	142			.198	.200	.188	.116	-.078	-.174
41	.350	60			-.054	-.034	.011	.064	.124	.200	94	.897	30			.408	.422	.455	.483	.498	.523
42	.350	F			-.077	-.040	.009	.035	.056	.055	95	.897	50			.343	.352	.366	.392	.417	.449
43	.350	F			-.044	-.018	-.064	-.195	-.381	-.740	96	.897	F			.270	.279	.297	.329	.350	.387
44	.350	F			.003	-.023	-.084	-.147	-.263	-.827	97	.897	F			.224	.224	.200	.143	.008	-.205
45	.350	135			.025	.009	-.013	-.019	-.019	-.131	98	.897	130			.314	.305	.300	.245	.116	-.024
46	.367	0			-.030	-.024	.008	.067	.146	.254	99	.897	150			.424	.416	.413	.356	.284	.187
47	.367	180			.037	.028	.034	.064	.086	.075	102	.984	0			-.564	-.562	-.555	-.526	-.495	-.496
48	.486	0			.009	.016	.070	.154	.254	.368	103	.984	30			-.666	-.655	-.579	-.514	-.464	-.436
49	.486	180			.043	.020	.021	.058	.080	.097	104	.984	60			-.690	-.683	-.697	-.678	-.641	-.702
50	.503	15			.012	.027	.092	.178	.274	.380	105	.984	90			-.700	-.707	-.719	-.716	-.704	-.742
51	.503	30			.014	.029	.094	.174	.259	.351	106	.984	135			-.720	-.715	-.713	-.603	-.615	-.689
52	.503	50			.017	.028	.094	.174	.259	.353	107	.984	180			-.592	-.599	-.610	-.592	-.585	-.629
53	.503	F			.028	.040	.100	.167	.233	.309											
54	.503	F			.020	.038	-.004	-.095	-.180	-.350											

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(c) $M = 0.95$ - ContinuedPressure coefficients C_p on -

Model orifice number	$\frac{x}{l}$	ϕ , deg	Fuselage								Model orifice number	$\frac{x}{l}$	ϕ , deg	Fuselage - Concluded							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$											$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
1	.026	0	.031	.093		.231	.326	.432	.550	.685	55	.503	F	.359	.206	.023	.016	-.226	-.203	-.242	
2	.026	90	.011	.253		.342	.301	.219	.090	-.091	56	.503	135	.400	.207	.009	.010	.046	.027	-.499	
3	.026	180	.655	.419		.203	.132	.091	.057	.029	57	.588	0	-.491	-.326	-.028	.098	.216	.332	-.448	
4	.084	0	-.101	-.029		.073	.162	.263	.382	.521	58	.588	180	.429	.212	-.059	-.212	-.321	-.401	-.560	
5	.084	45	-.344	-.052		.148	.217	.272	.319	.356	59	.604	15	-.677	-.375	-.051	.088	.216	.340	.461	
6	.084	90	-.177	.078		.197	.140	.087	-.054	-.220	60	.604	30	-.798	-.376	-.044	.052	.175	.328	.464	
7	.084	135	.349	.256		.107	.012	-.096	-.222	-.369	61	.604	50	-.940	-.401	-.034	.121	.255	.383	.506	
8	.084	180	.500	.264		.072	.009	-.035	-.068	-.105	62	.604	F	-1.008	-.478	-.038	.130	.275	.412	.541	
9	.134	J	.486	.277		.105	.057	.028	.008	-.022	63	.604	F	.521	.263	-.104	-.326	-.629	-.841	-.978	
10	.156	J	.406			.289	.237	.188	.127	.075	64	.604	135	.479	.238	-.091	-.278	-.410	-.671	-1.007	
12	.173	30	-.313	-.199		-.026	.054	.132	.216	.308	65	.690	0	-.829	-.532	-.012	.031	.099	.203	.319	
13	.173	60	-.535	-.204							66	.690	180	.279	.077	-.043	-.382	-.539	-.666	-.786	
14	.173	90	-.370	-.091		.049	.029	-.034	-.153	-.339	67	.706	15	-.881	-.498	.035	.059	.099	.193	.307	
15	.173	135	.181	.152		.111	.068	.023	-.014	-.044	68	.706	30	-.800	-.464	.059	.080	.113	.207	.319	
16	.173	J	.428	.295		.214	.182	.147	.099	.054	69	.706	50	-.736	-.497	.088	.097	.119	.210	.316	
17	.173	C	.522	.370		.255	.219	.202	.212	.221	70	.706	F	-.801	-.538	.107	.107	.122	.212	.316	
18	.190	J	.264	.161		.117	.096	.057	.002	-.062	71	.706	F	.252	.073	.063	-.128	-.519	-.797	-.929	
19	.190	C	.342	.188		.090	.061	.053	.073	.115	72	.706	135	.267	.084	.049	-.176	-.569	-.797	-.929	
20	.197	C	.094	-.133		-.313	-.348	-.354	-.344	-.321	73	.753	0	-.653	-.018	.171	.182	.165	.194	.287	
21	.207	C	.160	.039		-.013	-.027	-.022	.012	.031	74	.753	180	.235	.154	.156	.130	-.085	-.380	-.645	
23	.216	C	-.016	-.272		-.448	-.480	-.482	-.472	-.465	75	.770	15	-.599	.112	.362	.365	.331	.321	.398	
24	.226	23	-.171	-.272		-.127	-.040	.050	.153	.267	76	.770	30	-.628	.121	.279	.281	.238	.214	.297	
25	.226	45	-.195	-.315		-.095	-.021	.036	.088	.132	77	.770	50	-.491	.133	.227	.227	.169	.115	.197	
26	.226	68	-.584	-.339		-.109	-.061	-.050	-.072	-.131	78	.770	F	-.439	.143	.210	.209	.122	.086	.079	
27	.226	90	-.474	-.309		-.112	-.094	-.116	-.180	-.268	79	.770	F	-.489	.143	.247	.227	.065	-.280	-.376	
28	.226	135	.007	-.348		-.307	-.299	-.293	-.299	-.326	80	.770	F	-.055	.138	.211	.189	.090	-.332	-.560	
29	.226	C	-.060	-.487		-.870	-.896	-.886	-.853	-.791	81	.770	135	.141	.187	.235	.210	.091	-.202	-.735	
30	.263	C	.105	-.014		-.362	-.298	-.193	-.110	-.131	82	.820	15	-.081	.233	.220	.220	.196	.153	.226	
31	.283	0	-.201	-.127		-.231	-.164	-.055	.069	.211	83	.820	38	-.283	.206	.196	.202	.194	.166	.241	
32	.299	30	-.217	-.184		-.136	-.126	-.016	.100	.220	84	.820	F	-.232	.176	.177	.188	.189	.161	.233	
33	.299	60	-.477	-.304		-.132	-.127	-.015	.092	.191	85	.820	F	-.383	.163	.187	.202	.212	.194	.263	
34	.299	F	-1.023	-.436		-.123	-.134	-.044	.028	.078	86	.820	F	-.391	.181	.187	.202	.212	.194	.263	
35	.299	F	-.473	-.177		-.057	-.248	-.288	-.360	-.461	87	.820	F	-.359	.192	.197	.214	.252	.289	.387	
36	.299	F	.043	.022		-.126	-.507	-.707	-.881	-1.090	88	.820	F	-.061	.106	.181	.158	.057	-.213	-.667	
37	.299	135	.190	.042		-.092	-.144	-.256	-.338	-.450	89	.820	F	-.140	.109	.186	.169	.095	-.246	-.361	
38	.319	C	.205	.038		-.006	-.029	-.049	-.071	-.131	90	.820	F	-.156	.116	.201	.192	.130	-.076	-.342	
39	.350	0	-.157	-.065		-.042	.011	.073	.162	.287	91	.820	F	-.091	.131	.237	.240	.173	.010	-.696	
40	.350	20	-.056	-.100		-.036	.025	.078	.142	.245	92	.820	142			.267	.265	.212	.042	-.088	
41	.350	40	-.046	-.142		-.036	.026	.086	.151	.269	93	.820	165	.090	.192	.057	-.058	.026	.084	.162	
42	.350	60	-.893	.262		-.038	.032	.075	.073	.141	94	.897	30	-.364	-.057	-.062	-.032	.028	.098	.182	
43	.350	F	-.742	-.212			.001	-.060	-.184	-.540	95	.897	50	-.342	-.062	-.075	-.041	.029	.108	.191	
44	.350	F	.148	.071		-.053	-.136	-.209	-.599	-.946	96	.897	F	-.443	-.079	-.025	-.043	.107	-.290	-.496	
45	.350	135	.251	.103		-.022	-.054	-.039	-.055	-.108	97	.897	F	.026	-.046	-.004	-.017	.073	-.224	-.475	
46	.367	0	-.048	-.049		-.057	-.016	.048	.136	.266	98	.897	130	.014	-.032	-.012	-.032	.064	-.199	-.459	
47	.367	180	.285	.089		-.023	-.031	-.019	.004	-.027	99	.897	150	-.007	-.034	-.012	-.032	.064	-.199	-.459	
48	.486	0	-.021	-.019		-.016	.047	.141	.252	.375	102	.984	0	-.413	-.287	-.343	-.344	-.330	-.315	-.315	
49	.486	180	.351	.135		-.018	-.020	-.029	.006	-.069	103	.984	30	-.402	-.282	-.301	-.296	-.300	-.329	-.436	
50	.503	15	.000	.019		.002	.087	.192	.311	.434	104	.984	60	-.360	-.297	-.313	-.312	-.320	-.367	-.475	
51	.503	30	-.016	.018		.015	.109	.213	.323	.432	105	.984	90	-.363	-.304	-.354	-.338	-.293	-.321	-.447	
52	.503	50	.440	.041		.032	.120	.220	.323	.427	106	.984	135	-.434	-.310	-.421	-.406	-.362	-.403	-.523	
53	.503	F	-.258	-.086		.047	.133	.220	.302	.385	107	.984	180	-.372	-.387						
54	.503	F	-.483	-.111		.046	-.006	-.113	-.203	-.382											

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(c) $M = 0.95$ - ConcludedPressure coefficients C_p on -

Model orifice number	$\frac{x}{l}$	ϕ , deg	Fuselage								Model orifice number	$\frac{x}{l}$	ϕ , deg	Fuselage - Concluded												
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$					
			$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$											$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$												
1	.026	0			.173	.224	.323	.425	.543	.673	55	.503	F			.078	.039	.024	-.196	-.206	-.231					
2	.026	90			.339	.344	.309	.230	.104	-.076	56	.503	135			.066	.026	.015	.047	.057	-.463					
3	.026	180			.264	.214	.141	.097	.062	.030	57	.588	0			-.111	-.024	.107	.216	.330	-.444					
4	.084	0			.029	.067	.158	.257	.374	.509	58	.588	180			.042	-.029	-.194	-.310	-.382	-.551					
5	.084	45			.109	.142	.218	.269	.319	.356	59	.604	15			-.151	-.048	.099	.215	.338	.458					
6	.084	90			.171	.199	.148	.089	-.039	-.206	60	.604	30			-.047	-.043	.011	.093	.177	.357					
7	.084	135			.169	.121	.024	-.082	-.209	-.360	61	.604	50			-.151	-.035	.128	.255	.380	.501					
8	.084	180			.128	.083	.015	-.028	-.061	-.101	62	.604	F			-.176	-.044	.136	.274	.410	.536					
9	.134	J			.151	.115	.066	.034	.014	-.022	63	.604	F			.046	-.063	-.304	-.573	-.739	-.952					
10	.156	J			.327	.298	.245	.196	.134	.082	64	.604	135			.042	-.054	-.259	-.368	-.569	-.791					
12	.173	30			-.085	-.034	.053	.130	.213	.304	65	.690	0			.027	.034	.077	.111	.210	.321					
14	.173	90			.031	.050	.038	-.023	-.135	-.325	66	.690	180			.028	.014	-.122	-.353	-.569	-.735					
15	.173	135			.134	.117	.077	.030	-.009	-.041	67	.706	15			.072	.077	.110	.115	.204	.313					
16	.173	J			.241	.220	.187	.152	.108	.058	68	.706	30			.096	.099	.130	.129	.218	.325					
17	.173	C			.290	.262	.225	.205	.212	.227	69	.706	90			.128	.127	.148	.135	.219	.320					
18	.190	J			.131	.122	.103	.065	.011	-.055	70	.706	F			.145	.143	.159	.138	.220	.320					
19	.190	C			.118	.095	.067	.055	.073	.118	71	.706	F			.107	.103	.067	-.290	-.661	-.855					
20	.197	C			-.272	-.305	-.340	-.350	-.342	-.322	72	.706	135			.099	.090	.065	-.316	-.450	-.893					
21	.207	C			.002	-.008	-.023	-.021	.010	.035	73	.753	0			.194	.211	.229	.193	.225	.307					
23	.216	C			-.407	-.439	-.473	-.478	-.471	-.465	74	.753	180			.210	.201	.183	.069	-.180	-.593					
24	.226	23			-.176	-.131	-.040	.047	.150	.262	75	.770	15			.398	.404	.415	.365	.370	.431					
25	.226	45			-.156	-.102	-.020	.037	.089	.131	76	.770	30			.318	.324	.335	.274	.270	.332					
26	.226	68			-.163	-.112	-.059	-.047	-.064	-.122	77	.770	50			.270	.275	.285	.202	.182	.235					
27	.226	90			-.149	-.112	-.089	-.109	-.170	-.259	78	.770	F			.256	.260	.266	.148	.087	.117					
28	.226	135			-.318	-.306	-.295	-.290	-.293	-.321	79	.770	F			.292	.293	.284	.052	-.123	-.325					
29	.226	C			-.763	-.854	-.888	-.884	-.854	-.787	80	.770	F			.262	.262	.247	.068	-.243	-.495					
30	.263	C			-.276	-.303	-.254	-.153	-.096	-.111	81	.770	135			.288	.284	.266	.114	-.086	-.723					
31	.283	0			-.114	-.166	-.160	-.052	.064	.204	83	.820	38			.315	.316	.320	.296	.286	.327					
32	.299	30			-.140	-.104	-.091	-.014	.096	.215	84	.820	F			.286	.291	.301	.295	.297	.344					
33	.299	60			-.147	-.101	-.100	-.014	.089	.187	85	.820	F			.266	.271	.286	.287	.290	.335					
34	.299	F			-.164	-.096	-.117	-.041	.027	.077	86	.820	F			.267	.273	.292	.307	.311	.358					
35	.299	F			-.031	-.022	-.232	-.278	-.349	-.450	87	.820	F			.277	.280	.299	.350	.391	.461					
36	.299	F			-.047	-.103	-.383	-.702	-.862	-1.090	88	.820	F			.268	.265	.234	-.018	-.365	-.830					
37	.299	135			-.046	-.080	-.105	-.194	-.313	-.428	89	.820	F			.263	.262	.242	.075	-.169	-.560					
38	.319	C			.005	.000	-.021	-.043	.064	-.130	90	.820	F			.273	.275	.258	.108	-.167	-.301					
39	.350	20			-.061	-.041	.015	.077	.158	.284	91	.820	F			.290	.293	.280	.146	-.097	-.311					
40	.350	40			-.070	-.036	.027	.084	.142	.241	92	.820	142			.322	.328	.325	.188	-.005	-.565					
41	.350	60			-.072	-.036	.027	.092	.152	.264	93	.820	165			.353	.357	.344	.233	.048	-.066					
42	.350	F			-.098	-.042	.034	.081	.078	.141	94	.897	30			.499	.507	.517	.537	.551	.577					
43	.350	F			-.014	.008	-.047	-.186	-.519	-.657	95	.897	50			.410	.416	.421	.440	.461	.495					
44	.350	F			.001	-.038	-.123	-.190	-.491	-.932	96	.897	F			.328	.334	.342	.372	.397	.431					
45	.350	135			.016	-.011	-.046	-.042	-.045	-.100	97	.897	F			.289	.294	.265	.163	-.055	-.317					
46	.367	0			-.055	-.049	-.011	.050	.136	.258	98	.897	130			.385	.394	.374	.271	.099	-.360					
47	.367	180			.002	-.013	-.023	-.016	.005	-.026	99	.897	150			.501	.504	.471	.390	.229	-.059					
48	.486	0			-.014	-.008	.054	.141	.249	.369	102	.984	0			-.575	-.578	-.566	-.533	-.517	-.512					
49	.486	180			.013	-.004	-.009	-.024	.008	-.072	104	.984	60			-.698	-.697	-.676	-.649	-.659	-.614					
50	.503	15			-.033	-.024	.007	.063	.131	.309	105	.984	90			-.726	-.729	-.713	-.710	-.731	-.704					
51	.503	30			-.009	.020	.113	.213	.320	.428	106	.984	135			-.858	-.855	-.775	-.624	-.663	-.701					
52	.503	50			.013	.036	.125	.219	.320	.424	107	.984	180			-.615	-.616	-.611	-.588	-.608	-.640					
53	.503	F			.027	.051	.137	.220	.302	.382																
54	.503	F			.046	.059	.011	-.100	-.195	-.386																

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(d) $M = 1.03$ Pressure coefficients C_p on -

Model orifice number	$\frac{x}{l}$	β , deg	Fuselage								Model orifice number	$\frac{x}{l}$	β , deg	Fuselage - Concluded							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx -5^\circ; \delta_E = 0^\circ; \delta_V = 0^\circ; \text{brakes closed}$											$\beta \approx -5^\circ; \delta_E = 0^\circ; \delta_V = 0^\circ; \text{brakes closed}$							
1	.026	0				.311	.393	.492	.593	.702	55	.503	F	-.049	-.005	.020	-.078	-.174			
2	.026	90				.289	.248	.160	.054	-.114	56	.503	135	-.056	.003	.122	.009	-.300			
3	.026	180				.308	.231	.179	.126	.57	57	.588	0	.018	.157	.301	.408	.504			
4	.084	0				.158	.234	.324	.424	.536	58	.588	180	.020	-.104	-.188	-.264	-.367			
5	.084	45				.139	.153	.126	.123	.114	59	.604	15	.007	.140	.280	.382	.471			
6	.084	90				.135	.107	.028	-.077	-.274	60	.604	30	.014	.156	.298	.396	.479			
7	.084	135				.115	.081	.028	-.037	-.102	61	.604	50	.010	.162	.320	.430	.521			
8	.084	180				.162	.103	.049	.014	-.034	62	.604	F	.013	.179	.352	.472	.568			
9	.134	J				.207	.148	.104	.049	-.051	63	.604	F	.001	-.177	-.341	-.662	-.780			
10	.156	J				.167	.128	.111	.124	.152	64	.604	135	.011	-.147	-.295	-.451	-.603			
12	.173	30				.009	.029	.044	.067	.113	65	.690	0	-.122	.007	.184	.289	.385			
13	.173	60				.005	-.005	-.058	-.148	-.227	66	.690	180	-.119	-.262	-.385	-.526	-.667			
14	.173	90				.011	.001	-.060	-.170	-.387	67	.706	15	-.143	-.017	.163	.254	.342			
15	.173	135				.042	.046	.036	.012	-.041	68	.706	30	-.140	-.007	.169	.256	.339			
16	.173	J				.097	.077	.078	.105	.122	69	.706	50	-.138	.007	.182	.272	.358			
17	.173	C				.136	.102	.085	.093	.089	70	.706	F	-.137	.018	.195	.285	.370			
18	.190	J				.010	.016	.031	.070	.057	71	.706	F	-.165	-.309	-.442	-.678	-.869			
19	.190	C				.005	-.017	-.020	-.007	.027	72	.706	135	-.146	-.292	-.417	-.548	-.723			
20	.197	C				-.158	-.207	-.236	-.237	-.223	73	.753	0	.131	.196	.251	.292	.364			
21	.207	C				-.052	-.059	-.060	-.043	.029	74	.753	180	.085	.021	-.148	-.568	-.657			
23	.216	C				-.294	-.332	-.355	-.349	.339	75	.770	30	.032	.086	.137	.172	.212			
24	.226	23				-.060	-.031	-.002	.052	.094	76	.770	50	.029	.066	.102	.135	.178			
25	.226	45				-.059	-.055	-.070	-.081	-.132	77	.770	F	.028	.059	.080	.116	.167			
26	.226	68				-.086	-.088	-.139	-.210	-.386	78	.770	F	.062	.069	.037	.066	.117			
27	.226	90				-.113	-.096	-.133	-.204	-.218	79	.770	F	.087	.114	.003	-.246	-.477			
28	.226	135				-.340	-.305	-.275	-.291	-.344	80	.770	F	.054	.055	-.030	-.610	-.749			
29	.226	C				-.465	-.520	-.615	-.682	-.708	81	.770	135	.027	.041	-.123	-.561	-.734			
30	.263	C				-.294	-.291	-.255	-.190	-.150	82	.820	15	.036	.030	-.070	-.113	.028			
31	.283	0				-.151	-.088	-.014	.071	.208	83	.820	38	.046	.058	-.022	-.070	.030			
32	.299	30				-.148	-.123	-.089	-.113	.031	84	.820	F	.055	.068	.016	.029	.073			
33	.299	60				-.133	-.103	-.049	.023	.113	85	.820	F	.074	.086	.034	.065	.095			
34	.299	F				-.132	-.110	-.088	-.040	.019	86	.820	F								
35	.299	F				-.114	-.195	-.358	-.423	-.497	87	.820	F	.134	.143	.113	.137	.303			
36	.299	F				-.136	-.263	-.476	-.791	-.916	88	.820	F	.065	.077	-.021	-.065	-.211			
37	.299	135				-.083	-.131	-.201	-.265	-.286	89	.820	F	.057	.078	-.013	-.048	-.094			
38	.319	C				-.071	-.124	-.198	-.246	-.294	90	.820	F	.050	.066	-.004	-.021	-.077			
39	.350	20				-.114	-.119	-.081	.003	.156	91	.820	F	.027	.055	.001	.037	-.047			
40	.350	40				-.095	-.104	-.106	-.039	.087	92	.820	142	-.003	-.003	-.004	-.004	-.004			
41	.350	60				-.090	-.072	-.056	.037	.190	94	.897	30	-.014	.000	.008	.046	.152			
42	.350	F				-.093	-.080	-.123	-.073	.039	95	.897	50	-.017	-.005	.007	.058	.169			
43	.350	F				-.084	-.182	-.388	-.689	-.755	96	.897	F	-.014	-.002	.013	.069	.184			
44	.350	F				-.052	-.116	-.251	-.714	-.985	97	.897	F	-.026	-.055	-.149	-.197	-.291			
45	.350	135				-.016	-.040	-.102	-.141	.004	98	.897	130	-.011	-.032	-.114	-.138	-.227			
46	.367	C				-.110	-.092	-.032	.081	.266	99	.897	150	-.013	-.028	-.097	-.121	-.179			
47	.367	180				-.032	-.063	-.074	.016	.102	102	.984	0	-.359	-.387	-.408	-.390	-.368			
48	.486	0				-.070	-.051	.181	.311	.412	104	.984	60	-.307	-.331	-.381	-.461	-.565			
49	.486	180				-.077	-.083	.077	.126	.110	105	.984	90	-.329	-.352	-.407	-.478	-.552			
50	.503	15				-.064	-.039	.198	.299	.385	106	.984	135	-.345	-.414	-.463	-.536	-.633			
51	.503	30				-.057	-.030	.183	.266	.335	107	.984	180	-.417	-.447	-.507	-.553	-.617			
52	.503	50				-.050	.009	.213	.300	.373											
53	.503	F				-.044	.014	.204	.272	.332											
54	.503	F				-.054	-.030	.000	-.108	-.116											

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(a) $M = 1.03$ - ContinuedPressure coefficients C_p on -

Model orifice number	$\frac{x}{l}$	ϕ , deg	Fuselage								Model orifice number	$\frac{x}{l}$	ϕ , deg	Fuselage - Concluded							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx -5^\circ; \delta_E = 0^\circ; \delta_V = 0^\circ; \text{brakes open}$											$\beta \approx -5^\circ; \delta_E = 0^\circ; \delta_V = 0^\circ; \text{brakes open}$							
1	.026	0	.128	.176	.264	.315	.397	.496	.596	.708	55	.503	F	.324	.208	-.045	-.048	.009	.012	-.078	-.179
2	.026	90	-.134	.156	.260	.289	.234	.159	.048	-.124	56	.503	135	.341	.203	-.054	-.055	.020	.124	.004	-.316
3	.026	180	.726	.510	.354	.303	.221	.176	.141	.122	57	.588	0	-.375	-.199	-.051	.028	.170	.308	.414	.511
4	.084	0	-.026	.059	.123	.161	.236	.328	.428	.542	58	.588	180	.510	.309	.096	.014	-.109	-.189	-.263	-.377
5	.084	45	-.092	.058	.122	.131	.150	.129	.121	.113	59	.604	15	-.393	-.230	-.070	.016	.153	.287	.388	.479
6	.084	90	-.291	.025	.119	.123	.115	.076	.027	-.082	60	.604	30	-.460	-.237	-.066	.022	.170	.305	.401	.484
7	.084	135	.110	.119	.123	.115	.076	.027	-.038	-.105	61	.604	50	-.574	-.281	-.071	.019	.177	.328	.436	.528
8	.084	180	.565	.352	.204	.158	.099	.050	.012	-.035	62	.604	F	-.775	-.363	-.095	.021	.195	.360	.478	.575
9	.134	J	.551	.365	.240	.203	.141	.106	.045	-.056	63	.604	F	.576	.363	.103	-.006	-.186	-.331	-.673	-.784
10	.156	J	.407	.260	.185	.166	.123	.115	.125	.156	64	.604	135	.511	.324	.093	.004	-.156	-.302	-.457	-.610
12	.173	30	-.198	-.069	-.014	.011	.027	.047	.068	.117	65	.690	0	-.668	-.403	-.172	-.098	.076	.198	.298	.397
13	.173	60	-.271	-.109	-.017	.007	-.011	-.058	-.154	-.230	66	.690	180	.374	.171	-.045	-.123	-.273	-.388	-.512	-.666
14	.173	90	-.455	-.148	-.011	.013	-.004	-.061	-.180	-.396	67	.706	15	-.590	-.411	-.013	.013	.105	.180	.264	.355
15	.173	135	-.075	-.026	.026	.043	.042	.037	.009	-.040	68	.706	30	-.653	-.396	-.009	.030	.111	.185	.267	.352
16	.173	J	.101	.095	.075	.083	.106	.083	.106	.121	69	.706	50	-.722	-.409	-.011	.042	.118	.198	.283	.369
17	.173	C	.347	.205	.144	.135	.096	.087	.091	.093	70	.706	F	-.782	-.451	-.015	.038	.122	.208	.296	.381
18	.190	J	.099	.012	.006	.012	.015	.036	.071	.054	71	.706	F	.335	.154	.012	-.045	-.315	-.445	-.686	-.891
19	.190	C	.191	.071	.017	.004	-.020	-.017	-.008	.032	72	.706	135	.327	.151	.025	-.052	-.300	-.422	-.552	-.733
20	.197	C	.187	-.005	-.126	-.160	-.213	-.236	-.237	-.222	73	.753	0	-.656	-.137	.250	.290	.286	.281	.312	.387
21	.207	C	.049	-.036	-.054	-.052	-.062	-.059	-.041	.030	74	.753	180	.340	.238	.279	.264	.156	-.083	-.490	-.600
23	.216	C	.039	-.158	-.270	-.298	-.337	-.353	-.347	-.343	75	.770	15	-.555	-.133	.142	.148	.139	.146	.179	.218
24	.226	23	-.225	-.125	-.082	-.058	-.034	.000	.055	.099	76	.770	30	-.673	-.059	.177	.181	.150	.116	.143	.187
25	.226	45	-.323	-.158	-.078	-.057	-.058	-.067	-.078	-.132	77	.770	50	-.646	-.031	.189	.196	.173	.104	.125	.179
26	.226	68	-.414	-.232	-.111	-.085	-.092	-.139	-.215	-.392	78	.770	F	-.608	.040	.230	.220	.194	.084	.074	.127
27	.226	90	-.645	-.310	-.144	-.110	-.099	-.134	-.207	-.214	79	.770	F	-.679	-.052	.248	.239	.237	.072	-.187	-.382
28	.226	135	-.387	-.424	-.362	-.334	-.305	-.273	-.292	-.350	80	.770	F	.033	-.010	.202	.207	.184	.067	-.271	-.362
29	.226	C	-.233	-.349	-.437	-.469	-.526	-.618	-.684	-.706	81	.770	135	.128	.057	.162	.180	.172	.062	-.536	-.744
30	.263	C	-.046	-.270	-.298	-.295	-.292	-.251	-.184	-.152	82	.820	15	-.446	-.021	.203	.208	.199	.177	.190	.228
31	.283	0	-.221	-.190	-.181	-.148	-.087	-.010	.074	.217	83	.820	38	-.113	-.002	.203	.219	.220	.194	.209	.235
32	.299	30	-.264	-.152	-.156	-.148	-.124	-.085	-.111	.032	84	.820	F	-.041	.030	.215	.224	.227	.204	.232	.278
33	.299	60	-.176	-.211	-.153	-.132	-.105	-.047	.031	.117	85	.820	F	.005	.039	.231	.235	.238	.212	.240	.289
34	.299	F	-.452	-.316	-.165	-.132	-.115	-.088	-.037	.021	86	.820	F	-.036	.082						
35	.299	F	-.511	-.236							87	.820	F	-.041	.107	.270	.282	.272	.250	.282	.506
36	.299	F	-.010	-.050	-.087	-.136	-.267	-.469	-.799	-.917	88	.820	F	-.030	-.031	.203	.202	.188	.090	.032	-.237
37	.299	135	-.058	-.082	-.058	-.135	-.200	-.264	-.287	-.287	89	.820	F	-.038	-.040	.204	.200	.184	.087	.028	-.019
38	.319	C	.244	.018	-.053	-.076	-.131	-.198	-.249	-.248	90	.820	F	-.283	-.062	.201	.196	.174	.089	.035	.002
39	.350	20	.103	-.088	-.111	-.114	-.123	-.078	.013	.164	91	.820	F	-.240	-.087	.185	.188	.169	.094	.091	.068
40	.350	40	-.031	-.122	-.098	-.095	-.108	-.103	-.028	.094	92	.820	142	-.164	-.120	.173	.167	.157	.094	.090	.073
41	.350	60	.035	-.151	-.102	-.090	-.074	-.054	.048	.200	93	.820	165	-.090	-.142						
42	.350	F	-.759	-.211	-.112	-.092	-.083	-.127	-.065	.045	94	.897	30	-.135	-.092	.332	.385	.474	.499	.520	.559
43	.350	F	-.541	-.354	-.090	-.084	-.186	-.387	-.691	-.761	95	.897	50	-.154	-.117	.325	.339	.387	.411	.438	.484
44	.350	F	.073	-.033	-.033	-.056	-.124	-.254	-.727	-.984	96	.897	F	-.196	-.138	.287	.312	.346	.360	.388	.436
45	.350	135	.145	.014	-.003	-.017	-.046	-.104	-.122	.021	97	.897	F	-.063	-.069	.263	.244	.201	.187	.099	-.036
46	.367	0	.059	-.130	-.115	-.110	-.093	-.028	.097	.276	98	.897	130	-.052	-.069	.325	.343	.296	.313	.250	.129
47	.367	180	.336	.064	-.014	-.034	-.065	-.075	.046	.099	99	.897	150	-.053	-.083	.420	.382	.369	.411	.404	.284
48	.486	0	.088	.072	-.070	-.068	-.051	.189	.319	.420	102	.984	0	-.590	-.455	-.491	-.510	-.496	-.506	-.498	-.468
49	.486	180	.424	.195	-.075	-.077	-.084	.080	.124	.102	103	.984	30	-.591	-.449						
50	.503	15	.122	.160	-.052	-.063	-.014	.206	.304	.392	104	.984	60	-.460	-.376	-.653	-.627	-.653	-.635	-.702	-.738
51	.503	30	.016	.167	-.045	-.057	-.002	.190	.270	.340	105	.984	90	-.475	-.401	-.688	-.612	-.629	-.659	-.685	-.725
52	.503	50	-.249	.135	-.038	-.050	.034	.218	.304	.379	106	.984	135	-.513	-.478	-.732	-.605	-.698	-.852	-.804	-.705
53	.503	F	-.088	.037	-.032	-.043	.042	.209	.276	.337	107	.984	180	-.451	-.478	-.553	-.547	-.556	-.602	-.669	-.723
54	.503	F	-.094	.036	-.066	-.055	-.007	.002	-.118	-.119											

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(d) $M = 1.05$ - ContinuedPressure coefficients C_p on -

Model orifice number	$\frac{x}{l}$	β , deg	Fuselage								Model orifice number	$\frac{x}{l}$	β , deg	Fuselage - Concluded							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx 0^\circ$; $\delta_e = 0^\circ$; $\delta_v = 0^\circ$; brakes closed											$\beta \approx 0^\circ$; $\delta_e = 0^\circ$; $\delta_v = 0^\circ$; brakes closed							
1	.026	0	.140	.187		.325	.406	.503	.606	.719	55	.503	F	.382	.233		-.063	-.012	-.085	-.004	-.093
2	.026	90	-.014	.225		.335	.303	.233	.128	-.035	56	.503	135	.414	.238		-.059	.002	.149	.061	-.305
3	.026	180	.729	.508		.322	.254	.195	.160	.139	57	.588	0	-.227	-.164		.038	.182	.321	.430	.529
4	.084	0	.008	.062		.187	.265	.352	.451	.566	58	.588	180	.536	.327		.040	-.088	-.146	-.177	-.257
5	.084	45	-.119	.050		.178	.213	.254	.275	.289	59	.604	15	-.375	-.243		.020	.164	.306	.417	.517
6	.084	90	-.169	.064		.178	.145	.077	-.023	-.172	60	.604	30	-.625	-.253		.024	.174	.318	.430	.526
7	.084	135	.276	.218		.166	.097	.021	-.065	-.143	61	.604	50	-.708	-.274		.016	.173	.325	.443	.545
8	.084	180	.582	.365		.177	.117	.066	.033	.006	62	.604	F	-.778	-.383		.008	.182	.346	.472	.579
9	.134	J	.556	.362		.215	.159	.115	.063	-.010	63	.604	F	.589	.351		-.007	-.195	-.412	-.556	-.769
10	.156	J	.517	.367		.280	.231	.204	.188	.142	64	.604	135	.543	.326		.008	-.150	-.264	-.571	-.822
12	.173	30	-.195	-.088		.044	.088	.135	.172	.212	65	.690	0	-.593	-.375		-.106	.029	.189	.300	.402
13	.173	60	-.359	-.118		.044	.059	.039	-.020	-.071	66	.690	180	.390	.172		-.108	-.245	-.353	-.471	-.595
14	.173	90	-.367	-.089		.065	.050	-.010	-.107	-.333	67	.706	15	-.638	-.397		-.128	.006	.170	.274	.371
15	.173	135	.101	.103		.128	.107	.086	.063	.002	68	.706	30	-.655	-.485		-.125	.011	.176	.278	.372
16	.173	J	.376	.257		.208	.178	.167	.160	.130	69	.706	50	-.759	-.410		-.132	.010	.175	.278	.370
17	.173	C	.471	.323		.250	.214	.195	.199	.216	70	.706	F	-.835	-.457		-.134	.015	.182	.283	.374
18	.190	J	.214	.130		.120	.104	.097	.081	.045	71	.706	F	.338	.130		-.163	-.311	-.404	-.688	-.852
19	.190	C	.294	.160		.104	.075	.064	.074	.115	72	.706	135	.345	.139		-.145	-.291	-.410	-.608	-.851
20	.197	C	.209	.006		-.134	-.181	-.206	-.221	-.198	73	.753	0	-.637	-.438		.056	.138	.204	.270	.351
21	.207	C	.139	.043		.029	.012	.013	.034	.084	74	.753	180	.326	.170		-.030	-.231	-.410	-.539	-.620
23	.216	C	.066	-.144		-.268	-.307	-.333	-.345	-.323	75	.770	15	-.657	-.003		.189	.234	.253	.286	.346
24	.226	23	-.296	-.144		-.032	.021	.072	.130	.200	76	.770	30	-.704	-.061		.162	.186	.191	.218	.278
25	.226	45	-.324	-.189		-.027	.013	.030	.036	.025	77	.770	50	-.688	-.049		.143	.153	.142	.153	.217
26	.226	68	-.553	-.251		-.056	-.030	-.048	-.097	-.213	78	.770	F	-.718	.030		.148	.142	.106	.071	.133
27	.226	90	-.583	-.277		-.071	-.051	-.077	-.144	-.164	79	.770	F	-.003	-.003		.182	.122	.079	-.212	-.419
28	.226	135	-.282	-.333		-.269	-.247	-.230	-.234	-.293	80	.770	F	.046	.070		.136	.110	.043	-.241	-.437
29	.226	C	-.177	-.327		-.540	-.670	-.734	-.745	-.691	81	.770	135	.189	.115		.128	.084	-.111	-.617	-.833
30	.263	C	.091	-.174		-.241	-.245	-.231	-.212	-.088	82	.820	15	.159	.157		.178	.176	.135	.130	.195
31	.283	0	-.042	-.158		-.136	-.082	-.017	.090	.233	83	.820	38	.056	.137		.181	.184	.146	.142	.204
32	.299	30	-.233	-.203		-.136	-.081	-.007	.057	.162	84	.820	F	.108	.136		.182	.190	.159	.167	.246
33	.299	60	-.264	-.289		-.143	-.095	-.010	.066	.166	85	.820	F	-.058	.130		.179	.196	.162	.176	.253
34	.299	F	-.792	-.408		-.145	-.109	-.042	.010	.072	86	.820	F	-.109	.151		.199	.225	.194	.223	.315
35	.299	F	-.466	-.223		-.114	-.189	-.275	-.339	-.410	87	.820	F	-.116	.158		.216	.258	.233	.333	.486
36	.299	F	.020	-.039		-.171	-.329	-.618	-.753	-.920	88	.820	F	.057	.058		.166	.125	.041	-.294	-.477
37	.299	135	.154	.005		-.121	-.201	-.280	-.352	-.402	89	.820	F	.051	.057		.162	.144	.067	-.134	-.281
38	.319	C	.289	.051		-.067	-.102	-.139	-.182	-.199	90	.820	F	.026	.054		.163	.156	.099	-.075	-.062
39	.350	20	-.130	-.127		-.103	-.081	-.020	.077	.244	91	.820	F	-.013	.056		.165	.170	.120	-.042	-.069
40	.350	40	.001	-.147		-.101	-.082	-.041	.042	.184	92	.820	142				.168	.181	.144	.086	.025
41	.350	60	.037	-.185		-.113	-.076	-.031	.071	.236	93	.820	165	.058	.081						
42	.350	F	-.835	-.289		-.118	-.082	-.085	-.016	.103	94	.897	30	-.203	-.040		.010	.022	.028	.069	.174
43	.350	F	-.740	-.348		-.096	-.177	-.453	-.568	-.633	95	.897	50	-.201	-.048		.003	.015	.025	.084	.194
44	.350	F	.132	-.003		-.078	-.180	-.301	-.822	-.880	96	.897	F	-.330	-.057		-.006	.009	.024	.088	.201
45	.350	135	.232	.055		-.024	-.072	-.102	-.140	-.055	97	.897	F	-.003	-.003		.011	-.010	-.073	-.226	-.355
46	.367	0	.134	-.082		-.090	-.071	-.014	.089	.286	98	.897	130	-.014	-.035		.024	.010	-.048	-.140	-.308
47	.367	180	.358	.079		-.001	-.004	.002	.072	.164	99	.897	150	.016	.037		.025	.009	-.055	-.103	-.235
48	.486	0	.135	.158		-.057	-.037	.201	.332	.438	102	.984	0	-.548	-.441		-.328	-.356	-.380	-.367	-.343
49	.486	180	.446	.198		-.062	-.050	.152	.195	.206	103	.984	30	-.549	-.438		-.313	-.354	-.392	-.401	-.383
50	.503	15	.040	.142		-.054	-.003	.243	.354	.452	104	.984	60	-.469	-.374		-.261	-.296	-.382	-.471	-.544
51	.503	30	-.165	.147		-.054	.006	.238	.339	.424	105	.984	90	-.486	-.395		-.278	.319	-.382	-.462	-.532
52	.503	50	-.214	.158		-.059	.019	.244	.343	.428	106	.984	135	-.479	-.442		-.347	-.411	-.461	-.536	-.615
53	.503	F	-.148	-.030		-.056	.022	.238	.318	.389	107	.984	180	-.410	-.430		-.369	-.417	-.476	-.528	-.560
54	.503	F	-.119	-.038		-.060	-.079	-.018	-.049	-.179											

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(d) $M = 1.03$ - ContinuedPressure coefficients C_p on -

Model orifice number	$\frac{x}{l}$	β , deg	Fuselage							Model orifice number	$\frac{x}{l}$	β , deg	Fuselage - Concluded								
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$	
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$										$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$								
1	.026	0			.290	.327	.411	.497	.597	.719	55	.503	F			-.044	-.062	-.016	-.093	-.010	-.083
2	.026	90			.334	.341	.311	.234	.131	-.026	56	.503	135			-.042	-.058	-.002	.140	.067	-.295
3	.026	180			.375	.325	.259	.193	.157	.143	57	.588	0			-.039	.036	.183	.313	.422	.529
4	.084	0			.152	.190	.270	.348	.443	.565	58	.588	180			.131	.041	-.082	-.152	-.182	-.250
5	.084	45			.160	.180	.222	.250	.277	.298	59	.604	15			-.064	.022	.166	.299	.411	.518
6	.084	90			.174	.183	.157	.079	-.025	-.161	60	.604	30			-.064	.026	.175	.312	.424	.528
7	.084	135			.202	.170	.106	.020	-.068	-.141	61	.604	50			-.079	.019	.175	.318	.437	.546
8	.084	180			.231	.183	.125	.065	.030	.011	62	.604	F			-.104	.010	.183	.339	.465	.580
9	.134	J			.260	.220	.163	.112	.060	-.004	63	.604	F			.112	-.010	-.191	-.412	-.556	-.761
10	.156	J			.314	.287	.242	.204	.186	.148	64	.604	135			.113	.010	-.144	-.265	-.566	-.821
12	.173	30			.014	.048	.097	.128	.172	.240	65	.690	0			-.182	-.104	.031	.184	.295	.403
13	.173	60			.022	.051	.071	.040	-.017	-.059	66	.690	180			-.025	-.106	-.240	-.356	-.469	-.586
14	.173	90			.050	.073	.060	-.008	-.097	-.321	67	.706	15			.206	-.126	.007	.168	.272	.374
15	.173	135			.137	.136	.118	.085	.060	.009	68	.706	30			-.209	-.124	.013	.174	.275	.375
16	.173	J			.231	.216	.191	.165	.156	.137	69	.706	50			-.220	-.128	.012	.173	.273	.373
17	.173	C			.281	.258	.225	.198	.198	.225	70	.706	F			-.223	-.130	.017	.179	.279	.377
18	.190	J			.131	.128	.114	.097	.079	.053	71	.706	F			-.081	-.161	-.304	-.402	-.682	-.837
19	.190	C			.135	.113	.085	.064	.073	.122	72	.706	135			-.067	-.143	-.288	-.412	-.613	-.825
20	.197	C			-.096	-.133	-.176	-.218	-.228	-.196	73	.753	0			.064	.083	.147	.216	.277	.363
21	.207	C			.047	.036	.021	.013	.032	.087	74	.753	180			.059	.023	-.213	-.415	-.538	-.610
23	.216	C			-.233	-.268	-.308	-.337	-.351	-.319	75	.770	15			.206	.231	.263	.283	.311	.376
24	.226	23			-.058	-.027	.029	.070	.127	.203	76	.770	30			.187	.203	.215	.223	.243	.306
25	.226	45			-.057	-.018	.021	.029	.040	.033	77	.770	50			.177	.182	.179	.175	.178	.239
26	.226	68			-.088	-.051	-.021	-.049	-.098	-.184	78	.770	F			.176	.183	.163	.141	.094	.151
27	.226	90			-.103	-.064	-.041	-.077	-.146	-.162	79	.770	F			.169	.212	.126	.113	-.184	-.402
28	.226	135			-.278	-.261	-.238	-.233	-.237	-.287	80	.770	F			.174	.175	.125	.071	-.211	-.438
29	.226	C			-.458	-.541	-.668	-.738	-.749	-.690	81	.770	135			.182	.175	.104	-.078	-.377	-.774
30	.263	C			-.255	-.235	-.235	-.234	-.219	-.095	82	.820	15			.263	.253	.244	.209	.198	.260
31	.283	0			-.150	-.131	-.075	-.017	.085	.233	83	.820	38			.237	.234	.231	.197	.185	.245
32	.299	30			-.155	-.133	-.072	-.011	.052	.164	84	.820	F			.232	.229	.234	.204	.206	.284
33	.299	60			-.179	-.141	-.087	-.015	.060	.168	85	.820	F			.222	.223	.234	.202	.208	.281
34	.299	F			-.192	-.140	-.101	-.046	.005	.074	86	.820	F			.232	.236	.260	.226	.248	.334
35	.299	F			-.103	-.109	-.181	-.275	-.338	-.401	87	.820	F			.240	.250	.293	.262	.358	.492
36	.299	F			-.096	-.179	-.322	-.617	-.753	-.903	88	.820	F			.241	.203	.155	.079	-.186	-.284
37	.299	135			-.071	-.121	-.194	-.285	-.360	-.399	89	.820	F			.220	.205	.179	.103	-.062	-.142
38	.319	C			-.039	-.066	-.097	-.142	-.185	-.195	90	.820	F			.218	.208	.196	.142	-.033	-.049
39	.350	20			-.103	-.098	-.075	-.024	.072	.244	91	.820	F			.221	.218	.219	.170	.004	-.062
40	.350	40			-.111	-.099	-.076	-.044	.038	.188	92	.820	142			.234	.230	.241	.207	.135	.016
41	.350	60			-.127	-.108	-.071	-.036	.064	.236	94	.897	30			.053	.043	.047	.039	.068	.165
42	.350	F			-.150	-.114	-.076	-.087	-.019	.107	95	.897	50			.047	.038	.044	.042	.084	.192
43	.350	F			-.094	-.095	-.169	-.441	-.564	-.623	96	.897	F			.035	.028	.040	.043	.092	.203
44	.350	F			-.033	-.077	-.174	-.303	-.824	-.881	97	.897	F			.052	.047	.036	-.037	-.207	-.331
45	.350	135			.007	-.021	-.066	-.107	-.150	-.054	98	.897	130			.063	.062	.061	-.007	-.119	-.264
46	.367	0			-.086	-.085	-.066	-.018	.081	.285	99	.897	150			.058	.062	.058	-.011	-.087	-.198
47	.367	180			.019	.005	.003	-.005	.039	.165	102	.984	0			-.423	-.408	-.426	-.442	-.440	-.416
48	.486	0			-.052	-.053	-.032	.185	.325	.438	103	.984	30			-.420	-.405	-.432	-.468	-.485	-.467
49	.486	180			-.052	-.057	-.048	.142	.192	.208	104	.984	60			-.300	-.292	-.324	-.399	-.507	-.587
50	.503	15			-.042	-.052	-.009	.231	.348	.454	105	.984	90			-.318	-.312	-.351	-.419	-.506	-.576
51	.503	30			-.043	-.051	-.003	.228	.334	.428	106	.984	135			-.411	-.411	-.455	-.527	-.590	-.678
52	.503	50			-.046	-.056	.010	.232	.337	.431	107	.984	180			-.498	-.489	-.525	-.592	-.656	-.685
53	.503	F			-.051	-.053	.010	.228	.313	.390											
54	.503	F			-.083	-.061	-.096	-.026	-.054	-.183											

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(a) $M = 1.05$ - ContinuedPressure coefficients C_p on -

Model orifice number	$\frac{x}{l}$	β , deg	Fuselage								Model orifice number	$\frac{x}{l}$	β , deg	Fuselage - Concluded							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = -7.5^\circ; \text{brakes closed}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = -7.5^\circ; \text{brakes closed}$							
1	.026	G			.288	.329	.411	.506	.604	.715	55	.503	F	-.046	-.062	-.016	-.078	-.005	-.086		
2	.026	90			.327	.335	.306	.236	.132	-.032	56	.503	135	-.045	-.059	-.002	.148	.063	-.290		
3	.026	180			.377	.325	.260	.200	.162	.139	57	.588	0	-.041	.038	.181	.321	.429	.526		
4	.084	0			.151	.189	.270	.356	.451	.560	58	.588	180	.127	.041	-.080	-.143	-.171	-.250		
5	.084	45			.161	.183	.218	.257	.274	.284	59	.604	15	-.067	.021	.166	.307	.416	.513		
6	.084	90			.169	.178	.153	.080	-.018	-.168	60	.604	30	-.067	.024	.175	.320	.429	.523		
7	.084	135			.197	.169	.105	.028	-.057	-.138	61	.604	50	-.081	.017	.175	.326	.442	.541		
8	.084	180			.229	.182	.125	.074	.035	.006	62	.604	F	-.108	.009	.183	.347	.472	.576		
9	.134	J			.261	.218	.163	.122	.067	-.010	63	.604	F	.110	-.004	-.187	-.397	-.544	-.757		
10	.156	J			.305	.280	.232	.207	.190	.142	64	.604	135	.112	.012	-.142	-.256	-.551	-.807		
12	.173	30			.004	.033	.086	.126	.164	.221	65	.690	0	-.186	-.105	.031	.193	.302	.400		
13	.173	60			.019	.047	.066	.043	-.019	-.073	66	.690	180	-.026	-.105	-.236	-.343	-.458	-.578		
14	.173	90			.047	.066	.057	-.005	-.094	-.329	67	.706	15	-.208	-.127	.007	.176	.277	.370		
15	.173	135			.129	.129	.113	.090	.066	.004	68	.706	30	-.211	-.126	.011	.180	.281	.371		
16	.173	J			.223	.209	.185	.169	.163	.131	69	.706	50	-.222	-.131	.010	.179	.280	.370		
17	.173	C			.263	.239	.217	.197	.197	.211	70	.706	F	-.227	-.132	.015	.185	.285	.373		
18	.190	J			.123	.119	.109	.101	.085	.046	71	.706	F	-.082	-.160	-.301	-.392	-.672	-.837		
19	.190	C			.116	.107	.081	.066	.074	.110	72	.706	135	-.068	-.142	-.282	-.402	-.594	-.815		
20	.197	C			-.094	-.131	-.177	-.209	-.221	-.200	73	.753	0	.048	.078	.140	.224	.281	.362		
21	.207	C			.042	.031	.018	.014	.034	.080	74	.753	180	.017	-.001	-.260	-.401	-.521	-.602		
23	.216	C			-.235	-.267	-.307	-.328	-.344	-.320	75	.770	15	.141	.170	.214	.257	.292	.348		
24	.226	23			-.061	-.032	.028	.076	.128	.194	76	.770	30	.121	.142	.163	.194	.223	.277		
25	.226	45			-.060	-.026	.020	.032	.038	.020	77	.770	50	.110	.120	.123	.138	.155	.212		
26	.226	68			-.092	-.055	-.023	-.047	-.097	-.211	78	.770	F	.114	.126	.112	.093	.072	.131		
27	.226	90			-.108	-.069	-.044	-.073	-.142	-.165	79	.770	F	.101	.161	.110	.074	-.214	-.450		
28	.226	135			-.286	-.268	-.243	-.227	-.234	-.291	80	.770	F	.099	.114	.083	.027	-.258	-.448		
29	.226	C			-.453	-.534	-.649	-.717	-.733	-.690	81	.770	135	.105	.102	.047	-.124	-.594	-.826		
30	.263	C			-.269	-.242	-.236	-.229	-.215	-.106	82	.820	15	.074	.056	.015	-.130	-.162	.002		
31	.283	0			-.152	-.133	-.075	-.008	.089	.227	83	.820	38	.097	.086	.066	-.030	-.077	.089		
32	.299	30			-.154	-.134	-.073	-.005	.055	.155	84	.820	F	.110	.103	.094	.027	.010	.154		
33	.299	60			-.179	-.143	-.088	-.009	.064	.162	85	.820	F	.113	.109	.109	.050	.046	.187		
34	.299	F			-.194	-.145	-.105	-.041	.009	.068	86	.820	F	.139	.137	.150	.102	.126	.267		
35	.299	F			-.105	-.112	-.172	-.273	-.336	-.409	87	.820	F	.157	.161	.185	.150	.246	.469		
36	.299	F			-.095	-.167	-.318	-.610	-.742	-.909	88	.820	F	.139	.110	.084	.002	-.301	-.574		
37	.299	135			-.071	-.118	-.188	-.270	-.343	-.397	89	.820	F	.108	.102	.102	.025	-.167	-.353		
38	.319	C			-.043	-.068	-.096	-.138	-.178	-.197	90	.820	F	.091	.095	.104	.049	-.122	-.126		
39	.350	20			-.102	-.099	-.077	-.019	.074	.236	91	.820	F	.070	.083	.105	.055	-.078	-.122		
40	.350	40			-.110	-.099	-.077	-.041	.038	.178	92	.820	142	.045	.065	.093	.060	.026	-.035		
41	.350	60			-.126	-.109	-.072	-.030	.068	.229	94	.820	30	-.019	-.017	.005	.022	.067	.150		
42	.350	F			-.148	-.115	-.076	-.083	-.017	.097	95	.897	50	-.028	-.026	-.005	.017	.050	.165		
43	.350	F			-.096	-.093	-.168	-.415	-.565	-.635	96	.897	F	-.036	-.032	-.008	.019	.057	.174		
44	.350	F			-.035	-.075	-.169	-.292	-.789	-.878	97	.897	F	-.009	-.019	-.034	-.103	-.271	-.380		
45	.350	135			.004	-.021	-.062	-.097	-.142	-.069	98	.897	130	.001	-.008	-.019	-.085	-.177	-.347		
46	.367	0			-.087	-.086	-.066	-.011	.088	.281	99	.897	150	.005	-.001	-.017	-.087	-.147	-.258		
47	.367	180			.017	.003	.001	.003	.052	.165	102	.984	0	-.409	-.401	-.433	-.422	-.412	-.419		
48	.486	0			-.054	-.054	-.032	.193	.330	.434	103	.984	30	-.370	-.362	.417	-.435	-.447	-.443		
49	.486	180			-.053	-.060	-.048	.150	.199	.206	104	.984	60	-.304	-.291	-.316	-.380	-.453	-.540		
50	.503	15			-.045	-.052	-.010	.238	.353	.447	105	.984	90	-.314	-.304	-.335	-.399	-.471	-.537		
51	.503	30			-.044	-.052	-.004	.233	.337	.419	106	.984	135	-.353	-.329	-.400	-.439	-.541	-.622		
52	.503	50			-.048	-.056	.010	.239	.341	.423	107	.984	180	-.490	-.476	-.512	-.571	-.634	-.688		
53	.503	F			-.053	-.053	.012	.233	.317	.383											
54	.503	F			-.087	-.060	-.093	-.019	-.044	-.181											

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(d) M = 1.03 - Continued

Pressure coefficients C_p on -

Model orifice number	$\frac{x}{l}$	ϕ , deg	Fuselage							Model orifice number	$\frac{x}{l}$	ϕ , deg	Fuselage - Concluded									
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$		
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 7.5^\circ; \text{brakes closed}$										$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 7.5^\circ; \text{brakes closed}$									
1	.026	0				.289	.329	.412	.505	.604	.716	55	.503	F								
2	.026	90				.329	.338	.304	.241	.133	-.032	56	.503	135			-.048	-.062	-.012	-.080	-.005	-.088
3	.026	180				.374	.325	.255	.202	.161	.138	57	.588	0			-.045	-.058	.003	.149	.075	-.295
4	.084	0				.151	.189	.270	.356	.451	.562	58	.588	180			-.039	.039	.187	.323	.429	.528
5	.084	45				.161	.181	.218	.258	.279	.289	59	.604	15			.125	.043	-.085	-.143	-.174	-.254
6	.084	90				.171	.181	.148	.085	-.020	-.167	60	.604	30			-.061	.022	.170	.309	.417	.516
7	.084	135				.195	.168	.099	.028	-.060	-.142	61	.604	50			-.063	.025	.180	.322	.430	.527
8	.084	180				.227	.183	.120	.071	.035	.006	62	.604	F			-.078	.018	.179	.328	.443	.545
9	.134	J				.258	.217	.161	.121	.067	-.007	63	.604	F			-.106	.008	.188	.349	.472	.578
10	.156	J				.306	.283	.237	.210	.193	.146	64	.604	135			.102	-.010	-.199	-.396	-.560	-.769
12	.173	30				.015	.045	.096	.135	.176	.231	65	.690	0			.106	.011	-.149	-.257	-.560	-.821
13	.173	60				.021	.049	.067	.046	-.014	-.068	66	.690	180			-.179	-.093	.048	.199	.301	.405
14	.173	90				.048	.069	.057	-.002	-.086	-.327	67	.706	15			-.127	-.028	.069	-.186	-.459	-.589
15	.173	135				.132	.131	.114	.092	.064	.008	68	.706	30			-.116	-.022	.072	.190	.279	.375
16	.173	J				.223	.212	.185	.171	.164	.134	69	.706	50			-.097	-.011	.072	.189	.278	.373
17	.173	C				.277	.255	.220	.202	.202	.221	70	.706	F			-.078	-.009	.075	.195	.284	.378
18	.190	J				.124	.123	.111	.104	.085	.050	71	.706	F			-.051	-.097	-.262	-.396	-.646	-.848
19	.190	C				.128	.111	.081	.069	.076	.117	72	.706	135			-.036	-.084	-.286	-.403	-.601	-.835
20	.197	C				-.098	-.132	-.178	-.209	-.221	-.198	73	.753	0			.255	.273	.255	.248	.274	.366
21	.207	C				.041	.033	.018	.018	.035	.083	74	.753	180			.262	.267	.207	-.339	-.463	-.615
23	.216	C				-.236	-.268	-.309	-.330	-.344	-.323	75	.770	15			.302	.323	.322	.299	.292	.366
24	.226	23				-.060	-.029	.030	.076	.129	.200	76	.770	30			.285	.301	.281	.246	.225	.299
25	.226	45				-.052	-.019	.021	.034	.041	.022	77	.770	50			.279	.286	.256	.204	.161	.236
26	.226	68				-.089	-.051	-.024	-.045	-.094	-.138	78	.770	F			.284	.289	.249	.175	.079	.150
27	.226	90				-.105	-.068	-.045	-.072	-.140	-.180	79	.770	F			.290	.309	.234	.157	-.159	-.329
28	.226	135				-.285	-.268	-.241	-.227	-.230	-.295	80	.770	F			.266	.276	.218	.115	-.405	-.403
29	.226	C				-.462	-.537	-.663	-.728	-.742	-.699	81	.770	135			.270	.275	.220	-.005	-.320	-.659
30	.263	C				-.257	-.236	-.234	-.229	-.211	-.092	82	.820	15			.319	.318	.309	.270	.224	.299
31	.283	0				-.153	-.132	-.073	-.010	.093	.231	83	.820	38			.312	.315	.310	.275	.227	.295
32	.299	30				-.154	-.132	-.071	-.004	.055	.159	84	.820	F			.313	.315	.314	.284	.251	.329
33	.299	60				-.178	-.141	-.086	-.008	.065	.164	85	.820	F			.308	.310	.317	.286	.253	.324
34	.299	F				-.187	-.140	-.100	-.040	.012	.068	86	.820	F			.319	.323	.339	.307	.285	.364
35	.299	F				-.105	-.111	-.182	-.270	-.335	-.411	87	.820	F			.330	.335	.364	.337	.369	.503
36	.299	F				-.108	-.173	-.335	-.608	-.765	-.921	88	.820	F			.298	.284	.237	.140	-.242	-.479
37	.299	135				-.075	-.119	-.192	-.274	-.348	-.400	89	.820	F			.287	.284	.256	.161	-.065	-.292
38	.319	C				-.045	-.066	-.095	-.137	-.178	-.191	90	.820	F			.290	.288	.265	.193	-.081	-.090
39	.350	20				-.104	-.098	-.073	-.018	.077	.241	91	.820	F			.289	.294	.279	.214	-.046	-.144
40	.350	40				-.111	-.099	-.076	-.039	.041	.183	92	.820	142			.293	.296	.286	.233	.078	-.029
41	.350	60				-.129	-.109	-.070	-.029	.070	.234	94	.897	30			.480	.496	.535	.548	.562	.594
42	.350	F				-.150	-.114	-.075	-.081	-.016	.102	95	.897	50			.418	.431	.449	.459	.482	.524
43	.350	F				-.095	-.093	-.175	-.418	-.565	-.634	96	.897	F			.360	.370	.391	.401	.418	.463
44	.350	135				-.036	-.075	-.173	-.295	-.812	-.883	97	.897	F			.319	.315	.288	.229	.046	-.134
45	.350	0				-.003	-.021	-.063	-.099	-.138	-.050	98	.897	130			.401	.387	.384	.326	.243	.067
46	.367	0				-.089	-.086	-.064	-.011	.088	.285	99	.897	150			.507	.495	.493	.430	.415	.288
47	.367	180				.017	.005	.002	.003	.050	.165	102	.984	0			-.514	-.507	-.504	-.510	-.503	-.479
48	.486	0				-.054	-.054	-.032	.200	.331	.436	103	.984	30			-.457	-.450	-.463	-.491	-.496	-.513
49	.486	180				-.053	-.058	-.047	.152	.200	.203	104	.984	60			-.575	-.566	-.605	-.639	-.699	-.780
50	.503	15				-.045	-.052	-.002	.242	.354	.451	105	.984	90			-.529	-.540	-.577	-.678	-.679	-.749
51	.503	30				-.046	-.052	.004	.238	.339	.424	106	.984	135			-.780	-.817	-.731	-.698	-.789	-.722
52	.503	50				-.049	-.057	.015	.243	.343	.428	107	.984	180			-.565	-.568	-.584	-.634	-.691	-.701
53	.503	F				-.053	-.055	.020	.239	.319	.388											
54	.503	F				-.082	-.059	-.095	-.016	-.051	-.186											

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(d) $M = 1.03$ - ContinuedPressure coefficients C_p on -

Model orifice number	$\frac{x}{t}$	ϕ , deg	Fuselage								Model orifice number	$\frac{x}{t}$	ϕ , deg	Fuselage - Concluded															
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$								
			$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$											$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$															
1	.026	0	.124	.170		.313	.413	.514	.618	.736	55	.503	F	.442	.292		-.065	.026	-.036	-.042	-.127								
2	.026	90	.095	.317		.416	.374	.288	.165	-.003	56	.503	135	.479	.292		-.068	.042	.157	-.018	-.446								
3	.026	180	.713	.501		.285	.213	.172	.140	.122	57	.588	0	-.382	-.206		.027	.174	.320	.424	.527								
4	.084	0	-.041	.042		.160	.251	.347	.449	.571	58	.588	180	.515	.315		-.004	-.126	-.205	-.269	-.404								
5	.084	45	-.281	.014		.232	.301	.348	.384	.416	59	.604	15	-.624	-.246		.006	.166	.322	.433	.542								
6	.084	90	-.091	.142		.276	.215	.157	.017	-.137	60	.604	30	-.749	-.251		.083	.224	.380	.481	.571								
7	.084	135	.415	.330		.189	.086	-.031	-.156	-.296	61	.604	50	-.776	-.274		.028	.201	.362	.476	.583								
8	.084	180	.560	.347		.157	.091	.037	-.004	-.046	62	.604	F	-.792	-.479		.025	.212	.382	.505	.616								
9	.134	J	.547	.361		.191	.137	.100	.025	-.054	63	.604	F	.604	.369		-.042	-.242	-.428	-.564	-.789								
10	.156	J	.626	.486		.372	.316	.257	.186	.112	64	.604	135	.564	.342		-.030	-.191	-.293	-.691	-.821								
12	.173	30	-.246	-.136		.067	.147	.211	.272	.353	65	.690	0	-.642	-.419		-.111	.032	.204	.303	.407								
13	.173	60	-.448	-.135							66	.690	180	.376	.172		-.149	-.297	-.416	-.553	-.621								
14	.173	90	-.279	-.019		.141	.114	.033	-.141	-.288	67	.706	15	-.799	-.421		-.129	.031	.198	.293	.397								
15	.173	135	.255	.235		.204	.158	.110	.082	-.002	68	.706	30	-.935	-.416		-.117	.045	.212	.308	.407								
16	.173	J	.495	.379		.303	.267	.225	.175	.127	69	.706	50	-.820	-.427		-.109	.054	.218	.310	.405								
17	.173	C	.587	.453		.344	.306	.289	.297	.300	70	.706	F	-.780	-.427		-.113	.058	.223	.312	.407								
18	.190	J	.337	.250		.211	.187	.138	.077	.034	71	.706	F	.354	.162		-.181	-.333	-.525	-.634	-.761								
19	.190	C	.410	.279		.188	.158	.147	.166	.217	72	.706	135	.365	.171		-.164	-.313	-.424	-.739	-.891								
20	.197	C	.168	.023		-.198	-.233	-.244	-.238	-.211	73	.753	0	-.624	-.160		.147	.211	.246	.293	.376								
21	.207	C	.239	.145		.102	.086	.090	.124	.177	74	.753	180	.338	.222		.092	.000	-.170	-.594	-.638								
23	.216	C	.046	.153		-.312	-.345	-.351	-.347	-.333	75	.770	15	-.493	.006		.366	.394	.394	.410	.479								
24	.226	23	-.314	-.197		-.021	.060	.135	.206	.302	76	.770	30	-.695	.021		.295	.311	.299	.308	.383								
25	.226	45	-.427	-.251		.010	.079	.122	.147	.163	77	.770	50	-.537	.067		.242	.253	.224	.213	.288								
26	.226	68	-.616	-.269		-.001	.039	.034	-.007	-.110	78	.770	F	-.463	.128		.224	.225	.168	.107	.178								
27	.226	90	-.494	-.231		.000	.012	-.025	-.103	-.086	79	.770	F	-.317	.139		.247	.207	.057	-.141	-.255								
28	.226	135	-.151	-.235		-.180	-.171	-.174	-.196	-.274	80	.770	F	.069	.167		.222	.171	.022	-.278	-.514								
29	.226	C	-.151	.351		-.729	-.747	-.739	-.717	-.665	81	.770	135	.251	.230		.248	.167	.035	-.414	-.593								
30	.263	C	-.031	-.264		-.312	-.276	-.245	-.181	-.113	82	.820	15	.033	.247														
31	.283	0	-.200	-.190		-.143	-.069	.011	.100	.253	83	.820	38	-.443	.227		.274	.267	.236	.217	.299								
32	.299	30	-.225	-.284		-.144	-.051	.034	.130	.271	84	.820	F	-.114	.195		.255	.256	.237	.229	.313								
33	.299	60	-.497	-.367		-.128	-.041	.045	.136	.253	85	.820	F	-.248	.163		.240	.248	.229	.222	.306								
34	.299	F	-.904	-.478		-.112	-.042	.027	.086	.155	86	.820	F	-.249	.173		.254	.268	.253	.254	.336								
35	.299	F	-.423	-.204		-.061	-.149	-.199	-.268	-.338	87	.820	F	-.343	.174		.269	.291	.297	.349	.455								
36	.299	F	.082	-.007		-.192	-.468	-.598	-.764	-.919	88	.820	F	.103	.122		.240	.181	-.010	-.466	-.706								
37	.299	135	.224	.015		-.200	-.322	-.476	-.699	-.594	89	.820	F	-.010	.125		.240	.198	.061	-.241	-.577								
38	.319	C	.251	.011		-.088	-.146	-.221	-.280	-.336	90	.820	F	-.047	.138		.249	.216	.070	-.094	-.260								
39	.350	20	-.130	-.164		-.122	-.065	.034	.163	.331	91	.820	F	.027	.158		.264	.242	.161	-.120	-.223								
40	.350	40	.042	.215		-.118	-.058	.033	.146	.293	92	.820	142				.295	.290	.219	-.040	-.520								
41	.350	60	-.121	-.271		-.120	-.061	.029	.152	.314	93	.820	165	.198	.228		.341	.337	.250	.191	-.064								
42	.350	F	-.776	-.418		-.123	-.052	-.002	.086	.201	94	.897	30	-.326	-.022		.036	.053	.085	.151	.241								
43	.350	F	-.636	-.324		-.092	-.196	-.372	-.465	-.534	95	.897	50	-.283	-.029		.023	.044	.086	.166	.260								
44	.350	F	.211	.040		-.140	-.293	-.705	-.946	-.894	96	.897	F	-.470	-.056		.012	.039	.089	.176	.268								
45	.350	135	.309	.090		-.065	-.135	-.171	-.179	-.075	97	.897	F	.044	-.009		.062	.042	-.068	-.289	-.387								
46	.367	0	.068	.132		-.115	-.089	-.008	.127	.307	98	.897	130	.020	.019		.079	.068	-.004	-.256	-.494								
47	.367	180	.346	.067		-.041	-.067	-.089	.048	.068	99	.897	150	.003	.017		.078	.060	.006	-.163	-.441								
48	.486	0	.077	.079		-.081	-.054	.217	.330	.444	102	.984	0	-.606	-.473		-.385	-.412	-.423	-.402	-.375								
49	.486	180	.430	.209		-.080	-.069	.087	.112	.053	103	.984	30	-.602	-.459														
50	.503	15	.007	.132		-.070	.050	.282	.393	.503	104	.984	60	-.457	-.371		-.264	-.289	-.404	-.493	-.580								
51	.503	30	.019	.137		-.066	.070	.299	.401	.503	105	.984	90	-.492	-.394		-.279	-.313	-.406	-.487	-.574								
52	.503	50	-.272	.149		-.055	.085	.306	.403	.499	106	.984	135	-.482	-.450		-.377	-.427	-.491	-.614	-.587								
53	.503	F	.136	.131		-.045	.095	.306	.383	.461	107	.984	180	-.449	-.470		-.437	-.474	-.541	-.593	-.659								
54	.503	F	-.276	-.028		-.044	-.046	-.021	-.076	-.231																			

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(d) $M = 1.03$ - ConcludedPressure coefficients C_p on -

Model orifice number	$\frac{x}{l}$	ϕ , deg	Fuselage								Model orifice number	$\frac{x}{l}$	ϕ , deg	Fuselage - Concluded												
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$					
			$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$											$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$												
1	.026	0				.256	.307	.407	.505	.611	.727	55	.503	F				-.045	-.066	.018	-.057	-.037	-.118			
2	.026	90				.413	.415	.378	.300	.180	.007	56	.503	135				-.052	-.071	.031	.158	.020	-.422			
3	.026	180				.342	.289	.219	.179	.143	.123	57	.588	0				-.059	.019	.171	.313	.420	.524			
4	.084	0				.118	.153	.247	.339	.443	.564	58	.588	180				.082	.002	-.115	-.197	-.265	-.408			
5	.084	45				.190	.226	.298	.346	.385	.417	59	.604	15				-.089	-.001	.161	.313	.428	.538			
6	.084	90				.249	.276	.221	.167	.031	-.127	60	.604	30				.135	.202	.272	.388	.473	.481			
7	.084	135				.247	.193	.099	-.015	-.147	-.289	61	.604	50				-.090	.018	.194	.352	.471	.581			
8	.084	180				.208	.159	.097	.045	.001	-.045	62	.604	F				-.115	.013	.205	.372	.499	.614			
9	.134	J				.233	.193	.143	.106	.034	-.050	63	.604	F				.088	-.035	-.223	-.447	-.547	-.786			
10	.156	J				.405	.373	.321	.266	.196	.123	64	.604	135				.083	-.025	-.178	-.262	-.652	-.816			
12	.173	30				.008	.061	.147	.208	.269	.350	65	.690	0				-.183	-.105	.070	.201	.303	.408			
14	.173	90				.121	.138	.123	.047	-.127	-.278	66	.690	180				-.067	-.141	-.283	-.399	-.541	-.618			
15	.173	135				.223	.203	.164	.117	.084	.010	67	.706	15				-.107	-.031	.109	.200	.295	.401			
16	.173	J				.325	.303	.272	.232	.184	.135	68	.706	30				-.115	-.026	.124	.213	.309	.412			
17	.173	C				.373	.344	.311	.293	.299	.308	69	.706	50				-.066	-.004	.133	.220	.312	.410			
18	.190	J				.222	.211	.192	.147	.084	.037	70	.706	F				-.023	-.001	.134	.223	.314	.410			
19	.190	C				.211	.186	.161	.151	.166	.219	71	.706	F				-.074	-.127	-.309	-.487	-.583	-.713			
20	.197	C				-.163	-.198	-.228	-.241	-.236	-.213	72	.706	135				-.055	-.123	-.300	-.412	-.534	-.690			
21	.207	C				.111	.100	.089	.090	.124	.180	73	.753	0				.240	.279	.274	.268	.308	.391			
23	.216	C				-.279	-.313	-.342	-.349	-.345	-.334	74	.753	180				.277	.261	.127	-.096	-.413	-.601			
24	.226	23				-.080	-.029	.056	.129	.205	.300	75	.770	15				.453	.466	.459	.428	.437	.508			
25	.226	45				-.057	.004	.085	.119	.149	.167	76	.770	30				.392	.399	.384	.340	.337	.414			
26	.226	68				-.062	-.008	.039	.036	.000	-.091	77	.770	50				.349	.355	.334	.269	.245	.321			
27	.226	90				-.045	-.006	.014	-.018	-.092	-.129	78	.770	F				.337	.341	.311	.217	.143	.209			
28	.226	135				-.195	-.185	-.170	-.171	-.188	-.269	79	.770	F				.348	.357	.306	.111	-.108	-.181			
29	.226	C				-.633	-.729	-.745	-.740	-.717	-.667	80	.770	F				.337	.340	.277	.108	-.229	-.387			
30	.263	C				-.335	-.316	-.275	-.246	-.184	-.114	81	.770	135				.363	.363	.291	.117	-.552	-.511			
31	.283	0				-.177	-.149	-.072	.004	.092	.247	83	.820	38				.409	.403	.388	.350	.332	.395			
32	.299	30				-.193	-.151	-.055	.028	.125	.268	84	.820	F				.384	.382	.375	.350	.344	.409			
33	.299	60				-.189	-.135	-.045	.040	.132	.250	85	.820	F				.367	.367	.365	.343	.338	.400			
34	.299	F				-.191	-.119	-.041	.023	.085	.155	86	.820	F				.369	.372	.377	.363	.363	.420			
35	.299	F				-.068	-.065	-.143	-.191	-.258	-.331	87	.820	F				.380	.380	.392	.404	.444	.519			
36	.299	F				-.107	-.191	-.444	-.579	-.741	-.912	88	.820	F				.371	.354	.303	.069	-.419	-.681			
37	.299	135				-.139	-.197	-.309	-.454	-.693	-.910	89	.820	F				.361	.356	.319	.154	-.227	-.513			
38	.319	C				-.067	-.089	-.136	-.212	-.269	-.270	90	.820	F				.368	.368	.335	.177	-.067	-.264			
39	.350	20				-.149	-.126	-.069	.026	.155	.326	91	.820	F				.380	.384	.359	.256	-.088	-.209			
40	.350	40				-.162	-.124	-.059	.025	.140	.289	92	.820	142				.404	.414	.400	.315	-.015	-.461			
41	.350	60				-.168	-.125	-.056	.021	.145	.309	93	.820	165				.437	.448	.444	.340	.216	-.060			
42	.350	F				-.194	-.129	-.052	.009	.081	.198	94	.897	30				.576	.583	.592	.598	.618	.648			
43	.350	F				-.102	-.094	-.178	-.360	-.454	-.528	95	.897	50				.489	.494	.499	.505	.531	.569			
44	.350	F				-.061	-.136	-.278	-.638	-.935	-.891	96	.897	F				.417	.423	.429	.439	.470	.507			
45	.350	135				-.016	-.064	-.129	-.166	-.173	-.068	97	.897	F				.381	.381	.343	.234	-.023	-.158			
46	.367	0				-.119	-.117	-.088	-.016	.117	.300	98	.897	130				.467	.471	.441	.355	.080	-.294			
47	.367	180				-.020	-.042	-.064	-.083	.049	.075	99	.897	150				.581	.580	.548	.425	.322	.012			
48	.486	0				-.080	-.083	-.054	.203	.326	.439	102	.984	0				-.515	-.524	-.506	-.512	-.495	-.460			
49	.486	180				-.083	-.084	-.071	.085	.117	.064	104	.984	60				-.581	-.581	-.548	-.620	-.655	-.646			
50	.503	15				.097	.135	.186	.296	.313	.224	105	.984	90				-.589	-.615	-.522	-.657	-.674	-.739			
51	.503	30				-.081	-.069	.049	.288	.399	.500	106	.984	135				-.872	-.881	-.836	-.849	-.724	-.746			
52	.503	50				-.068	-.059	.067	.295	.401	.497	107	.984	180				-.559	-.567	-.550	-.609	-.683	-.712			
53	.503	F				-.071	-.050	.074	.297	.383	.459															
54	.503	F				-.068	-.046	-.076	-.010	-.067	-.240															

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(e) $M = 1.18$ Pressure coefficients C_p on -

Model orifice number	$\frac{x}{l}$	β , deg	Fuselage								Model orifice number	$\frac{x}{l}$	β , deg	Fuselage - Concluded							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$											$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
1	.026	0				.256	.332	.435	.551	.689	55	.503	F				-.041	-.064	-.199	-.457	-.319
2	.026	90				.241	.220	.135	-.008	-.801	56	.503	135				-.046	-.048	-.072	-.192	-.499
3	.026	180				.256	.200	.155	.122	.089	57	.588	0				.036	.152	.321	.430	.540
4	.084	0				.158	.212	.311	.423	.548	58	.588	180				.017	-.042	-.083	-.095	-.198
5	.084	45				.126	.137	.150	.138	.154	59	.604	15				.037	.191	.324	.425	.523
6	.084	90				.117	.097	.029	-.069	-.220	60	.604	30				.043	.203	.344	.446	.537
7	.084	135				.093	.065	.022	-.023	-.084	61	.604	50				.040	.201	.349	.463	.584
8	.084	180				.137	.085	.030	.000	-.030	62	.604	F				.043	.219	.390	.524	.640
9	.134	J				.098	.045	.016	-.007	-.085	63	.604	F				.047	-.080	-.220	-.454	-.567
10	.156	J				.181	.121	.093	.095	.117	64	.604	135				.051	-.060	-.158	-.245	-.397
12	.173	30				.004	.022	.050	.099	.155	65	.690	0				-.045	.094	.223	.364	.475
13	.173	60				.003	-.019	-.061	-.108	-.171	66	.690	180				-.033	-.151	-.263	-.369	-.415
14	.173	90				.030	-.006	-.078	-.189	-.364	67	.706	15				-.063	.069	.203	.340	.442
15	.173	135				.089	.070	.054	.001	-.069	68	.706	30				-.062	.073	.214	.347	.444
16	.173	J				.153	.115	.105	.119	.100	69	.706	50				-.063	.077	.224	.362	.465
17	.173	C				.194	.149	.125	.111	.083	70	.706	F				-.063	.086	.243	.380	.481
18	.190	J				.061	.060	.082	.102	.042	71	.706	F				-.066	-.187	-.503	-.474	-.614
19	.190	C				.060	.032	.039	.034	.040	72	.706	135				-.050	-.175	-.280	-.361	-.486
20	.197	C				-.078	-.127	-.153	-.171	-.192	73	.753	0				-.063	.083	.289	.401	.483
21	.207	C				.025	.027	.023	.023	.058	74	.753	180				-.080	-.193	-.305	-.406	-.411
23	.216	C				-.191	-.223	-.239	-.243	-.276	75	.770	15				.018	.164	.269	.322	.363
24	.226	23				-.019	-.010	.011	.061	.126	76	.770	30				-.020	.128	.233	.287	.330
25	.226	45				-.005	-.018	-.059	-.081	-.094	77	.770	50				-.046	.085	.196	.261	.317
26	.226	68				-.020	-.032	-.098	-.232	-.337	78	.770	F				-.060	.046	.153	.216	.272
27	.226	90				-.024	-.024	-.064	-.155	-.322	79	.770	F				.003	.000	-.069	-.160	-.277
28	.226	135				-.196	-.172	-.154	-.187	-.268	80	.770	F				-.077	-.178	-.315	-.379	-.510
29	.226	C				-.308	-.360	-.479	-.552	-.568	81	.770	135				-.072	-.205	-.318	-.396	-.513
30	.263	C				-.264	-.248	-.220	-.195	-.238	82	.820	15				-.053	-.041	-.004	.044	.100
31	.283	0				-.069	-.032	.009	.068	.159	83	.820	38				-.023	-.017	-.016	.005	.060
32	.299	30				-.078	-.066	-.090	-.120	-.084	84	.820	F				.000	.009	.015	.013	.066
33	.299	60				-.078	-.036	.019	.048	-.004	85	.820	F				.034	.040	.046	-.004	-.011
34	.299	F				-.083	-.052	-.002	.031	.054	86	.820	F								
35	.299	F				-.078	-.139	-.213	-.275	-.349	87	.820	F								
36	.299	F				-.110	-.204	-.422	-.549	-.634	88	.820	F				.109	.142	.174	.156	.078
37	.299	135				-.064	-.101	-.163	-.235	-.331	89	.820	F				.065	.001	-.067	.010	-.223
38	.319	C				-.079	-.126	-.182	-.233	-.259	90	.820	F				.050	.008	-.011	.063	-.059
39	.350	20				-.091	-.066	-.012	.069	.017	91	.820	F				.034	.012	.028	.099	-.024
40	.350	40				-.079	-.065	-.026	.038	.113	92	.820	142				.005	-.002	.033	.074	-.077
41	.350	60				-.075	-.043	.009	.095	.169	94	.897	30				.007	.188	.007	.181	.008
42	.350	F				-.078	-.060	-.045	.028	.113	95	.897	50				.030	.044	.076	.146	.231
43	.350	F				-.081	-.144	-.380	-.468	-.535	96	.897	F				.024	.031	.068	.143	.247
44	.350	F				-.059	-.118	-.222	-.620	-.726	97	.897	F				.022	.030	.066	.148	.264
45	.350	135				-.007	-.038	-.091	-.145	-.239	98	.897	130				-.036	-.052	-.119	-.142	-.212
46	.367	0				-.089	-.048	.029	.132	.196	99	.897	150				.007	.010	-.041	-.063	-.096
47	.367	180				-.006	-.048	-.064	-.085	-.209	102	.984	0				.013	.023	-.002	-.006	-.025
48	.486	0				-.044	-.050	-.015	.082	.193	104	.984	60				-.320	-.322	-.297	-.261	-.226
49	.486	180				-.058	-.074	-.078	-.088	-.173	105	.984	90				-.227	-.223	-.236	-.310	-.389
50	.503	15				-.042	-.065	-.045	.028	.126	106	.984	135				-.248	-.242	-.259	-.324	-.381
51	.503	30				-.040	-.070	-.065	-.024	.068	107	.984	180				-.296	-.278	-.294	-.373	-.456
52	.503	50				-.038	-.059	-.049	-.015	.083							-.365	-.348	-.387	-.428	-.455
53	.503	F				-.031	-.056	-.073	-.071	.017											
54	.503	F				-.041	-.112	-.264	-.308	-.227											

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(e) M = 1.18 - Continued

Pressure coefficients C_p on -

Model orifice number	$\frac{x}{l}$	β , deg	Fuselage								Model orifice number	$\frac{x}{l}$	β , deg	Fuselage - Concluded											
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				
			$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$											$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$											
1	.026	C	.083	.147	.214	.259	.343	.441	.560	.697	55	.503	F	.011	-.073	-.053	-.040	-.068	-.219	-.474	-.321				
2	.026	90	-.217	.125	.242	.242	.217	.132	-.023	-.213	56	.503	135	.051	-.059	-.056	-.045	-.045	-.074	-.202	-.527				
3	.026	180	.710	.441	.301	.254	.192	.151	.114	.086	57	.588	0	-.179	-.087	-.019	.038	.172	.327	.433	.545				
4	.084	0	-.020	.035	.112	.157	.222	.316	.428	.554	58	.588	180	.549	.307	.080	.015	-.046	-.087	-.094	-.204				
5	.084	45	-.076	.037	.102	.126	.142	.150	.136	.155	59	.604	15	-.180	-.108	-.030	.042	.208	.330	.427	.528				
6	.084	90	-.246	.029	.109	.116	.095	.022	-.079	-.240	60	.604	30	-.249	-.112	-.024	.049	.220	.351	.449	.541				
7	.084	135	.142	.109	.102	.093	.062	.020	-.029	-.093	61	.604	50	-.394	-.152	-.026	.045	.219	.356	.469	.590				
8	.084	180	.572	.319	.185	.135	.080	.030	-.003	-.030	62	.604	F	-.577	-.244	-.049	.048	.236	.396	.529	.646				
9	.134	J	.586	.330	.140	.096	.045	.015	-.008	-.095	63	.604	F	.649	.379	.137	.042	-.090	-.246	-.513	-.578				
10	.156	J	.472	.298	.214	.182	.118	.093	.095	.115	64	.604	135	.580	.345	.127	.047	-.065	-.164	-.257	-.410				
12	.173	30	-.145	-.019	-.009	.006	.027	.055	-.103	-.156	65	.690	0	-.478	-.271	-.123	-.040	.109	.229	.368	.480				
13	.173	60	-.202	-.043	.004	.003	.018	-.064	-.113	-.178	66	.690	180	.467	.217	.047	-.036	.159	.268	.374	.426				
14	.173	90	-.396	-.062	.024	.030	.011	-.088	-.204	-.386	67	.706	15	-.382	-.279	-.136	-.059	.084	.211	.344	.447				
15	.173	135	.015	.051	.084	.088	.069	.049	-.005	-.078	68	.706	30	-.418	-.265	-.136	-.056	.089	.222	.352	.448				
16	.173	J			.166	.151	.116	.102	.122	.092	69	.706	50	-.472	-.277	-.137	-.058	.094	.232	.367	.470				
17	.173	C	.422	.271	.211	.195	.148	.124	.106	.082	70	.706	F	-.518	-.300	-.147	-.061	.104	.251	.385	.485				
18	.190	J	.181	.073	.058	.062	.064	.083	.098	.033	71	.706	F	.451	.206	.018	-.068	-.196	-.311	-.489	-.638				
19	.190	C	.272	.132	.077	.058	.034	.040	.027	.036	72	.706	135	.439	.200	.023	-.052	-.181	-.286	-.365	-.500				
20	.197	C	.274	.067	-.038	-.079	-.128	-.154	-.176	-.199	73	.753	0	-.510	-.297	-.136	-.057	.105	.299	.407	.491				
21	.207	C	.151	.061	.033	.025	.033	.022	.022	.057	74	.753	180	.465	.221	-.004	-.083	-.193	-.150	-.405	-.421				
23	.216	C	.125	-.065	-.162	-.190	-.218	-.240	-.244	-.281	75	.770	15	-.394	-.208	-.071	.024	.182	.275	.324	.367				
24	.226	23	-.114	-.044	-.021	-.018	-.008	.015	.062	.128	76	.770	30	-.435	-.286	-.091	-.015	.144	.237	.288	.332				
25	.226	45	-.218	-.065	-.011	-.004	-.019	-.062	-.087	-.099	77	.770	50	-.418	-.257	-.111	-.039	.099	.200	.263	.320				
26	.226	68	-.300	-.124	-.032	-.018	-.033	-.104	-.242	-.347	78	.770	F	-.379	-.247	-.115	-.055	.057	.157	.218	.274				
27	.226	90	-.562	-.191	-.047	-.023	-.022	-.071	-.164	-.335	79	.770	F	-.408	-.039	.010	.003	-.002	-.075	-.164	-.295				
28	.226	135	-.286	-.280	-.221	-.194	-.167	-.154	-.191	-.279	80	.770	F	.200	.086	-.024	-.078	-.184	-.321	-.384	-.525				
29	.226	C	-.093	-.218	-.285	-.311	-.363	-.489	-.562	-.568	81	.770	135	.288	.172	.004	-.075	-.210	-.321	-.410	-.526				
30	.263	C	.003	-.185	-.259	-.261	-.244	-.218	-.196	-.249	82	.820	15	-.418	-.184	.116	.090	-.034	-.001	.044	.100				
31	.283	0	-.243	-.139	-.100	-.069	-.028	.009	.068	.164	83	.820	38	-.345	-.057	.116	.117	-.001	-.016	.006	.067				
32	.299	30	-.289	-.108	-.094	-.074	-.065	-.095	-.126	-.084	84	.820	F	-.062	.013	.129	.119	.033	.020	.016	.093				
33	.299	60	-.321	-.161	-.100	-.076	-.028	.021	.046	-.015	85	.820	F	.125	.040	.139	.123	.058	.051	.007	.025				
34	.299	F	-.352	-.232	-.117	-.081	-.044	.000	.033	.052	86	.820	F	.111	.066										
35	.299	F	-.470	-.155							87	.820	F	.070	.065	.159	.158	.159	.184	.175	.280				
36	.299	F	-.091	-.008	-.059	-.112	-.220	-.451	-.583	-.647	88	.820	F	.022	.115	.097	.079	.004	-.048	.023	-.209				
37	.299	135			-.063	-.063	-.106	-.173	-.247	-.353	89	.820	F	-.011	.073	.074	.064	.012	.004	.068	-.061				
38	.319	C	.073	-.076	-.071	-.080	-.129	-.185	-.246	-.264	90	.820	F	-.159	.023	.048	.050	.020	.038	.103	-.032				
39	.350	20	-.101	-.074	-.098	-.090	-.061	-.007	.069	.016	91	.820	F	-.062	-.032	.001	.026	.010	.038	.075	-.075				
40	.350	40	-.412	-.095	-.088	-.079	-.063	-.023	.040	.113	92	.820	142	.007	-.059	-.038	-.019	-.013	-.022	-.083	-.463				
41	.350	60	-.320	-.127	-.092	-.073	-.040	.012	.098	.170	93	.820	165	.072	-.047										
42	.350	F	-.493	-.168	-.104	-.077	-.059	-.043	.031	.115	94	.897	30	.026	.057	.195	.287	.442	.522	.587	.658				
43	.350	F	-.654	-.270	-.092	-.080	-.154	-.406	-.487	-.554	95	.897	50	-.015	.024	.213	.293	.374	.450	.513	.589				
44	.350	F	.087	.021	.029	-.060	-.125	-.243	-.657	-.738	96	.897	F	-.080	-.018	.197	.310	.378	.425	.476	.553				
45	.350	135	.127	.072	.016	-.009	-.041	-.098	-.153	-.253	97	.897	F	.048	-.086	.288	.256	.215	.248	.197	.012				
46	.367	0	-.197	-.114	-.107	-.088	-.040	.033	.133	.192	98	.897	130	.049	-.043	.309	.322	.301	.363	.316	.167				
47	.367	180	.257	.108	.019	-.007	-.047	-.064	-.088	-.209	99	.897	150	-.008	-.030	.364	.293	.329	.420	.445	.316				
48	.486	0	-.180	-.104	-.044	-.042	-.047	-.012	.084	.198	102	.984	0	-.417	-.345	-.353	-.384	-.369	-.358	-.337	-.299				
49	.486	180	.185	.000	-.050	-.056	-.077	-.080	-.099	-.182	103	.984	30	-.415	-.331										
50	.503	15	-.141	-.032	-.036	-.044	-.063	-.043	.031	.138	104	.984	60	-.335	-.241	-.396	-.409	-.400	-.425	-.469	-.480				
51	.503	30	-.185	-.026	-.031	-.041	-.069	-.063	-.021	.075	105	.984	90	-.349	-.257	-.425	-.201	-.283	-.466	-.509	-.515				
52	.503	50	-.414	-.059	-.029	-.039	-.058	-.048	-.013	.087	106	.984	135	-.354	-.318	-.520	-.501	-.603	-.710	-.645	-.505				
53	.503	F	-.340	-.215	-.031	-.031	-.057	-.073	-.071	.021	107	.984	180	-.348	-.378	-.431	-.403	-.406	-.452	-.515	-.540				
54	.503	F	-.284	-.221	-.063	-.041	-.126	-.297	-.291	-.228															

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(e) M = 1.18 - Continued

Pressure coefficients C_p on -

Model orifice number	$\frac{x}{t}$	β , deg	Fuselage								Model orifice number	$\frac{x}{t}$	β , deg	Fuselage - Concluded							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
1	.026	0	.113	.166		.272	.356	.462	.577	.710	55	.503	F	.079	-.033		-.036	-.069	-.283	-.376	-.311
2	.026	90	-.049	.191		.280	.262	.210	.069	-.079	56	.503	135	.130	-.015		-.029	-.031	-.032	-.146	-.401
3	.026	180	.712	.463		.267	.213	.182	.153	.117	57	.588	0	.018	-.037		.030	.169	.358	.471	.585
4	.084	0	.006	.060		.167	.252	.353	.462	.583	58	.588	180	.592	.367		.041	-.023	-.021	.026	.027
5	.084	45	-.109	.052		.160	.202	.259	.297	.329	59	.604	15	-.142	-.120		.037	.203	.360	.475	.587
6	.084	90	-.123	.094		.158	.157	.108	.001	-.130	60	.604	30	-.442	-.135		.039	.209	.370	.489	.600
7	.084	135	.300	.217		.144	.093	.039	-.052	-.133	61	.604	50	-.491	-.167		.031	.206	.373	.496	.617
8	.084	180	.587	.357		.163	.109	.073	.031	.004	62	.604	F	-.519	-.231		.027	.211	.389	.527	.656
9	.134	J	.589	.344		.114	.073	.050	.011	-.014	63	.604	F	.663	.394		.011	-.127	-.275	-.506	-.546
10	.156	J	.577	.417		.287	.228	.189	.151	.098	64	.604	135	.615	.375		.024	-.080	-.146	-.341	-.684
12	.173	30	-.117	-.038		.027	.053	.113	.178	.269	65	.690	0	-.372	-.247		-.020	.133	.261	.394	.507
13	.173	60	-.272	-.059		.038	.053	.054	.022	-.008	66	.690	180	.491	.256		-.020	-.134	-.220	-.289	-.380
14	.173	90	-.276	-.018		.080	.034	-.026	-.131	-.279	67	.706	15	-.420	-.272		-.039	.109	.243	.376	.484
15	.173	135	.185	.177		.177	.142	.104	.057	-.022	68	.706	30	-.340	-.194		-.039	.110	.248	.380	.486
16	.173	J	.450	.330		.264	.223	.193	.167	.110	69	.706	50	-.532	-.283		-.048	.109	.249	.377	.485
17	.173	C	.541	.396		.313	.271	.240	.229	.230	70	.706	F	-.589	-.307		-.057	.109	.255	.385	.492
18	.190	J	.297	.201		.171	.160	.149	.118	.050	71	.706	F	.452	.221		-.060	-.176	-.283	-.566	-.612
19	.190	C	.372	.231		.157	.132	.125	.132	.151	72	.706	135	.457	.226		-.043	-.166	-.265	-.398	-.473
20	.197	C	.294	.089		-.063	-.104	-.126	-.152	-.147	73	.753	0	-.434	-.287		-.042	.115	.260	.389	.479
21	.207	C	.233	.143		.107	.105	.112	.134	.144	74	.753	180	.451	.196		-.066	-.171	-.261	-.345	-.422
23	.216	C	.149	-.038		-.159	-.185	-.208	-.225	-.217	75	.770	15	-.456	-.282		.059	.217	.347	.423	.490
24	.226	23	-.187	-.063		.006	.040	.084	.151	.233	76	.770	30	-.512	-.290		.001	.145	.277	.356	.424
25	.226	45	-.201	-.089		.016	.042	.038	.046	.065	77	.770	50	-.512	-.271		-.040	.084	.212	.291	.363
26	.226	68	-.415	-.150		.007	.023	-.009	-.111	-.175	78	.770	F	-.474	-.258		-.065	.030	.139	.215	.288
27	.226	90	-.449	-.170		.015	.028	-.003	-.094	-.353	79	.770	F	.008	.007		.039	.001	-.015	-.175	-.320
28	.226	135	-.181	-.190		-.137	-.113	-.102	-.130	-.231	80	.770	F	.210	.073		-.076	-.161	-.304	-.471	-.549
29	.226	C	-.042	-.203		-.415	-.520	-.569	-.581		81	.770	135	.339	.182		-.069	-.199	-.306	-.402	-.619
30	.263	C	-.030	-.179		-.202	-.187	-.170	-.141		82	.820	15	-.064	.162		.160	.151	.160	.171	.232
31	.283	0	-.105	-.110		-.063	-.018	.035	.096	.188	83	.820	38	-.046	.107		.170	.173	.183	.151	.188
32	.299	30	-.271	-.136		-.063	-.005	.043	.043		84	.820	F	-.023	.116		.175	.188	.213	.192	.190
33	.299	60	-.264	-.211		-.087	-.002	.080	.141	.162	85	.820	F	.024	.109		.175	.195	.226	.217	.153
34	.299	F	-.594	-.310		-.091	-.010	.061	.109	.137	86	.820	F	-.038	.130		.192	.224	.266	.269	.315
35	.299	F	-.415	-.156		-.047	-.094	-.121	-.180	-.259	87	.820	F	-.056	.133		.202	.252	.320	.326	.515
36	.299	F	.015	.005		-.115	-.268	-.407	-.519	-.671	88	.820	F	.173	.138		.157	.071	-.021	-.101	-.250
37	.299	135	-.004	.024		-.098	-.162	-.225	-.316	-.366	89	.820	F	.158	.123		.146	.094	.043	.005	-.137
38	.319	C	.071	-.012		-.063	-.088	-.111	-.155	-.183	90	.820	F	.008	.128		.148	.122	.080	.060	-.102
39	.350	20	-.311	-.106		-.050	-.001	.072	.152	.234	91	.820	F	.039	.129		.152	.136	.098	.183	-.062
40	.350	40	-.204	-.130		-.053	-.013	.052	.121	.190	92	.820	142				.153	.147	.096	.194	-.015
41	.350	60	-.226	-.168		-.065	-.018	.056	.140	.230	93	.820	165	.219	.139						
42	.350	F	-.659	-.262		-.078	-.030	.013	.071	.139	94	.897	30	-.103	.042		.069	.071	.099	.164	.253
43	.350	F	-.536	-.262		-.067	-.135	-.286	-.375	-.454	95	.897	50	-.120	.028		.057	.062	.093	.165	.264
44	.350	F	.131	.039		-.087	-.180	-.343	-.691	-.768	96	.897	F	-.250	.013		.044	.056	.088	.164	.269
45	.350	135	.210	.107		-.012	-.054	-.087	-.155	-.203	97	.897	F	.008	.007		.054	.044	-.014	-.140	-.262
46	.367	0	-.051	-.072		-.049	-.004	.078	.170	.273	98	.897	130	.105	.019		.069	.068	.019	-.100	-.235
47	.367	180	.299	.138		.042	.031	.018	-.008	-.028	99	.897	150	.103	.013		.074	.077	.022	-.051	-.143
48	.486	0	-.117	-.023		-.025	-.019	.011	.112	.225	102	.984	0	-.380	-.337		-.308	-.303	-.280	-.246	-.213
49	.486	180	.220	.034		-.030	-.021	-.017	-.015	-.067	103	.984	30	-.376	-.338		-.298	-.296	-.288	-.270	-.244
50	.503	15	-.190	-.047		-.019	-.004	.019	.090	.198	104	.984	60	-.322	-.260		-.221	-.228	-.266	-.325	-.387
51	.503	30	-.282	-.053		-.016	-.007	.012	.061	.160	105	.984	90	-.327	-.278		-.241	-.247	-.264	-.317	-.378
52	.503	50	-.355	-.058		-.021	-.017	.005	.042	.141	106	.984	135	-.303	-.293		-.286	-.299	-.325	-.389	-.467
53	.503	F	-.289	-.281		-.022	-.021	-.012	-.007	.081	107	.984	180	-.272	-.333		-.335	-.331	-.356	-.410	-.415
54	.503	F	-.216	-.315		-.027	-.114	-.253	-.263	-.229											

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(e) M = 1.18 - Continued

Pressure coefficients C_p on -

Model orifice number	$\frac{x}{t}$	β , deg	Fuselage								Model orifice number	$\frac{x}{t}$	β , deg	Fuselage - Concluded							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
1	.026	0			.242	.274	.351	.452	.566	.697	55.	.503	F			-.017	-.031	-.072	-.295	-.380	-.321
2	.026	90			.279	.286	.260	.202	.065	-.084	56	.503	135			-.013	-.028	-.037	-.043	-.154	-.401
3	.026	180			.315	.269	.210	.170	.142	.109	57	.588	0			-.009	.028	.158	.344	.458	.570
4	.084	0			.131	.170	.248	.343	.450	.569	58	.588	180			.096	.041	-.027	-.033	.018	.019
5	.084	45			.149	.165	.201	.251	.292	.324	59	.604	15			-.020	.036	.190	.347	.462	.573
6	.084	90			.157	.162	.156	.099	-.006	-.132	60	.604	30			-.022	.037	.197	.358	.477	.587
7	.084	135			.176	.146	.092	.028	-.061	-.144	61	.604	50			-.036	.028	.194	.361	.484	.603
8	.084	180			.203	.165	.108	.063	.022	-.007	62	.604	F			-.058	.024	.198	.376	.514	.643
9	.134	J			.156	.116	.070	.038	.001	-.025	63	.604	F			.108	.013	-.125	-.279	-.499	-.553
10	.156	J			.321	.293	.228	.179	.142	.086	64	.604	135			.115	.027	-.085	-.156	-.350	-.691
12	.173	30			.007	.032	.079	.135	.194	.263	65	.690	0			-.094	-.021	.127	.249	.381	.494
13	.173	60			.033	.043	.053	.047	.019	-.012	66	.690	180			.064	-.020	-.138	-.233	-.296	-.383
14	.173	90			.075	.084	.033	-.036	-.135	-.282	67	.706	15			-.110	-.040	.103	.232	.365	.472
15	.173	135			.181	.182	.143	.094	.050	-.030	68	.706	30			-.112	-.040	.106	.236	.369	.474
16	.173	J			.283	.268	.223	.182	.157	.100	69	.706	50			-.124	-.048	.105	.238	.368	.473
17	.173	J			.338	.318	.272	.232	.225	.220	70	.706	F			-.138	-.056	.106	.243	.375	.480
18	.190	J			.181	.176	.160	.140	.110	.048	71	.706	F			.021	-.060	-.179	-.295	-.571	-.609
19	.190	C			.182	.163	.135	.120	.126	.141	72	.706	135			.035	-.045	-.168	-.275	-.408	-.676
20	.197	C			-.028	-.064	-.110	-.145	-.166	-.158	73	.753	0			-.129	-.042	.112	.257	.388	.475
21	.207	C			.125	.111	.104	.106	.127	.134	74	.753	180			.006	-.063	-.175	-.269	-.353	-.426
23	.216	C			-.129	-.162	-.195	-.220	-.237	-.228	75	.770	15			-.023	.068	.221	.349	.430	.496
24	.226	23			-.012	.010	.039	.077	.143	.224	76	.770	30			-.072	.011	.148	.279	.360	.429
25	.226	45			-.003	.021	.044	.035	.041	.061	77	.770	50			-.103	-.032	.086	.212	.294	.364
26	.226	68			-.013	.011	.022	-.016	-.115	-.184	78	.770	F			-.102	-.063	.028	.134	.214	.285
27	.226	90			-.008	.016	.026	-.011	-.101	-.359	79	.770	F			.036	.047	.003	-.011	-.170	-.296
28	.226	135			-.144	-.135	-.114	-.112	-.140	-.242	80	.770	F			-.044	-.065	-.151	-.310	-.483	-.357
29	.226	C			-.326	-.421	-.525	-.575	.007	-.570	81	.770	135			.001	-.054	-.191	-.258	-.408	-.618
30	.263	C			-.214	-.201	-.193	-.183	-.151	-.150	82	.820	15			.254	.261	.261	.261	.252	.298
31	.283	0			-.077	-.060	-.021	.028	.084	.175	83	.820	38			.236	.253	.255	.257	.238	.244
32	.299	30			-.087	-.061	-.008	.033	.044	.034	84	.820	F			.235	.253	.262	.274	.271	.273
33	.299	60			-.115	-.087	-.009	.069	.132	.158	85	.820	F			.228	.249	.260	.276	.285	.280
34	.299	F			-.128	-.092	-.022	.051	.101	.131	86	.820	F			.232	.261	.282	.305	.326	.378
35	.299	F			-.052	-.046	-.103	-.128	-.185	-.266	87	.820	F			.236	.270	.306	.343	.406	.568
36	.299	F			-.068	-.112	-.266	-.408	-.520	-.672	88	.820	F			.248	.231	.141	.035	-.056	-.263
37	.299	135			-.060	-.098	-.168	-.237	-.330	-.383	89	.820	F			.243	.226	.171	.097	.065	-.055
38	.319	C			-.046	-.064	-.094	-.119	-.164	-.192	90	.820	F			.237	.231	.199	.141	.099	-.080
39	.350	20			-.069	-.047	-.006	.064	.143	.223	91	.820	F			.241	.240	.214	.153	.124	-.039
40	.350	40			-.077	-.052	-.016	.045	.113	.180	92	.820	142			.256	.250	.228	.146	.277	.090
41	.350	60			-.093	-.065	-.022	.046	.130	.219	94	.897	30			.111	.106	.100	.104	.157	.236
42	.350	F			-.115	-.072	-.030	.008	.067	.132	95	.897	50			.099	.095	.095	.105	.165	.249
43	.350	F			-.079	-.065	-.130	-.288	-.377	-.456	96	.897	F			.086	.084	.089	.102	.165	.255
44	.350	F			-.041	-.086	-.183	-.360	.008	-.778	97	.897	F			.105	.104	.097	.060	-.079	-.212
45	.350	135			.018	-.011	-.060	-.099	-.165	-.211	98	.897	130			.112	.115	.117	.089	-.024	-.181
46	.367	0			-.068	-.048	-.007	.069	.162	.264	99	.897	150			.110	.119	.121	.092	.000	-.120
47	.367	180			.064	.045	.025	.007	-.016	-.045	102	.984	0			-.361	-.365	-.365	-.339	-.308	-.272
48	.486	0			-.035	-.028	-.025	-.002	.100	.209	103	.984	30			-.362	-.369	-.372	-.359	-.334	-.321
49	.486	180			-.030	-.033	-.031	-.028	-.028	-.079	104	.984	60			-.233	-.237	-.247	-.281	-.354	-.423
50	.503	15			-.027	-.020	-.014	.011	.081	.186	105	.984	90			-.249	-.258	-.267	-.290	-.352	-.412
51	.503	30			-.029	.016	-.008	.004	.053	.149	106	.984	135			-.299	-.304	-.321	-.355	-.413	-.508
52	.503	50			-.030	.019	-.019	-.004	.034	.129	107	.984	180			-.457	-.448	-.428	-.470	-.513	-.550
53	.503	F			-.033	.020	-.022	-.021	.015	.069											
54	.503	F			-.054	-.025	-.111	-.259	-.276	-.243											

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(e) $M = 1.18$ - ContinuedPressure coefficients C_p on -

Model orifice number	$\frac{x}{t}$	ϕ , deg	Fuselage								Model orifice number	$\frac{x}{t}$	ϕ , deg	Fuselage - Concluded							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = -7.5^\circ; \text{brakes closed}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = -7.5^\circ; \text{brakes closed}$							
1	.026	0			.239	.276	.351	.451	.563	.693	55	.503	F			-.021	-.039	-.078	-.298	-.385	-.339
2	.026	90			.273	.281	.253	.195	.056	-.094	56	.503	135			-.019	-.031	-.038	-.045	-.164	-.403
3	.026	180			.317	.270	.207	.169	.140	.107	57	.588	0			-.005	.028	.162	.348	.458	.570
4	.084	0			.128	.167	.248	.343	.449	.568	58	.588	180			.100	.043	-.029	-.031	.020	.019
5	.084	45			.141	.167	.195	.254	.282	.310	59	.604	15			-.023	.036	.198	.349	.461	.571
6	.084	90			.152	.157	.157	.094	-.012	-.141	60	.604	30			-.025	.040	.206	.360	.476	.585
7	.084	135			.173	.144	.087	.023	-.061	-.137	61	.604	50			-.040	.032	.202	.363	.484	.602
8	.084	180			.206	.165	.106	.058	.019	-.007	62	.604	F			-.060	.031	.207	.380	.515	.642
9	.134	J			.160	.118	.067	.037	-.001	-.026	63	.604	F			.116	.017	-.129	-.279	-.508	-.546
10	.156	J			.314	.286	.220	.174	.137	.088	64	.604	135			.124	.028	-.086	-.155	-.343	-.661
12	.173	30			.008	.028	.079	.130	.184	.255	65	.690	0			-.099	-.024	.126	.249	.380	.494
13	.173	60			.031	.041	.047	.038	.006	-.022	66	.690	180			.069	-.017	-.140	-.228	-.292	-.379
14	.173	90			.073	.079	.028	-.040	-.283	.104	67	.706	15			-.115	-.043	.104	.232	.364	.472
15	.173	135			.176	.177	.135	.092	.045	-.031	68	.706	30			-.117	-.043	.105	.237	.368	.473
16	.173	J			.274	.263	.213	.179	.152	.101	69	.706	50			-.128	-.052	.103	.238	.367	.473
17	.173	C			.318	.303	.261	.225	.209	.214	70	.706	F			-.143	-.060	.103	.244	.375	.479
18	.190	J			.174	.169	.152	.135	.107	.038	71	.706	F			.024	-.058	-.182	-.294	-.566	-.611
19	.190	C			.167	.157	.127	.111	.115	.131	72	.706	135			.039	-.040	-.172	-.276	-.398	-.660
20	.197	C			-.025	-.064	-.111	-.143	-.164	-.169	73	.753	0			-.135	-.047	.110	.259	.390	.476
21	.207	C			.122	.108	.098	.098	.120	.128	74	.753	180			.009	-.064	-.178	-.271	-.351	-.420
23	.216	C			-.127	-.160	-.197	-.220	-.235	-.227	75	.770	15			-.043	.053	.216	.347	.425	.485
24	.226	23			-.015	.008	.034	.074	.136	.215	76	.770	30			-.092	-.001	.145	.279	.357	.419
25	.226	45			-.006	.017	.040	.029	.031	.049	77	.770	50			-.119	-.044	.084	.212	.291	.356
26	.226	68			-.019	.008	.016	-.024	-.125	-.193	78	.770	F			-.115	-.068	.027	.136	.213	.281
27	.226	90			-.015	.015	.020	-.017	-.104	-.362	79	.770	F			.024	.036	-.009	-.028	-.194	-.330
28	.226	135			-.151	-.139	-.119	-.116	-.139	-.242	80	.770	F			-.044	-.075	-.175	-.314	-.474	-.569
29	.226	C			-.312	-.408	-.515	-.572	-.586	-.568	81	.770	135			.002	-.067	-.208	-.316	-.406	-.605
30	.263	C			-.217	-.210	-.193	-.184	-.161	-.151	82	.820	15			.025	.014	-.004	-.006	.000	.025
31	.283	0			-.081	-.062	-.021	.025	.084	.174	83	.820	38			.080	.081	.071	.059	.025	.052
32	.299	30			-.088	-.063	-.012	.040	.025	.023	84	.820	F			.103	.113	.115	.115	.073	.078
33	.299	60			-.115	-.085	-.003	.066	.125	.145	85	.820	F			.111	.128	.137	.146	.111	.013
34	.299	F			-.129	-.088	-.009	.047	.093	.122	86	.820	F			.135	.160	.183	.204	.183	.141
35	.299	F			-.055	-.045	-.099	-.137	-.195	-.274	87	.820	F			.150	.180	.221	.271	.244	.360
36	.299	F			-.064	-.110	-.257	-.417	-.526	-.661	88	.820	F			.160	.138	.040	-.054	-.136	-.271
37	.299	135			-.056	-.097	-.159	-.233	-.323	-.370	89	.820	F			.144	.124	.067	.013	-.036	-.184
38	.319	C			-.042	-.060	-.092	-.122	-.167	-.193	90	.820	F			.115	.112	.083	.047	.037	-.133
39	.350	20			-.073	-.051	-.009	.059	.137	.218	91	.820	F			.090	.093	.079	.052	.131	-.070
40	.350	40			-.081	-.054	-.019	.040	.106	.173	92	.820	142			.052	.062	.059	.022	.088	-.094
41	.350	60			-.097	-.068	-.023	.044	.125	.214	94	.897	30			.034	.042	.051	.082	.158	.240
42	.350	F			-.120	-.076	-.032	.005	.060	.126	95	.897	50			.022	.027	.035	.069	.155	.246
43	.350	F			-.085	-.068	-.144	-.301	-.389	-.465	96	.897	F			.006	.014	.024	.058	.155	.251
44	.350	F			-.043	-.085	-.183	-.348	-.682	-.762	97	.897	F			-.009	-.009	-.021	-.094	-.215	-.330
45	.350	135			.017	-.011	-.062	-.098	-.166	-.217	98	.897	130			.028	.025	.015	-.045	-.134	-.286
46	.367	0			.071	-.048	-.009	.070	.158	.254	99	.897	150			.053	.048	.032	-.034	-.080	-.163
47	.367	180			.065	.042	.021	.008	-.015	-.041	102	.984	0			-.342	-.350	-.365	-.333	-.305	-.261
48	.486	0			-.026	-.022	-.021	.009	.100	.209	103	.984	30			-.333	-.345	-.367	-.343	-.328	-.303
49	.486	180			-.025	-.025	-.026	-.026	-.032	-.080	104	.984	60			-.251	-.249	-.257	-.274	-.340	-.408
50	.503	15			-.025	-.017	-.010	.007	.077	.162	105	.984	90			-.265	-.265	-.272	-.294	-.352	-.415
51	.503	30			-.026	-.016	-.015	.002	.047	.143	106	.984	135			-.298	-.300	-.309	-.329	-.411	-.487
52	.503	50			-.032	-.021	-.025	-.008	.028	.123	107	.984	180			-.427	-.447	-.441	-.463	-.501	-.532
53	.503	F			-.037	-.023	-.030	-.026	-.021	.063											
54	.503	F			-.060	-.028	-.121	-.265	-.274	-.238											

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(e) M = 1.18 - Continued

Pressure coefficients C_p on -

Model orifice number	$\frac{x}{l}$	ϕ , deg	Fuselage								Model orifice number	$\frac{x}{l}$	ϕ , deg	Fuselage - Concluded							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx 0^\circ$; $\delta_e = 0^\circ$; $\delta_v = 7.5^\circ$; brakes closed											$\beta \approx 0^\circ$; $\delta_e = 0^\circ$; $\delta_v = 7.5^\circ$; brakes closed							
1	.026	0			.242	.280	.357	.455	.571	.697	55	.503	F			-.015	-.030	-.070	-.287	-.376	-.303
2	.026	90			.279	.289	.263	.206	.067	-.084	56	.503	135			-.013	-.032	-.031	-.040	-.150	-.391
3	.026	180			.324	.278	.215	.178	.148	.112	57	.588	0			-.001	.033	.173	.351	.465	.573
4	.084	0			.131	.171	.254	.346	.454	.570	58	.588	180			.114	.051	-.020	-.030	.021	.018
5	.084	45			.147	.177	.203	.255	.293	.321	59	.604	15			-.019	.046	.210	.353	.469	.576
6	.084	90			.155	.165	.161	.096	-.004	-.129	60	.604	30			-.022	.049	.216	.364	.483	.589
7	.084	135			.181	.151	.100	.029	-.057	-.137	61	.604	50			-.037	.040	.211	.368	.491	.606
8	.084	180			.214	.169	.114	.064	.025	-.001	62	.604	F			-.061	.035	.216	.384	.522	.646
9	.134	J			.165	.125	.076	.043	.006	-.021	63	.604	F			.128	.025	-.126	-.266	-.489	-.546
10	.156	J			.326	.300	.230	.179	.145	.093	64	.604	135			.137	.036	-.075	-.151	-.339	-.667
12	.173	30			.016	.042	.095	.145	.200	.270	65	.690	0			-.096	-.021	.129	.253	.385	.494
13	.173	60			.036	.049	.056	.046	.019	-.010	66	.690	180			.080	-.009	-.132	-.228	-.292	-.380
14	.173	90			.080	.090	.040	-.032	-.133	-.273	67	.706	15			-.114	-.038	.107	.235	.368	.471
15	.173	135			.184	.182	.143	.097	.051	-.022	68	.706	30			-.117	-.039	.108	.240	.371	.474
16	.173	J			.283	.271	.222	.184	.160	.111	69	.706	50			-.130	-.051	.107	.241	.371	.473
17	.173	C			.341	.321	.269	.234	.223	.229	70	.706	F			-.143	-.059	.109	.247	.378	.480
18	.190	J			.183	.175	.160	.142	.113	.049	71	.706	F			.034	-.051	-.177	-.289	-.558	-.634
19	.190	C			.184	.164	.136	.121	.126	.140	72	.706	135			.049	-.034	-.166	-.269	-.401	-.657
20	.197	C			-.020	-.059	-.106	-.137	-.160	-.158	73	.753	0			-.132	-.045	.116	.253	.381	.466
21	.207	C			.127	.110	.105	.107	.128	.133	74	.753	180			.018	-.058	-.175	-.266	-.348	-.423
23	.216	C			-.122	-.157	-.191	-.214	-.230	-.226	75	.770	15			-.036	.057	.216	.338	.415	.478
24	.226	23			-.010	.012	.041	.082	.143	.220	76	.770	30			-.086	.000	.144	.268	.347	.413
25	.226	45			-.006	.022	.047	.039	.040	.059	77	.770	50			-.113	-.041	.083	.204	.283	.350
26	.226	68			-.016	.010	.024	-.014	-.114	-.179	78	.770	F			-.108	-.066	.029	.130	.208	.276
27	.226	90			-.012	.019	.027	-.008	-.093	-.353	79	.770	F			.040	.046	.002	-.020	-.176	-.322
28	.226	135			-.148	-.136	-.114	-.108	-.133	-.236	80	.770	F			-.038	-.067	-.156	-.308	-.476	-.566
29	.226	C			-.309	-.406	-.519	-.576	-.585	-.569	81	.770	135			.010	-.058	-.201	-.310	-.402	-.592
30	.263	C			-.210	-.197	-.188	-.176	-.151	-.143	82	.820	15			.232	.220	.177	.169	.178	.289
31	.283	0			-.073	-.057	-.012	.037	.091	.178	83	.820	38			.230	.226	.195	.190	.174	.292
32	.299	30			-.083	-.058	-.004	.051	.042	.033	84	.820	F			.230	.230	.211	.221	.235	.335
33	.299	60			-.112	-.081	-.003	.072	.133	.159	85	.820	F			.221	.225	.215	.232	.261	.353
34	.299	F			-.129	-.086	-.005	.053	.099	.132	86	.820	F			.221	.235	.241	.269	.310	.434
35	.299	F			-.050	-.039	-.092	-.131	-.186	-.263	87	.820	F			.220	.240	.264	.316	.365	.599
36	.299	F			-.055	-.101	-.260	-.409	-.514	-.657	88	.820	F			.206	.195	.102	-.012	-.106	-.303
37	.299	135			-.049	-.091	-.155	-.228	-.319	-.374	89	.820	F			.196	.188	.129	.048	.012	-.073
38	.319	C			-.037	-.057	-.083	-.118	-.160	-.188	90	.820	F			.184	.193	.159	.089	.053	-.102
39	.350	20			-.069	-.046	-.001	.066	.144	.223	91	.820	F			.179	.198	.175	.112	.177	-.041
40	.350	40			-.077	-.052	-.014	.047	.114	.180	92	.820	142			.173	.197	.182	.108	.202	.032
41	.350	60			-.094	-.064	-.020	.050	.132	.220	94	.897	30			.470	.506	.564	.604	.648	.699
42	.350	F			-.117	-.073	-.029	.011	.068	.133	95	.897	50			.429	.455	.488	.520	.569	.631
43	.350	F			-.081	-.064	-.137	-.288	-.375	-.453	96	.897	F			.389	.416	.446	.478	.518	.572
44	.350	F			-.035	-.076	-.174	-.344	-.677	-.767	97	.897	F			.370	.362	.348	.317	.155	-.023
45	.350	135			.025	-.005	-.057	-.095	-.157	-.206	98	.897	130			.444	.420	.423	.401	.341	.151
46	.367	0			-.067	-.050	-.003	.073	.163	.266	99	.897	150			.541	.524	.543	.522	.483	.345
47	.367	180			.069	.047	.027	.010	-.010	-.037	102	.984	0			-.360	-.358	-.364	-.359	-.337	-.303
48	.486	0			-.021	-.022	-.021	.010	.102	.210	103	.984	30			-.317	-.314	.352	-.372	-.382	-.366
49	.486	180			-.023	-.025	-.017	-.024	-.026	-.070	104	.984	60			-.397	-.404	-.399	-.437	-.470	-.488
50	.503	15			-.019	-.013	-.006	.012	.083	.188	105	.984	90			-.229	-.218	.336	-.453	-.492	-.488
51	.503	30			-.020	-.012	-.011	.004	.054	.149	106	.984	135			-.599	-.645	-.628	-.582	-.601	-.529
52	.503	50			-.026	-.017	-.021	-.003	.034	.129	107	.984	180			-.420	-.423	-.438	-.481	-.510	-.527
53	.503	F			-.031	-.016	-.024	-.020	-.013	.070											
54	.503	F			-.058	-.023	-.117	-.254	-.260	-.227											

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Continued

(e) M = 1.18 - Continued

Pressure coefficients C_p on -

Model orifice number	$\frac{x}{l}$	β , deg	Fuselage								Model orifice number	$\frac{x}{l}$	β , deg	Fuselage - Concluded							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$											$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
1	.026	0	.093	.158		.255	.344	.448	.577	.725	55	.503	F	.158	.004		-.047	-.088	-.396	-.262	-.323
2	.026	90	.092	.276		.365	.333	.244	.126	.006	56	.503	135	.202	.024		-.049	-.056	-.069	-.139	-.467
3	.026	180	.703	.441		.242	.191	.152	.112	.088	57	.588	0	-.190	-.081		.009	.114	.309	.431	.560
4	.084	0	-.028	.035		.154	.225	.326	.445	.581	58	.588	180	.556	.334		.002	-.053	-.092	-.119	-.193
5	.084	45	-.241	.028		.212	.282	.337	.392	.441	59	.604	15	-.414	-.133		.010	.150	.327	.460	.591
6	.084	90	-.037	.153		.239	.216	.152	.052	-.084	60	.604	30	-.536	-.146		.059	.123	.288	.445	.607
7	.084	135	.430	.312		.177	.093	-.016	-.132	-.269	61	.604	50	-.595	-.195		.024	.174	.362	.507	.638
8	.084	180	.567	.324		.139	.078	.030	-.008	-.036	62	.604	F	-.576	-.343		.025	.191	.380	.535	.674
9	.134	J	.584	.337		.083	.040	.003	-.023	-.080	63	.604	F	.665	.382		-.005	-.153	-.372	-.466	-.571
10	.156	J	.683	.526		.376	.298	.214	.131	.062	64	.604	135	.622	.363		-.005	-.107	-.175	-.429	-.704
12	.173	30	-.180	-.076		.056	.136	.210	.293	.380	65	.690	0	-.435	-.271		-.028	.113	.242	.378	.494
13	.173	60	-.398	.057							66	.690	180	.468	.223		-.063	-.172	-.277	-.384	-.412
14	.173	90	.189	.057		.144	.078	.008	-.093	-.227	67	.706	15	-.620	-.269		-.043	.108	.239	.377	.491
15	.173	135	.333	.299		.249	.193	.109	.049	-.071	68	.706	30	-.750	-.266		-.033	.120	.254	.391	.504
16	.173	J	.565	.441		.352	.304	.237	.164	.090	69	.706	50	-.645	-.269		-.028	.128	.262	.397	.506
17	.173	C	.656	.516		.408	.367	.326	.317	.280	70	.706	F	-.568	-.204		.030	.132	.271	.403	.509
18	.190	J	.418	.323		.268	.241	.182	.110	.027	71	.706	F	.458	.216		-.087	-.195	-.394	-.501	-.543
19	.190	C	.488	.349		.256	.234	.214	.224	.241	72	.706	135	.467	.220		-.077	-.173	-.277	-.487	-.706
20	.197	C	.248	.052		-.101	-.136	-.167	-.183	-.192	73	.753	0	-.426	-.318		-.040	.117	.306	.406	.489
21	.207	C	.344	.256		.214	.207	.203	.232	.249	74	.753	180	.461	.202		-.090	-.191	-.300	-.416	-.426
23	.216	C	.128	-.053		.047	.112	.117	.152	.188	75	.770	15	-.508	-.064		.230	.370	.463	.529	.595
24	.226	23	-.237	-.103		.047	.112	.117	.152	.188	76	.770	30	-.505	-.064		.091	.250	.397	.425	.496
25	.226	45	-.324	-.147		.059	.099	.054	-.018	-.067	77	.770	50	-.466	-.140		-.019	.123	.255	.329	.405
26	.226	68	-.523	-.159		.080	.098	.054	-.018	-.067	78	.770	F	-.434	-.131		-.063	.027	.147	.227	.303
27	.226	90	-.378	-.107		.080	.098	.054	-.018	-.067	79	.770	F	-.259	-.045		.112	.100	-.024	-.141	-.194
28	.226	135	-.085	-.093		-.055	-.042	-.051	-.129	-.266	80	.770	F	.225	.080		.005	.001	-.253	-.174	-.454
29	.226	C	-.025	-.229		-.573	-.578	-.579	-.564	-.549	81	.770	135	.390	.233		-.028	-.087	-.286	-.499	-.430
30	.263	C	.010	-.168		-.273	-.245	-.214	-.192	-.244	82	.820	15	.191	.230						
31	.283	0	-.249	-.137		-.076	-.030	.019	.089	.202	83	.820	38	-.328	.204		.330	.319	.306	.318	.375
32	.299	30	-.296	-.192		-.065	.017	.093	.167	.149	84	.820	F	.047	.195		.317	.314	.312	.332	.395
33	.299	60	-.433	-.271		-.072	.026	.115	.199	.250	85	.820	F	-.100	.171		.304	.305	.307	.331	.392
34	.299	F	-.746	-.358		-.070	.029	.108	.168	.208	86	.820	F	-.153	.155		.311	.319	.339	.372	.436
35	.299	F	-.351	-.120		-.033	-.029	-.063	-.122	-.201	87	.820	F	-.374	.121		.325	.332	.397	.482	.575
36	.299	F	.077	.045		-.117	-.277	-.378	-.499	-.655	88	.820	F	.338	.264		.287	.204	-.098	-.290	-.509
37	.299	135	.158	.031		-.125	-.245	-.461	-.532	-.520	89	.820	F	.223	.231		.281	.215	.063	-.111	-.583
38	.319	C	.047	-.081		-.096	-.129	-.185	-.237	-.276	90	.820	F	.178	.230		.294	.232	.079	.051	-.179
39	.350	20	-.328	-.152		-.077	-.002	.079	.161	.259	91	.820	F	.219	.242		.313	.256	.180	.021	-.190
40	.350	40	-.200	-.203		-.077	-.003	.072	.163	.249	92	.820	142				.337	.294	.162	.111	-.412
41	.350	60	-.640	-.228		-.080	-.014	.069	.173	.280	93	.820	165	.348	.317		.386	.313	.142	.178	.113
42	.350	F	-.712	-.340		-.084	-.020	.051	.132	.203	94	.897	30	-.268	.093		.113	.111	.121	.194	.303
43	.350	F	-.486	-.227		-.047	-.138	-.214	-.286	-.380	95	.897	50	-.184	.085		.098	.099	.117	.195	.320
44	.350	F	.178	.033		-.147	-.262	-.591	-.724	-.794	96	.897	F	-.302	.067		.086	.093	.117	.197	.330
45	.350	135	.257	.070		-.093	-.158	-.197	-.232	-.221	97	.897	F	.120	.081		.143	.125	.045	-.146	-.214
46	.367	0	-.180	-.127		-.072	-.013	.063	.168	.248	98	.897	130	.138	.085		.154	.154	.113	-.129	-.277
47	.367	180	.274	.106		-.005	-.037	-.066	-.096	-.218	99	.897	150	.138	.080		.161	.156	.132	-.030	-.258
48	.486	0	-.176	-.101		-.055	-.048	-.011	.091	.211	102	.984	0	-.440	-.345		-.331	-.326	-.309	-.276	-.231
49	.486	180	.202	.017		-.057	-.069	-.085	-.125	-.215	103	.984	30	-.426	-.334						
50	.503	15	-.181	-.049		-.045	-.026	.007	.108	.266	104	.984	60	-.335	-.257		-.219	-.210	-.273	-.355	-.431
51	.503	30	-.205	-.053		-.040	-.015	.014	.111	.269	105	.984	90	-.358	-.273		-.239	-.228	-.271	-.338	-.407
52	.503	50	-.514	-.068		-.029	-.015	-.001	.089	.274	106	.984	135	-.302	-.285		-.275	-.291	-.334	-.450	-.349
53	.503	F	-.335	-.286		-.029	-.006	.010	.052	.231	107	.984	180	-.340	-.370		-.378	-.348	-.397	-.444	-.478
54	.503	F	-.336	-.306		-.034	-.118	-.300	-.427	-.328											

TABLE II.- PRESSURE COEFFICIENTS MEASURED ON FUSELAGE - Concluded

(e) $M = 1.18$ - ConcludedPressure coefficients C_p on -

Model orifice number	$\frac{x}{t}$	β , deg	Fuselage								Model orifice number	$\frac{x}{t}$	β , deg	Fuselage - Concluded							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$											$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
1	.026	0			.211	.253	.341	.447	.575	.714	55	.503	F			-.018	-.050	-.091	-.389	-.250	-.316
2	.026	90			.361	.362	.334	.247	.132	.013	56	.503	135			-.027	-.054	-.060	-.070	-.126	-.431
3	.026	180			.295	.241	.185	.151	.115	.087	57	.588	0			-.030	.007	.108	.307	.429	.551
4	.084	0			.111	.154	.222	.325	.441	.572	58	.588	180			.070	-.002	-.055	-.091	-.123	-.199
5	.084	45			.178	.203	.280	.336	.393	.439	59	.604	15			-.050	.007	.142	.323	.456	.582
6	.084	90			.235	.233	.217	.156	.057	-.074	60	.604	30			.197	.298	.170	.233	.344	.471
7	.084	135			.232	.170	.095	-.012	-.129	-.264	61	.604	50			-.058	.021	.168	.359	.504	.631
8	.084	180			.197	.138	.079	.032	-.006	-.036	62	.604	F			-.082	.020	.186	.377	.532	.667
9	.134	J			.139	.083	.039	.005	-.018	-.082	63	.604	F			.098	-.012	-.156	-.367	-.475	-.567
10	.156	J			.423	.377	.300	.217	.135	.068	64	.604	135			.093	-.007	-.109	-.177	-.421	-.694
12	.173	30			.016	.040	.133	.209	.290	.374	65	.690	0			-.112	-.034	.108	.239	.375	.489
14	.173	90			.146	.136	.080	.011	-.089	-.218	66	.690	180			.030	-.063	-.173	-.280	-.394	-.406
15	.173	135			.270	.244	.195	.111	.053	-.060	67	.706	15			-.124	-.047	.102	.236	.374	.485
16	.173	J			.378	.351	.305	.238	.170	.094	68	.706	30			-.117	-.037	.115	.251	.389	.499
17	.173	C			.438	.406	.367	.326	.319	.286	69	.706	50			-.122	-.035	.122	.259	.395	.502
18	.190	J			.283	.265	.241	.184	.116	.033	70	.706	F			-.129	-.039	.125	.267	.401	.504
19	.190	C			.281	.252	.234	.215	.226	.245	71	.706	F			.015	-.088	-.196	-.392	-.508	-.536
20	.197	C			-.062	-.108	-.137	-.165	-.182	-.192	72	.706	135			.028	-.077	-.175	-.279	-.469	-.676
21	.207	C			.226	.210	.204	.201	.231	.257	73	.753	0			-.130	-.043	.114	.307	.405	.490
23	.216	C			-.160	-.200	-.222	-.229	-.232	-.257	74	.753	180			-.014	-.093	-.143	-.104	-.420	-.412
24	.226	23			-.027	.004	.075	.133	.219	.308	75	.770	15			.159	.239	.374	.467	.532	.602
25	.226	45			.001	.026	.107	.115	.152	.184	76	.770	30			.050	.109	.257	.361	.429	.503
26	.226	68			.014	.056	.097	.057	-.016	-.065	77	.770	50			-.040	-.004	.130	.260	.331	.411
27	.226	90			.045	.076	.095	.055	-.139	-.300	78	.770	F			-.030	-.055	.027	.150	.229	.306
28	.226	135			-.062	-.058	-.045	-.052	-.124	-.257	79	.770	F			.112	.117	.116	.004	-.121	-.178
29	.226	C			-.530	-.584	-.585	-.583	-.569	-.552	80	.770	F			-.014	.022	.046	.001	-.152	-.429
30	.263	C			-.258	-.282	-.248	-.211	-.187	-.236	81	.770	135			.044	.001	.031	-.099	-.340	-.411
31	.283	0			-.094	-.077	-.033	.020	.088	.196	82	.820	38			.357	.360	.342	.333	.371	.457
32	.299	30			-.115	-.072	.012	.091	.166	.149	84	.820	F			.338	.347	.337	.338	.387	.476
33	.299	60			-.131	-.080	.021	.113	.196	.259	85	.820	F			.324	.334	.327	.333	.387	.472
34	.299	F			-.131	-.089	.023	.105	.166	.210	86	.820	F			.320	.337	.338	.358	.423	.506
35	.299	F			-.008	-.041	-.032	-.062	-.120	-.192	87	.820	F			.333	.348	.348	.408	.523	.621
36	.299	F			-.038	-.127	-.275	-.377	-.499	-.638	88	.820	F			.326	.318	.253	-.004	-.202	-.508
37	.299	135			-.063	-.125	-.243	-.462	-.535	-.530	89	.820	F			.328	.316	.265	.123	-.021	-.544
38	.319	C			-.079	-.104	-.131	-.186	-.237	-.280	90	.820	F			.330	.328	.279	.130	.081	-.119
39	.350	20			-.111	-.078	-.006	.078	.160	.254	91	.820	F			.341	.347	.296	.205	.069	-.129
40	.350	40			-.125	-.078	-.006	.072	.162	.245	92	.820	142			.359	.370	.330	.205	.158	-.417
41	.350	60			-.126	-.082	-.017	.068	.172	.273	93	.820	165			.401	.416	.342	.194	.218	.092
42	.350	F			-.149	-.086	-.022	.051	.133	.200	94	.897	30			.632	.641	.652	.670	.708	.753
43	.350	F			-.063	-.062	-.139	-.212	-.286	-.369	95	.897	50			.524	.535	.550	.577	.621	.674
44	.350	F			-.066	-.148	-.262	-.599	-.731	-.787	96	.897	F			.461	.478	.494	.520	.560	.614
45	.350	135			-.043	-.097	-.161	-.200	-.232	-.219	97	.897	F			.441	.435	.382	.268	.097	-.027
46	.367	0			-.085	-.074	-.018	.063	.164	.243	98	.897	130			.516	.514	.481	.373	.172	-.077
47	.367	180			.033	-.007	-.041	-.066	-.095	-.224	99	.897	150			.645	.643	.582	.442	.394	.138
48	.486	0			-.057	-.054	-.051	-.012	.088	.204	102	.984	0			-.329	-.387	-.376	-.364	-.321	-.284
49	.486	180			-.054	-.066	-.072	-.085	-.124	-.212	104	.984	60			-.376	-.397	-.402	-.428	-.473	-.464
50	.503	15			.116	.176	.079	.063	.073	.095	105	.984	90			-.244	-.225	-.330	-.503	-.473	-.597
51	.503	30			-.069	-.042	-.019	.012	.109	.242	106	.984	135			-.681	-.744	-.693	-.657	-.584	-.568
52	.503	50			-.056	-.029	-.017	-.001	.088	.245	107	.984	180			-.423	-.404	-.409	-.452	-.529	-.494
53	.503	F			-.052	-.026	-.009	-.003	.050	.204											
54	.503	F			-.053	-.052	-.118	-.296	-.437	-.348											

TABLE III.- PRESSURE COEFFICIENTS MEASURED ON RIGHT WING

(a) $M = 0.60$ Pressure coefficients C_p on -

Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Lower surface								Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Upper surface								
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$	
			$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$											$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$								
200	.180	.000				.630	.131	-1.013	-2.040	-1.853	202	.180	.100				-.043	-.526	-1.975	-1.762	-1.788	
201	.180	.100				-.029	.223	.418	.549	.634	204	.180	.200				-.060	-.268	-.351	-1.405	-1.653	
203	.180	.200				-.060	.135	.286	.414	.519	206	.180	.300				-.079	-.208	-.270	-1.064	-1.421	
205	.180	.300				-.078	.072	.204	.317	.422	208	.180	.400				-.084	-.176	-.270	-.804	-1.111	
207	.180	.400				-.094	.026	.135	.233	.333	210	.180	.500				-.095	-.164	-.251	-.668	-.955	
209	.180	.500				-.094	.000	.086	.168	.257	212	.180	.600				-.099	-.149	-.226	-.588	-.823	
211	.180	.600				-.087	-.016	.051	.113	.189	214	.180	.700				-.060	.095	-.181	-.568	-.686	
213	.180	.700				-.046	.009	.053	.093	.147	216	.180	.800				-.036	.058	-.179	-.329	-.581	
215	.180	.800				-.014	.022	.048	.062	.088	218	.180	.900				-.022	.037	-.163	-.158	-.491	
217	.180	.900				-.003	.017	.026	.008	-.015												
$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$																						
200	.180	.000	-1.646	-1.240	.387	.632	.080	-1.073	-2.022	-1.826	202	.180	.100	.625	.417	.136	-.040	-.577	-2.038	-1.708	-1.773	
201	.180	.100	-.599	-2.143	-.226	-.007	.245	.435	.562	.645	204	.180	.200	.508	.290	.068	-.054	-.279	-.437	-1.439	-1.645	
203	.180	.200	-1.494	-.316	-.173	.037	.156	.303	.430	.534	206	.180	.300	.413	.207	.028	-.068	-.201	-.265	-1.107	-1.416	
205	.180	.300	-1.307	-.294	-.152	-.055	.096	.221	.332	.437	208	.180	.400	.317	.138	.003	-.068	-.166	-.258	-.844	-1.107	
207	.180	.400	-1.110	-.258	-.140	-.071	.047	.153	.249	.350	210	.180	.500	.249	.088	-.018	-.077	-.151	-.239	-.697	-.947	
209	.180	.500	-.946	-.227	-.122	-.070	.023	.106	.185	.275	212	.180	.600	.169	.039	-.035	-.079	-.131	-.219	-.618	-.815	
211	.180	.600	-.800	-.218	-.097	-.062	.008	.070	.131	.207	214	.180	.700	.126	.030	-.012	-.037	-.078	-.187	-.562	-.675	
213	.180	.700	-.679	-.188	-.043	-.019	.031	.071	.110	.165	216	.180	.800	.069	.020	.006	.014	-.036	-.203	-.311	-.578	
215	.180	.800	-.560	-.223	-.004	.013	.046	.067	.079	.108	218	.180	.900	-.019	-.001	.018	.008	-.012	-.167	-.169	-.496	
217	.180	.900	-.449	-.183	.017	.027	.042	.044	.026	.004												
$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$																						
200	.180	.000	-1.532	-1.042		.714	.232	-.768	-1.707	-1.636	202	.180	.100	.670	.432		-.049	-.638	-1.768	-1.548	-1.590	
201	.180	.100	-1.492	-1.715		-.029	.243	.442	.578	.676	204	.180	.200	.534	.283		-.068	-.275	-.928	-1.272	-1.542	
203	.180	.200	-1.453	-.995		-.059	.143	.303	.429	.542	206	.180	.300	.428	.191		-.079	-.227	-.307	-1.010	-1.380	
205	.180	.300	-1.308	-.337		-.074	.080	.209	.323		208	.180	.400	.329	.129		-.080	-.191	-.256	-.806	-.1086	
207	.180	.400	-1.052	-.271		-.087	.033	.138	.231	.342	210	.180	.500	.253	.082		-.088	-.170	-.252	-.693	-.863	
209	.180	.500	-.817	-.239		-.083	.010	.089	.163	.263	212	.180	.600	.172	.035		-.087	-.142	-.232	-.612	-.753	
211	.180	.600	-.691	-.205		-.073	-.004	.052	.105	.192	214	.180	.700	.121	.037		-.042	-.084	-.177	-.535	-.693	
213	.180	.700	-.631	-.155		-.023	.023	.057	.080	.147	216	.180	.800	.053	.028		-.010	-.038	-.127	-.473	-.646	
215	.180	.800	-.594	-.106		.013	.043	.053	.041	.085	218	.180	.900	-.055	.007		.009	-.007	-.084	-.408	-.587	
217	.180	.900	-.538	-.065		.030	.044	.033	-.035	-.027												
$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$																						
200	.180	.000			.429	.703	.289	-.736	-1.670	-1.593	202	.180	.100			.143	-.039	-.558	-1.722	-1.505	-1.547	
201	.180	.100			-.272	-.044	.235	.439	.574	.670	204	.180	.200			.069	-.055	-.261	-.921	-1.231	-1.506	
203	.180	.200			-.206	-.067	.135	.303	.421	.539	206	.180	.300			.025	-.071	-.219	-.289	-.971	-1.349	
205	.180	.300			-.182	-.081	.075	.210	.321	.433	208	.180	.400			-.004	-.071	-.180	-.253	-.770	-1.057	
207	.180	.400			-.163	-.090	.032	.138	.231	.340	210	.180	.500			-.024	-.077	-.159	-.243	-.671	-.841	
209	.180	.500			-.138	-.085	.009	.091	.165	.262	212	.180	.600			-.037	-.074	-.131	-.220	-.593	-.735	
211	.180	.600			-.106	-.071	-.002	.057	.108	.192	214	.180	.700			-.004	-.034	-.071	-.164	-.526	-.679	
213	.180	.700			-.046	-.020	.028	.062	.082	.149	216	.180	.800			.018	.000	-.025	-.116	-.471	-.636	
215	.180	.800			.001	.019	.050	.061	.046	.089	218	.180	.900			.031	.021	.009	-.067	-.408	-.578	
217	.180	.900			.030	.039	.055	.043	-.030	-.023												

TABLE III.- PRESSURE COEFFICIENTS MEASURED ON RIGHT WING - Continued

(a) $M = 0.60$ - ConcludedPressure coefficients C_p on -

Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Lower surface								Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Upper surface										
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$			
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = -7.5^\circ; \text{brakes closed}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = -7.5^\circ; \text{brakes closed}$										
200	.180	.000			.437	.700	.255	-.769	-1.709	-1.704	202	.180	.100			.141	-.040	-.557	-1.764	-1.578	-1.644			
201	.180	.100			-.270	-.045	.235	.438	.571	.670	203	.180	.200			.066	-.062	-.278	-.855	-1.298	-1.574			
203	.180	.200			-.208	-.071	.133	.300	.418	.536	204	.180	.300			.022	-.076	-.229	-.270	-1.030	-1.376			
205	.180	.300			-.183	-.084	.074	.208	.317	.433	206	.180	.400			-.005	-.080	-.186	-.247	-.813	-1.082			
207	.180	.400			-.168	-.096	.026	.136	.229	.338	208	.180	.500			-.028	-.089	-.169	-.245	-.691	-.874			
209	.180	.500			-.144	-.092	.002	.086	.160	.259	210	.180	.600			-.045	-.090	-.149	-.230	-.603	-.762			
211	.180	.600			-.116	-.081	.011	.050	.101	.186	212	.180	.700			-.018	-.048	-.088	-.176	-.522	-.701			
213	.180	.700			-.058	-.034	.016	.053	.077	.142	214	.180	.800			.001	-.019	-.044	-.130	-.454	-.652			
215	.180	.800			-.016	.001	.033	.048	.038	.079	216	.180	.900			.009	-.004	-.015	-.088	-.387	-.590			
217	.180	.900			.008	.017	.032	.026	-.037	-.034	218	.180												
$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 7.5^\circ; \text{brakes closed}$																								
200	.180	.000			.437	.705	.280	-.737	-1.646	-1.581	202	.180	.100			.151	-.025	-.478	-1.593	-1.508	-1.540			
201	.180	.100			-.253	.035	.243	.442	.573	.674	203	.180	.200			.078	-.043	-.271	-.880	-1.214	-1.498			
203	.180	.200			-.191	-.057	.143	.308	.421	.544	204	.180	.300			.034	-.056	-.219	-.326	-.943	-1.337			
205	.180	.300			-.167	-.070	.085	.219	.322	.440	206	.180	.400			.008	-.059	-.174	-.254	-.738	-1.045			
207	.180	.400			-.150	-.079	.042	.151	.238	.349	208	.180	.500			-.010	-.066	-.152	-.235	-.640	-.831			
209	.180	.500			-.122	-.072	.020	.105	.170	.272	210	.180	.600			-.022	-.063	-.125	-.204	-.565	-.725			
211	.180	.600			-.090	-.056	.011	.073	.115	.201	212	.180	.700			.009	-.022	-.066	-.153	-.502	-.669			
213	.180	.700			-.029	-.006	.040	.081	.093	.158	214	.180	.800			.031	.012	-.018	-.103	-.445	-.627			
215	.180	.800			.018	.033	.062	.083	.057	.100	216	.180	.900			.043	.037	.018	-.045	-.366	-.566			
217	.180	.900			.047	.055	.068	.064	-.017	-.010	218	.180												
$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$																								
200	.180	.000	-1.898	-.715		.762	.207	-.560	-1.587	-1.675	202	.180	.100	.706	.455		-.097	-.997	-1.177	-1.467	-1.606			
201	.180	.100	-1.495	-1.139		-.010	.281	.476	.613	.723	203	.180	.200	.555	.302		-.110	-.316	-1.019	-1.080	-1.535			
203	.180	.200	-1.321	-.995		-.040	.170	.321	.450	.577	204	.180	.300	.445	.214		-.113	-.221	-.866	-.809	-1.278			
205	.180	.300	-.923	-.845		-.056	.103	.229	.343	.463	206	.180	.400	.342	.147		-.108	-.185	-.660	-.709	-.932			
207	.180	.400	-.778	-.670		-.068	.054	.154	.249	.371	208	.180	.500	.263	.093		-.103	-.160	-.514	-.674	-.825			
209	.180	.500	-.755	-.523		-.066	.030	.104	.177	.288	210	.180	.600	.176	.050		-.094	-.133	-.410	-.647	-.794			
211	.180	.600	-.739	-.384		-.056	.017	.063	.114	.211	212	.180	.700	.121	.044		-.036	-.068	-.318	-.621	-.758			
213	.180	.700	-.708	-.271		-.005	.046	.064	.082	.164	214	.180	.800	.046	.034		.004	-.020	-.231	-.578	-.717			
215	.180	.800	-.668	-.181		.035	.065	.053	.037	.095	216	.180	.900	-.084	-.001		.022	.017	-.156	-.525	-.660			
217	.180	.900	-.622	-.105		.059	.071	.018	-.063	-.032	218	.180												
$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$																								
200	.180	.000			.543	.770	.291	-.501	-1.460	-1.754	202	.180	.100			.141	-.056	-.830	-1.159	-1.398	-1.550			
201	.180	.100			-.236	-.030	.269	.466	.611	.719	203	.180	.200			.069	-.061	-.265	-1.016	-1.006	-1.466			
203	.180	.200			-.183	-.045	.164	.315	.451	.572	204	.180	.300			.032	-.068	-.207	-.827	-.753	-1.191			
205	.180	.300			-.159	-.058	.101	.228	.344	.462	206	.180	.400			.006	-.065	-.172	-.600	-.656	-.865			
207	.180	.400			-.140	-.068	.059	.157	.255	.370	208	.180	.500			-.006	-.064	-.144	-.447	-.624	-.779			
209	.180	.500			-.110	-.058	.039	.110	.184	.289	210	.180	.600			-.012	-.053	-.116	-.344	-.597	-.755			
211	.180	.600			-.076	-.042	.031	.075	.124	.215	212	.180	.700			.026	-.003	-.046	-.251	-.571	-.725			
213	.180	.700			-.010	.012	.062	.079	.098	.171	214	.180	.800			.055	.036	.005	-.172	-.536	-.685			
215	.180	.800			.043	.058	.087	.076	.057	.106	216	.180	.900			.076	.064	.043	-.103	-.484	-.628			
217	.180	.900			.077	.085	.094	.048	-.038	-.016	218	.180												

TABLE III.- PRESSURE COEFFICIENTS MEASURED ON RIGHT WING - Continued

(b) $M = 0.90$ Pressure coefficients C_p on -

Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Lower surface								Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Upper surface									
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$		
			$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$											$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$									
200	.180	.000				.687	.460	-.034	-.742	-1.165	202	.180	.100				-.036	-.652	-1.287	-1.223	-1.380		
201	.180	.100				-.013	.278	.472	.614	.703	204	.180	.200				-.076	-.501	-1.221	-1.149	-1.270		
203	.180	.200				-.059	.184	.348	.484	.591	206	.180	.300				-.105	-.405	-.514	-1.014	-1.098		
205	.180	.300				-.100	.106	.253	.384	.496	208	.180	.400				-.123	-.388	-.481	-.849	-.986		
207	.180	.400				-.138	.040	.171	.296	.409	210	.180	.500				-.145	-.348	-.515	-.795	-.975		
209	.180	.500				-.140	-.001	.108	.224	.335	212	.180	.600				-.155	-.181	-.492	-.759	-.987		
211	.180	.600				-.127	-.032	.050	.158	.265	214	.180	.700				-.071	-.083	-.387	-.687	-.939		
213	.180	.700				-.049	.003	.045	.133	.229	216	.180	.800				-.022	-.028	-.311	-.653	-.792		
215	.180	.800				.005	.034	.033	.098	.180	218	.180	.900				.006	-.002	-.282	-.619	-.745		
217	.180	.900				.028	.040	-.019	.022	.085													
			$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$											$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$									
200	.180	.000	-1.265	-.190		.586	.697	.425	-.067	-.782	-1.187	202	.180	.100	.701	.470	.157	-.042	-.685	-1.251	-1.091	-1.306	
201	.180	.100	-1.303	-.013		-.248	.022	.302	.485	.622	.709	204	.180	.200	.590	.346	.080	-.073	-1.145	-1.057	-1.196		
203	.180	.200	-1.013	-1.208		-.207	-.026	.207	.363	.495	.601	206	.180	.300	.499	.257	.030	-.095	-.395	-.659	-.934	-1.025	
205	.180	.300	-1.044	-.381		-.219	-.066	.129	.268	.395	.508	208	.180	.400	.410	.176	-.007	-.105	-.316	-.521	-.799	-.952	
207	.180	.400	-.924	-.490		-.218	-.101	.067	.187	.308	.423	210	.180	.500	.336	.108	-.041	-.120	-.200	-.473	-.748	-.941	
209	.180	.500	-.903	-.508		-.166	-.100	.029	.126	.238	.350	212	.180	.600	.252	.034	-.070	-.118	-.173	-.430	-.708	-.948	
211	.180	.600	-.869	-.467		-.112	-.087	.001	.072	.173	.281	214	.180	.700	.207	.020	-.016	-.039	-.083	-.344	-.656	-.873	
213	.180	.700	-.845	-.381		-.025	-.014	.033	.068	.149	.246	216	.180	.800	.148	-.002	.024	.014	-.021	-.290	-.613	-.748	
215	.180	.800	-.794	-.310		.032	.040	.061	.056	.114	.197	218	.180	.900	.047	-.061	.047	.042	.023	-.238	-.578	-.759	
217	.180	.900	-.664	-.277		.060	.065	.067	.005	.037	.102												
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$									
200	.180	.000	-1.150	-.018		.772	.600	.169	-.413			202	.180	.100	.744	.492		-.039	-.566	-1.131	-1.133		
201	.180	.100	-1.185	-1.015		-.021	.294	.479	.629			204	.180	.200	.616	.352		-.085	-.446	-1.135	-1.020		
203	.180	.200	-1.138	-1.007		-.073	.177	.346	.486			206	.180	.300	.514	.264		-.119	-.441	-.896	-.915		
205	.180	.300	-1.044	-.872		-.113	.099	.246	.383			208	.180	.400	.418	.178		-.135	-.435	-.633	-.814		
207	.180	.400	-.959	-.726		-.155	.032	.160	.290			210	.180	.500	.343	.111		-.158	-.410	-.529	-.761		
209	.180	.500	-.910	-.637		-.151	-.016	.097	.214			212	.180	.600	.257	.034		-.155	-.158	-.479	-.715		
211	.180	.600	-.876	-.559		-.130	-.037	.038	.146			214	.180	.700	.209	.020		-.058	-.044	-.400	-.683		
213	.180	.700	-.861	-.459		-.039	.010	.039	.123			216	.180	.800	.150	.000		-.001	.011	-.341	-.654		
215	.180	.800	-.777	-.373		.025	.052	.031	.089			218	.180	.900	.045	-.057		.035	.043	-.281	-.612		
217	.180	.900	-.713	-.288		.058	.070	-.017	.013														
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$									
200	.180	.000				.670	.773	.611	.183	-.392	-1.002	202	.180	.100				.157	-.033	-.514	-1.095	-1.106	-1.213
201	.180	.100				-.287	-.020	.291	.481	.630	.740	204	.180	.200				.076	-.079	-.426	-1.100	-1.012	-1.171
203	.180	.200				-.256	-.069	.178	.349	.491	.612	206	.180	.300				.015	-.113	-.428	-.874	-.901	-1.069
205	.180	.300				-.286	-.110	.098	.248	.383	.508	208	.180	.400				-.027	-.130	-.426	-.646	-.799	-.939
207	.180	.400				-.303	-.150	.031	.163	.291	.414	210	.180	.500				-.068	-.150	-.400	-.532	-.746	-.895
209	.180	.500				-.214	-.145	-.007	.100	.216	.336	212	.180	.600				-.094	-.141	-.142	-.482	-.698	-.899
211	.180	.600				-.127	-.118	-.031	.043	.148	.264	214	.180	.700				-.026	-.045	-.035	-.402	-.667	-.851
213	.180	.700				-.034	-.028	.017	.044	.125	.228	216	.180	.800				.023	.014	.022	-.337	-.641	-.792
215	.180	.800				.033	.038	.062	.037	.093	.181	218	.180	.900				.055	.053	.058	-.276	-.603	-.713
217	.180	.900				.073	.074	.084	-.008	.018	.089												

TABLE III.- PRESSURE COEFFICIENTS MEASURED ON RIGHT WING - Continued

(b) $M = 0.90$ - ConcludedPressure coefficients C_p on -

Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Lower surface								Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Upper surface							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = -7.5^\circ; \text{brakes closed}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = -7.5^\circ; \text{brakes closed}$							
200	.180	.000			.666	.769	.602	.176	-.409	-1.010	202	.180	.100			.158	-.032	-.523	-1.163	-1.156	-1.34
201	.180	.100			-.295	-.026	.290	.483	.627	.738	204	.180	.200			.075	-.075	-.422	-1.135	-1.052	-1.273
203	.180	.200			-.258	-.074	.176	.350	.488	.607	206	.180	.300			.013	-.112	-.426	-.895	-.948	-1.102
205	.180	.300			-.287	-.116	.097	.250	.382	.505	208	.180	.400			-.029	-.129	-.423	-.558	-.838	-.978
207	.180	.400			-.309	-.158	.028	.164	.290	.414	210	.180	.500			-.072	-.156	-.418	-.511	-.779	-.928
209	.180	.500			-.232	-.154	-.013	.101	.214	.334	212	.180	.600			-.105	-.157	-.178	-.477	-.730	-.938
211	.180	.600			-.136	-.131	-.041	.043	.145	.261	214	.180	.700			-.041	-.061	-.047	-.391	-.695	-.881
213	.180	.700			-.045	-.041	.004	.042	.121	.226	216	.180	.800			.004	-.003	.006	-.328	-.663	-.827
215	.180	.800			.018	.021	.045	.033	.087	.175	218	.180	.900			.029	.028	.034	-.282	-.621	-.708
217	.180	.900			.051	.051	.062	-.016	.012	.082											
$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 7.5^\circ; \text{brakes closed}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 7.5^\circ; \text{brakes closed}$										
200	.180	.000			.668	.777	.582	.167	-.413	-1.009	202	.180	.100			.168	-.041	-.655	-1.042	-1.060	-1.200
201	.180	.100			-.282	.002	.310	.491	.637	.744	204	.180	.200			.086	-.085	-.553	-1.034	-.993	-1.158
203	.180	.200			-.242	-.046	.196	.361	.500	.616	206	.180	.300			.028	-.117	-.513	-.852	-.883	-1.052
205	.180	.300			-.267	-.086	.119	.261	.394	.513	208	.180	.400			-.012	-.128	-.463	-.650	-.786	-.922
207	.180	.400			-.258	-.121	.054	.177	.301	.421	210	.180	.500			-.046	-.136	-.356	-.552	-.733	-.877
209	.180	.500			-.167	-.109	.018	.117	.230	.344	212	.180	.600			-.068	-.112	-.171	-.489	-.690	-.882
211	.180	.600			-.102	-.084	-.004	.063	.163	.274	214	.180	.700			-.002	-.022	-.051	-.405	-.650	-.838
213	.180	.700			-.011	.001	.041	.064	.141	.239	216	.180	.800			.044	.036	.007	-.330	-.624	-.761
215	.180	.800			.056	.064	.084	.058	.108	.191	218	.180	.900			.078	.076	.020	-.259	-.589	-.698
217	.180	.900			.095	.099	.101	.013	.032	.099											
$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$											$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$										
200	.180	.000	-1.018	.115		.848	.658	.222	-.359	-.972	202	.180	.100	.784	.518		-.095	-1.113	-1.018	-1.167	-1.269
201	.180	.100	-.986	-.949		.014	.332	-.539	.687	.797	204	.180	.200	.639	.377		-.128	-.504	-.916	-1.091	-1.142
203	.180	.200	-1.004	-.845		-.034	.216	.391	.541	.659	206	.180	.300	.534	.272		-.156	-.486	-.838	-.985	-1.089
205	.180	.300	-1.000	-.731		-.081	.130	.285	.429	.553	208	.180	.400	.432	.187		-.163	-.478	-.731	-.869	-.995
207	.180	.400	-.980	-.661		-.126	.056	.194	.331	.456	210	.180	.500	.353	.117		-.171	-.496	-.666	-.813	-.979
209	.180	.500	-.959	-.602		-.124	.014	.127	.255	.375	212	.180	.600	.265	.038		-.135	-.147	-.579	-.762	-.956
211	.180	.600	-.930	-.532		-.101	-.013	.065	.182	.300	214	.180	.700	.219	.031		-.033	-.016	-.491	-.721	-.914
213	.180	.700	-.894	-.459		-.009	.036	.062	.155	.260	216	.180	.800	.159	.007		.031	.039	-.415	-.684	-.860
215	.180	.800	-.850	-.388		.058	.080	.049	.120	.209	218	.180	.900	.051	-.057		.076	.073	-.346	-.641	-.817
217	.180	.900	-.797	-.318		.099	.105	-.008	.037	.111											
$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$											$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$										
200	.180	.000			.777	.846	.673	.249	-.316	-.950	202	.180	.100			.154	-.061	-1.085	-.949	-1.033	-1.145
201	.180	.100			-.224	.015	.328	.539	.683	.795	204	.180	.200			.079	-.097	-.457	-.825	-1.010	-1.068
203	.180	.200			-.213	-.032	.217	.390	.537	.656	206	.180	.300			.021	-.121	-.457	-.746	-.944	-1.022
205	.180	.300			-.249	-.072	.135	.287	.427	.551	208	.180	.400			-.017	-.124	-.448	-.650	-.824	-.929
207	.180	.400			-.238	-.105	.068	.200	.333	.456	210	.180	.500			-.046	-.124	-.286	-.584	-.762	-.919
209	.180	.500			-.142	-.093	.034	.137	.258	.378	212	.180	.600			-.053	-.094	-.076	-.501	-.706	-.902
211	.180	.600			-.084	-.065	.015	.080	.189	.304	214	.180	.700			.020	.001	.009	-.421	-.649	-.865
213	.180	.700			.011	.024	.068	.081	.166	.266	216	.180	.800			.075	.065	.066	-.350	-.598	-.830
215	.180	.800			.083	.091	.115	.073	.131	.218	218	.180	.900			.114	.112	.106	-.283	-.559	-.780
217	.180	.900			.131	.135	.141	.020	.050	.119											

TABLE III.- PRESSURE COEFFICIENTS MEASURED ON RIGHT WING - Continued

(c) $M = 0.95$ Pressure coefficients C_p on -

Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Lower surface								Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Upper surface							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$											$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
200	.180	.000				.706	.574	.132	-.551	-1.001	202	.180	.100				-.012	-.442	-1.063	-1.279	-1.345
201	.180	.100				.000	.274	.478	.637	.734	203	.180	.200				-.056	-.353	-1.011	-1.312	-1.211
203	.180	.200				-.049	.174	.357	.510	.625	204	.180	.300				-.097	-.342	-.436	-.993	-1.048
205	.180	.300				-.101	.093	.264	.412	.533	206	.180	.400				-.129	-.334	-.481	-.750	-.971
207	.180	.400				-.165	.022	.181	.325	.448	208	.180	.500				-.170	-.361	-.524	-.720	-.979
209	.180	.500				-.200	-.026	.116	.253	.375	210	.180	.600				-.237	-.426	-.552	-.583	-.978
211	.180	.600				-.200	-.074	.056	.188	.307	212	.180	.700				-.110	-.419	-.452	-.492	-.985
213	.180	.700				-.052	-.037	.054	.167	.272	214	.180	.800				.002	-.075	-.398	-.560	-.967
215	.180	.800				.030	.015	.050	.138	.228	216	.180	.900				.038	.027	-.335	-.633	-.833
217	.180	.900				.060	.048	.009	.072	.143	218	.180	.900								
$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$																					
200	.180	.000	-1.111	-.030	.619	.712	.544	.095	-.592	-1.023	202	.180	.100	.734	.481	.178	-.006	-.485	-1.116	-1.272	-1.248
201	.180	.100	-1.412	-1.102	-.210	.020	.304	.497	.645	.738	204	.180	.200	.627	.360	.099	-.049	-.394	-1.065	-1.271	-1.201
203	.180	.200	-.013	-1.140	-.194	.030	.209	.379	.520	.632	206	.180	.300	.537	.272	.043	-.081	-.356	-.403	-.852	-1.057
205	.180	.300	-1.075	-.457	-.218	.077	.129	.283	.422	.541	208	.180	.400	.452	.190	.002	-.102	-.346	-.376	-.634	-.987
207	.180	.400	-1.014	-.519	-.278	.130	.063	.202	.336	.458	210	.180	.500	.382	.127	-.042	-.128	-.371	-.400	-.672	-1.013
209	.180	.500	-.993	-.568	-.301	.127	.022	.138	.265	.386	212	.180	.600	.299	.046	-.094	-.158	-.430	-.486	-.685	-1.011
211	.180	.600	-.988	-.553	-.142	.108	-.014	.082	.201	.319	214	.180	.700	.256	.037	-.018	-.025	-.145	-.438	-.631	-1.029
213	.180	.700	-1.001	-.449	.015	.002	.028	.080	.180	.285	216	.180	.800	.202	.018	.042	.040	.025	-.344	-.626	-1.020
215	.180	.800	-.979	-.434	.071	.063	.073	.074	-.150	.239	218	.180	.900	.109	-.027	.074	.073	.082	-.225	-.611	-.739
217	.180	.900	-.832	-.336	.098	.092	.096	.031	.082	.154											
$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$																					
200	.180	.000	-.993	.121		.806	.697	.300	-.242	-.860	202	.180	.100	.774	.502		-.015	-.497	-1.067	-1.267	-1.397
201	.180	.100	-.013	-.924		.004	.294	.497	.655	.768	204	.180	.200	.650	.358		-.065	-.334	-.991	-1.215	-1.334
203	.180	.200	-1.357	-.883		-.052	.181	.365	.518	.641	206	.180	.300	.553	.272		-.109	-.349	-.889	-1.235	-1.136
205	.180	.300	-1.159	-.844		-.102	.099	.265	.413	.541	208	.180	.400	.455	.183		-.144	-.355	-.450	-.860	-.956
207	.180	.400	-1.002	-.640		-.169	.021	.178	.322	.449	210	.180	.500	.382	.113		-.180	-.383	-.477	-.337	-.933
209	.180	.500	-.961	-.566		-.215	-.031	.111	.249	.374	212	.180	.600	.297	.026		-.268	-.440	-.520	-.578	-.946
211	.180	.600	-.938	-.548		-.239	-.092	.045	.178	.301	214	.180	.700	.254	.025		-.144	-.447	-.476	-.660	-.900
213	.180	.700	-.779	-.553		-.041	-.043	.044	.157	.268	216	.180	.800	.201	.008		.032	-.077	-.383	-.655	-.723
215	.180	.800	-.804	-.514		.054	.031	.043	.129	.223	218	.180	.900	.102	-.030		.072	.051	-.316	-.615	-.780
217	.180	.900	-.813	-.324		.093	.075	.004	.064	.137											
$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$																					
200	.180	.000			.715	.796	.709	.315	-.228	-.845	202	.180	.100			.178	.004	-.405	-1.066	-1.269	-1.362
201	.180	.100			-.230	-.013	.283	.492	.655	.768	204	.180	.200			.094	-.044	-.315	-1.002	-1.218	-1.307
203	.180	.200			-.219	-.061	.176	.364	.519	.642	206	.180	.300			.029	-.091	-.335	-.887	-1.227	-1.139
205	.180	.300			-.248	-.111	.094	.262	.414	.541	208	.180	.400			-.023	-.124	-.342	-.435	-.761	-.924
207	.180	.400			-.301	-.177	.016	.174	.323	.450	210	.180	.500			-.077	-.164	-.369	-.449	-.523	-.888
209	.180	.500			-.342	-.224	-.033	.108	.248	.374	212	.180	.600			-.172	-.251	-.428	-.485	-.645	-.888
211	.180	.600			-.371	-.238	-.089	.045	.180	.302	214	.180	.700			-.058	-.068	-.410	-.420	-.646	-.720
213	.180	.700			-.082	-.016	-.027	.045	.159	.269	216	.180	.800			.034	.045	-.021	-.365	-.620	-.764
215	.180	.800			.069	.067	.049	.045	.131	.225	218	.180	.900			.079	.087	.080	-.288	-.612	-.792
217	.180	.900			.113	.107	.094	.009	.066	.141											

TABLE III.- PRESSURE COEFFICIENTS MEASURED ON RIGHT WING - Continued

(c) M = 0.95 - Concluded

Pressure coefficients C_p on -

Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Lower surface								Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Upper surface							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = -7.5^\circ; \text{brakes closed}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = -7.5^\circ; \text{brakes closed}$							
200	.180	.000			.725	.795	.687	.292	-.238	-.856	202	.180	.100			.175	-.005	-.486	-1.065	-1.261	-1.397
201	.180	.100			-.249	-.011	.290	.498	.657	.764	204	.180	.200			.093	-.054	-.342	-.995	-1.215	-1.330
203	.180	.200			-.224	-.062	.177	.368	.521	.639	206	.180	.300			.027	-.100	-.356	-.866	-1.250	-1.107
205	.180	.300			-.247	-.112	.096	.266	.418	.538	208	.180	.400			-.022	-.135	-.363	-.457	-.870	-.930
207	.180	.400			-.299	-.178	.019	.178	.325	.448	210	.180	.500			-.078	-.173	-.389	-.514	-.622	-.923
209	.180	.500			-.339	-.223	-.032	.112	.252	.371	212	.180	.600			-.177	-.260	-.447	-.583	-.436	-.938
211	.180	.600			-.369	-.246	-.090	.047	.182	.298	214	.180	.700			-.103	-.145	-.452	-.604	-.492	-.959
213	.180	.700			-.172	-.045	-.041	.045	.162	.266	216	.180	.800			.013	.025	-.079	-.527	-.621	-.927
215	.180	.800			.048	.050	.028	.044	.134	.221	218	.180	.900			.058	.063	.045	-.258	-.674	-.644
217	.180	.900			.094	.085	.068	.007	.068	.136											
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 7.5^\circ; \text{brakes closed}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 7.5^\circ; \text{brakes closed}$							
200	.180	.000			.712	.793	.690	.309	-.228	-.855	202	.180	.100			.188	.014	-.412	-.963	-1.029	-1.308
201	.180	.100			-.222	-.004	.297	.497	.653	.767	204	.180	.200			.105	-.033	-.315	-.862	-1.000	-1.250
203	.180	.200			-.207	-.053	.190	.370	.519	.643	206	.180	.300			.042	-.077	-.336	-.752	-.935	-1.087
205	.180	.300			-.244	-.100	.111	.271	.414	.542	208	.180	.400			-.006	-.109	-.341	-.379	-.771	-.877
207	.180	.400			-.294	-.160	.040	.185	.324	.453	210	.180	.500			-.055	-.141	-.367	-.375	-.704	-.730
209	.180	.500			-.333	-.178	-.002	.121	.250	.377	212	.180	.600			-.118	-.191	-.422	-.421	-.647	-.777
211	.180	.600			-.273	-.109	-.036	.060	.184	.307	214	.180	.700			-.007	-.006	-.128	-.370	-.599	-.843
213	.180	.700			.025	.014	.027	.064	.164	.274	216	.180	.800			.063	.062	.065	-.299	-.560	-.826
215	.180	.800			.098	.087	.091	.064	.137	.230	218	.180	.900			.103	.104	.120	-.176	-.489	-.748
217	.180	.900			.135	.127	.127	.026	.071	.144											
			$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$											$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
200	.180	.000	-.902	.265		.876	.762	.366	-.178	-.819	202	.180	.100	.816	.524		-.070	-.889	-1.068	-1.287	-1.379
201	.180	.100	-.013	-1.084		.036	.326	.547	.704	.825	204	.180	.200	.674	.385		-.107	-.404	-.972	-1.228	-1.364
203	.180	.200	-.013	-.819		-.019	.214	.400	.561	.692	206	.180	.300	.572	.284		-.151	-.395	-.952	-1.165	-1.270
205	.180	.300	-.013	-.700		-.071	.125	.294	.451	.589	208	.180	.400	.473	.196		-.181	-.392	-.828	-1.133	-1.092
207	.180	.400	-.013	-.688		-.147	.043	.202	.354	.494	210	.180	.500	.396	.125		-.205	-.423	-.617	-.876	-.881
209	.180	.500	-.851	-.667		-.196	-.012	.132	.277	.413	212	.180	.600	.309	.039		-.301	-.468	-.577	-.848	-.780
211	.180	.600	-.964	-.667		-.227	-.074	.064	.206	.339	214	.180	.700	.268	.037		-.098	-.497	-.449	-.680	-.890
213	.180	.700	-.915	-.667		-.001	-.010	.065	.184	.301	216	.180	.800	.209	.028		.066	-.061	-.266	-.701	-.877
215	.180	.800	-.858	-.364		.085	.058	.062	.153	.252	218	.180	.900	.108	-.007		.110	.075	-.230	-.645	-.832
217	.180	.900	-.820	-.147		.129	.103	.020	.083	.164											
			$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$											$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
200	.180	.000			.819	.871	.781	.395	-.138	-.793	202	.180	.100			.172	-.015	-.870	-1.049	-1.034	-1.368
201	.180	.100			-.192	.020	.318	.539	.697	.819	204	.180	.200			.096	-.064	-.372	-.953	-1.024	-1.359
203	.180	.200			-.198	-.036	.213	.395	.554	.683	206	.180	.300			.028	-.105	-.371	-.858	-.999	-1.152
205	.180	.300			-.234	-.086	.129	.291	.445	.581	208	.180	.400			-.023	-.137	-.374	-.634	-.909	-.975
207	.180	.400			-.286	-.153	.053	.202	.351	.486	210	.180	.500			-.068	-.162	-.407	-.495	-.888	-.936
209	.180	.500			-.330	-.185	.011	.137	.277	.411	212	.180	.600			-.119	-.197	-.452	-.441	-.788	-.924
211	.180	.600			-.249	-.082	-.022	.075	.207	.337	214	.180	.700			.026	.029	-.125	-.401	-.712	-.875
213	.180	.700			.059	.040	.050	.078	.188	.302	216	.180	.800			.092	.095	.086	.362	-.633	-.842
215	.180	.800			.130	.116	.117	.076	.158	.255	218	.180	.900			.141	.144	.150	-.298	-.559	-.823
217	.180	.900			.170	.162	.159	.032	.088	.167											

TABLE III.- PRESSURE COEFFICIENTS MEASURED ON RIGHT WING - Continued

(a) $M = 1.03$ Pressure coefficients C_p on -

Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Lower surface								Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Upper surface							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$											$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
200	.180	.000				.742	.640	.187	-.426	-.797	202	.180	.100				.031	-.374	-.997	-1.151	-1.208
201	.180	.100				.057	.348	.578	.715	.797	203	.180	.200				.001	-.306	-.984	-1.061	-.963
203	.180	.200				.014	.254	.462	.595	.696	204	.180	.300				-.038	-.264	-.381	-.740	-.847
205	.180	.300				-.032	.173	.369	.502	.609	206	.180	.400				-.076	-.262	-.385	-.617	-.800
207	.180	.400				-.095	.098	.288	.420	.529	208	.180	.500				-.110	-.287	-.423	-.634	-.817
209	.180	.500				-.138	.048	.225	.351	.459	210	.180	.600				-.187	-.346	-.480	-.662	-.820
211	.180	.600				-.178	-.020	.162	.290	.395	212	.180	.700				-.208	-.361	-.485	-.672	-.834
213	.180	.700				-.198	-.038	.156	.269	.365	214	.180	.800				-.204	-.356	-.494	-.682	-.831
215	.180	.800				-.186	-.027	.150	.243	.326	216	.180	.900				-.220	-.361	-.488	-.689	-.837
217	.180	.900				-.166	-.025	.115	.184	.249	218	.180	.900								
			$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$											$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
200	.180	.000	-.909	.037	.690	.748	.611	.169	-.448	-.814	202	.180	.100	.804	.587	.215	.017	-.400	-.988	-1.162	-1.213
201	.180	.100	-1.199	-.954	-.134	.077	.367	.586	.721	.801	203	.180	.200	.704	.471	.140	-.011	-.324	-.980	-1.052	-.964
203	.180	.200	-.968	-.961	-.127	.031	.274	.472	.602	.701	204	.180	.300	.621	.385	.086	-.046	-.278	-.403	-.754	-.851
205	.180	.300	-.861	-.360	-.153	-.021	.193	.378	.510	.615	206	.180	.400	.535	.305	.036	-.081	-.270	-.383	-.633	-.807
207	.180	.400	-.834	-.406	-.200	-.083	.119	.298	.427	.536	208	.180	.500	.471	.240	-.007	-.112	-.294	-.419	-.648	-.831
209	.180	.500	-.841	-.449	-.234	-.126	.070	.234	.360	.469	210	.180	.600	.394	.158	-.093	-.188	-.351	-.473	-.676	-.828
211	.180	.600	-.846	-.474	-.264	-.164	.009	.174	.298	.405	212	.180	.700	.354	.143	-.117	-.208	-.366	-.482	-.685	-.838
213	.180	.700	-.858	-.495	-.256	-.182	.029	.169	.278	.375	214	.180	.800	.307	.124	-.046	-.093	-.305	-.489	-.692	-.836
215	.180	.800	-.861	-.500	-.016	.002	.073	.163	.251	.334	216	.180	.900	.220	.076	.122	.133	-.029	-.303	-.696	-.844
217	.180	.900	-.855	-.362	.121	.157	.095	.127	.192	.257	218	.180	.900								
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
200	.180	.000	-.798	.205		.841	.773	.373	-.130	-.668	202	.180	.100	.842	.590		.033	-.638	-.919	-1.094	-1.192
201	.180	.100	-1.198	-.842		.055	.362	.584	.732	.835	203	.180	.200	.724	.458		-.004	-.264	-.862	-1.049	-1.121
203	.180	.200	-1.148	-.848		.002	.254	.459	.603	.714	204	.180	.300	.633	.376		-.044	-.266	-.783	-1.044	-.922
205	.180	.300	-.972	-.746		-.032	.173	.362	.504	.620	206	.180	.400	.543	.287		-.082	-.272	-.340	-.627	-.786
207	.180	.400	-.851	-.471		-.097	.095	.276	.417	.534	208	.180	.500	.473	.218		-.104	-.297	-.377	-.530	-.789
209	.180	.500	-.794	-.425		-.144	.045	.210	.346	.463	210	.180	.600	.393	.130		-.184	-.350	-.446	-.590	-.802
211	.180	.600	-.787	-.460		-.182	-.027	.141	.279	.393	212	.180	.700	.354	.117		-.217	-.373	-.464	-.608	-.809
213	.180	.700	-.795	-.488		-.198	-.052	.136	.259	.363	214	.180	.800	.303	.096		-.213	-.366	-.481	-.616	-.823
215	.180	.800	-.808	-.500		-.195	-.046	.132	.233	.323	216	.180	.900	.214	.042		-.225	-.308	-.470	-.622	-.830
217	.180	.900	-.799	-.414		-.181	-.041	.100	.176	.247	218	.180	.900								
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
200	.180	.000		.793	.833	.767	.388	-.117	-.650		202	.180	.100		.233	.029	-.642	-.908	-1.093	-1.154	
201	.180	.100		-.133	.051	.362	.575	.724	.836		203	.180	.200		.155	-.008	-.259	-.860	-1.045	-1.097	
203	.180	.200		-.144	.002	.255	.452	.597	.717		204	.180	.300		.096	-.048	-.259	-.784	-1.043	-.921	
205	.180	.300		-.155	-.031	.176	.354	.496	.621		206	.180	.400		.046	-.088	-.267	-.337	-.641	-.774	
207	.180	.400		-.204	-.095	.097	.268	.409	.535		208	.180	.500		.002	-.110	-.293	-.376	-.530	-.772	
209	.180	.500		-.244	-.143	.047	.202	.338	.462		210	.180	.600		-.092	-.188	-.345	-.446	-.588	-.781	
211	.180	.600		-.277	-.179	-.025	.135	.272	.395		212	.180	.700		-.131	-.218	-.369	-.464	-.608	-.793	
213	.180	.700		-.291	-.195	-.049	.131	.254	.364		214	.180	.800		-.134	-.214	-.359	-.480	-.616	-.803	
215	.180	.800		-.281	-.192	-.044	.129	.229	.326		216	.180	.900		-.154	-.222	-.242	-.353	-.607	-.808	
217	.180	.900		-.149	-.171	-.038	.098	.173	.250		218	.180	.900								

TABLE III.- PRESSURE COEFFICIENTS MEASURED ON RIGHT WING - Continued

(a) $M = 1.03$ - ConcludedPressure coefficients C_p on -

Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Lower surface								Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Upper surface							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = -7.5^\circ; \text{brakes closed}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = -7.5^\circ; \text{brakes closed}$							
200	.180	.000			.793	.834	.769	.387	-.114	-.660	202	.180	.100			.232	.040	-.602	-.893	-1.061	-1.166
201	.180	.100			-.136	.051	.363	.584	.731	.831	204	.180	.200			.156	.003	-.258	-.841	-1.028	-1.112
203	.180	.200			-.147	.001	.253	.459	.602	.711	206	.180	.300			.095	-.041	-.256	-.741	-1.022	-.906
205	.180	.300			-.159	-.032	.175	.363	.503	.618	208	.180	.400			.047	-.077	-.263	-.327	-.592	-.761
207	.180	.400			-.208	-.096	.096	.277	.415	.531	210	.180	.500			.002	-.099	-.287	-.375	-.519	-.759
209	.180	.500			-.246	-.145	.047	.213	.345	.459	212	.180	.600			-.091	-.179	-.340	-.439	-.572	-.777
211	.180	.600			-.280	-.180	-.024	.144	.279	.391	214	.180	.700			-.130	-.213	-.363	-.459	-.594	-.793
213	.180	.700			-.293	-.198	-.050	.139	.260	.361	216	.180	.800			-.131	-.207	-.357	-.472	-.601	-.805
215	.180	.800			-.286	-.194	-.045	.136	.237	.321	218	.180	.900			-.155	-.223	-.352	-.476	-.611	-.812
217	.180	.900			-.227	-.181	-.043	.105	.179	.246											
$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 7.5^\circ; \text{brakes closed}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 7.5^\circ; \text{brakes closed}$										
200	.180	.000			.784	.827	.753	.390	-.101	-.663	202	.180	.100			.221	.032	-.654	-.897	-1.107	-1.184
201	.180	.100			-.136	.054	.369	.584	.729	.834	204	.180	.200			.142	-.004	-.279	-.846	-1.058	-1.130
203	.180	.200			-.137	.010	.261	.461	.602	.716	206	.180	.300			.083	-.048	-.282	-.759	-1.021	-.928
205	.180	.300			-.156	-.032	.181	.363	.502	.619	208	.180	.400			.031	-.087	-.278	-.325	-.635	-.784
207	.180	.400			-.207	-.094	.103	.279	.415	.535	210	.180	.500			-.011	-.109	-.303	-.370	-.529	-.786
209	.180	.500			-.243	-.141	.053	.214	.345	.463	212	.180	.600			-.104	-.188	-.355	-.438	-.573	-.796
211	.180	.600			-.274	-.177	-.017	.147	.279	.396	214	.180	.700			-.136	-.218	-.373	-.456	-.590	-.798
213	.180	.700			-.286	-.193	-.043	.144	.261	.365	216	.180	.800			-.127	-.179	-.257	-.470	-.628	-.801
215	.180	.800			-.086	-.088	.005	.144	.236	.326	218	.180	.900			.120	.140	-.053	-.201	-.446	-.605
217	.180	.900			.143	.169	.077	.113	.178	.250											
$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$											$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$										
200	.180	.000	-.708	.309		.910	.818	.422	-.077	-.628	202	.180	.100	.883	.626		-.021	-.774	-.937	-1.120	-1.179
201	.180	.100	-.849	.000		.085	.407	.646	.782	.889	204	.180	.200	.751	.490		-.065	-.380	-.892	-1.091	-1.220
203	.180	.200	-.659	.000		.036	.296	.505	.646	.761	206	.180	.300	.654	.393		-.082	-.331	-.874	-1.040	-1.125
205	.180	.300	-.550	.000		-.005	.211	.404	.541	.664	208	.180	.400	.558	.304		-.119	-.318	-.761	-.981	-.935
207	.180	.400	-.554	.000		-.072	.127	.315	.450	.573	210	.180	.500	.488	.233		-.146	-.337	-.696	-.797	-.865
209	.180	.500	-.931	-.559		-.124	.071	.245	.376	.498	212	.180	.600	.405	.147		-.209	-.379	-.660	-.628	-.833
211	.180	.600	-.832	-.563		-.161	-.006	.177	.307	.428	214	.180	.700	.365	.139		-.246	-.400	-.682	-.491	-.772
213	.180	.700	-.605	-.576		-.190	-.034	.174	.286	.394	216	.180	.800	.315	.121		-.245	-.390	-.695	-.280	-.521
215	.180	.800	-.626	-.538		-.194	-.013	.169	.258	.351	218	.180	.900	.222	.071		-.191	-.226	-.206	-.447	-.592
217	.180	.900	-.685	-.249		-.137	.028	.129	.194	.270											
$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$											$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$										
200	.180	.000			.881	.901	.834	.458	-.033	-.614	202	.180	.100			.202	.001	-.738	-.893	-1.110	-1.174
201	.180	.100			-.130	.073	.392	.632	.776	.884	204	.180	.200			.135	-.042	-.335	-.840	-1.077	-1.213
203	.180	.200			-.155	.019	.286	.491	.640	.755	206	.180	.300			.089	-.077	-.306	-.827	-1.024	-1.131
205	.180	.300			-.164	-.018	.200	.391	.536	.657	208	.180	.400			.019	-.111	-.299	-.707	-.977	-.945
207	.180	.400			-.214	-.081	.118	.303	.443	.569	210	.180	.500			-.019	-.135	-.321	-.469	-.745	-.871
209	.180	.500			-.247	-.133	.067	.236	.373	.496	212	.180	.600			-.118	-.203	-.362	-.434	-.628	-.705
211	.180	.600			-.285	-.169	-.009	.169	.305	.426	214	.180	.700			-.166	-.239	-.386	-.459	-.643	-.745
213	.180	.700			-.291	-.192	-.037	.168	.286	.395	216	.180	.800			-.146	-.156	-.332	-.419	-.523	-.662
215	.180	.800			-.002	-.076	.065	.168	.260	.352	218	.180	.900			.157	.172	.070	-.149	-.514	-.679
217	.180	.900			.177	.198	.143	.132	.197	.271											

TABLE III.- PRESSURE COEFFICIENTS MEASURED ON RIGHT WING - Continued

(e) $M = 1.18$ Pressure coefficients C_p on -

Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Lower surface								Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Upper surface							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$											$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
200	.180	.000				.775	.773	.461	-.089	-.540	202	.180	.100				.050	-.187	-.616	-.779	-.863
201	.180	.100				.066	.341	.585	.757	.869	204	.180	.200				.011	-.159	-.601	-.680	-.744
203	.180	.200				.025	.275	.496	.651	.774	206	.180	.300				.015	-.162	-.282	-.449	-.565
205	.180	.300				.016	.214	.415	.568	.693	208	.180	.400				-.008	-.161	-.255	-.410	-.534
207	.180	.400				-.034	.150	.342	.493	.618	210	.180	.500				-.023	-.166	-.285	-.440	-.554
209	.180	.500				-.067	.104	.286	.431	.556	212	.180	.600				-.083	-.210	-.324	-.472	-.565
211	.180	.600				-.096	.057	.218	.372	.498	214	.180	.700				-.116	-.226	-.337	-.505	-.584
213	.180	.700				-.122	.039	.210	.363	.478	216	.180	.800				-.103	-.221	-.340	-.511	-.594
215	.180	.800				-.109	.053	.217	.351	.451	218	.180	.900				-.105	-.224	-.338	-.504	-.596
217	.180	.900				-.092	.067	.204	.314	.393											
			$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$											$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
200	.180	.000	-.629	.380	.762	.778	.761	.441	-.122	-.556	202	.180	.100	.873	.563	.205	.044	-.252	-.656	-.838	-.913
201	.180	.100	-.909	-.552	-.082	.075	.368	.596	.763	.873	204	.180	.200	.781	.480	.148	.006	-.178	-.656	-.688	-.749
203	.180	.200	-.894	.008	-.093	.035	.298	.503	.657	.778	206	.180	.300	.705	.413	.123	.010	-.183	-.307	-.471	-.598
205	.180	.300	-.620	-.351	-.100	.020	.234	.423	.573	.698	208	.180	.400	.628	.353	.100	-.013	-.174	-.273	-.440	-.571
207	.180	.400	-.576	-.330	-.124	-.029	.169	.350	.498	.623	210	.180	.500	.572	.302	.084	-.028	-.178	-.300	-.470	-.577
209	.180	.500	-.567	-.338	-.157	-.064	.122	.293	.437	.561	212	.180	.600	.501	.215	.016	-.089	-.221	-.338	-.506	-.583
211	.180	.600	-.575	-.350	-.184	-.090	.075	.225	.378	.504	214	.180	.700	.471	.204	-.025	-.119	-.233	-.347	-.533	-.595
213	.180	.700	-.597	-.358	-.197	-.112	.059	.216	.368	.483	216	.180	.800	.435	.193	-.019	-.106	-.230	-.349	-.545	-.606
215	.180	.800	-.610	-.372	-.192	-.112	.070	.224	.355	.455	218	.180	.900	.366	.150	-.022	-.108	-.238	-.355	-.550	-.614
217	.180	.900	-.622	-.377	-.177	-.089	.083	.210	.318	.396											
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
200	.180	.000	-.522	.455		.885	.907	.628	.175	-.401	202	.180	.100	.916	.596		.062	-.434	-.543	-.795	-.878
201	.180	.100	-.862	-.548		.072	.338	.591	.776	.913	204	.180	.200	.804	.491		-.004	-.198	-.542	-.815	-.892
203	.180	.200	-.861	-.465		.017	.257	.488	.660	.797	206	.180	.300	.719	.425		-.006	-.185	-.541	-.891	-.973
205	.180	.300	-.686	-.398		.012	.209	.407	.572	.708	208	.180	.400	.633	.356		-.036	-.183	-.372	-.578	-.640
207	.180	.400	-.543	-.434		-.040	.159	.340	.492	.628	210	.180	.500	.570	.303		-.057	-.195	-.268	-.400	-.529
209	.180	.500	-.515	-.420		-.077	.110	.289	.430	.563	212	.180	.600	.496	.211		-.087	-.222	-.303	-.422	-.543
211	.180	.600	-.530	-.402		-.107	.068	.220	.366	.500	214	.180	.700	.466	.203		-.119	-.235	-.326	-.435	-.548
213	.180	.700	-.543	-.383		-.122	.037	.205	.358	.479	216	.180	.800	.430	.193		-.112	-.224	-.327	-.447	-.567
215	.180	.800	-.569	-.368		-.110	.059	.210	.349	.453	218	.180	.900	.355	.146		-.111	-.222	-.332	-.464	-.588
217	.180	.900	-.569	-.354		-.092	.055	.203	.314	.396											
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
200	.180	.000				.871	.855	.867	.615	.183	202	.180	.100			.196	.062	-.403	-.491	-.758	-.869
201	.180	.100				-.082	.049	.322	.577	.761	204	.180	.200			.130	-.004	-.190	-.428	-.757	-.887
203	.180	.200				-.102	.012	.246	.477	.646	206	.180	.300			.097	-.004	-.181	-.410	-.515	-.704
205	.180	.300				-.103	.015	.205	.401	.557	208	.180	.400			.054	-.037	-.175	-.396	-.586	-.642
207	.180	.400				-.137	-.040	.155	.328	.478	210	.180	.500			.054	-.060	-.191	-.360	-.401	-.532
209	.180	.500				-.160	-.078	.104	.277	.416	212	.180	.600			.000	-.088	-.216	-.360	-.424	-.546
211	.180	.600				-.192	-.107	.064	.208	.353	214	.180	.700			-.035	-.117	-.233	-.361	-.438	-.550
213	.180	.700				-.202	-.123	.033	.194	.345	216	.180	.800			-.020	-.111	-.222	-.350	-.449	-.569
215	.180	.800				-.195	-.109	.053	.198	.336	218	.180	.900			-.044	-.109	-.212	-.341	-.456	-.586
217	.180	.900				-.168	-.091	.050	.191	.302											

TABLE III.- PRESSURE COEFFICIENTS MEASURED ON RIGHT WING - Concluded

(e) $M = 1.18$ - ConcludedPressure coefficients C_p on -

Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Lower surface								Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Upper surface							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = -7.5^\circ; \text{brakes closed}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = -7.5^\circ; \text{brakes closed}$							
200	.180	.000			.856	.852	.862	.601	.163	-.405	202	.180	.100			.217	.070	-.434	-.558	-.782	-.856
201	.180	.100			-.083	.048	.329	.581	.760	.896	204	.180	.200			.142	.006	-.200	-.555	-.794	-.867
203	.180	.200			-.099	.021	.254	.481	.645	.782	206	.180	.300			.102	-.002	-.186	-.533	-.487	-.673
205	.180	.300			-.106	.012	.211	.403	.557	.693	208	.180	.400			.065	-.034	-.186	-.356	-.383	-.521
207	.180	.400			-.141	-.045	.155	.329	.478	.614	210	.180	.500			.067	-.048	-.197	-.270	-.401	-.515
209	.180	.500			-.164	-.081	.105	.278	.416	.549	212	.180	.600			.011	-.082	-.226	-.315	-.425	-.530
211	.180	.600			-.196	-.111	.063	.208	.352	.488	214	.180	.700			-.026	-.115	-.240	-.334	-.439	-.545
213	.180	.700			-.204	-.126	.031	.195	.345	.467	216	.180	.800			-.014	-.108	-.230	-.338	-.452	-.564
215	.180	.800			-.200	-.114	.054	.200	.336	.440	218	.180	.900			-.038	-.107	-.224	-.338	-.466	-.576
217	.180	.900			-.175	-.095	.049	.192	.303	.384											
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 7.5^\circ; \text{brakes closed}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 7.5^\circ; \text{brakes closed}$							
200	.180	.000			.856	.855	.865	.607	.166	-.398	202	.180	.100			.222	.074	-.439	-.521	-.756	-.854
201	.180	.100			-.086	.046	.335	.587	.769	.901	204	.180	.200			.155	.017	-.194	-.482	-.758	-.866
203	.180	.200			-.099	.029	.265	.487	.654	.787	206	.180	.300			.114	.006	-.185	-.516	-.518	-.683
205	.180	.300			-.105	.007	.219	.407	.564	.696	208	.180	.400			.079	-.024	-.183	-.391	-.383	-.530
207	.180	.400			-.142	-.046	.158	.333	.485	.616	210	.180	.500			.080	-.040	-.197	-.303	-.393	-.521
209	.180	.500			-.168	-.079	.109	.281	.421	.550	212	.180	.600			.020	-.077	-.222	-.317	-.418	-.535
211	.180	.600			-.195	-.105	.070	.212	.359	.488	214	.180	.700			-.018	-.107	-.234	-.330	-.435	-.548
213	.180	.700			-.200	-.119	.043	.197	.350	.468	216	.180	.800			-.010	-.103	-.223	-.332	-.444	-.569
215	.180	.800			-.194	-.112	.055	.203	.340	.441	218	.180	.900			-.028	-.098	-.221	-.328	-.451	-.579
217	.180	.900			-.171	-.090	.058	.196	.304	.384											
			$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$											$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
200	.180	.000	-.423	.584		.951	.957	.667	.196	-.347	202	.180	.100	.947	.606		.042	-.445	-.594	-.752	-.857
201	.180	.100	.007	.006		.067	.357	.617	.812	.949	204	.180	.200	.818	.492		-.024	-.224	-.469	-.708	-.859
203	.180	.200	.007	-.522		.015	.265	.495	.683	.825	206	.180	.300	.724	.411		-.027	-.220	-.437	-.692	-.827
205	.180	.300	.007	-.406		-.001	.204	.410	.586	.729	208	.180	.400	.634	.343		-.059	-.210	-.420	-.656	-.703
207	.180	.400	.007	-.377		-.047	.133	.334	.500	.645	210	.180	.500	.570	.288		-.094	-.228	-.434	-.577	-.630
209	.180	.500	-.667	-.382		-.087	.086	.282	.434	.577	212	.180	.600	.494	.191		-.135	-.255	-.436	-.490	-.594
211	.180	.600	-.594	-.387		-.116	.068	.213	.371	.512	214	.180	.700	.466	.170		-.154	-.268	-.426	-.466	-.588
213	.180	.700	-.579	-.383		-.120	.043	.200	.363	.491	216	.180	.800	.429	.173		-.130	-.243	-.407	-.457	-.585
215	.180	.800	-.584	-.379		-.095	.061	.214	.356	.462	218	.180	.900	.356	.140		-.130	-.236	-.384	-.456	-.581
217	.180	.900	-.563	-.365		-.084	.057	.214	.317	.400											
			$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$											$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
200	.180	.000			.943	.945	.954	.674	.212	-.323	202	.180	.100	.210	.051		-.448	-.609	-.771	-.853	
201	.180	.100			-.229	.055	.347	.612	.807	.940	204	.180	.200	.142	-.011		-.246	-.466	-.729	-.848	
203	.180	.200			-.132	.008	.258	.490	.678	.816	206	.180	.300	.100	-.020		-.215	-.420	-.707	-.821	
205	.180	.300			-.124	-.006	.198	.406	.581	.720	208	.180	.400	.050	-.051		-.208	-.387	-.677	-.709	
207	.180	.400			-.158	-.055	.127	.330	.496	.636	210	.180	.500	.025	-.084		-.227	-.407	-.603	-.636	
209	.180	.500			-.170	-.091	.081	.280	.432	.570	212	.180	.600	-.019	-.128		-.254	-.427	-.515	-.592	
211	.180	.600			-.204	-.120	.061	.211	.367	.506	214	.180	.700	-.055	-.146		-.268	-.428	-.480	-.583	
213	.180	.700			-.198	-.119	.039	.195	.360	.485	216	.180	.800	-.040	-.126		-.243	-.416	-.471	-.576	
215	.180	.800			-.186	-.106	.054	.209	.354	.457	218	.180	.900	-.058	-.126		-.233	-.392	-.474	-.567	
217	.180	.900			-.166	-.091	.051	.210	.316	.397											

TABLE IV.- PRESSURE COEFFICIENTS MEASURED ON RIGHT HORIZONTAL TAIL

(a) $M = 0.60$ Pressure coefficients C_p on -

Model orifice number	y b/2	x c	Lower surface								Model orifice number	y b/2	x c	Upper surface							
			α = -20°	α = -10°	α = -3°	α = 0°	α = 5°	α = 10°	α = 15°	α = 20°				α = -20°	α = -10°	α = -3°	α = 0°	α = 5°	α = 10°	α = 15°	α = 20°
			β ≈ -5°; δ _e = 0°; δ _v = 0°; brakes closed											β ≈ -5°; δ _e = 0°; δ _v = 0°; brakes closed							
304	.260	.200				-.146	-.212	-.258	-.255	-.421	300	.260	.000				.258	-.115	-.492	-.943	-1.366
306	.260	.300				-.151	-.206	-.259	-.299	-.375	301	.260	.100				-.042	.043	.121	.201	.301
308	.260	.400				-.163	-.211	-.262	-.311	-.388	303	.260	.200				-.064	.012	.085	.165	.274
310	.260	.500				-.175	-.216	-.260	-.315	-.393	307	.260	.400				-.117	-.056	.014	.099	.200
314	.260	.700				-.149	-.177	-.217	-.261	-.333	309	.260	.500				-.138	-.085	-.020	.064	.157
316	.260	.800				-.135	-.159	-.191	-.231	-.290	313	.260	.700				-.150	-.109	-.059	.011	.085
											315	.260	.800				-.143	-.112	-.074	-.017	.039
322	.620	.200				-.181	-.301	-.748	-1.373	-1.400	317	.260	.900				-.131	-.108	-.082	-.042	-.009
324	.620	.400				-.184	-.246	-.320	-.399	-1.094											
326	.620	.600				-.151	-.190	-.217	-.218	-.426	323	.620	.400				-.124	-.048	.033	.134	.224
328	.620	.800				-.095	-.117	-.139	-.158	-.292	327	.620	.800				-.105	-.081	-.043	.016	.045
			β ≈ -5°; δ _e = 0°; δ _v = 0°; brakes open											β ≈ -5°; δ _e = 0°; δ _v = 0°; brakes open							
304	.260	.200	-.117	-.126	-.054	-.085	-.165	-.178	-.177	-.401	300	.260	.000	-.301	.031	.356	.221	-.182	-.507	-.962	-1.335
306	.260	.300	-.063	-.118	-.058	-.089	-.161	-.176	-.218	-.324	301	.260	.100	-.077	-.020	.010	.057	.130	.196	.266	.350
308	.260	.400	-.046	-.121	-.072	-.107	-.158	-.182	-.230	-.335	303	.260	.200	-.093	-.067	-.002	.041	.098	.164	.238	.331
310	.260	.500	-.050	-.130	-.094	-.118	-.175	-.189	-.240	-.343	307	.260	.400	-.150	-.147	-.027	.004	.053	.112	.182	.265
314	.260	.700	-.041	-.115	-.102	-.119	-.169	-.175	-.221	-.302	309	.260	.500	-.174	-.170	-.044	-.018	.026	.081	.148	.223
316	.260	.800	-.039	-.108	-.112	-.124	-.169	-.171	-.211	-.278	313	.260	.700	-.187	-.173	-.075	-.057	.027	.019	.074	.132
											315	.260	.800	-.178	-.160	-.102	-.089	-.066	-.029	.017	.060
322	.620	.200	-.068	-.067	-.123	-.173	-.351	-.721	-1.444	-1.247	317	.260	.900	-.153	-.137	-.129	-.122	-.110	-.079	-.044	-.017
324	.620	.400	-.030	-.104	-.145	-.174	-.247	-.302	-.401	-1.048											
326	.620	.600	-.018	-.110	-.145	-.161	-.207	-.244	-.228	-.461	323	.620	.400	-.281	-.164	-.108	-.074	-.012	.061	.145	.222
328	.620	.800	-1.881	-1.880	-.113	-.124	-.151	-.183	-.187	-.333	327	.620	.800	-.208	-.122	-.133	-.119	-.098	-.060	-.006	.020
			β ≈ 0°; δ _e = 0°; δ _v = 0°; brakes closed											β ≈ 0°; δ _e = 0°; δ _v = 0°; brakes closed							
304	.260	.200	.079	-.066		-.082	-.157	-.226	-.254	-.693	300	.260	.000	-.319	.204	.365	.153	-.365	-1.121	-1.588	
306	.260	.300	.074	-.083		-.095	-.157	-.226	-.281	-.388	301	.260	.100	-.248	-.072	-.048	.044	.139	.269	.358	
308	.260	.400	.060	-.104		-.112	-.169	-.231	-.300	-.404	303	.260	.200	-.242	-.093	-.073	.008	.100	.228	.323	
310	.260	.500	.033	-.125		-.135	-.181	-.240	-.308	-.406	307	.260	.400	-.267	-.131	-.111	-.055	.026	.140	.230	
314	.260	.700	.012	-.116		-.114	-.150	-.197	-.262	-.354	309	.260	.500	-.275	-.149	-.131	-.081	-.008	.096	.179	
316	.260	.800	-.004	-.109		-.103	-.138	-.179	-.235	-.311	313	.260	.700	-.265	-.157	-.138	-.104	-.051	.030	.092	
											315	.260	.800	-.245	-.148	-.132	-.106	-.067	-.007	.038	
322	.620	.200	.136	-.159		-.114	-.248	-.639	-1.358	-1.306	317	.260	.900	-.211	-.130	-.114	-.099	-.076	-.040	-.017	
324	.620	.400	.107	-.147		-.140	-.225	-.315	-.968	-1.645											
326	.620	.600	.053	-.123		-.126	-.174	-.215	-.225	-.620	323	.620	.400	-.465	-.151	-.140	-.056	.048	.172	.250	
328	.620	.800	-.007	-.091		-.076	-.108	-.133	-.132	-.349	327	.620	.800	-.337	-.140	-.107	-.079	-.039	.020	.037	
			β ≈ 0°; δ _e = 0°; δ _v = 0°; brakes open											β ≈ 0°; δ _e = 0°; δ _v = 0°; brakes open							
304	.260	.200			-.034	-.056	-.120	-.197	-.238	-.690	300	.260	.000			.430	.384	.182	-.306	-1.083	-1.559
306	.260	.300			-.052	-.072	-.123	-.192	-.247	-.368	301	.260	.100			-.069	-.031	.057	.152	.281	.365
308	.260	.400			-.074	-.091	-.135	-.199	-.268	-.381	303	.260	.200			-.090	-.052	.024	.109	.238	.329
310	.260	.500			-.100	-.115	-.153	-.213	-.282	-.385	307	.260	.400			-.118	-.095	-.039	.036	.150	.234
314	.260	.700			-.088	-.097	-.129	-.178	-.240	-.336	309	.260	.500			-.134	-.114	-.068	.002	.106	.183
316	.260	.800			-.081	-.089	-.116	-.159	-.216	-.296	313	.260	.700			-.138	-.126	-.093	-.043	.038	.095
											315	.260	.800			-.129	-.122	-.097	-.061	.001	.040
322	.620	.200			-.062	-.095	-.219	-.619	-1.279	-1.245	317	.260	.900			-.111	-.105	-.092	-.071	-.033	-.015
324	.620	.400			-.102	-.122	-.198	-.280	-.941	-1.574											
326	.620	.600			-.102	-.113	-.155	-.197	-.215	-.614	323	.620	.400			-.168	-.131	-.052	.051	.175	.250
328	.620	.800			-.063	-.067	-.095	-.123	-.119	-.342	327	.620	.800			-.112	-.098	-.074	-.034	.025	.042

TABLE IV.- PRESSURE COEFFICIENTS MEASURED ON RIGHT HORIZONTAL TAIL - Continued

(a) $M = 0.60$ - ConcludedPressure coefficients C_p on -

Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Lower surface								Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Upper surface							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = -7.5^\circ; \text{brakes closed}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = -7.5^\circ; \text{brakes closed}$							
304	.260	.200			-.072	-.103	-.178	-.254	-.260	-.647	300	.260	.000			.408	.360	.152	-.351	-1.113	-1.567
306	.260	.300			-.090	-.116	-.179	-.249	-.302	-.403	301	.260	.100			-.122	-.079	.016	.121	.247	.342
308	.260	.400			-.108	-.133	-.188	-.252	-.319	-.418	303	.260	.200			-.144	-.103	-.016	.084	.207	.308
310	.260	.500			-.130	-.154	-.199	-.258	-.326	-.423	307	.260	.400			-.164	-.137	-.073	.011	.124	.218
314	.260	.700			-.111	-.129	-.164	-.213	-.273	-.368	309	.260	.500			-.179	-.156	-.098	-.021	.082	.170
316	.260	.800			-.103	-.116	-.145	-.187	-.242	-.322	313	.260	.700			-.173	-.158	-.119	-.059	.020	.086
322	.620	.200			-.081	-.126	-.258	-.626	-1.429	-1.366	315	.260	.800			-.157	-.146	-.118	-.074	-.012	.035
324	.620	.400			-.125	-.152	-.236	-.332	-.925	-1.656	317	.260	.900			-.132	-.125	-.108	-.080	-.043	-.020
326	.620	.600			-.118	-.136	-.183	-.223	-.212	-.598	323	.620	.400			-.199	-.158	-.070	.039	.163	.246
328	.620	.800			-.073	-.084	-.112	-.140	-.132	-.335	327	.620	.800			-.129	-.115	-.087	-.042	.016	.040
$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 7.5^\circ; \text{brakes closed}$																					
304	.260	.200			.007	-.017	-.087	-.143	-.213	-.675	300	.260	.000			.491	.445	.141	-.342	-1.072	-1.538
306	.260	.300			-.006	-.027	-.084	-.142	-.211	-.340	301	.260	.100			.014	.048	.129	.213	.324	.403
308	.260	.400			-.025	-.043	-.091	-.149	-.228	-.348	303	.260	.200			.008	.038	.104	.176	.288	.375
310	.260	.500			-.053	-.071	-.114	-.162	-.248	-.356	307	.260	.400			-.003	.016	.062	.124	.219	.296
314	.260	.700			-.065	-.078	-.109	-.152	-.235	-.326	309	.260	.500			-.017	-.001	.035	.091	.178	.249
316	.260	.800			-.077	-.088	-.116	-.151	-.226	-.300	313	.260	.700			-.045	-.037	-.012	.029	.092	.145
322	.620	.200			-.057	-.093	-.220	-.647	-1.246	-1.186	315	.260	.800			-.074	-.070	-.053	-.021	.029	.069
324	.620	.400			-.101	-.127	-.201	-.284	-.885	-1.478	317	.260	.900			-.106	-.103	-.095	-.072	-.041	-.017
326	.620	.600			-.117	-.131	-.175	-.212	-.239	-.625	323	.620	.400			-.111	-.084	-.017	.071	.180	.254
328	.620	.800			-.089	-.097	-.128	-.159	-.165	-.387	327	.620	.800			-.120	-.112	-.093	-.059	.000	.018
$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$																					
304	.260	.200	.223	-.033		-.028	-.113	-.202	-.647	-1.272	300	.260	.000	-.839	.041		.522	.327	-.360	-1.341	-1.702
306	.260	.300	.205	-.028		-.055	-.121	-.206	-.294	-.545	301	.260	.100	-.608	-.021		.051	.054	.195	.362	.435
308	.260	.400	.175	-.036		-.080	-.135	-.217	-.283	-.477	303	.260	.200	-.448	-.061		-.081	.026	.154	.306	.394
310	.260	.500	.130	-.050		-.109	-.153	-.226	-.295	-.447	307	.260	.400	-.459	-.134		-.102	-.035	.073	.196	.278
314	.260	.700	.069	-.046		-.100	-.128	-.191	-.259	-.383	309	.260	.500	-.459	-.158		-.119	-.059	.032	.139	.219
316	.260	.800	.032	-.051		-.090	-.115	-.172	-.232	-.342	313	.260	.700	-.430	-.168		-.127	-.087	-.018	.055	.113
322	.620	.200	.283	.061		-.090	-.224	-.982	-1.153	-1.260	315	.260	.800	-.398	-.158		-.118	-.091	-.042	.009	.048
324	.620	.400	.211	.019		-.124	-.210	-.336	-1.434	-1.329	317	.260	.900	-.347	-.138		-.102	-.088	-.061	-.037	-.025
326	.620	.600	.103	-.016		-.115	-.167	-.184	-.662	-1.159	323	.620	.400	-.774	-.288		-.154	-.049	.090	.206	.286
328	.620	.800	-.010	-.032		-.069	-.101	-.106	-.219	-.612	327	.620	.800	-.542	-.198		-.107	-.073	-.023	.010	.025
$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$																					
304	.260	.200			.063	.041	-.042	-.130	-.626	-1.256	300	.260	.000			.565	.538	.312	-.314	-1.284	-1.612
306	.260	.300			.043	.026	-.046	-.132	-.221	-.514	301	.260	.100			.013	.052	.140	.255	.417	.476
308	.260	.400			.020	.003	-.061	-.141	-.209	-.431	303	.260	.200			.011	.036	.123	.215	.367	.441
310	.260	.500			-.012	-.027	-.083	-.156	-.231	-.399	307	.260	.400			.017	.034	.085	.164	.277	.344
314	.260	.700			-.030	-.039	-.082	-.149	-.225	-.358	309	.260	.500			.006	.020	.060	.129	.227	.286
316	.260	.800			-.045	-.053	-.090	-.146	-.216	-.329	313	.260	.700			-.017	-.013	.011	.058	.127	.171
322	.620	.200			-.011	-.051	-.187	-.887	-1.074	-1.142	315	.260	.800			-.045	-.043	-.028	.005	.056	.084
324	.620	.400			-.060	-.086	-.187	-.289	-1.378	-1.271	317	.260	.900			-.077	-.078	-.073	-.055	-.023	-.014
326	.620	.600			-.087	-.100	-.159	-.192	-.655	-1.104	323	.620	.400			-.118	-.085	.001	.112	.226	.294
328	.620	.800			-.068	-.076	-.114	-.136	-.258	-.654	327	.620	.800			-.107	-.100	-.077	-.035	.011	.014

TABLE IV.- PRESSURE COEFFICIENTS MEASURED ON RIGHT HORIZONTAL TAIL - Continued

(b) $M = 0.90$ Pressure coefficients C_p on -

Model orifice number	y b/2	x c	Lower surface								Model orifice number	y b/2	x c	Upper surface							
			α = -20°	α = -10°	α = -3°	α = 0°	α = 5°	α = 10°	α = 15°	α = 20°				α = -20°	α = -10°	α = -3°	α = 0°	α = 5°	α = 10°	α = 15°	α = 20°
			β ≈ -5°; δ _e = 0°; δ _v = 0°; brakes closed											β ≈ -5°; δ _e = 0°; δ _v = 0°; brakes closed							
304	.260	.200				-.156	-.217	-.242	-.311	-.774	300	.260	.000				.295	.022	-.329	-.934	-1.182
306	.260	.300				-.172	-.229	-.252	-.323	-.452	301	.260	.100				-.032	.043	.131	.211	.282
308	.260	.400				-.197	-.249	-.276	-.344	-.462	303	.260	.200				-.062	.011	.108	.195	.265
310	.260	.500				-.246	-.302	-.325	-.379	-.482	307	.260	.400				-.136	-.069	.031	.129	.205
314	.260	.700				-.270	-.328	-.348	-.369	-.441	309	.260	.500				-.173	-.107	-.007	.092	.165
316	.260	.800				-.270	-.322	-.346	-.344	-.405	313	.260	.700				-.248	-.182	-.071	.024	.090
											315	.260	.800				-.275	-.204	-.105	-.020	.035
322	.620	.200				-.263	-.428	-.836	-1.001	-1.059	317	.260	.900				-.269	-.224	-.146	-.077	-.037
324	.620	.400				-.307	-.383	-.511	-1.080	-1.209											
326	.620	.600				-.327	-.415	-.422	-.520	-.785	323	.620	.400				-.206	-.112	.027	.145	.212
328	.620	.800				-.199	-.224	-.228	-.370	-.547	327	.620	.800				-.278	-.219	-.115	-.043	-.005
			β ≈ -5°; δ _e = 0°; δ _v = 0°; brakes open											β ≈ -5°; δ _e = 0°; δ _v = 0°; brakes open							
304	.260	.200	-.072	-.134	.005	-.026	-.102	-.120	-.248	-1.000	300	.260	.000	-.342	-.460	.395	.282	-.013	-.402	-.909	-1.177
306	.260	.300	-.066	-.147	.003	-.026	-.091	-.113	-.221	-.498	301	.260	.100	-.301	.057	.079	.124	.190	.262	.327	.387
308	.260	.400	-.074	-.171	-.007	-.035	-.095	-.113	-.222	-.438	303	.260	.200	-.275	-.030	.070	.113	.179	.254	.318	.378
310	.260	.500	-.093	-.204	-.030	-.056	-.110	-.125	-.234	-.399	307	.260	.400	-.248	-.127	.058	.095	.149	.217	.278	.335
314	.260	.700	-.082	-.218	-.050	-.069	-.120	-.138	-.245	-.381	309	.260	.500	-.238	-.159	.046	.080	.131	.192	.249	.300
316	.260	.800	-.087	-.217	-.075	-.090	-.138	-.158	-.254	-.380	313	.260	.700	-.216	-.220	.018	.043	.080	.130	.174	.214
											315	.260	.800	-.206	-.233	-.021	-.003	.025	.065	.100	.131
322	.620	.200	-.034	-.215	-.096	-.142	-.411	-.817	-.860	-.919	317	.260	.900	-.190	-.236	-.092	-.082	-.061	-.029	-.003	.018
324	.620	.400	-.010	-.221	-.136	-.167	-.227	-.395	-.960	-1.108											
326	.620	.600	-.035	-.206	-.186	-.206	-.270	-.285	-.508	-.829	323	.620	.400	-.332	-.143	-.050	-.014	.048	.132	.200	.249
328	.620	.800	-.552	-.551	-.216	-.236	-.288	-.306	-.434	-.647	327	.620	.800	-.276	-.185	-.220	-.203	-.169	-.111	-.061	-.027
			β ≈ 0°; δ _e = 0°; δ _v = 0°; brakes closed											β ≈ 0°; δ _e = 0°; δ _v = 0°; brakes closed							
304	.260	.200	.046	-.112		-.075	-.140	-.188	-.538		300	.260	.000	-.573	-.201		.386	.228	-.180	-1.021	
306	.260	.300	.043	-.113		-.102	-.158	-.206	-.283		301	.260	.100	-.389	.076		-.030	.060	.159	.288	
308	.260	.400	.031	-.125		-.132	-.186	-.232	-.301		303	.260	.200	-.389	.030		-.060	.016	.123	.251	
310	.260	.500	.003	-.151		-.186	-.242	-.291	-.341		307	.260	.400	-.331	-.075		-.133	-.064	.048	.168	
314	.260	.700	-.004	-.150		-.211	-.264	-.306	-.349		309	.260	.500	-.329	-.122		-.171	-.102	.008	.122	
316	.260	.800	-.022	-.152		-.211	-.262	-.293	-.332		313	.260	.700	-.310	-.188		-.246	-.174	-.059	.044	
											315	.260	.800	-.299	-.200		-.269	-.197	-.094	-.004	
322	.620	.200	.113	-.118		-.166	-.318	-.833	-.955		317	.260	.900	-.276	-.198		-.259	-.215	-.134	-.065	
324	.620	.400	.084	-.117		-.232	-.332	-.466	-1.098												
326	.620	.600	.031	-.123		-.263	-.365	-.417	-.784		323	.620	.400	-.512	-.208		-.226	-.116	.049	.178	
328	.620	.800	-.035	-.114		-.173	-.194	-.203	-.439		327	.620	.800	-.453	-.197		-.276	-.226	-.109	-.036	
			β ≈ 0°; δ _e = 0°; δ _v = 0°; brakes open											β ≈ 0°; δ _e = 0°; δ _v = 0°; brakes open							
304	.260	.200			-.021	-.037	-.100	-.154	-.610	-.991	300	.260	.000			.460	.415	.246	-.197	-1.015	-1.163
306	.260	.300			-.049	-.065	-.118	-.170	-.274	-.501	301	.260	.100			-.022	.006	.086	.185	.303	.355
308	.260	.400			-.081	-.096	-.146	-.196	-.278	-.472	303	.260	.200			-.054	-.022	.044	.152	.268	.325
310	.260	.500			-.135	-.150	-.201	-.251	-.315	-.477	307	.260	.400			-.123	-.100	-.041	.067	.179	.244
314	.260	.700			-.158	-.172	-.226	-.265	-.325	-.450	309	.260	.500			-.160	-.138	-.083	.025	.131	.197
316	.260	.800			-.165	-.178	-.228	-.266	-.313	-.424	313	.260	.700			-.233	-.211	-.153	-.047	.048	.109
											315	.260	.800			-.258	-.235	-.180	-.085	-.003	.048
322	.620	.200			-.104	-.137	-.278	-.782	-.1057	-1.057	317	.260	.900			-.257	-.242	-.205	-.130	-.065	-.030
324	.620	.400			-.177	-.200	-.291	-.457	-1.023	-1.090											
326	.620	.600			-.218	-.234	-.319	-.380	-.794	-.983	323	.620	.400			-.244	-.204	-.102	.058	.182	.249
328	.620	.800			-.157	-.161	-.189	-.212	-.438	-.658	327	.620	.800			-.283	-.268	-.217	-.105	-.035	.000

Pressure coefficients C_p on -

496-T

TABLE IV.- PRESSURE COEFFICIENTS MEASURED ON RIGHT HORIZONTAL TAIL - Continued

(c) $M = 0.95$ Pressure coefficients C_p on -

Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Lower surface								Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Upper surface							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$											$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
304	.260	.200				-.123	-.156	-.207	-.380	-.809	300	.260	.000				.324	.160	-.177	-.883	-.986
306	.260	.300				-.143	-.166	-.198	-.333	-.536	301	.260	.100				-.006	.046	.112	.163	.256
308	.260	.400				-.163	-.194	-.228	-.329	-.501	303	.260	.200				-.035	.017	.092	.160	.238
310	.260	.500				-.215	-.242	-.275	-.393	-.519	307	.260	.400				-.110	-.064	.023	.116	.191
314	.260	.700				-.261	-.292	-.316	-.433	-.536	309	.260	.500				-.153	-.100	-.011	.085	.163
316	.260	.800				-.268	-.299	-.332	-.440	-.537	313	.260	.700				-.222	-.170	-.071	.030	.104
											315	.260	.800				-.263	-.217	-.097	-.008	.058
322	.620	.200				-.243	-.318	-.658	-.931	-.979	317	.260	.900				-.283	-.232	-.130	-.058	-.007
324	.620	.400				-.282	-.332	-.359	-1.074	-1.129											
326	.620	.600				-.347	-.393	-.422	-.584	-.897	323	.620	.400				-.185	-.115	.020	.144	.224
328	.620	.800				-.310	-.369	-.413	-.511	-.761	327	.620	.800				-.304	-.230	-.125	-.043	.012
			$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$											$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
304	.260	.200	-.215	-.217	.043	.013	-.037	-.074	-.290	-.651	300	.260	.000	-.360	-.401	.418	.328	.108	-.330	-.899	-1.008
306	.260	.300	-.203	-.157	.046	.020	-.029	-.063	-.220	-.420	301	.260	.100	-.442	.029	.120	.153	.210	.283	.338	.375
308	.260	.400	-.195	-.152	.037	.013	-.034	-.060	-.212	-.375	303	.260	.200	-.419	-.031	.111	.143	.203	.278	.333	.374
310	.260	.500	-.220	-.175	.018	-.007	-.050	-.074	-.214	-.368	307	.260	.400	-.391	-.115	.099	.131	.183	.250	.306	.350
314	.260	.700	-.164	-.211	.002	-.014	-.056	-.082	-.214	-.365	309	.260	.500	-.380	-.147	.090	.119	.168	.230	.281	.324
316	.260	.800	-.162	-.227	-.019	-.033	-.072	-.101	-.221	-.357	313	.260	.700	-.332	-.205	.067	.090	.128	.177	.215	.249
											315	.260	.800	-.312	-.235	.034	.049	.079	.120	.148	.176
322	.620	.200	-.123	-.211	-.047	-.086	-.258	-.665	-.747	-.837	317	.260	.900	-.292	-.256	-.031	-.022	.001	.034	.051	.070
324	.620	.400	-.122	-.245	-.081	-.108	-.165	-.340	-.860	-1.002											
326	.620	.600	-.128	-.279	-.126	-.147	-.202	-.219	-.514	-.777	323	.620	.400	-.480	-.146	.003	.034	.092	.175	.237	.277
328	.620	.800	-.437	-.442	-.155	-.171	-.219	-.244	-.464	-.697	327	.620	.800	-.385	-.279	-.148	-.135	-.103	-.053	-.016	.013
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
304	.260	.200	-.117	-.097		-.043	-.077	-.155	-.537	-1.184	300	.260	.000	-.409	-.076		.471	.339	-.074	-.859	-1.112
306	.260	.300	-.106	-.107		-.070	-.099	-.168	-.296	-.795	301	.260	.100	-.504	.094		.007	.072	.147	.241	.323
308	.260	.400	-.073	-.129		-.101	-.129	-.197	-.312	-.637	303	.260	.200	-.592	.035		-.024	.031	.116	.213	.298
310	.260	.500	-.083	-.169		-.159	-.188	-.251	-.368	-.574	307	.260	.400	-.534	-.068		-.103	-.053	.042	.151	.238
314	.260	.700	-.039	-.197		-.196	-.236	-.292	-.409	-.595	309	.260	.500	-.531	-.112		-.143	-.091	.006	.114	.198
316	.260	.800	-.044	-.212		-.211	-.250	-.320	-.420	-.608	313	.260	.700	-.476	-.188		-.217	-.158	-.055	.050	.125
											315	.260	.800	-.433	-.238		-.252	-.204	-.081	.011	.074
322	.620	.200	.045	-.188		-.141	-.239	-.583	-.911	-1.019	317	.260	.900	-.368	-.258		-.271	-.218	-.119	-.042	.007
324	.620	.400	.033	-.205		-.206	-.275	-.397	-1.098	-1.105											
326	.620	.600	-.003	-.224		-.291	-.348	-.433	-.822	-1.207	323	.620	.400	-.649	-.175		-.196	-.110	.048	.181	.260
328	.620	.800	-.049	-.197		-.256	-.331	-.419	-.536	-1.011	327	.620	.800	-.538	-.294		-.312	-.227	-.117	-.029	.029
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
304	.260	.200			.007	-.003	-.038	-.122	-.485	-1.072	300	.260	.000			.481	.447	.320	-.087	-.893	-1.065
306	.260	.300			-.022	-.032	-.061	-.133	-.278	-.688	301	.260	.100			.025	.040	.100	.175	.262	.332
308	.260	.400			-.057	-.065	-.092	-.160	-.291	-.654	303	.260	.200			-.010	.012	.060	.144	.235	.308
310	.260	.500			-.116	-.126	-.151	-.219	-.343	-.513	307	.260	.400			-.083	-.069	-.025	.064	.163	.243
314	.260	.700			-.147	-.158	-.186	-.258	-.375	-.547	309	.260	.500			-.122	-.109	-.068	.024	.124	.202
316	.260	.800			-.163	-.171	-.210	-.274	-.386	-.551	313	.260	.700			-.192	-.177	-.139	-.041	.056	.125
											315	.260	.800			-.230	-.221	-.178	-.072	.014	.073
322	.620	.200			-.093	-.111	-.207	-.611	-.879	-.969	317	.260	.900			-.253	-.242	-.197	-.112	-.042	.005
324	.620	.400			-.170	-.182	-.239	-.359	-1.015	-1.020											
326	.620	.600			-.236	-.253	-.311	-.396	-.797	-1.096	323	.620	.400			-.200	-.173	-.098	.057	.186	.261
328	.620	.800			-.216	-.220	-.293	-.387	-.514	-.894	327	.620	.800			-.305	-.285	-.212	-.114	-.028	.027

TABLE IV.- PRESSURE COEFFICIENTS MEASURED ON RIGHT HORIZONTAL TAIL - Continued

(c) $M = 0.95$ - ConcludedPressure coefficients C_p on -

Model orifice number	y b/2	x c	Lower surface								Model orifice number	y b/2	x c	Upper surface									
			α = -20°	α = -10°	α = -5°	α = 0°	α = 5°	α = 10°	α = 15°	α = 20°				α = -20°	α = -10°	α = -5°	α = 0°	α = 5°	α = 10°	α = 15°	α = 20°		
			β ≈ 0°; δ _e = 0°; δ _v = -7.5°; brakes closed											β ≈ 0°; δ _e = 0°; δ _v = -7.5°; brakes closed									
304	.260	.200			-.064	-.082	-.120	-.177	-.615	-1.196	300	.260	.000			.433	.396	.282	.003	-.723	-1.107		
306	.260	.300			-.091	-.106	-.143	-.191	-.304	-.796	301	.260	.100			-.049	-.035	.029	.109	.204	.298		
308	.260	.400			-.118	-.132	-.169	-.219	-.331	-.630	303	.260	.200			-.089	-.066	-.006	.080	.182	.276		
310	.260	.500			-.171	-.184	-.225	-.270	-.380	-.567	307	.260	.400			-.146	-.132	-.081	.007	.123	.215		
314	.260	.700			-.216	-.229	-.285	-.329	-.438	-.607	309	.260	.500			-.190	-.176	-.116	-.030	.088	.176		
316	.260	.800			-.234	-.255	-.286	-.352	-.454	-.626	313	.260	.700			-.260	-.244	-.195	-.099	.026	.105		
322	.620	.200			-.144	-.162	-.274	-.510	-.959	-1.027	315	.260	.800			-.311	-.301	-.243	-.124	-.010	.058		
324	.620	.400			-.214	-.243	-.322	-.419	-1.122	-1.154	317	.260	.900			-.326	-.310	-.258	-.145	-.057	-.005		
326	.620	.600			-.310	-.324	-.391	-.467	-.820	-1.229	323	.620	.400			-.247	-.221	-.137	.006	.162	.248		
328	.620	.800			-.263	-.289	-.368	-.446	-.566	-1.037	327	.620	.800			-.374	-.355	-.262	-.138	-.037	.021		
			β ≈ 0°; δ _e = 0°; δ _v = 7.5°; brakes closed											β ≈ 0°; δ _e = 0°; δ _v = 7.5°; brakes closed									
304	.260	.200			.113	.099	.053	-.037	-.258	-1.012	300	.260	.000			.540	.506	.329	-.189	-.866	-1.050		
306	.260	.300			.107	.094	.054	-.027	-.181	-.570	301	.260	.100			.138	.157	.222	.315	.389	.449		
308	.260	.400			.095	.083	.044	-.034	-.171	-.435	303	.260	.200			.135	.155	.207	.302	.377	.437		
310	.260	.500			.067	.057	.017	-.054	-.188	-.401	307	.260	.400			.139	.155	.193	.272	.338	.393		
314	.260	.700			.052	.045	.005	-.067	-.202	-.414	309	.260	.500			.133	.146	.180	.251	.309	.359		
316	.260	.800			.027	.022	-.017	-.093	-.222	-.404	313	.260	.700			.108	.118	.144	.196	.240	.278		
322	.620	.200			.030	.007	-.093	-.587	-.786	-.870	315	.260	.800			.067	.077	.096	.138	.170	.200		
324	.620	.400			-.028	-.042	-.111	-.316	-.851	-.890	317	.260	.900			-.004	.005	.017	.048	.069	.091		
326	.620	.600			-.083	-.095	-.152	-.243	-.549	-.957	323	.620	.400			.013	.038	.096	.203	.276	.323		
328	.620	.800			-.119	-.122	-.178	-.247	-.449	-.809	327	.620	.800			-.136	-.123	-.093	-.033	.010	.043		
			β ≈ 5°; δ _e = 0°; δ _v = 0°; brakes closed											β ≈ 5°; δ _e = 0°; δ _v = 0°; brakes closed									
304	.260	.200	.108	.024		.039	-.010	-.178	-1.124	-1.377	300	.260	.000			-.706	.185		.604	.500	-.045	-.971	-.017
306	.260	.300	.113	-.010		.002	-.041	-.165	-.391	-1.176	301	.260	.100			-.634	.002		.029	.086	.214	.330	.415
308	.260	.400	.107	-.049		-.034	-.078	-.199	-.389	-.811	303	.260	.200			-.496	-.038		-.012	.057	.176	.299	.393
310	.260	.500	.086	-.103		-.100	-.143	-.256	-.433	-.596	307	.260	.400			-.531	-.121		-.077	.026	.088	.211	.312
314	.260	.700	.076	-.146		-.130	-.183	-.301	-.474	-.591	309	.260	.500			-.547	-.154		-.120	-.065	.045	.166	.266
316	.260	.800	.054	-.163		-.147	-.207	-.324	-.490	-.566	313	.260	.700			-.558	-.218		-.189	-.129	-.022	.091	.181
322	.620	.200	.210	-.148		-.079	-.199	-.552	-.980	-1.058	315	.260	.800			-.529	-.253		-.225	-.162	-.054	.043	.123
324	.620	.400	.182	-.186		-.148	-.244	-.568	-1.073	-1.074	317	.260	.900			-.484	-.287		-.239	-.173	-.095	.018	.046
326	.620	.600	.116	-.207		-.230	-.327	-.493	-1.178	-1.068	323	.620	.400			-.871	-.247		-.197	-.083	.080	.235	.330
328	.620	.800	.027	-.192		-.201	-.308	-.456	-.945	-.965	327	.620	.800			-.645	-.383		-.301	-.204	-.107	-.004	.070
			β ≈ 5°; δ _e = 0°; δ _v = 0°; brakes open											β ≈ 5°; δ _e = 0°; δ _v = 0°; brakes open									
304	.260	.200			.183	.175	.120	.010	-.978	-1.407	300	.260	.000			.669	.649	.531	-.313	-1.047	-.015		
306	.260	.300			.169	.162	.112	-.007	-.293	-1.046	301	.260	.100			.148	.172	.233	.393	.461	.522		
308	.260	.400			.148	.146	.099	-.025	-.252	-.653	303	.260	.200			.139	.160	.228	.369	.441	.506		
310	.260	.500			.120	.115	.067	-.053	-.249	-.515	307	.260	.400			.165	.181	.220	.323	.383	.446		
314	.260	.700			.105	.104	.051	-.061	-.255	-.493	309	.260	.500			.163	.176	.209	.296	.351	.408		
316	.260	.800			.082	.081	.030	-.083	-.269	-.487	313	.260	.700			.145	.155	.176	.236	.274	.319		
322	.620	.200			.092	.024	-.046	-.570	-.839	-1.033	315	.260	.800			.109	.116	.132	.174	.203	.238		
324	.620	.400			.031	-.032	-.075	-.723	-.880	-1.039	317	.260	.900			.039	.046	.052	.080	.097	.124		
326	.620	.600			-.034	-.049	-.119	-.361	-.973	-1.134	323	.620	.400			.015	.045	.121	.256	.325	.382		
328	.620	.800			-.054	-.774	-.131	-.265	-.712	-1.025	327	.620	.800			-.090	-.085	-.062	.004	.046	.087		

TABLE IV.- PRESSURE COEFFICIENTS MEASURED ON RIGHT HORIZONTAL TAIL - Continued

(d) $M = 1.03$ Pressure coefficients C_p on -

Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Lower surface								Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Upper surface							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$											$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
304	.260	.200				-.005	-.056	-.199	-.701	-.956	300	.260	.000				.390	.235	-.153	-.605	-.815
306	.260	.300				-.026	-.064	-.207	-.340	-.855	301	.260	.100				.106	.141	.160	.221	.377
308	.260	.400				-.046	-.096	-.225	-.341	-.537	303	.260	.200				.083	.116	.140	.183	.342
310	.260	.500				-.094	-.134	-.257	-.346	-.445	307	.260	.400				.014	.040	.070	.134	.270
314	.260	.700				-.127	-.175	-.273	-.353	-.480	309	.260	.500				-.031	-.004	.032	.113	.236
316	.260	.800				-.126	-.184	-.279	-.322	-.502	313	.260	.700				-.092	-.073	.026	.081	.181
											315	.260	.800				-.133	-.092	.037	.058	.144
322	.620	.200				-.127	-.228	-.601	-.859	-.938	317	.260	.900				-.138	-.112	-.059	.023	.092
324	.620	.400				-.162	-.230	-.460	-1.029	-1.091											
326	.620	.600				-.215	-.284	-.374	-.673	-.851	323	.620	.400				-.057	-.018	.069	.183	.281
328	.620	.800				-.183	-.261	-.365	-.459	-.851	327	.620	.800				-.156	-.110	-.034	.042	.108
			$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$											$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
304	.260	.200	-.022	-.090	.152	.128	.071	-.003	-.435	-.939	300	.260	.000	-.240	-.294	.473	.412	.199	-.211	-.735	-.785
306	.260	.300	-.047	-.075	.152	.132	.079	-.002	-.154	-.631	301	.260	.100	-.380	-.088	.228	.262	.302	.338	.388	.489
308	.260	.400	-.065	-.059	.143	.125	.074	.000	-.118	-.431	303	.260	.200	-.345	-.082	.210	.250	.296	.333	.383	.475
310	.260	.500	-.116	-.104	.126	.104	.057	-.003	-.147	-.323	307	.260	.400	-.356	-.131	.196	.232	.278	.315	.363	.433
314	.260	.700	-.145	-.134	.110	.099	.056	.000	-.120	-.290	309	.260	.500	-.360	-.147	.187	.221	.265	.299	.344	.407
316	.260	.800	-.136	-.146	.088	.080	.040	-.015	-.113	-.288	313	.260	.700	-.372	-.179	.166	.195	.229	.257	.292	.339
											315	.260	.800	-.336	-.225	.136	.158	.185	.208	.234	.272
322	.620	.200	-.133	-.106	.061	.028	-.143	-.524	-.733	-.788	317	.260	.900	-.336	-.241	.078	.097	.114	.131	.149	.178
324	.620	.400	-.143	-.162	.033	.010	-.054	-.324	-.973	-.946											
326	.620	.600	-.113	-.210	-.006	-.026	-.085	-.139	-.379	-.815	323	.620	.400	-.494	-.195	.109	.147	.197	.247	.298	.355
328	.620	.800	-.292	-.290	-.029	-.044	-.094	-.154	-.287	-.608	327	.620	.800	-.485	-.288	-.023	-.004	.021	.050	.080	.119
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
304	.260	.200	.048	-.012		.052	.004	-.138	-.855	-1.097	300	.260	.000	-.256	-.055		.526	.369	-.015	-.536	-.004
306	.260	.300	.013	-.040		.030	-.014	-.128	-.228	-.957	301	.260	.100	-.520	.016		.106	.168	.189	.250	.383
308	.260	.400	-.017	-.075		.003	-.038	-.139	-.177	-.544	303	.260	.200	-.454	-.196		.079	.125	.153	.216	.351
310	.260	.500	-.076	-.128		-.050	-.091	-.177	-.242	-.451	307	.260	.400	-.411	-.072		.008	.042	.077	.163	.288
314	.260	.700	-.103	-.164		-.077	-.137	-.232	-.331	-.501	309	.260	.500	-.437	-.099		-.036	.000	.041	.135	.256
316	.260	.800	-.063	-.175		-.083	-.136	-.250	-.360	-.515	313	.260	.700	-.478	-.156		-.097	-.070	-.002	.092	.202
											315	.260	.800	-.487	-.197		-.133	-.099	-.031	.070	.165
322	.620	.200	.026	-.179		-.035	-.149	-.485	-.790	-.967	317	.260	.900	-.494	-.222		-.132	-.101	-.045	.036	.109
324	.620	.400	.012	-.190		-.106	-.197	-.343	-.941	-1.069											
326	.620	.600	-.014	-.209		-.169	-.251	-.366	-.914	-1.148	323	.620	.400	-.568	-.156		-.073	-.014	.090	.209	.326
328	.620	.800	-.031	-.203		-.123	-.227	-.357	-.539	-1.005	327	.620	.800	-.576	-.298		-.166	-.105	-.026	.054	.128
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
304	.260	.200			.109	.088	.037	-.091	-.796	-.996	300	.260	.000			.556	.506	.321	-.017	-.546	-.896
306	.260	.300			.081	.063	.025	-.082	-.076	-.714	301	.260	.100			.124	.135	.200	.215	.271	.397
308	.260	.400			.049	.035	-.001	-.099	-.147	-.487	303	.260	.200			.097	.110	.155	.180	.240	.366
310	.260	.500			-.007	-.021	-.055	-.154	-.244	-.456	307	.260	.400			.036	.038	.071	.097	.170	.297
314	.260	.700			-.032	-.042	-.089	-.195	-.327	-.485	309	.260	.500			-.004	-.002	.030	.059	.138	.263
316	.260	.800			-.046	-.051	-.092	-.214	-.348	-.481	313	.260	.700			-.070	-.069	-.033	.003	.088	.203
											315	.260	.800			-.092	-.085	-.060	-.022	.062	.162
322	.620	.200			.004	-.011	-.112	-.461	-.768	-.865	317	.260	.900			-.117	-.111	-.082	-.042	.028	.108
324	.620	.400			-.058	-.070	-.145	-.311	-.907	-.917											
326	.620	.600			-.131	-.134	-.198	-.340	-.818	-1.061	323	.620	.400			-.072	-.056	.012	.095	.205	.328
328	.620	.800			-.099	-.103	-.188	-.319	-.559	-.912	327	.620	.800			-.166	-.149	-.090	-.025	.046	.128

TABLE IV.- PRESSURE COEFFICIENTS MEASURED ON RIGHT HORIZONTAL TAIL - Continued

(a) $M = 1.03$ - ConcludedPressure coefficients C_p on -

Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Lower surface							Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Upper surface								
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$				$\alpha = 20^\circ$	$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = -7.5^\circ; \text{brakes closed}$										$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = -7.5^\circ; \text{brakes closed}$								
304	.260	.200			.031	.013	-.024	-.167	-.830	-1.080	300	.260	.000			.493	.451	.325	.018	-.442	-.856
306	.260	.300			.012	-.001	-.037	-.158	-.391	-1.028	301	.260	.100			.058	.065	.124	.157	.204	.358
308	.260	.400			-.014	-.025	-.061	-.180	-.278	-.685	303	.260	.200			.024	.041	.088	.130	.177	.326
310	.260	.500			-.062	-.070	-.103	-.203	-.263	-.481	307	.260	.400			-.025	-.017	.023	.068	.133	.262
314	.260	.700			-.090	-.103	-.157	-.251	-.350	-.515	309	.260	.500			-.069	-.060	-.017	.035	.102	.229
316	.260	.800			-.104	-.114	-.163	-.271	-.369	-.530	313	.260	.700			-.125	-.113	-.078	-.036	.047	.176
322	.620	.200			-.051	-.058	-.165	-.444	-.798	-1.039	315	.260	.800			-.159	-.151	-.129	-.056	.037	.142
324	.620	.400			-.121	-.133	-.205	-.367	-.948	-1.107	317	.260	.900			-.184	-.175	-.132	-.081	.015	.094
326	.620	.600			-.193	-.193	-.270	-.392	-.893	-1.102	323	.620	.400			-.110	-.087	-.018	.065	.182	.309
328	.620	.800			-.148	-.152	-.244	-.377	-.559	-1.053	327	.620	.800			-.224	-.204	-.138	-.053	.041	.119
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 7.5^\circ; \text{brakes closed}$										$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 7.5^\circ; \text{brakes closed}$								
304	.260	.200			.227	.210	.145	.025	-.823	-1.136	300	.260	.000			.619	.576	.332	-.018	-.752	-.003
306	.260	.300			.219	.203	.148	.048	-.081	-.747	301	.260	.100			.248	.268	.337	.365	.432	.503
308	.260	.400			.206	.191	.140	.042	-.078	-.416	303	.260	.200			.245	.263	.321	.355	.424	.496
310	.260	.500			.181	.165	.120	.028	-.078	-.291	307	.260	.400			.247	.261	.302	.334	.399	.463
314	.260	.700			.163	.154	.107	.023	-.075	-.275	309	.260	.500			.239	.251	.289	.320	.377	.436
316	.260	.800			.140	.133	.087	.000	-.087	-.255	313	.260	.700			.213	.225	.254	.279	.322	.367
322	.620	.200			.140	.111	-.035	-.602	-.731	-.864	315	.260	.800			.174	.186	.210	.229	.261	.298
324	.620	.400			.088	.069	-.011	-.119	-.919	-.832	317	.260	.900			.110	.120	.138	.150	.172	.200
326	.620	.600			.037	.025	-.045	-.155	-.509	-.938	323	.620	.400			.128	.153	.217	.278	.352	.407
328	.620	.800			.004	-.002	-.067	-.174	-.344	-.754	327	.620	.800			-.008	.005	.040	.074	.117	.154
			$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$										$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$								
304	.260	.200	.051	.102		.110	.047	-.149	-.980	-1.198	300	.260	.000	-.348	.153		.651	.466	-.018	-.692	-.005
306	.260	.300	.098	.062		.081	.025	-.136	-.257	-1.137	301	.260	.100	-.547	-.037		.111	.175	.254	.366	.468
308	.260	.400	.116	.021		.050	-.005	-.147	-.318	-.709	303	.260	.200	-.419	-.062		.074	.142	.226	.343	.448
310	.260	.500	.108	-.043		-.008	-.064	-.198	-.359	-.500	307	.260	.400	-.430	-.127		.012	.063	.150	.269	.379
314	.260	.700	.135	-.088		-.034	-.107	-.242	-.432	-.548	309	.260	.500	-.455	-.158		-.033	.026	.114	.231	.339
316	.260	.800	.125	-.097		-.040	-.103	-.273	-.436	-.567	313	.260	.700	-.524	-.210		-.094	-.037	.062	.170	.267
322	.620	.200	.238	-.052		.015	-.137	-.576	-.911	-1.012	315	.260	.800	-.557	-.237		-.124	-.043	.040	.132	.217
324	.620	.400	.232	-.106		-.049	-.186	-.624	-1.088	-1.111	317	.260	.900	-.567	-.261		-.115	-.062	.011	.083	.150
326	.620	.600	.186	-.128		-.127	-.236	-.431	-1.098	-1.048	323	.620	.400	-.690	-.243		-.087	.024	.172	.305	.406
328	.620	.800	.115	-.137		-.085	-.216	-.401	-.836	-1.029	327	.620	.800	-.724	-.340		-.192	-.079	.011	.099	.173
			$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$										$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$								
304	.260	.200			.289	.274	.204	-.016	-1.102	-1.204	300	.260	.000			.752	.718	.542	-.047	-.855	-.005
306	.260	.300			.274	.263	.201	.040	-.231	-1.130	301	.260	.100			.253	.280	.340	.424	.509	.575
308	.260	.400			.258	.247	.189	.026	-.172	-.680	303	.260	.200			.246	.268	.331	.409	.498	.564
310	.260	.500			.228	.218	.162	.003	-.164	-.362	307	.260	.400			.267	.283	.322	.380	.456	.517
314	.260	.700			.209	.204	.146	.013	-.160	-.363	309	.260	.500			.265	.278	.311	.360	.428	.485
316	.260	.800			.187	.186	.124	.002	-.157	-.354	313	.260	.700			.249	.258	.281	.314	.364	.408
322	.620	.200			.197	.177	.029	-.489	-.886	-.986	315	.260	.800			.214	.220	.239	.262	.299	.335
324	.620	.400			.138	.126	.021	-.549	-.942	-1.063	317	.260	.900			.149	.154	.165	.179	.203	.231
326	.620	.600			.076	.076	-.021	-.196	-.892	-1.092	323	.620	.400			.129	.159	.238	.327	.412	.466
328	.620	.800			.057	.061	-.042	-.180	-.588	-.920	327	.620	.800			.029	.034	.063	.106	.156	.195

TABLE IV.- PRESSURE COEFFICIENTS MEASURED ON RIGHT HORIZONTAL TAIL - Continued

(e) $M = 1.18$ Pressure coefficients C_p on -

Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Lower surface								Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Upper surface							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$											$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
304	.260	.200				-.060	-.148	-.381	-.561	-.711	300	.260	.000				.422	.142	-.128	-.408	-.655
306	.260	.300				-.066	-.130	-.213	-.242	-.637	301	.260	.100				.064	.118	.170	.242	.437
308	.260	.400				-.082	-.138	-.232	-.272	-.459	303	.260	.200				.034	.075	.127	.271	.411
310	.260	.500				-.112	-.151	-.250	-.296	-.394	307	.260	.400				-.042	-.020	.143	.208	.348
314	.260	.700				-.122	-.162	-.230	-.313	-.421	309	.260	.500				-.064	.052	.096	.181	.319
316	.260	.800				-.112	-.138	-.198	-.318	-.402	313	.260	.700				-.018	.007	.047	.129	.282
322	.620	.200				-.131	-.286	-.464	-.644	-.748	315	.260	.800				-.042	-.017	.021	.099	.260
324	.620	.400				-.159	-.225	-.517	-.704	-.797	317	.260	.900				-.064	-.033	.007	.107	.227
326	.620	.600				-.191	-.249	-.373	-.693	-.801	323	.620	.400				.057	.072	.142	.215	.369
328	.620	.800				-.130	-.148	-.282	-.396	-.692	327	.620	.800				-.063	-.007	.032	.162	.250
			$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$											$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
304	.260	.200	.011	-.057	.025	-.009	-.133	-.376	-.616	-.737	300	.260	.000	-.157	-.340	.454	.362	.117	-.150	-.487	-.658
306	.260	.300	-.020	-.044	.059	.038	-.054	-.187	-.205	-.726	301	.260	.100	-.295	.007	.130	.167	.179	.277	.441	.592
308	.260	.400	.024	-.051	.109	.092	.030	-.071	-.220	-.497	303	.260	.200	-.293	-.052	.101	.152	.244	.325	.450	.579
310	.260	.500	.006	-.096	.166	.151	.094	.107	-.106	-.311	307	.260	.400	-.280	-.133	.118	.239	.319	.375	.456	.551
314	.260	.700	-.026	-.100	.212	.202	.178	.146	-.030	-.181	309	.260	.500	-.270	-.139	.126	.254	.330	.383	.447	.531
316	.260	.800	-.033	-.083	.203	.192	.178	.142	.025	-.154	313	.260	.700	-.251	-.037	.235	.275	.333	.371	.415	.478
322	.620	.200	-.004	-.073	.171	.139	-.347	-.506	-.695	-.738	315	.260	.800	-.262	-.038	.231	.259	.306	.336	.370	.420
324	.620	.400	-.026	-.128	.149	.112	.122	-.558	-.812	-.846	317	.260	.900	-.265	-.047	.205	.220	.254	.277	.302	.340
326	.620	.600	-.053	-.135	.128	.104	.090	.095	-.341	-.709	323	.620	.400	-.455	-.094	.225	.264	.326	.368	.417	.477
328	.620	.800	-.066	-.073	.121	.107	.049	.064	-.088	-.475	327	.620	.800	-.445	-.103	.147	.162	.190	.210	.243	.277
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
304	.260	.200	.156	.077		.094	.013	-.355	-.665	-.815	300	.260	.000	-.170	.126		.558	.333	.014	-.315	-.609
306	.260	.300	.114	.040		.074	.012	-.127	-.368	-.800	301	.260	.100	-.555	-.097		.154	.213	.248	.351	.461
308	.260	.400	.086	.011		.046	-.011	-.102	-.187	-.569	303	.260	.200	-.337	-.082		.120	.165	.218	.307	.424
310	.260	.500	.046	-.021		-.004	-.049	-.078	-.209	-.383	307	.260	.400	-.315	-.070		.038	.089	.135	.230	.354
314	.260	.700	.018	-.032		-.027	-.061	-.096	-.233	-.386	309	.260	.500	-.302	-.077		.012	.042	.099	.190	.323
316	.260	.800	.015	-.042		-.043	-.073	-.132	-.241	-.392	313	.260	.700	-.322	-.091		-.037	-.001	.069	.138	.289
322	.620	.200	.056	.019		.009	-.139	-.422	-.624	-.795	315	.260	.800	-.336	-.103		-.055	-.026	.033	.096	.273
324	.620	.400	.038	-.039		-.053	-.127	-.563	-.728	-.831	317	.260	.900	-.345	-.118		-.067	-.041	.022	.131	.244
326	.620	.600	.053	-.087		-.144	-.197	-.253	-.735	-.793	323	.620	.400	-.580	-.148		-.012	.069	.167	.267	.405
328	.620	.800	.108	-.098		-.085	-.154	-.249	-.493	-.825	327	.620	.800	-.441	-.168		-.101	-.034	.060	.198	.277
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
304	.260	.200			.169	.137	.072	-.297	-.664	-.807	300	.260	.000			.603	.541	.300	-.044	-.376	-.637
306	.260	.300			.140	.126	.056	.020	-.179	-.739	301	.260	.100			.168	.205	.258	.293	.370	.464
308	.260	.400			.106	.098	.054	-.008	-.117	-.399	303	.260	.200			.144	.176	.221	.250	.325	.426
310	.260	.500			.063	.058	.020	.001	-.152	-.313	307	.260	.400			.086	.097	.122	.155	.236	.346
314	.260	.700			.009	.008	-.021	-.079	-.189	-.343	309	.260	.500			.039	.050	.075	.125	.196	.310
316	.260	.800			.012	-.004	-.045	-.097	-.209	-.355	313	.260	.700			-.015	-.005	.024	.080	.139	.271
322	.620	.200			.077	.045	-.073	-.303	-.549	-.736	315	.260	.800			-.036	-.025	-.006	.045	.095	.253
324	.620	.400			.005	-.009	-.084	-.400	-.644	-.803	317	.260	.900			-.050	-.041	-.024	.016	.113	.230
326	.620	.600			-.075	-.074	-.135	-.300	-.658	-.806	323	.620	.400			-.012	-.022	.090	.175	.271	.394
328	.620	.800			-.045	-.046	-.118	-.225	-.417	-.750	327	.620	.800			-.100	-.081	-.018	.056	.177	.266

TABLE IV.- PRESSURE COEFFICIENTS MEASURED ON RIGHT HORIZONTAL TAIL - Concluded

(e) $M = 1.18$ - ConcludedPressure coefficients C_p on -

Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Lower surface								Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Upper surface							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = -7.5^\circ; \text{brakes closed}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = -7.5^\circ; \text{brakes closed}$							
304	.260	.200			.048	.030	-.055	-.375	-.666	-.769	300	.260	.000			.544	.486	.276	.009	-.273	-.506
306	.260	.300			.014	-.004	-.078	-.242	-.527	-.736	301	.260	.100			.026	.078	.148	.170	.243	.404
308	.260	.400			-.015	-.032	-.093	-.203	-.266	-.667	303	.260	.200			.002	.045	.080	.118	.191	.383
310	.260	.500			-.063	-.067	-.124	-.205	-.304	-.456	307	.260	.400			-.043	-.037	-.013	.010	.203	.329
314	.260	.700			-.086	-.091	-.144	-.191	-.304	-.433	309	.260	.500			-.085	-.084	-.064	.100	.171	.295
316	.260	.800			-.097	-.097	-.125	-.188	-.298	-.436	313	.260	.700			-.059	-.036	-.006	.069	.130	.242
											315	.260	.800			-.083	-.065	-.025	.032	.090	.218
322	.620	.200			-.030	-.053	-.205	-.452	-.644	-.779	317	.260	.900				-.081	-.045	.016	.101	.203
324	.620	.400			-.094	-.098	-.205	-.443	-.715	-.820											
326	.620	.600			-.176	-.176	-.252	-.360	-.711	-.818	323	.620	.400			-.123	-.096	.066	.155	.249	.366
328	.620	.800			-.143	-.145	-.208	-.317	-.624	-.775	327	.620	.800			-.145	-.106	-.038	.049	.159	.250
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 7.5^\circ; \text{brakes closed}$											$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 7.5^\circ; \text{brakes closed}$							
304	.260	.200			.198	.183	.107	-.241	-.647	-.820	300	.260	.000			.600	.553	.308	-.004	-.381	.014
306	.260	.300			.221	.214	.152	-.099	-.315	-.761	301	.260	.100			.203	.229	.279	.336	.471	.592
308	.260	.400			.238	.232	.186	.184	-.036	-.476	303	.260	.200			.148	.152	.161	.167	.175	.580
310	.260	.500			.243	.235	.191	.131	.034	-.236	307	.260	.400			.300	.316	.355	.408	.471	.558
314	.260	.700			.262	.255	.212	.155	.116	-.140	309	.260	.500			.311	.325	.363	.412	.466	.540
316	.260	.800			.249	.244	.202	.144	.109	-.120	313	.260	.700			.310	.325	.357	.396	.439	.491
											315	.260	.800			.286	.302	.329	.359	.394	.436
322	.620	.200			.208	.171	.003	-.310	-.539	-.722	317	.260	.900			.238	.254	.275	.297	.321	.355
324	.620	.400			.181	.167	.098	-.431	-.709	-.839											
326	.620	.600			.153	.151	.074	-.001	-.400	-.748	323	.620	.400			.239	.266	.330	.398	.460	.516
328	.620	.800			.156	.150	.075	-.006	-.144	-.514	327	.620	.800			.166	.174	.202	.232	.268	.309
			$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$											$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
304	.260	.200	.159	.163		.189	.112	-.311	-.607	-.804	300	.260	.000	-.082	.294		.705	.465	.031	-.275	.007
306	.260	.300	.128	.133		.170	.104	-.047	-.270	-.721	301	.260	.100	-.451	.034		.208	.267	.305	.376	.509
308	.260	.400	.111	.103		.142	.086	.013	-.111	-.685	303	.260	.200	-.382	.043		.161	.224	.267	.338	.475
310	.260	.500	.078	.053		.095	.025	-.020	-.191	-.430	307	.260	.400	-.272	.019		.101	.135	.185	.272	.418
314	.260	.700	.067	.012		.048	-.005	-.113	-.228	-.401	309	.260	.500	-.291	-.020		.050	.081	.149	.250	.389
316	.260	.800	.090	.000		.045	-.022	-.119	-.214	-.425	313	.260	.700	-.347	-.086		-.010	.025	.112	.202	.349
											315	.260	.800	-.385	-.119		-.031	.007	.076	.200	.321
322	.620	.200	.203	.014		.110	-.052	-.347	-.573	-.762	317	.260	.900	-.405	-.140		-.052	-.011	.056	.196	.279
324	.620	.400	.220	-.031		.047	-.073	-.449	-.690	-.789											
326	.620	.600	.241	-.071		-.034	-.142	-.382	-.711	-.797	323	.620	.400	-.493	-.143		.012	.109	.226	.352	.486
328	.620	.800	.221	-.047		-.006	-.111	-.263	-.501	-.731	327	.620	.800	-.524	-.209		-.105	-.004	.101	.224	.309
			$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$											$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
304	.260	.200			.301	.292	.216	-.068	-.610	-.814	300	.260	.000			.780	.738	.525	.065	-.324	.007
306	.260	.300			.306	.301	.232	.131	-.029	-.699	301	.260	.100			.246	.269	.331	.411	.517	.633
308	.260	.400			.307	.303	.236	.118	.026	-.577	303	.260	.200			.261	.271	.335	.414	.507	.625
310	.260	.500			.298	.295	.225	.086	.055	-.294	307	.260	.400			.327	.341	.380	.448	.522	.600
314	.260	.700			.311	.307	.238	.104	.024	-.284	309	.260	.500			.344	.356	.388	.447	.512	.580
316	.260	.800			.298	.295	.224	.108	.007	-.282	313	.260	.700			.354	.364	.388	.427	.476	.528
											315	.260	.800			.334	.341	.360	.389	.427	.470
322	.620	.200			.270	.261	.089	-.323	-.563	-.690	317	.260	.900			.284	.290	.302	.321	.348	.382
324	.620	.400			.234	.229	.103	-.424	-.716	-.805											
326	.620	.600			.191	.198	.088	-.104	-.522	-.773	323	.620	.400			.231	.264	.349	.433	.511	.576
328	.620	.800			.182	.189	.084	-.026	-.251	-.666	327	.620	.800			.170	.183	.219	.259	.304	.348

TABLE V.- PRESSURE COEFFICIENTS MEASURED ON VERTICAL TAIL AND SPEED BRAKES

(a) $M = 0.60$ Pressure coefficients C_p on -

Upper tail										Lower tail											
Model orifice number	$\frac{y}{b/2}$	α	$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$	Model orifice number	$\frac{y}{b/2}$	α	$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
$\beta = -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$										$\beta = -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$											
401	.110	.647				-.204	-.207	-.221			441	.130	.647				-.209	-.181	-.153	-.113	-.062
402	.110	.681				-.122	-.140	-.141			442	.130	.682				-.098	-.078	-.045	.001	.056
403	.110	.698				-.115	-.135	-.134			443	.130	.700				-.096	-.075	-.044	.001	.054
404	.110	.748				-.138	-.152	-.151			444	.130	.748				-.085	-.068	-.042	.001	.052
405	.110	.832				-.159	-.173	-.158			445	.130	.832				-.130	-.117	-.059	-.065	-.027
406	.110	.949				-.214	-.228	-.196			446	.130	.949				-.177	-.168	-.159	-.138	-.111
410	.250	.000				.658	.626	.601			450	.300	.000				.444	.503	.559	.641	.678
411	.250	.200				.577	.593	.568			451	.300	.200				.401	.423	.418	.493	.531
412	.250	.400				-.069	-.091	-.151			452	.300	.400				-.051	-.027	.004	.049	.103
413	.250	.600				-.115	-.136	-.182			453	.300	.600				-.102	-.080	-.051	-.007	.043
414	.250	.630				-.214	-.231	-.252			454	.300	.633				-.194	-.166	-.144	-.103	.061
415	.250	.668				-.101	-.118	-.184			455	.300	.668				-.090	-.072	-.039	.008	.057
416	.250	.685				-.101	-.120	-.176			456	.300	.686				-.085	-.069	-.039	.005	.054
417	.250	.738				-.121	-.140	-.188			457	.300	.738				-.099	-.084	-.057	-.018	.026
418	.250	.825				-.142	-.158	-.204			458	.300	.825				-.106	-.094	-.076	-.042	.004
419	.250	.947				-.208	-.223	-.245			459	.300	.947				-.165	-.158	-.149	-.125	-.092
420	.500	.000				.615	.547	.489			460	.600	.000				.364	.461	.546	.643	.683
421	.500	.100				-.172	-.250	-.280			461	.600	.100				-.097	-.116	-.129	-.074	-.082
422	.500	.200				-.068	-.095	-.119			462	.600	.200				-.045	-.011	.026	.063	.107
423	.500	.300				-.065	-.085	-.114			463	.600	.300				-.042	.015	.020	.055	.101
424	.500	.400				-.066	-.092	-.115			464	.600	.400				-.054	-.027	.005	.042	.090
425	.500	.500				-.077	-.099	-.118			465	.600	.500				-.065	-.038	.007	.031	.076
426	.500	.600				-.089	-.106	-.121			466	.600	.600				-.082	-.061	-.032	.005	.052
427	.500	.700									467	.600	.700				-.084	-.066	-.042	-.008	.032
428	.500	.800				-.132	-.143	-.152			468	.600	.800				-.102	-.089	-.072	-.037	.003
429	.500	.900				-.164	-.175	-.180			469	.600	.900				-.130	-.123	-.114	-.086	-.045
430	.830	.000				.654	.583	.527			470	.800	.000				.539	.619	.691	.768	.813
431	.830	.200				-.111	-.129	-.142			471	.800	.200				-.072	-.036	.005	.044	.091
432	.830	.400				-.084	-.088	-.102			472	.800	.400				-.062	-.036	-.007	.035	.082
433	.830	.600				-.089	-.096	-.104			473	.800	.600				-.074	-.056	-.028	.012	.057
434	.830	.800				-.120	-.125	-.127			474	.800	.800				-.098	-.089	-.067	-.027	.010
$\beta = -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$																					
401	.110	.647	-.125	-.186	.516	.493	.306	.506	.566	.478	441	.130	.647	-.128	-.106	.408	.487	.507	.516	.524	.524
402	.110	.681	-.059	-.106	.764	.410	.742	.850	.773	.442	.130	.682	-.128	-.108	.745	.818	.823	.816	.781	.771	
403	.110	.698	-.059	-.103	.714	.709	.383	.718	.770	.721	.443	.130	.700	-.126	-.108	.658	.735	.741	.733	.708	
404	.110	.748	-.084	-.126	.501	.499	.538	.601	.547	.457	.444	.130	.748	-.148	-.123	.495	.532	.539	.542	.539	
405	.110	.832	-.118	-.153	.357	.357	.238	.444	.396	.293	445	.130	.832	-.219	-.166	.390	.393	.385	.387	.391	
406	.110	.949	-.184	-.215	.130	.135	.048	.196	.164	.068	446	.130	.949	-.219	-.166	.174	.156	.143	.141	.145	
410	.250	.000	.849	.704	.678	.657	.616	.638	.679	.662	450	.300	.000	.522	.236	.408	.460	.515	.573	.659	
411	.250	.200	-.009	-.069	.000	.005	-.027	-.087	-.075	-.071	451	.300	.200	.151	-.115	.024	.037	.054	.081	.121	
412	.250	.400	-.005	-.055	.081	.074	.048	.019	.025	-.025	452	.300	.400	-.146	-.106	.086	.102	.123	.150	.186	
413	.250	.600	-.056	-.105	.351	.355	.261	.323	.384	.352	453	.300	.600	-.168	-.138	.327	.356	.367	.375	.404	
414	.250	.630	-.152	-.193	.435	.438	.345	.412	.465	.429	454	.300	.633	-.272	-.187	.416	.434	.431	.427	.466	
415	.250	.668	-.038	-.088	.659	.689	.611	.642	.777	.772	455	.300	.668	-.156	-.141	.757	.734	.701	.670	.651	
416	.250	.685	-.044	-.093	.594	.622	.527	.534	.665	.677	456	.300	.686	-.144	-.133	.660	.648	.621	.600	.592	
417	.250	.738	-.067	-.110	.444	.447	.330	.363	.470	.430	457	.300	.738	-.157	-.139	.457	.471	.474	.476	.479	
418	.250	.825	-.096	-.138	.311	.301	.190	.303	.348	.284	458	.300	.825	-.157	-.148	.323	.325	.335	.354	.372	
419	.250	.947	-.170	-.211	.105	.101	.024	.138	.147	.094	459	.300	.947	-.218	-.192	.118	.113	.118	.131	.156	
420	.500	.000	.877	.717	.654	.611	.535	.496	.469	.432	460	.600	.000	.364	.142	.288	.367	.444	.553	.655	
421	.500	.100	-.074	-.251	-.108	-.084	-.203	-.166	-.200	-.200	461	.600	.100	-.223	-.021	.020	.011	-.050	-.070	-.016	
422	.500	.200	-.009	-.030	.019	.005	-.024	-.033	-.127	-.252	462	.600	.200	-.197	-.087	.012	.035	.068	.103	.137	
423	.500	.300	.001	-.028	.049	.039	.006	.004	-.089	-.165	463	.600	.300	-.192	-.093	.061	.062	.088	.121	.153	
424	.500	.400	-.004	-.041	.080	.069	.027	.048	-.055	-.092	464	.600	.400	-.204	-.094	.059	.084	.110	.141	.171	
425	.500	.500	-.017	-.055	.115	.104	.056	.099	-.011	-.013	465	.600	.500	-.212	-.094	.089	.118	.144	.170	.197	
426	.500	.600	-.035	-.089	.110	.098	.040	.093	-.016	-.020	466	.600	.600	-.260	-.105	.132	.159	.184	.211		
427	.500	.700	-.056	-.091	-.223	-.227	-.247	-.205	-.232	-.252	467	.600	.700	-.233	-.103	-.067	-.051	-.030	-.003	.054	
428	.500	.800	-.083	-.119	-.306	-.312	-.318	-.276	-.285	-.333	468	.600	.800	-.232	-.113	-.218	-.211	-.194	-.172	.051	
429	.500	.900	-.119	-.160	-.306	-.312	-.318	-.276	-.285	-.333	469	.600	.900	-.232	-.113	-.297	-.291	-.283	-.265	-.231	
430	.830	.000	.895	.775	.701	.659	.576	.531	.527	.491	470	.800	.000	.433	.371	.486	.544	.623	.700	.777	
431	.830	.200	.001	-.063	-.024	-.054	-.069	-.070	-.074	-.109	471	.800	.200	-.131	-.122	-.022	.005	.041	.081	.117	
432	.830	.400	-.020	-.057	.018	.008	-.017	-.004	-.023	-.103	472	.800	.400	-.146	-.061	.039	.078	.115	.179	.241	
433	.830	.600	-.023	-.073	-.012	-.022	-.044	-.032	-.055	-.141	473	.800	.600	-.164	-.083	.019	.034	.054	.084	.118	
434	.830	.800	-.046	-.110	-.122	-.123	-.145	-.153	-.194	-.308	474	.800	.800			-.151	-.135	-.117	-.097	-.068	

TABLE V.- PRESSURE COEFFICIENTS MEASURED ON VERTICAL TAIL AND SPEED BRAKES - Continued

(a) $M = 0.60$ - ContinuedPressure coefficients C_p on -

Model orifice number	y b/2	x c	Upper tail								Model orifice number	y b/2	x c	Lower tail								
			α = -20°	α = -10°	α = -3°	α = 0°	α = 3°	α = 10°	α = 15°	α = 20°				α = -20°	α = -10°	α = -3°	α = 0°	α = 3°	α = 10°	α = 15°	α = 20°	
			β = 0°; δ _e = 0°; δ _v = 0°; brakes closed																			
401	.110	.647	-.088	-.155	-.151	-.167	-.192	-.230	-.298	441	.130	.647	-.292	-.191	-.168	-.154	-.136	-.097	-.044			
402	.110	.681	-.011	-.073	-.069	-.081	-.114	-.179	-.239	442	.130	.682	-.198	-.077	-.063	-.047	-.017	.034	.082			
403	.110	.698	-.013	-.073	-.068	-.080	-.110	-.172	-.237	443	.130	.700	-.189	-.074	-.062	-.047	-.020	.029	.074			
404	.110	.748	-.045	-.103	-.093	-.105	-.135	-.189	-.258	444	.130	.748	-.212	-.096	-.084	-.069	-.048	-.004	.039			
405	.110	.832	-.086	-.134	-.121	-.128	-.157	-.203	-.283	445	.130	.832	-.231	-.115	-.105	-.098	-.084	-.051	-.018			
406	.110	.949	-.162	-.197	-.180	-.186	-.209	-.236	-.327	446	.130	.949	-.271	-.165	-.148	-.143	-.145	-.129	-.104			
β = 0°; δ _e = 0°; δ _v = 0°; brakes open																						
410	.250	.000	.940	.899	.876	.879	.904	.935	.924	450	.300	.000			.804	.831	.854	.888	.915			
411	.250	.200	.105	.058	.059	.045	.011	-.061	-.126	451	.300	.200	-.104	.021	.057	.080	.106	.140	.186			
412	.250	.400	.063	.007	.008	-.007	-.035	-.097	-.164	452	.300	.400	-.144	-.020	.011	.034	.061	.102	.153			
413	.250	.600	-.009	-.068	-.063	-.076	-.102	-.146	-.222	453	.300	.600	-.200	-.081	-.066	-.049	-.022	.023	.073			
414	.250	.830	-.119	-.175	-.163	-.181	-.205	-.240	-.312	454	.300	.830	-.291	-.178	-.155	-.143	-.130	-.096	.050			
415	.250	.668	.013	-.048	-.048	-.056	-.081	-.125	-.206	455	.300	.668	-.185	-.066	-.056	-.040	-.008	.042	.090			
416	.250	.685	.003	-.052	-.052	-.061	-.084	-.131	-.211	456	.300	.686	-.180	-.061	-.052	-.036	-.011	.037	.082			
417	.250	.738	-.026	-.080	-.077	-.085	-.109	-.151	-.234	457	.300	.738	-.193	-.076	-.069	-.055	-.031	.010	.056			
418	.250	.825	-.063	-.110	-.102	-.107	-.128	-.169	-.256	458	.300	.825	-.203	-.089	-.082	-.075	-.059	-.023	.012			
419	.250	.947	-.144	-.185	-.177	-.186	-.197	-.220	-.309	459	.300	.947	-.236	-.210	-.140	-.136	-.133	-.112	-.081			
β = 0°; δ _e = 0°; δ _v = 0°; brakes open																						
420	.500	.000	.937	.862	.781	.750	.742	.757	.770	460	.600	.000	.734	.739	.793	.834	.870	.912	.945			
421	.500	.100	.131	.078	.072	.052	.017	-.033	-.099	461	.600	.100	-.104	.018	.070	.097	.122	.153	.196			
422	.500	.200	.096	.048	.041	.023	-.005	-.053	-.125	462	.600	.200	-.118	-.002	.035	.061	.089	.127	.173			
423	.500	.300	.076	.027	.019	.001	-.023	-.071	-.139	463	.600	.300	-.126	-.013	.012	.038	.068	.110	.159			
424	.500	.400	.060	.006	-.001	-.016	-.040	-.087	-.156	464	.600	.400	-.142	-.031	-.011	.016	.048	.093	.142			
425	.500	.500	.039	-.013	-.020	-.036	-.057	-.100	-.172	465	.600	.500	-.152	-.043	-.026	-.002	.028	.076	.123			
426	.500	.600	.018	-.033	-.036	-.050	-.069	-.106	-.180	466	.600	.600			-.026	.045	.024	.064	.093			
427	.500	.700	-.005	-.056	-.057	-.068	-.084	-.123	-.198	467	.600	.700	-.171	-.066	-.053	-.035	-.012	.032	.071			
428	.500	.800	-.041	-.089	-.087	-.100	-.111	-.143	-.219	468	.600	.800	-.189	-.088	-.073	-.058	-.039	-.006	.032			
429	.500	.900	-.091	-.134	-.128	-.140	-.154	-.176	-.255	469	.600	.900	-.217	-.124	-.109	-.099	-.088	-.060	.027			
β = 0°; δ _e = 0°; δ _v = 0°; brakes open																						
430	.830	.000	.943	.847	.749	.705	.678	.667	.674	470	.800	.000	.725	.723	.781	.825	.870	.924	.961			
431	.830	.200	.100	.039	.023	.009	-.039	-.099	-.164	471	.800	.200	-.104	.018	.023	.051	.085	.126	.172			
432	.830	.400	.063	.007	-.013	-.035	-.061	-.082	-.139	472	.800	.400	-.135	-.039	-.012	.015	.049	.094	.142			
433	.830	.600	.039	.015	-.037	-.052	-.069	-.093	-.143	473	.800	.600	-.151	-.060	-.038	-.015	.013	.061	.103			
434	.830	.800	-.014	-.066	-.081	-.094	-.107	-.125	-.177	474	.800	.800	-.173	-.089	-.071	-.053	-.031	.008	.043			
β = 0°; δ _e = 0°; δ _v = 0°; brakes open																						
401	.110	.647	-.149	-.148	-.148	-.146	-.175	-.238	-.323	441	.130	.647	-.186	-.184	-.175	-.152	-.114	-.087	-.034			
402	.110	.681	-.063	-.057	-.057	-.055	-.095	-.167	-.253	442	.130	.682	-.056	-.054	-.045	-.028	.012	.051	.097			
403	.110	.698	-.064	-.057	-.057	-.055	-.093	-.160	-.252	443	.130	.700	-.059	-.057	-.047	-.031	.007	.044	.090			
404	.110	.748	-.100	-.092	-.092	-.091	-.125	-.187	-.287	444	.130	.748	-.085	-.083	-.078	-.064	-.030	.000	.046			
405	.110	.832	-.136	-.127	-.127	-.125	-.158	-.209	-.315	445	.130	.832	-.112	-.113	-.110	-.103	-.077	-.054	.000			
406	.110	.949	-.196	-.189	-.189	-.186	-.208	-.243	-.343	446	.130	.949	-.171	-.171	-.170	-.163	-.143	-.130	.000			
β = 0°; δ _e = 0°; δ _v = 0°; brakes open																						
410	.250	.000	.872	.863	.871	.888	.875	.919	.919	450	.300	.000	.801	.815	.843	.873	.900	.930	.960			
411	.250	.200	.164	.161	.154	.113	.021	-.052	.051	451	.300	.200	.129	.140	.157	.173	.189	.222	.252			
412	.250	.400	.049	.051	.050	.015	-.064	-.148	.452	452	.300	.400	.031	.036	.050	.070	.107	.146	.185			
413	.250	.600	-.068	-.062	-.062	-.090	-.164	-.269	.453	453	.300	.600	-.071	-.068	-.060	-.041	-.007	.025	.064			
414	.250	.830	-.194	-.184	-.181	-.217	-.285	-.383	.454	454	.300	.830	-.192	-.189	-.184	-.168	-.137	.018	.067			
415	.250	.668	-.071	-.060	-.060	-.084	-.173	-.288	.455	455	.300	.668	-.068	-.067	-.062	-.052	-.022	.006	.055			
416	.250	.685	-.071	-.057	-.055	-.090	-.171	-.287	.456	456	.300	.686	-.061	-.062	-.056	-.043	-.016	.007	.056			
417	.250	.738	-.102	-.091	-.087	-.119	-.200	-.317	.457	457	.300	.738	-.080	-.080	-.077	-.068	-.042	.022	.071			
418	.250	.825	-.137	-.124	-.116	-.149	-.230	-.353	.458	458	.300	.825	-.099	-.100	-.099	-.094	-.077	.064	.113			
419	.250	.947	-.191	-.187	-.183	-.198	-.252	-.357	.459	459	.300	.947	-.163	-.160	-.155	-.148	-.133	-.126	.000			
β = 0°; δ _e = 0°; δ _v = 0°; brakes open																						
420	.500	.000	.691	.675	.648	.646	.672	.703	.660	460	.600	.000	.762	.784	.823	.873	.917	.957	.996			
421	.500	.100	.279	.270	.258	.230	.180	.118	.461	461	.600	.100	.217	.232	.257	.277	.299	.334	.369			
422	.500	.200	.177	.178	.172	.142	.092	.014	.462	462	.600	.200	.118	.130	.149	.175	.205	.241	.276			
423	.500	.300	.117	.117	.111	.090	.041	-.035	.463	463	.600	.300	.077	.085	.106	.130	.158	.196	.233			
424	.500	.400	.078	.080	.076	.052	.007	-.075	.464	464	.600	.400	.046	.053	.073	.094	.128	.168	.208			
425	.500	.500	.047	.050	.045	.031	-.013	-.099	.465	465	.600	.500	.026	.032	.050	.074	.110	.146	.182			
426	.500	.600	.026	.032	.028	.014	-.030	-.116	.466	466	.600	.600	.001	.007	.023	.046	.080	.115	.150			
427	.500	.700	-.002	.002	.002	-.012	-.053	-.143	.467	467	.600	.700	-.009	-.005	.008	.027	.060	.092	.127			
428	.500	.800	-.034	-.032	-.028	-.042	-.079	-.171	.468	468	.600	.800	-.032	-.030	-.020	.002	.026	.053	.088			
429	.500	.900	-.077	-.075	-.066	-.079	-.115	-.208	.469	469	.600	.900	-.068	-.065	-.059	-.047	-.023	.001	.036			
β = 0°; δ _e = 0°; δ _v = 0°; brakes open																						
430	.830	.000	.595	.555	.501	.465	.456	.469	.470	470	.800	.000	.687	.713	.761	.824	.885	.935	.985			
431	.830	.200	.053	.046	.032	.011	-.018	-.064	.471	471	.800	.200	.087	.100	.127	.162	.198	.243	.288			
432	.830	.400	.025	.016	.008	-.010	-.029	-.076	.472	472	.800	.400	.026	.038	.065	.100	.144	.188	.233			
433	.830	.600	-.019	-.025	-.037	-.050	-.061	-.113	.473	473	.800	.600	-.008	.002	.025	.060	.103	.145	.187			
434	.830	.800							.474	474	.800	.800	-.039	-.032	-.014	.013	.053	.089	.134			

TABLE V.- PRESSURE COEFFICIENTS MEASURED ON VERTICAL TAIL AND SPEED BRAKES - Continued

(a) $M = 0.60$ - ContinuedPressure coefficients C_p on -

Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Upper tail								Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Lower tail																							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$																
			$\beta = 0^\circ; \delta_e = 0^\circ; \delta_v = -7.5^\circ; \text{brakes closed}$																																		
401	.110	.647										.130	.647																								
402	.110	.681										.130	.682																								
403	.110	.698										.130	.700																								
404	.110	.748										.130	.748																								
405	.110	.832										.130	.832																								
406	.110	.949										.130	.949																								
410	.250	.000										.300	.000																								
411	.250	.200										.300	.200																								
412	.250	.400										.300	.400																								
413	.250	.600										.300	.600																								
414	.250	.630										.300	.633																								
415	.250	.668										.300	.668																								
416	.250	.685										.300	.686																								
417	.250	.738										.300	.738																								
418	.250	.825										.300	.825																								
419	.250	.947										.300	.947																								
420	.500	.000										.600	.000																								
421	.500	.100										.600	.100																								
422	.500	.200										.600	.200																								
423	.500	.300										.600	.300																								
424	.500	.400										.600	.400																								
425	.500	.500										.600	.500																								
426	.500	.600										.600	.600																								
427	.500	.700										.600	.700																								
428	.500	.800										.600	.800																								
429	.500	.900										.600	.900																								
430	.830	.000										.800	.000																								
431	.830	.200										.800	.200																								
432	.830	.400										.800	.400																								
433	.830	.600										.800	.600																								
434	.830	.800										.800	.800																								
$\beta = 0^\circ; \delta_e = 0^\circ; \delta_v = 7.5^\circ; \text{brakes closed}$																			$\beta = 0^\circ; \delta_e = 0^\circ; \delta_v = 7.5^\circ; \text{brakes closed}$																		
401	.110	.647										.130	.647																								
402	.110	.681										.130	.682																								
403	.110	.698										.130	.700																								
404	.110	.748										.130	.748																								
405	.110	.832										.130	.832																								
406	.110	.949										.130	.949																								
410	.250	.000										.300	.000																								
411	.250	.200										.300	.200																								
412	.250	.400										.300	.400																								
413	.250	.600										.300	.600																								
414	.250	.630										.300	.633																								
415	.250	.668										.300	.668																								
416	.250	.685										.300	.686																								
417	.250	.738										.300	.738																								
418	.250	.825										.300	.825																								
419	.250	.947										.300	.947																								
420	.500	.000										.600	.000																								
421	.500	.100										.600	.100																								
422	.500	.200										.600	.200																								
423	.500	.300										.600	.300																								
424	.500	.400										.600	.400																								
425	.500	.500										.600	.500																								
426	.500	.600										.600	.600																								
427	.500	.700										.600	.700																								
428	.500	.800										.600	.800																								
429	.500	.900										.600	.900																								
430	.830	.000										.800	.000																								
431	.830	.200										.800	.200																								
432	.830	.400										.800	.400																								
433	.830	.600										.800	.600																								
434	.830	.800										.800	.800																								
$\beta = 0^\circ; \delta_e = 0^\circ; \delta_v = 7.5^\circ; \text{brakes closed}$																			$\beta = 0^\circ; \delta_e = 0^\circ; \delta_v = 7.5^\circ; \text{brakes closed}$																		
401	.110	.647										.130	.647																								
402	.110	.681										.130	.682																								
403	.110	.698										.130	.700																								
404	.110	.748										.130	.748																								
405	.110	.832										.130	.832																								
406	.110	.949										.130	.949																								
410	.250	.000										.300	.000																								
411	.250	.200										.300	.200																								
412	.250	.400										.300	.400																								
413	.250	.600										.300	.600																								
414	.250	.630										.300	.633																								
415	.250	.668										.300	.668																								
416	.250	.685										.300	.686																								
417	.250	.738										.300	.738																								
418	.250	.825										.300	.825																								
419	.250	.947										.300	.947																								
420	.500	.000										.600	.000																								
421	.500	.100										.600	.100																								
422	.500	.200																																			

TABLE V.- PRESSURE COEFFICIENTS MEASURED ON VERTICAL TAIL AND SPEED BRAKES - Continued

(a) $M = 0.60$ - ConcludedPressure coefficients C_p on -

Model orifice number	$\frac{y}{b/2}$	α	Upper tail								Model orifice number	$\frac{y}{b/2}$	α	Lower tail							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta = 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$																		
			$\beta = 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$																		
401	.110	.647	-.054	-.126	-.123	-.130	-.152	-.233	-.339	441	.130	.647	-.326	-.169	-.174	-.151	-.119	-.076	-.025		
402	.110	.681	.041	-.023	-.009	-.014	-.058	-.142	-.274	442	.130	.682	-.242	-.042	-.019	-.004	.028	.070	.121		
403	.110	.698	.038	-.023	-.008	-.012	-.056	-.141	-.269	443	.130	.700	-.235	-.039	-.034	-.005	.026	.066	.111		
404	.110	.748	-.003	-.058	-.079	-.079	-.113	-.193	-.312	444	.130	.748	-.254	-.061	-.047	-.037	-.008	.026	.071		
405	.110	.832	-.057	-.099	-.079	-.079	-.113	-.193	-.312	445	.130	.832	-.276	-.087	-.076	-.069	-.052	-.028	.006		
406	.110	.949	-.152	-.187	-.172	-.167	-.188	-.263	-.360	446	.130	.949	-.306	-.145	-.147	-.142	-.136	-.125	-.098		
410	.250	.000	.833	.740	.539	.552	.606	.744	.688	450	.300	.000	.353	.688	.747	.797	.855	.899	.933		
411	.250	.200	.229	.195	.209	.196	.136	.016	-.150	451	.300	.200	-.139	.141	.172	.190	.208	.230	.283		
412	.250	.400	.143	.086	.092	.083	.044	-.060	-.189	452	.300	.400	-.182	.045	.063	.086	.118	.162	.216		
413	.250	.600	.047	-.016	-.011	-.018	-.048	-.129	-.248	453	.300	.600	-.233	-.042	-.034	-.013	.021	.065	.115		
414	.250	.630	-.090	-.151	.004	.000	-.030	-.106	-.236	454	.300	.633	-.317	-.166	-.165	-.143	-.113	-.078	-.030		
415	.250	.668	.068	.004	-.001	-.005	-.036	-.110	-.242	455	.300	.668	-.220	-.030	-.016	.003	.035	.078	.129		
416	.250	.685	.058	-.006	.004	.000	-.036	-.110	-.242	456	.300	.686	-.217	-.024	.012	.003	.035	.074	.122		
417	.250	.738	.021	-.037	-.033	-.036	-.061	-.132	-.258	457	.300	.738	-.225	-.044	-.042	-.020	.009	.047	.087		
418	.250	.825	-.028	-.076	-.066	-.067	-.089	-.160	-.281	458	.300	.825	-.230	-.062	-.054	-.044	-.023	.003	.042		
419	.250	.947	-.132	-.175	-.163	-.161	-.174	-.243	-.344	459	.300	.947	-.287	-.140	-.154	-.129	-.117	-.101	-.071		
420	.500	.000	.530	.481	.471	.481	.414	.075	.460	.600	.000	.610	.651	.700	.771	.852	.897	.953			
421	.500	.100	.303	.267	.271	.250	.205	.084	-.069	.461	.600	.100	-.043	.189	.238	.257	.272	.289	.336		
422	.500	.200	.208	.164	.177	.160	.123	.022	-.105	.462	.600	.200	-.108	.092	.127	.152	.179	.210	.268		
423	.500	.300	.161	.108	.099	.100	.069	-.017	-.133	.463	.600	.300	-.128	.043	.075	.108	.139	.180	.234		
424	.500	.400	.139	.067	.071	.071	.024	-.153	-.153	.464	.600	.400	-.145	.022	.041	.061	.080	.124	.177		
425	.500	.500	.108	.039	.037	.026	-.001	-.060	-.167	.465	.600	.500	-.155	-.004	.012	.041	.080	.124	.177		
426	.500	.600	.080	.023	.005	.003	-.017	-.073	-.176	.466	.600	.600	-.168	-.028	-.028	.012	.049	.089	.141		
427	.500	.700			-.016	-.023	-.040	-.090	-.192	.467	.600	.700	-.171	-.036	-.023	-.004	.028	.067	.114		
428	.500	.800	.001	-.051	-.054	-.056	-.069	-.120	-.221	.468	.600	.800	-.190	-.063	-.049	-.034	-.008	.025	.066		
429	.500	.900	-.063	-.109	-.109	-.106	-.113	-.167	-.261	.469	.600	.900	-.219	-.099	-.091	-.080	-.063	-.042	-.004		
430	.830	.000	.863	.745	.543	.487	.444	.402	.392	470	.800	.000	.624	.610	.696	.765	.843	.904	.949		
431	.830	.200	.194	.132	.081	.066	.039	-.014	-.081	471	.800	.200	-.100	.039	.063	.110	.160	.198	.258		
432	.830	.400	.147	.068	-.152	-.016	-.018	-.050	-.106	472	.800	.400	-.123	-.009	.015	.050	.098	.143	.203		
433	.830	.600	.111	.033	-.015	-.028	-.043	-.065	-.120	473	.800	.600	-.136	-.041	-.019	.011	.054	.098	.155		
434	.830	.800	.064	-.026	-.060	-.071	-.080	-.101	-.152	474	.800	.800	-.155	-.071	-.056	-.032	-.001	.038	.087		
			$\beta = 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$											$\beta = 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
401	.110	.647			.620	.633	.644	.635	.659	.365	441	.130	.647			.599	.599	.596	.600	.610	.623
402	.110	.681			.856	.868	.889	.736	.676	.587	442	.130	.682			.926	.925	.914	.903	.896	.895
403	.110	.698			.786	.793	.790	.680	.646	.548	443	.130	.700			.793	.795	.792	.791	.795	.796
404	.110	.748			.458	.477	.494	.468	.310	.122	444	.130	.748			.624	.618	.604	.594	.595	.598
406	.110	.949			.227	.244	.260	.247	.071	-.107	445	.130	.832			.511	.500	.478	.461	.459	.459
410	.250	.000			.558	.542	.539	.580	.758	.712	450	.300	.000			.727	.751	.797	.849	.905	.933
411	.250	.200			.281	.277	.263	.206	.045	-.109	451	.300	.200			.254	.255	.262	.273	.294	.334
412	.250	.400			.235	.234	.223	.185	.065	-.077	452	.300	.400			.217	.221	.233	.252	.290	.329
413	.250	.600			.466	.468	.462	.419	.352	.230	453	.300	.600			.449	.450	.457	.469	.492	.513
414	.250	.630			.802	.804	.808	.764	.714	.540	454	.300	.633			.519	.519	.526	.535	.554	.566
415	.250	.668			.708	.706	.699	.683	.653	.494	455	.300	.668			.840	.847	.850	.854	.871	.893
416	.250	.685			.485	.486	.486	.506	.444	.337	456	.300	.686			.700	.707	.716	.727	.744	.758
417	.250	.738			.352	.355	.358	.384	.312	.192	457	.300	.738			.498	.500	.504	.515	.538	.556
418	.250	.825			.157	.159	.167	.186	.126	-.005	458	.300	.825			.375	.375	.376	.386	.409	.430
419	.250	.947									459	.300	.947			.178	.175	.178	.186	.208	.227
420	.500	.000			.566	.530	.472	.468	.442	.112	459	.300	.947								
421	.500	.100			.330	.325	.306	.257	.129	-.029	460	.600	.000			.661	.695	.765	.844	.916	.953
422	.500	.200			.254	.250	.234	.192	.081	-.057	461	.600	.200			.289	.298	.312	.321	.336	.374
423	.500	.300			.214	.210	.193	.158	.062	-.065	462	.600	.300			.201	.207	.225	.246	.275	.317
424	.500	.400			.203	.199	.185	.154	.071	-.050	463	.600	.400			.175	.185	.204	.224	.265	.305
425	.500	.500			.208	.204	.193	.165	.096	-.008	464	.600	.500			.173	.181	.199	.226	.265	.303
426	.500	.600			.174	.167	.155	.139	.090	-.004	465	.600	.600			.184	.193	.210	.236	.276	.312
427	.500	.700			-.022	-.031	-.043	-.042	-.080	-.143	466	.600	.700			.150	.158	.176	.205	.247	.279
428	.500	.800			-.195	-.204	-.211	-.204	-.264	-.402	467	.600	.800			-.039	-.031	-.008	.022	.065	.104
429	.500	.900			-.287	-.293	-.298	-.301	-.341	-.399	468	.600	.900			-.200	-.197	-.181	-.158	-.119	-.085
430	.830	.000			.592	.556	.490	.458	.428	.391	469	.600	.900			-.282	-.282	-.269	-.255	-.226	-.198
431	.830	.200			.164	.147	.120	.095	.036	-.041	470	.800	.000			.646	.691	.759	.832	.905	.948
432	.830	.400			.673	.594	.528	.441	.392	-.382	471	.800	.200			.137	.149	.176	.214	.258	.305
433	.830	.600			.037	.021	.001	-.010	-.033	-.084	472	.800	.400			.112	.123	.151	.187	.234	.280
434	.830	.800			-.094	-.109	-.126	-.139	-.152	-.192	473	.800	.600			.045	.057	.085	.122	.175	.220
											474	.800	.800			-.146	-.138	-.112	-.077	-.028	.018

TABLE V.- PRESSURE COEFFICIENTS MEASURED ON VERTICAL TAIL AND SPEED BRAKES - Continued

(b) $M = 0.90$ Pressure coefficients C_p on -

Model orifice number	y b/2	x c	Upper tail								Model orifice number	y b/2	x c	Lower tail								
			α = -20°	α = -10°	α = -5°	α = 0°	α = 5°	α = 10°	α = 15°	α = 20°				α = -20°	α = -10°	α = -5°	α = 0°	α = 5°	α = 10°	α = 15°	α = 20°	
			β = -5°; δ ₀ = 0°; δ _v = 0°; brakes closed																			
401	.110	.647				-.274	-.256	-.286	-.339	-.402	441	.130	.647				-.273	-.238	-.183	-.130	-.081	
402	.110	.681				-.149	-.187	-.165	-.197	-.270	442	.130	.682				-.130	-.089	-.027	-.028	-.080	
403	.110	.698				-.153	-.189	-.167	-.205	-.262	443	.130	.700				-.128	-.083	-.022	-.029	-.078	
404	.110	.748				-.193	-.223	-.210	-.252	-.307	444	.130	.748				-.172	-.128	-.066	-.010	-.037	
405	.110	.832				-.221	-.254	-.251	-.293	-.353	445	.130	.832				-.209	-.174	-.113	-.062	-.021	
406	.110	.949				-.246	-.278	-.224	-.274	-.373	446	.130	.949				-.224	-.217	-.193	-.168	-.150	
410	.250	.000				.899	.878	.735	.835	.645	450	.300	.000				.679	.722	.778	.826	.854	
411	.250	.200				-.081	-.119	-.208	-.266	-.390	451	.300	.200				-.039	-.018	-.015	-.004	.035	
412	.250	.400				-.063	-.093	-.140	-.143	.451	452	.300	.400				-.029	.002	.029	.050	.096	
413	.250	.600				-.143	-.168	-.190	-.187	-.274	453	.300	.600				-.125	-.089	-.043	-.001	.048	
414	.250	.830				-.263	-.282	-.317	-.342	-.416	454	.300	.830				-.250	-.207	-.156	-.115	-.076	
415	.250	.668				-.125	-.167	-.211	-.168	-.267	455	.300	.668				-.125	-.085	-.026	.028	.078	
416	.250	.685				-.136	-.173	-.189	-.174	-.255	456	.300	.686				-.120	-.083	-.026	.024	.072	
417	.250	.738				-.172	-.209	-.227	-.219	-.286	457	.300	.738				-.149	-.110	-.051	-.002	.042	
418	.250	.825				-.198	-.232	-.260	-.244	-.320	458	.300	.825				-.172	-.139	-.085	-.038	.003	
419	.250	.947				-.239	-.267	-.255	-.263	-.373	459	.300	.947				-.210	-.203	-.174	-.147	-.120	
420	.500	.000				.808	.754	.702	.468	.688	460	.600	.000				.653	.719	.793	.855	.888	
421	.500	.100				-.423	-.495	-.432	-.459	-.432	461	.600	.100				-.390	-.366	-.309	-.122	-.221	
422	.500	.200				-.059	-.064	-.137	-.403	-.479	462	.600	.200				-.017	.020	.019	.030	.069	
423	.500	.300				-.034	-.074	-.121	-.272	-.356	463	.600	.300				-.009	.026	.039	.049	.088	
424	.500	.400				-.049	-.084	-.110	-.243	-.464	464	.600	.400				-.037	-.006	.022	.046	.085	
425	.500	.500				-.081	-.111	-.134	-.250	-.243	465	.600	.500				-.072	-.034	.004	.039	.080	
426	.500	.600				-.109	-.141	-.160	-.252	-.271	466	.600	.600				-.112	-.070	.025	.014	.057	
427	.500	.700									467	.600	.700				-.127	-.080	.038	.004	.044	
428	.500	.800				-.171	-.201	-.226	-.277	-.295	468	.600	.800				-.158	-.123	-.074	-.029	.005	
429	.500	.900				-.195	-.219	-.237	-.286	-.318	469	.600	.900				-.181	-.160	-.122	-.085	-.053	
430	.830	.000				.772	.711	.669	.651	.719	470	.800	.000				.720	.793	.867	.926	.971	
431	.830	.200				-.137	-.262	-.283	-.283	-.197	471	.800	.200				-.111	-.018	.006	.038	.069	
432	.830	.400				-.074	-.075	-.096	-.145	-.177	472	.800	.400				-.041	-.029	.006	.030	.074	
433	.830	.600				-.110	-.118	-.123	-.151	-.205	473	.800	.600				-.101	-.074	-.025	.015	.061	
434	.830	.800				-.152	-.163	-.165	-.185	-.222	474	.800	.800				-.145	-.120	-.071	-.026	.007	
β = -5°; δ ₀ = 0°; δ _v = 0°; brakes open																						
401	.110	.647	-.181	-.227	.588	.560	.375	.629	.504	.472	441	.130	.647				.486	.579	.620	.625	.619	.625
402	.110	.681	-.084	-.124	.829	.840	.483	.907	.815	.804	442	.130	.682	-.168	-.108	.819	.929	.924	.898	.865	.859	
403	.110	.698	-.087	-.130	.827	.820	.480	.847	.882	.783	443	.130	.700	-.178	-.121	.720	.859	.871	.842	.803	.789	
404	.110	.748	-.124	-.174	.567	.567	.461	.702	.619	.462	444	.130	.748	-.209	-.151	.558	.641	.660	.656	.643	.638	
405	.110	.832	-.159	-.206	.458	.441	.379	.553	.433	.248	445	.130	.832	-.227	-.180	.483	.508	.508	.507	.503	.501	
406	.110	.949	-.244	-.273	.282	.270	.217	.334	.242	.065	446	.130	.949	-.255	-.205	.337	.317	.307	.301	.296	.295	
410	.250	.000	1.031	.967	.908	.893	.846	.749	.834	.601	450	.300	.000	.409	.576	.647	.684	.724	.783	.821	.852	
411	.250	.200	.073	-.060	.017	.013	-.024	-.024	-.108	-.410	451	.300	.200	-.462	-.133	.049	.057	.058	.066	.068	.109	
412	.250	.400	.071	-.052	.143	.131	.099	.057	.002	-.123	452	.300	.400	-.265	-.087	.156	.173	.198	.212	.226	.264	
413	.250	.600	.106	-.127	.451	.443	.327	.441	.458	.327	453	.300	.600	-.286	-.157	.412	.460	.467	.469	.483	.509	
414	.250	.830	.217	-.245	.533	.518	.374	.544	.472	.415	454	.300	.830	-.384	-.230	.522	.543	.541	.527	.532	.555	
415	.250	.668	.059	-.097	.757	.742	.527	.813	.829	.720	455	.300	.668	-.326	-.171	.872	.854	.805	.738	.701	.728	
416	.250	.685	.073	-.109	.706	.732	.487	.704	.635	.456	456	.300	.686	-.257	-.159	.780	.780	.717	.681	.659	.684	
417	.250	.738	.102	-.145	.556	.547	.361	.517	.571	.406	457	.300	.738	-.267	-.183	.560	.588	.582	.571	.560	.570	
418	.250	.825	.132	-.179	.418	.397	.254	.432	.421	.277	458	.300	.825	-.274	-.202	.433	.440	.457	.465	.466	.467	
419	.250	.947	.218	-.257	.236	.226	.128	.269	.248	.135	459	.300	.947	-.315	-.247	.261	.248	.256	.267	.278	.286	
420	.500	.000	1.012	.922	.840	.803	.739	.691	.443	.756	460	.600	.000	.732	.524	.595	.648	.714	.790	.843	.880	
421	.500	.100	.088	-.298	-.348	-.363	-.399	-.183	-.449	-.442	461	.600	.100	-.404	-.559	-.210	-.236	-.268	-.262	-.238	-.184	
422	.500	.200	.032	-.006	.060	.054	.025	-.041	-.291	-.474	462	.600	.200	-.317	-.052	.068	.090	.111	.111	.113	.153	
423	.500	.300	.057	-.030	.113	.096	.055	.022	-.172	-.244	463	.600	.300	-.158	-.059	.113	.138	.171	.180	.181	.216	
424	.500	.400	.065	-.050	.153	.137	.097	.036	-.078	-.098	464	.600	.400	-.188	-.067	.134	.165	.199	.208	.216	.249	
425	.500	.500	.070	-.073	.210	.187	.152	.166	.024	-.005	465	.600	.500	-.200	-.088	.173	.212	.246	.256	.265	.296	
426	.500	.600	.078	-.096	.218	.198	.143	.166	.064	.038	466	.600	.600	1.178	1.184	.184	.216	.247	.261	.270	.298	
427	.500	.700	.092	-.118							467	.600	.700	-.204	-.129	.008	.031	.051	.081	.105	.130	
428	.500	.800	.120	-.154	-.187	-.203	-.227	-.186	-.212	-.326	468	.600	.800	-.204	-.157	.190	.174	-.154	-.123	-.103	-.087	
429	.500	.900	.155	-.191	-.315	-.332	-.347	-.279	-.297	-.436	469	.600	.900			.318	-.309	-.284	-.257	-.251	-.244	
430	.830	.000	1.024	.913	.821	.782	.712	.669	.640	.733	470	.800	.000	.756	.594	.674	.729	.805	.873	.925	.969	
431	.830	.200	.025	-.005	-.033	-.113	-.198	-.172	-.111	-.232	471	.800	.200	-.152	-.196	-.018	-.008	.010	.009	.118	.160	
432	.830	.400	.073	-.069	.101	.087	.072	.069	.013	-.076	472	.800	.400	-.163	-.093	.143	.152	.163	.178	.192	.225	
433	.830	.600	.057	-.091	.074	.051	.047	.041	-.002	-.055	473	.800	.600	-.171	-.099	.118	.139	.155	.179	.200	.229	
434	.830	.800	.099	-.133	-.081	-.095	-.124	-.136	-.184	-.200	474	.800	.800			.136	-.106	-.071	-.047	-.025	.001	

TABLE V.- PRESSURE COEFFICIENTS MEASURED ON VERTICAL TAIL AND SPEED BRAKES - Continued

(b) $M = 0.90$ - ContinuedPressure coefficients C_p on -

Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Upper tail								Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Lower tail							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta = 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$																		
			$\beta = 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$																		
401	.110	.647	-.139	-.199		-.207	-.228	-.255	-.283	441	.130	.647	-.351	-.249		-.242	-.213	-.165	-.108		
402	.110	.681	-.014	-.080		-.090	-.113	-.149	-.187	442	.130	.682	-.215	-.095		-.106	-.061	.002	.070		
403	.110	.698	-.019	-.087		-.093	-.116	-.147	-.180	443	.130	.700	-.181	-.091		-.106	-.060	.001	.065		
404	.110	.748	-.060	-.131		-.138	-.159	-.193	-.221	445	.130	.748	-.223	-.128		-.149	-.106	-.042	.022		
405	.110	.832	-.101	-.160		-.163	-.180	-.210	-.234	446	.130	.832	-.244	-.160		-.190	-.152	-.093	-.036		
406	.110	.949	-.197	-.216		-.197	-.202	-.224	-.252	447	.130	.949	-.282	-.187		-.190	-.184	-.165	-.140		
410	.250	.000	1.026	.992		.970	.978	1.024	1.021	450	.300	.000				.898	.929	.946	.965		
411	.250	.200	.019	.051		.095	.087	.038	-.098	451	.300	.200	-.255	.041		.090	.115	.112	.108		
412	.250	.400	-.001	-.005		.017	.003	-.030	-.124	452	.300	.400	-.216	-.006		.006	.040	.068	.103		
413	.250	.600	-.043	-.085		-.085	-.101	-.122	-.167	453	.300	.600	-.223	-.095		-.103	-.067	-.018	.036		
414	.250	.630	-.185	-.224		-.222	-.243	-.275	-.316	454	.300	.633	-.343	-.236		-.230	-.199	-.156	-.112		
415	.250	.668	.013	-.049		-.063	-.081	-.099	-.125	455	.300	.668	-.195	-.077		-.230	-.199	-.156	-.112		
416	.250	.685	-.004	-.064		-.078	-.095	-.117	-.145	456	.300	.686	-.188	-.074		-.098	-.053	.007	.069		
417	.250	.738	-.037	-.102		-.116	-.136	-.155	-.178	457	.300	.738	-.199	-.105		-.130	-.085	-.023	.038		
418	.250	.825	-.072	-.132		-.143	-.156	-.174	-.196	458	.300	.825	-.211	-.123		-.159	-.120	-.061	-.006		
419	.250	.947	-.170	-.196		-.185	-.187	-.200	-.239	459	.300	.947	-.247	-.198		-.178	-.167	-.139	-.108		
420	.500	.000	1.020	.955		.866	.840	.836	.849	460	.600	.000	.864	.840		.882	.926	.960	.992		
421	.500	.100	.077	.079		.108	.098	.054	-.045	461	.600	.100	-.198	.037		.102	.131	.125	.119		
422	.500	.200	.019	.033		.065	.053	.013	-.086	462	.600	.200	-.252	.017		.056	.086	.095	.104		
423	.500	.300	.006	.012		.031	.019	-.013	-.097	463	.600	.300	-.163	-.020		.024	.057	.079	.103		
424	.500	.400	.006	-.004		-.001	-.014	-.040	-.103	464	.600	.400	-.182	-.020		-.014	.023	.058	.096		
425	.500	.500	.002	-.028		-.033	-.049	-.071	-.114	465	.600	.500	-.173	-.047		-.047	.007	.039	.087		
426	.500	.600	-.008	-.050		-.059	-.078	-.100	-.125	466	.600	.600				-.086	-.043	.008	.062		
427	.500	.700	-.020	-.069		-.089	-.107	-.126	-.135	467	.600	.700	-.171	-.087		-.103	-.059	-.003	.052		
428	.500	.800	-.051	-.104		-.117	-.132	-.145	-.156	468	.600	.800	-.188	-.111		-.134	-.094	-.038	.015		
429	.500	.900	-.098	-.137		-.143	-.151	-.159	-.171	469	.600	.900	-.203	-.132		-.152	-.128	-.083	-.041		
430	.830	.000	1.035	.946		.832	.784	.756	.745	470	.800	.000	.839	.815		.865	.917	.961	1.004		
431	.830	.200	.042	.029						471	.800	.200				.037	.071	.090	.108		
432	.830	.400	.019	-.006		-.024	-.048	-.069	-.104	472	.800	.400	-.155	-.037		-.023	.017	.057	.098		
433	.830	.600	.018	-.026		-.056	-.079	-.095	-.102	473	.800	.600	-.159	-.077		-.080	-.033	.019	.074		
434	.830	.800	-.022	-.069		-.099	-.119	-.125	-.120	474	.800	.800	-.163	-.110		-.126	-.084	-.028	.024		
$\beta = 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$																					
401	.110	.647	-.209	-.203		-.214	-.252	-.311	-.414	441	.130	.647	-.251	-.255		-.245	-.203	-.154	-.112		
402	.110	.681	-.060	-.057		-.070	-.114	-.180	-.306	442	.130	.682	-.086	-.076		-.050	-.001	.047	.091		
403	.110	.698	-.068	-.064		-.077	-.115	-.167	-.297	443	.130	.700	-.087	-.078		-.052	-.005	.041	.083		
404	.110	.748	-.122	-.119		-.133	-.170	-.217	-.342	444	.130	.748	-.138	-.132		-.107	-.056	-.008	.030		
405	.110	.832	-.170	-.161		-.171	-.207	-.252	-.401	445	.130	.832	-.190	-.184		-.164	-.115	-.070	-.039		
406	.110	.949	-.206	-.200		-.202	-.216	-.239	-.454	446	.130	.949	-.196	-.197		-.195	-.178	-.162	-.149		
410	.250	.000	.986	.975		.970	1.024	1.013	.983	450	.300	.000				.907	.918	.943	.964		
411	.250	.200	.212	.213		.209	.144	.019	-.142	451	.300	.200				.185	.191	.200	.184		
412	.250	.400	.073	.075		.069	.028	-.121	-.280	452	.300	.400				.056	.059	.073	.087		
413	.250	.600	-.070	-.066		-.071	-.103	-.194	-.323	453	.300	.600				-.088	-.085	-.030	.016		
414	.250	.630	-.258	-.253		-.257	-.278	-.348	-.437	454	.300	.633				-.243	-.246	-.236	-.202		
415	.250	.668	-.061	-.053		-.062	-.111	-.206	-.406	455	.300	.668				-.098	-.092	-.071	-.032		
416	.250	.685	-.066	-.061		-.070	-.113	-.203	-.436	456	.300	.686				-.093	-.086	-.062	-.020		
417	.250	.738	-.120	-.113		-.123	-.165	-.249	-.468	457	.300	.738				-.128	-.122	-.100	-.058		
418	.250	.825	-.164	-.153		-.161	-.202	-.292	-.451	458	.300	.825				-.166	-.161	-.141	-.099		
419	.250	.947	-.201	-.190		-.191	-.210	-.316	-.520	459	.300	.947				-.182	-.183	-.179	-.159		
420	.500	.000	.886	.862		.828	.843	.879	.869	460	.600	.000				.890	.909	.947	.987		
421	.500	.100	.332	.325		.316	.277	.191	.102	461	.600	.100				.277	.288	.305	.295		
422	.500	.200	.212	.216		.211	.175	.061	-.078	462	.600	.200				.153	.159	.175	.176		
423	.500	.300	.139	.137		.136	.103	-.003	-.168	463	.600	.300				.101	.106	.124	.132		
424	.500	.400	.089	.090		.086	.062	-.025	-.144	464	.600	.400				.056	.063	.084	.104		
425	.500	.500	.053	.050		.046	.027	-.036	-.169	465	.600	.500				.020	.028	.052	.084		
426	.500	.600	-.019	.020		-.012	-.003	-.048	-.157	466	.600	.600				-.029	-.016	.011	.051		
427	.500	.700	-.016	.016		-.016	-.026	-.064	-.173	467	.600	.700				-.043	-.032	.013	.032		
428	.500	.800	-.056	.054		-.059	-.070	-.095	-.192	468	.600	.800				-.089	-.080	-.051	.003		
429	.500	.900	-.089	-.084		-.086	-.090	-.112	-.213	469	.600	.900				-.114	-.108	-.088	-.047		
430	.830	.000		.761	.719		.661	.642	.649	466	.800	.000				.827	.851	.899	.954		
431	.830	.200		.052	.038		.021	.002	-.034	470	.800	.200				.111	.118	.145	.159		
432	.830	.400		.031	.020		.010	.001	-.030	471	.800	.400				.026	.038	.056	.071		
433	.830	.600		.014	.002		-.020	-.029	-.039	472	.800	.600				-.042	-.027	.009	.062		
434	.830	.800		-.039	-.050		-.067	-.069	-.059	473	.800	.800				-.099	-.084	-.046	.011		
$\beta = 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$																					
401	.110	.647	-.209	-.203		-.214	-.252	-.311	-.414	441	.130	.647	-.251	-.255		-.245	-.203	-.154	-.112		
402	.110	.681	-.060	-.057		-.070	-.114	-.180	-.306	442	.130	.682	-.086	-.076		-.050	-.001	.047	.091		
403	.110	.698	-.068	-.064		-.077	-.115	-.167	-.297	443	.130	.700	-.087	-.078		-.052	-.005	.041	.083		
404	.110	.748	-.122	-.119		-.133	-.170	-.217	-.342	444	.130	.748	-.138	-.132		-.107	-.056	-.008	.030		
405	.110	.832	-.170	-.161		-.171	-.207	-.252	-.401	445	.130	.832	-.190	-.184		-.164	-.115	-.070	-.039		
406	.110	.949	-.206	-.200		-.202	-.216	-.239	-.454	446	.130	.949	-.196	-.197		-.195	-.178	-.162	-.149		
410	.250	.000	.986	.975		.970	1.024	1.013	.983	450	.300	.000				.907	.918	.943	.964		
411	.250	.200	.212	.213		.209	.144	.019	-.142	451	.300	.200				.185	.191	.200	.184		
412	.250	.400	.073	.075		.069	.028	-.121	-.280	452	.300	.400				.056	.059	.073	.087		
413	.25																				

TABLE V.- PRESSURE COEFFICIENTS MEASURED ON VERTICAL TAIL AND SPEED BRAKES - Continued

(b) M = 0.90 - Continued

Pressure coefficients C_p on -

Model number	Upper tail										Model number	Lower tail											
	Y b/2	X c	α = -20°	α = -10°	α = -5°	α = 0°	α = 5°	α = 10°	α = 15°	α = 20°		Y b/2	X c	α = -20°	α = -10°	α = -5°	α = 0°	α = 5°	α = 10°	α = 15°	α = 20°		
β = 0°; θ ₀ = 0°; θ _y = -7.5°; brakes closed																							
401	110	647				-260	-271	-294	-301	-316	-402	441	130	647				-275	-272	-246	-181	-121	-066
402	110	681				-132	-143	-173	-204	-254	-312	442	130	682				-164	-159	-106	-033	-025	-073
403	110	698				-139	-150	-177	-207	-242	-317	443	130	700				-166	-142	-109	-036	-019	-066
404	110	748				-182	-190	-222	-253	-279	-373	444	130	748				-205	-195	-155	-077	-018	-031
405	110	832				-191	-198	-224	-252	-277	-383	445	130	832				-233	-226	-188	-114	-060	-011
406	110	949				-186	-186	-194	-222	-231	-327	446	130	949				-188	-187	-179	-157	-134	-111
410	250	000				1.002	0.993	0.995	1.042	1.043	1.001	450	300	000				0.887	0.899	0.925	0.950	0.969	0.903
411	250	200				-0.072	-0.079	-0.091	-1.042	-1.051	-0.436	451	300	200				-0.082	-0.064	-0.038	-0.041	-0.085	-0.122
412	250	400				-0.049	-0.062	-0.084	-1.03	-1.07	-0.150	452	300	400				-0.078	-0.060	-0.021	-0.022	-0.100	-0.169
413	250	600				-0.130	-0.137	-0.158	-1.03	-1.07	-0.173	453	300	600				-0.157	-0.146	-0.127	-0.126	-0.166	-0.233
414	250	630				-0.258	-0.289	-0.309	-0.293	-0.330	-0.398	454	300	633				-0.268	-0.264	-0.218	-0.116	-0.126	-0.091
415	250	668				-0.139	-0.149	-0.168	-0.167	-0.186	-0.270	455	300	668				-0.190	-0.173	-0.128	-0.036	-0.028	-0.082
416	250	738				-0.152	-0.157	-0.176	-0.185	-0.208	-0.287	456	300	738				-0.171	-0.160	-0.126	-0.039	-0.025	-0.074
417	250	738				-0.178	-0.183	-0.206	-0.219	-0.235	-0.325	457	300	738				-1.198	-1.188	-1.153	-0.665	-0.005	-0.044
418	250	825				-0.180	-0.189	-0.212	-0.218	-0.233	-0.328	458	300	825				-0.205	-0.199	-0.160	-0.083	-0.023	-0.073
419	250	947				-0.162	-0.169	-0.177	-0.184	-0.179	-0.275	459	300	947				-0.174	-0.168	-0.148	-0.103	-0.057	-0.042
420	500	000				0.890	0.868	0.838	0.844	0.876	0.868	460	600	000				0.701	0.727	0.769	0.808	0.840	0.885
421	500	100				-0.178	-0.178	-0.158	-0.438	-0.465	-0.465	461	600	100				-1.108	-1.100	-1.077	-0.893	-0.870	-0.867
422	500	200				-0.048	-0.061	-0.085	-0.126	-0.278	-0.476	462	600	200				-0.042	-0.026	-0.012	-0.039	-0.057	-0.082
423	500	300				-0.221	-0.235	-0.263	-0.082	-0.121	-0.205	463	600	300				-0.044	-0.027	-0.014	-0.053	-0.102	-0.149
424	500	400				-0.035	-0.055	-0.083	-0.101	-0.138	-0.153	464	600	400				-0.076	-0.060	-0.017	-0.042	-0.091	-0.132
425	500	500				-0.070	-0.089	-0.119	-0.130	-0.158	-0.221	465	600	500				-0.110	-0.092	-0.050	-0.016	-0.069	-0.110
426	500	600				-0.115	-0.135	-0.162	-0.195	-0.252	-0.400	466	600	600				-0.160	-0.143	-0.093	-0.033	-0.017	-0.059
427	500	700				-0.163	-0.178	-0.202	-0.217	-0.234	-0.305	467	600	700				-0.185	-0.175	-0.137	-0.072	-0.035	-0.085
428	500	800				-0.180	-0.191	-0.215	-0.223	-0.240	-0.312	468	600	800				-0.195	-0.187	-0.155	-0.094	-0.048	-0.106
429	500	900				-0.156	-0.162	-0.179	-0.183	-0.196	-0.280	469	600	900				-0.179	-0.171	-0.145	-0.096	-0.063	-0.042
430	830	000				0.827	0.791	0.736	0.716	0.713	0.734	470	800	000				0.711	0.745	0.802	0.860	0.903	0.941
431	830	200										471	800	200				0.023	0.037	0.036	-0.025	-0.048	-0.048
432	830	400				-0.063	-0.074	-0.092	-0.107	-0.143	-0.175	472	800	400				-0.045	-0.056	-0.013	-0.026	-0.053	-0.092
433	830	600				-0.105	-0.117	-0.135	-0.146	-0.165	-0.215	473	800	600				-0.073	-0.078	-0.084	-0.035	-0.001	-0.043
434	830	800				-0.147	-0.153	-0.164	-0.173	-0.184	-0.245	474	800	800				-0.178	-0.170	-0.135	-0.094	-0.065	-0.054
β = 0°; θ ₀ = 0°; θ _y = 7.5°; brakes closed																							
401	110	647				0.608	0.613	0.612	0.568	0.493	0.302	441	130	647				0.638	0.643	0.650	0.654	0.647	0.655
402	110	681				0.857	0.864	0.859	0.803	0.737	0.582	442	130	682				0.976	0.976	0.954	0.926	0.889	0.891
403	110	698				0.873	0.883	0.876	0.812	0.691	0.722	443	130	700				0.876	0.890	0.918	0.919	0.886	0.886
404	110	748				0.653	0.646	0.594	0.507	0.435	0.454	444	130	748				0.679	0.689	0.699	0.695	0.703	0.705
405	110	832				0.514	0.522	0.465	0.385	0.325	0.182	445	130	832				0.576	0.572	0.542	0.534	0.534	0.535
406	110	949				0.336	0.358	0.317	0.269	0.231	0.066	446	130	949				0.391	0.375	0.346	0.341	0.338	0.338
410	250	000				0.986	0.987	0.994	1.033	1.045	0.982	450	300	000				0.909	0.922	0.944	0.964	0.984	1.015
411	250	200				0.188	0.187	0.167	0.106	-0.065	-0.220	451	300	200				0.184	0.194	0.206	0.205	0.204	0.247
412	250	400				0.212	0.212	0.188	0.141	0.010	-0.159	452	300	400				0.212	0.222	0.239	0.259	0.281	0.315
413	250	600				0.538	0.533	0.515	0.473	0.407	0.316	453	300	600				0.516	0.525	0.540	0.560	0.569	0.585
414	250	630				0.552	0.546	0.529	0.533	0.534	0.457	454	300	633				0.603	0.599	0.590	0.593	0.587	0.631
415	250	668				0.847	0.863	0.888	0.903	0.888	0.749	455	300	668				0.941	0.930	0.908	0.887	0.880	0.941
416	250	685				0.854	0.850	0.831	0.806	0.786	0.801	456	300	686				0.855	0.862	0.871	0.877	0.884	0.884
417	250	738				0.593	0.593	0.578	0.540	0.552	0.491	457	300	738				0.608	0.611	0.615	0.632	0.654	0.662
418	250	825				0.436	0.444	0.435	0.404	0.427	0.287	458	300	825				0.471	0.467	0.460	0.473	0.490	0.506
419	250	947				0.267	0.275	0.268	0.256	0.280	0.146	459	300	947				0.282	0.277	0.274	0.287	0.303	0.319
420	500	000				0.910	0.890	0.855	0.851	0.856	0.837	460	600	000				0.885	0.909	0.945	0.979	1.012	1.048
421	500	100				0.194	0.188	0.165	0.113	-0.016	-0.160	461	600	100				0.174	0.189	0.207	0.204	0.199	0.243
422	500	200				0.172	0.169	0.145	0.095	-0.041	-0.229	462	600	200				0.160	0.172	0.190	0.198	0.206	0.247
423	500	300				0.174	0.174	0.145	0.102	-0.021	-0.179	463	600	300				0.170	0.181	0.199	0.216	0.233	0.269
424	500	400				0.204	0.200	0.178	0.137	0.034	-0.117	464	600	400				0.205	0.206	0.226	0.247	0.270	0.305
425	500	500				0.256	0.250	0.229	0.194	0.111	-0.001	465	600	500				0.247	0.250	0.276	0.301	0.323	0.350
426	500	600				0.262	0.257	0.235	0.214	0.176	0.091	466	600	600				0.256	0.266	0.285	0.306	0.330	0.358
427	500	700				0.042	0.036	0.011	0.002	-0.005	-0.067	467	600	700				0.040	0.051	0.073	0.107	0.139	0.168
428	500	800				0.176	0.176	0.164	0.154	0.119	-0.289	468	600	800				-0.176	-0.170	-0.150	-0.113	-0.083	-0.057
429	500	900				-0.307	-0.309	-0.325	-0.326	-0.338	-0.415	469	600	900				-0.311	-0.308	-0.293	-0.261	-0.236	-0.217
430	830	000				0.880	0.852	0.800	0.773	0.746	0.736	470	800	000				0.866	0.892	0.935	0.978	1.021	1.061
431	830	200										471	800	200				0.146	0.160	0.181	0.196	0.209	0.250
432	830	400				0.151	0.140	0.109	0.078	0.009	-0.084	472	800	400				0.169	0.183	0.206	0.229	0.254	0.289
433	830	600				0.116	0.105	0.074	0.052	0.027	-0.027	473	800	600				0.184	0.195	0.219	0.252	0.281	0.319
434	830	800				-0.061	-0.077	-0.113	-0.125	-0.126	-0.162	474	800	800				-0.119	-0.103	-0.074	-0.027	0.014	0.049

TABLE V.- PRESSURE COEFFICIENTS MEASURED ON VERTICAL TAIL AND SPEED BRAKES - Continued

(b) $M = 0.90$ - ConcludedPressure coefficients C_p on -

Model orifice number	y b/2	x c	Upper tail								Model orifice number	y b/2	x c	Lower tail												
			α = -20°	α = -10°	α = -5°	α = 0°	α = 5°	α = 10°	α = 15°	α = 20°				α = -20°	α = -10°	α = -5°	α = 0°	α = 5°	α = 10°	α = 15°	α = 20°					
			β = 5°; δ _e = 0°; δ _v = 0°; brakes closed																							
401	.110	.647	-.089	-.155		-.171	-.186	-.207	-.304	-.411	441	.130	.647	-.372	-.223		-.237	-.211	-.141	-.089	-.035					
402	.110	.681	.062	.004		-.004	-.033	-.081	-.172	-.281	442	.130	.682	-.272	-.062		-.047	-.011	.068	.124	.174					
403	.110	.698	.056	-.003		-.011	-.034	-.083	-.175	-.277	443	.130	.700	-.250	-.056		-.047	-.011	.064	.117	.167					
404	.110	.748	.013	-.049							444	.130	.748	-.277	-.094		-.098	-.062	.014	.066	.115					
405	.110	.832	-.040	-.089		-.093	-.112	-.149	-.235	-.351	445	.130	.832	-.298	-.123		-.139	-.107	-.040	.004	.047					
406	.110	.949	-.156	-.171		-.148	-.155	-.191	-.270	-.401	446	.130	.949	-.334	-.161		-.156	-.158	-.134	-.120	-.100					
β = 5°; δ _e = 0°; δ _v = 0°; brakes closed																										
410	.250	.000	.989	.947		.840	.821	.764	.874	.989	450	.300	.000	.235	.811		.902	.937	.974	1.004	1.038					
411	.250	.200	.179	.194		.271	.262	.162	-.021	-.295	451	.300	.200	-.244	.172		.233	.239	.227	.225	.277					
412	.250	.400	.103	.086		.120	.108	.043	-.105	-.286	452	.300	.400	-.244	.050		.082	.104	.143	.177	.231					
413	.250	.600	.036	-.013		-.017	-.035	-.075	-.166	-.290	453	.300	.600	-.252	-.062		-.060	.027	.040	.090	.142					
414	.250	.630	-.145	-.193							454	.300	.633	-.366	-.219		-.227	-.198	-.135	-.089	-.039					
415	.250	.668	.090	.032		.009	-.014	-.050	-.128	-.245	455	.300	.668	-.220	-.052		-.050	-.008	.071	.127	.179					
416	.250	.685	.072	.014		-.003	-.027	-.062	-.142	-.252	456	.300	.686	-.234	-.048		-.046	.008	.067	.121	.171					
417	.250	.738	.034	-.025		-.046	-.069	-.098	-.173	-.291	457	.300	.738	-.248	-.076		-.083	-.043	.031	.084	.134					
418	.250	.825	-.010	-.061		-.074	-.095	-.118	-.192	-.310	458	.300	.825	-.256	-.093		-.108	-.073	-.005	.038	.082					
419	.250	.947	-.122	-.146		-.135	-.143	-.164	-.236	-.348	459	.300	.947	-.295	-.143		-.139	-.135	-.101	-.081	-.054					
β = 5°; δ _e = 0°; δ _v = 0°; brakes open																										
420	.500	.000				.782	.731	.676	.389	.902	460	.600	.000	.738	.797		.854	.907	.966	1.011	1.055					
421	.500	.100	.275	.274		.327	.310	.250	.105	-.099	461	.600	.100	-.087	.227		.296	.304	.285	.285	.337					
422	.500	.200	.155	.153		.223	.210	.146	-.001	-.245	462	.600	.200	-.235	.107		.160	.180	.187	.206	.267					
423	.500	.300	.111	.102		.140	.125	.070	-.070	-.256	463	.600	.300	-.168	.054		.098	.126	.158	.185	.238					
424	.500	.400	.107	.078		.084	.065	.020	-.088	-.221	464	.600	.400	-.169	.019		.046	.079	.127	.167	.218					
425	.500	.500	.091	.052		.038	.015	-.019	-.100	-.210	465	.600	.500	-.167	-.018		.002	.039	.099	.148	.200					
426	.500	.600	.074	.025		.004	-.021	-.045	-.111	-.209	466	.600	.600	-.186	-.054		-.045	-.004	.066	.117	.168					
427	.500	.700				-.026	-.051	-.066	-.116	-.223	467	.600	.700	-.180	-.062		-.063	-.020	.051	.101	.148					
428	.500	.800	.017	-.034		-.061	-.081	-.091	-.138	-.243	468	.600	.800	-.200	-.085		-.094	-.055	.012	.059	.102					
429	.500	.900	-.039	-.077		-.091	-.105	-.111	-.155	-.238	469	.600	.900	-.214	-.106		-.114	-.091	-.040	-.004	.031					
430	.830	.000	1.009	.911		.688	.628	.590	.584	.741	470	.800	.000	.795	.750		.823	.885	.957	1.010	1.055					
431	.830	.200	.148	.119		.090	.072	.030	-.063	-.204	471	.800	.200	-.204	.042		.089	.123	.167	.197	.258					
432	.830	.400	.121	.066		.010	-.020	-.045	-.093	-.116	472	.800	.400	-.132	-.015		.011	.056	.114	.161	.219					
433	.830	.600	.107	.040		-.024	-.055	-.064	-.085	-.117	473	.800	.600	-.141	-.063		-.053	-.004	.070	.126	.180					
434	.830	.800	.060	-.009		-.073	-.099	-.095	-.102	-.137	474	.800	.800	-.155	-.096		-.101	-.056	.017	.068	.117					
β = 5°; δ _e = 0°; δ _v = 0°; brakes open																										
401	.110	.647				.714	.728	.747	.594	.457	.245	441	.130	.647			.733	.727	.724	.715	.722	.742				
402	.110	.681				.963	.980	.997	.823	.701	.439	442	.130	.682			1.055	1.046	1.031	1.009	.993	1.000				
403	.110	.698				.932	.933	.899	.814	.739	.450	443	.130	.700			.929	.928	.934	.945	.940	.942				
404	.110	.748										444	.130	.748			.746	.738	.728	.726	.722	.724				
405	.110	.832				.577	.595	.625	.553	.333	-.178	445	.130	.832			.635	.619	.600	.589	.579	.582				
406	.110	.949				.394	.408	.426	.366	.148	-.296	446	.130	.949			.426	.412	.398	.392	.383	.384				
410	.250	.000				.848	.828	.836	.764	.871	.703	450	.300	.000			.887	.903	.941	.982	1.010	1.044				
411	.250	.200				.351	.353	.346	.234	.014	-.281	451	.300	.200			.330	.325	.326	.313	.313	.357				
412	.250	.400				.305	.307	.295	.210	.014	-.230	452	.300	.400			.287	.288	.302	.325	.344	.382				
413	.250	.600				.599	.598	.583	.512	.407	.163	453	.300	.600			.575	.576	.582	.600	.612	.630				
414	.250	.630										454	.300	.633			.657	.653	.655	.663	.675	.693				
415	.250	.668				.908	.915	.952	.830	.751	.723	455	.300	.668			.978	.976	.976	.977	.980	1.011	.988			
416	.250	.685				.865	.859	.846	.794	.756	.750	456	.300	.686			.848	.853	.865	.882	.889	.898				
417	.250	.738				.615	.614	.620	.612	.522	.272	457	.300	.738			.627	.626	.631	.649	.659	.673				
418	.250	.825				.470	.472	.487	.480	.366	.060	458	.300	.825			.496	.493	.495	.510	.522	.539				
419	.250	.947				.296	.296	.309	.316	.229	-.096	459	.300	.947			.309	.305	.310	.330	.341	.360				
420	.500	.000				.813	.775	.727	.682	.423	.892	460	.600	.000			.832	.860	.914	.975	1.016	1.059				
421	.500	.100				.398	.395	.382	.305	.133	-.130	461	.600	.100			.366	.369	.377	.361	.358	.401				
422	.500	.200				.319	.318	.307	.228	.045	-.228	462	.600	.200			.270	.269	.283	.288	.298	.349				
423	.500	.300				.279	.275	.262	.189	.008	-.219	463	.600	.300			.242	.247	.266	.287	.306	.348				
424	.500	.400				.281	.278	.262	.198	.048	-.166	464	.600	.400			.252	.259	.277	.305	.328	.365				
425	.500	.500				.314	.310	.294	.241	.122	-.113	465	.600	.500			.292	.298	.316	.345	.366	.398				
426	.500	.600				.298	.292	.279	.240	.158	-.103	466	.600	.600			.277	.284	.304	.338	.361	.392				
427	.500	.700				.063	.052	.040	.020	-.035	-.110	467	.600	.700			.039	.051	.078	.120	.155	.192				
428	.500	.800				-.164	-.173	-.179	-.183	-.241	-.319	468	.600	.800			-.167	-.167	-.144	-.104	-.071	-.035				
429	.500	.900				-.298	-.305	-.304	-.302	-.353	-.425	469	.600	.900			-.309	-.302	-.287	-.251	-.222	-.193				
430	.830	.000				.755	.708	.637	.601	.577	.754	470	.800	.000			.789	.824	.891	.964	1.013	1.061				
431	.830	.200				.215	.200	.183	.126	-.007	-.234	471	.800	.200			.203	.203	.205	.289	.339	.395				
432	.830	.400				.073	.051	.031	.014	-.010	-.014	472														

TABLE V.- PRESSURE COEFFICIENTS MEASURED ON VERTICAL TAIL AND SPEED BRAKES - Continued

(c) $M = 0.95$ Pressure coefficients C_p on -

Model orifice number	y b/2	x c	Upper tail								Model orifice number	y b/2	x c	Lower tail																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																														
			α = -20°	α = -10°	α = -5°	α = 0°	α = 5°	α = 10°	α = 15°	α = 20°				α = -20°	α = -10°	α = -5°	α = 0°	α = 5°	α = 10°	α = 15°	α = 20°																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																							
			β ≈ -5°; δ _E = 0°; δ _V = 0°; brakes closed																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																									
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401	.110	.647										.130	.647																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																															

TABLE V. - PRESSURE COEFFICIENTS MEASURED ON VERTICAL TAIL AND SPEED BRAKES - Continued

(c) $M = 0.95$ - ContinuedPressure coefficients C_p on -

Model orifice number	y b/2	x c	Upper tail								Model orifice number	y b/2	x c	Lower tail													
			α = -20°	α = -10°	α = -5°	α = 0°	α = 5°	α = 10°	α = 15°	α = 20°				α = -20°	α = -10°	α = -5°	α = 0°	α = 5°	α = 10°	α = 15°	α = 20°						
			β = 0°; δ _θ = 0°; δ _v = 0°; brakes closed																								
401	.110	.647	-.274	-.230		-.220	-.224	-.243	-.283	-.390	441	.130	.647	-.345	-.231		-.230	-.221	-.193	-.134	-.068						
402	.110	.681	-.129	-.061		-.052	-.065	-.099	-.148	-.242	442	.130	.682	-.234	-.065		-.075	-.038	.019	.077	.134						
403	.110	.698	-.122	-.068		-.064	-.072	-.103	-.139	-.236	443	.130	.700	-.297	-.056		-.070	-.035	.018	.075	.127						
404	.110	.748	-.160	-.129		-.126	-.138	-.172	-.210	-.375	444	.130	.748	-.295	-.111		-.124	-.091	.030	.030	.085						
405	.110	.832	-.113	-.180		-.165	-.183	-.222	-.272	-.427	445	.130	.832	-.325	-.169		-.184	-.147	-.081	-.021	-.029						
406	.110	.949	-.182	-.245		-.227	-.271	-.343	-.368	-.508	446	.130	.949	-.386	-.237		-.245	-.212	-.152	-.112	-.082						
410	.250	.000	1.052	1.014		1.013	1.020	1.014	.958	.875	450	.300	.000				.931	.954	.963	.983	1.020						
411	.250	.200	-.011	.055		.129	.138	.126	.061	.102	451	.300	.200	-.063	.133		.125	.137	.107	.044	.127						
412	.250	.400	-.102	.004		.047	.050	.036	-.036	-.075	452	.300	.400	.093	.059		.040	.061	.071	.070	.138						
413	.250	.600	-.143	-.076		-.061	-.065	-.076	-.121	-.261	453	.300	.600	-.228	-.053		-.074	-.050	.015	.024	.084						
414	.250	.630	-.289	-.246		-.237	-.239	-.247	-.287	-.438	454	.300	.633	-.329	-.213		-.229	-.212	-.181	-.136	-.079						
415	.250	.668	-.105	-.030		-.032	-.038	-.052	-.110	-.288	455	.300	.668	-.284	-.044		-.077	-.032	.025	.083	.140						
416	.250	.685	-.113	-.048		-.049	-.057	-.078	-.121	-.299	456	.300	.686	-.276	-.042		-.062	-.030	.021	.075	.131						
417	.250	.738	-.135	-.099		-.103	-.113	-.133	-.172	-.329	457	.300	.738	-.274	-.085		-.105	-.071	-.011	.047	.100						
418	.250	.825	-.093	-.146		-.143	-.160	-.181	-.226	-.369	458	.300	.825	-.291	-.128		-.150	-.113	-.049	.009	.059						
419	.250	.947	-.155	-.215		-.246	-.265	-.279	-.297	-.408	459	.300	.947	-.320	-.225		-.222	-.184	-.122	-.079	-.048						
420	.500	.000	1.040	.973		.897	.881	.860	.837	.850	460	.600	.000	.837	.860		.912	.950	.974	1.005	1.051						
421	.500	.100	.070	.080		.142	.149	.157	.106	.161	461	.600	.100	.112	.139		.137	.153	.114	.057	.144						
422	.500	.200	-.023	.036		.094	.101	.103	.052	.087	462	.600	.200	.050	.103		.088	.107	.092	.044	.127						
423	.500	.300	-.068	.026		.060	.066	.061	.006	.006	463	.600	.300	-.016	.072		.055	.078	.081	.059	.129						
424	.500	.400	-.096	.004		.028	.026	.022	-.033	-.071	464	.600	.400	-.080	.038		.017	.043	.062	.065	.130						
425	.500	.500	-.116	-.019		-.010	-.014	-.017	-.068	-.150	465	.600	.500	-.140	.000		-.020	.009	.042	.068	.127						
426	.500	.600	-.122	-.048		-.047	-.055	-.060	-.104	-.221	466	.600	.600	.000			-.074	-.039	.008	.053	.107						
427	.500	.700	-.127	-.063		-.074	-.085	-.096	-.128	-.275	467	.600	.700	-.229	-.067		-.066	-.046	.006	.056	.106						
428	.500	.800	-.108	-.135		-.120	-.134	-.151	-.180	-.296	468	.600	.800	-.251	-.114		-.129	-.089	-.027	.027	.073						
429	.500	.900	-.092	-.135		-.146	-.160	-.179	-.215	-.314	469	.600	.900	-.291	-.165		-.179	-.137	-.067	-.019	.019						
430	.830	.000	1.052	.959		.851	.820	.794	.771	.763	470	.800	.000	.818	.839		.899	.942	.978	1.017	1.065						
431	.830	.200	-.007	.034		-.005	-.011	-.011	-.041	-.073	471	.800	.200	.000	.051		.069	.092	.089	.055	.133						
432	.830	.400	-.081	.002		-.044	-.057	-.063	-.086	-.176	472	.800	.400	-.074	.018		.004	.034	.060	.070	.132						
433	.830	.600	-.093	-.024		-.044	-.057	-.063	-.086	-.176	473	.800	.600	-.074	.018		-.068	-.027	.023	.065	.119						
434	.830	.800	-.088	-.071		-.105	-.124	-.138	-.143	-.214	474	.800	.800	-.225	-.118		-.124	-.082	-.018	.036	.080						
β ≈ 0°; δ _θ = 0°; δ _v = 0°; brakes open																											
401	.110	.647	-.216	-.212	-.208	-.225	-.212	-.208	-.279	-.401	441	.130	.647	-.228	-.233	-.231	-.215	-.178	-.130								
402	.110	.681	-.025	-.023	-.027	-.065	-.050	-.043	-.150	-.344	442	.130	.682	-.047	-.043	-.024	.012	.054	.106								
403	.110	.698	-.037	-.037	-.038	-.070	-.050	-.043	-.128	-.338	443	.130	.700	-.045	-.042	-.024	.010	.050	.101								
404	.110	.748	-.105	-.106	-.106	-.143	-.109	-.106	-.209	-.376	444	.130	.748	-.107	-.105	-.088	-.045	.001	.047								
405	.110	.832	-.163	-.160	-.167	-.209	-.291	-.436	-.445	-.536	445	.130	.832	-.166	-.165	-.146	-.100	-.051	-.012								
406	.110	.949	-.232	-.224	-.244	-.311	-.378	-.456	-.466	-.546	446	.130	.949	-.244	-.243	-.221	-.168	-.134	-.108								
410	.250	.000	1.013	1.012	1.020	1.026	.976	.900	.850	.800	450	.300	.000	.939	.947	.968	.979	.997	1.035								
411	.250	.200	.242	.243	.252	.229	.144	.182	.451	.300	451	.300	.200	.222	.223	.227	.183	.110	.181								
412	.250	.400	.100	.103	.111	.093	.000	-.060	.452	.300	452	.300	.400	.093	.092	.098	.091	.073	.132								
413	.250	.600	-.041	-.037	-.033	-.052	-.129	-.297	.453	.300	453	.300	.600	-.050	-.052	-.042	-.022	.001	.044								
414	.250	.630	-.233	-.230	-.225	-.227	-.278	-.408	.454	.300	454	.300	.633	-.220	-.223	-.215	-.190	-.156	-.120								
415	.250	.668	-.030	-.023	-.024	-.069	-.220	-.406	.455	.300	455	.300	.668	-.066	-.066	-.054	-.024	.006	.045								
416	.250	.685	-.036	-.034	-.033	-.068	-.214	-.404	.456	.300	456	.300	.686	-.054	-.052	-.036	-.006	.026	.064								
417	.250	.738	-.101	-.098	-.098	-.136	-.214	-.385	.457	.300	457	.300	.738	-.097	-.096	-.081	-.046	-.008	.031								
418	.250	.825	-.154	-.146	-.151	-.197	-.286	-.443	.458	.300	458	.300	.825	-.141	-.140	-.124	-.083	-.044	-.012								
419	.250	.947	-.259	-.260	-.272	-.303	-.368	-.544	.459	.300	459	.300	.947	-.213	-.211	-.191	-.149	-.121	-.105								
420	.500	.000	.911	.900	.879	.851	.842	.802	.600	.600	460	.600	.000	.924	.941	.975	1.006	1.041	1.081								
421	.500	.100	.359	.353	.358	.365	.312	.248	.461	.600	461	.600	.100	.312	.317	.328	.290	.242	.314								
422	.500	.200	.239	.242	.252	.257	.199	.217	.462	.600	462	.600	.200	.187	.187	.197	.178	.117	.197								
423	.500	.300	.162	.163	.174	.179	.114	.093	.463	.600	463	.600	.300	.137	.138	.149	.133	.092	.162								
424	.500	.400	.114	.115	.123	.125	.057	.007	.464	.600	464	.600	.400	.091	.093	.106	.079	.051	.150								
425	.500	.500	.076	.075	.079	.081	.016	-.076	.465	.600	465	.600	.500	.053	.055	.072	.088	.097	.147								
426	.500	.600	.036	.037	.040	.039	-.017	-.143	.466	.600	466	.600	.600	.002	.004	.025	.052	.081	.090								
427	.500	.700	.001	.002	-.001	-.096	-.197	-.370	.467	.600	467	.600	.700	-.020	-.020	.010	.030	.050	.077								
428	.500	.800	-.046	-.048	-.052	-.051	-.105	-.218	.468	.600	468	.600	.800	-.072	-.066	-.042	.008	.051	.090								
429	.500	.900	-.075	-.076	-.080	-.097	-.138	-.254	.469	.600	469	.600	.900	-.120	-.114	-.084	-.031	.012	.042								
430	.830	.000		.798	.757	.715	.660	.646	.592	.470	.800	.000	.861	.884	.927	.973	1.016	.972	1.063								
431	.830	.200		.072	.063	.058	.058	.030	.040	.472	.800	.200	.145	.145	.168	.160	.119	.200	.119								
432	.830	.400		.028	.016	.005	.004	-.016	-.086	.473	.800	.400	.060	.060	.069	.089	.108	.110	.172								
433	.830	.600		.032	-.045	-.064	-.069	-.074	-.138	.474	.800	.600	-.016	-.007	.021	.063	.101	.154	.116								
434	.830	.800									.800	.800	-.088	-.077	-.042	.018	.072	.111	.064								

TABLE V.- PRESSURE COEFFICIENTS MEASURED ON VERTICAL TAIL AND SPEED BRAKES - Continued

(c) $M = 0.95$ - ContinuedPressure coefficients C_p on -

Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Upper tail								Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Lower tail							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta = 0^\circ; \delta_e = 0^\circ; \delta_v = -7.5^\circ; \text{brakes closed}$																		
401	.110	.647	-.245	-.249	-.259	-.278	-.302	-.395	441	.130	.647	-.251	-.252	-.237	-.223	-.146	-.072				
402	.110	.681	-.097	-.107	-.121	-.153	-.188	-.315	442	.130	.682	-.131	-.125	-.091	-.030	.026	.085				
403	.110	.698	-.112	-.120	-.136	-.159	-.191	-.323	443	.130	.700	-.139	-.132	-.089	-.035	.020	.081				
404	.110	.748	-.179	-.192	-.207	-.231	-.265	-.395	444	.130	.748	-.189	-.182	-.146	-.090	.018	.080				
405	.110	.832	-.215	-.227	-.251	-.284	-.325	-.461	445	.130	.832	-.236	-.230	-.194	-.125	.051	.009				
406	.110	.949	-.231	-.240	-.294	-.352	-.388	-.512	446	.130	.949	-.269	-.269	-.231	-.146	.112	.080				
410	.250	.000	1.026	1.024	1.029	1.022	.975	.856	450	.300	.000	.924	.931	.947	.970	.994	1.023				
411	.250	.200	1.026	1.024	1.029	1.022	.975	.856	451	.300	.200	.924	.931	.947	.970	.994	1.023				
412	.250	.400	1.026	1.024	1.029	1.022	.975	.856	452	.300	.400	.924	.931	.947	.970	.994	1.023				
413	.250	.600	1.026	1.024	1.029	1.022	.975	.856	453	.300	.600	.924	.931	.947	.970	.994	1.023				
414	.250	.800	1.026	1.024	1.029	1.022	.975	.856	454	.300	.800	.924	.931	.947	.970	.994	1.023				
415	.250	.000	1.026	1.024	1.029	1.022	.975	.856	455	.300	.000	.924	.931	.947	.970	.994	1.023				
416	.250	.685	1.026	1.024	1.029	1.022	.975	.856	456	.300	.685	.924	.931	.947	.970	.994	1.023				
417	.250	.738	1.026	1.024	1.029	1.022	.975	.856	457	.300	.738	.924	.931	.947	.970	.994	1.023				
418	.250	.825	1.026	1.024	1.029	1.022	.975	.856	458	.300	.825	.924	.931	.947	.970	.994	1.023				
419	.250	.947	1.026	1.024	1.029	1.022	.975	.856	459	.300	.947	.924	.931	.947	.970	.994	1.023				
420	.500	.000	.918	.904	.875	.852	.848	.814	460	.600	.000	.753	.775	.807	.842	.895	.914				
421	.500	.100	-.189	-.180	-.155	-.100	-.212	-.138	461	.600	.100	-.097	-.090	-.152	-.278	-.375	-.278				
422	.500	.200	-.025	-.036	-.046	-.026	-.102	-.024	462	.600	.200	-.008	.002	.022	.048	.003	.071				
423	.500	.300	.015	.000	-.016	-.013	-.064	.054	463	.600	.300	-.005	.003	.032	.069	.057	.131				
424	.500	.400	-.008	-.021	-.040	-.037	-.077	-.106	464	.600	.400	-.038	-.031	.000	.045	.054	.124				
425	.500	.500	-.043	-.059	-.080	-.077	-.133	-.145	465	.600	.500	-.076	-.066	-.008	.016	.045	.109				
426	.500	.600	-.104	-.120	-.139	-.138	-.154	-.236	466	.600	.600	-.124	-.113	-.043	.001	.045	.131				
427	.500	.700	-.165	-.183	-.202	-.203	-.233	-.340	467	.600	.700	-.193	-.183	-.145	-.093	-.041	.017				
428	.500	.800	-.195	-.211	-.226	-.237	-.263	-.360	468	.600	.800	-.206	-.200	-.171	-.107	-.052	.003				
429	.500	.900	-.173	-.193	-.226	-.241	-.271	-.348	469	.600	.900	-.227	-.219	-.172	-.084	-.050	-.026				
430	.830	.000	.858	.818	.775	.741	.722	.694	470	.800	.000	.761	.789	.833	.880	.940	.973				
431	.830	.200	-.037	-.047	-.059	-.051	-.075	-.114	471	.800	.200	.067	.071	.048	-.012	.053	.029				
432	.830	.400	-.095	-.107	-.120	-.119	-.135	-.214	472	.800	.400	-.039	-.028	.005	.025	.033	.082				
433	.830	.600	-.157	-.168	-.185	-.197	-.205	-.283	473	.800	.600	-.126	-.116	-.079	-.042	-.002	.043				
434	.830	.800							474	.800	.800	-.193	-.182	-.146	-.102	-.056	-.035				
$\beta = 0^\circ; \delta_e = 0^\circ; \delta_v = 7.5^\circ; \text{brakes closed}$																					
401	.110	.647	.640	.646	.644	.592	.508	.395	441	.130	.647	.666	.669	.677	.673	.651	.669				
402	.110	.681	.884	.900	.887	.824	.718	.542	442	.130	.682	.991	.997	.983	.941	.902	.900				
403	.110	.698	.920	.921	.899	.826	.682	.476	443	.130	.700	.990	.913	.948	.950	.936	.933				
404	.110	.748	.692	.689	.654	.555	.562	.485	444	.130	.748	.706	.716	.724	.731	.735	.735				
405	.110	.832	.548	.569	.519	.415	.410	.232	445	.130	.832	.610	.601	.582	.572	.565	.565				
406	.110	.949	.388	.413	.381	.306	.300	.062	446	.130	.949	.445	.431	.402	.393	.385	.383				
410	.250	.000	1.023	1.019	1.023	1.044	1.040	1.003	450	.300	.000	.939	.946	.967	.981	.998	1.033				
411	.250	.200	.220	.219	.213	.154	.045	.024	451	.300	.200	.218	.222	.234	.215	.194	.240				
412	.250	.400	.525	.524	.524	.513	.070	-.140	452	.300	.400	.247	.252	.267	.280	.286	.326				
413	.250	.600	.472	.473	.472	.458	.053	-.368	453	.300	.600	.244	.254	.272	.288	.290	.366				
414	.250	.800	.579	.591	.595	.598	.050	-.403	454	.300	.800	.268	.282	.291	.291	.291	.366				
415	.250	.000	.864	.881	.898	.910	.906	.794	455	.300	.000	.970	.959	.936	.897	.897	.961				
416	.250	.685	.887	.887	.879	.847	.855	.821	456	.300	.685	.980	.891	.904	.903	.919	.906				
417	.250	.738	.627	.629	.621	.587	.603	.507	457	.300	.738	.641	.642	.647	.666	.683	.691				
418	.250	.825	.471	.477	.465	.444	.446	.445	458	.300	.825	.509	.505	.496	.508	.517	.533				
419	.250	.947	.323	.321	.326	.302	.328	.172	459	.300	.947	.332	.327	.321	.332	.343	.361				
420	.500	.000	.927	.911	.890	.879	.863	.885	460	.600	.000	.914	.933	.967	.977	1.024	1.065				
421	.500	.100	.222	.221	.212	.165	.093	.102	461	.600	.100	.209	.217	.234	.209	.185	.234				
422	.500	.200	.202	.201	.202	.144	.047	.001	462	.600	.200	.195	.201	.216	.211	.199	.244				
423	.500	.300	.208	.207	.197	.156	-.084	-.084	463	.600	.300	.210	.210	.217	.235	.235	.274				
424	.500	.400	.242	.235	.222	.182	.091	-.084	464	.600	.400	.231	.238	.255	.272	.281	.314				
425	.500	.500	.295	.290	.276	.238	.173	.055	465	.600	.500	.284	.292	.311	.331	.343	.369				
426	.500	.600	.501	.526	.562	.620	.228	.142	466	.600	.600	.291	.298	.317	.337	.354	.381				
427	.500	.700	.067	.031	-.070	-.035	.031	-.693	467	.600	.700	.065	.075	.098	.132	.159	.190				
428	.500	.800	-.159	-.166	-.186	-.197	-.196	-.263	468	.600	.800	-.157	-.159	-.171	-.179	-.187	-.209				
429	.500	.900	-.306	-.312	-.332	-.337	-.337	-.403	469	.600	.900	-.320	-.313	-.293	-.256	-.228	-.209				
430	.830	.000	.905	.873	.838	.806	.783	.777	470	.800	.000	.894	.915	.956	.998	1.033	1.079				
431	.830	.200	.190	.184	.167	.137	.086	-.023	471	.800	.200	.180	.187	.206	.210	.206	.255				
432	.830	.400	.149	.149	.126	.100	.078	.015	472	.800	.400	.216	.207	.216	.224	.247	.286				
433	.830	.600	-.023	-.039	-.069	-.086	-.089	-.138	473	.800	.600	.189	.200	.224	.254	.276	.307				
434	.830	.800							474	.800	.800	-.088	-.076	-.045	.000	.037	.075				

TABLE V.- PRESSURE COEFFICIENTS MEASURED ON VERTICAL TAIL AND SPEED BRAKES - Continued

(c) $M = 0.95$ - ConcludedPressure coefficients C_p on -

Model orifice number	Y b/2	X c	Upper tail								Model orifice number	Y b/2	X c	Lower tail																
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$									
			$\beta = 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$											$\beta = 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$																
401	.110	.647	-.137	-.198		-.171	-.163	-.176	-.281	-.406	441	.130	.647	-.377	-.183		-.209	-.202	-.171	-.121	-.042									
402	.110	.681	.049	-.001		.020	.008	-.058	-.199	-.329	442	.130	.682	-.392	-.042		-.025	.008	.066	.125	.195									
403	.110	.698	.043	-.013		.008	.000	-.066	-.203	-.309	443	.130	.700	-.368	-.042		-.017	.009	.062	.120	.188									
404	.110	.748	.003	-.072							444	.130	.748	-.357	-.089		-.080	-.052	.010	.069	.138									
405	.110	.832	-.032	-.111		-.095	-.117	-.185	-.309	-.463	445	.130	.832	-.380	-.146		-.136	-.106	-.042	.019	.080									
406	.110	.949	-.114	-.181		-.149	-.167	-.245	-.367	-.578	446	.130	.949	-.399	-.213		-.207	-.165	-.120	-.081	-.045									
410	.250	.000	1.025	.973		.892	.904	.794	.674	.969	450	.300	.000	.494	.771		.938	.966	.992	1.020	1.060									
411	.250	.200	.141	.226		.303	.319	.264	.112	-.053	451	.300	.200	.048	.273		.263	.258	.231	.261	.261									
412	.250	.400	.015	.092		.145	.157	.123	-.007	-.196	452	.300	.400	-.160	.130		.109	.120	.129	.222	.222									
413	.250	.600	-.004	-.018		.003	.002	-.035	-.162	-.292	453	.300	.600	-.269	-.021		-.033	-.012	.028	.147	.147									
414	.250	.830	-.201	-.219							454	.300	.830	-.357	-.183		-.209	-.195	.159	.202	.202									
415	.250	.668	.079	.026		.031	.021	-.026	-.157	-.288	455	.300	.668	-.324	-.028		.036	.004	.067	.191	.191									
416	.250	.685	.057	.005		.014	.004	-.043	-.172	-.302	456	.300	.685	-.323	-.024		-.019	.010	.065	.154	.154									
417	.250	.738	.024	-.045		-.040	-.052	-.096	-.214	-.346	457	.300	.738	-.345	-.067		-.067	-.034	.026	.112	.112									
418	.250	.825	-.025	-.083		-.077	-.097	-.149	-.262	-.425	458	.300	.825	-.336	-.110		-.104	-.069	-.008	.068	.068									
419	.250	.947	-.086	-.167		-.144	-.171	-.228	-.321	-.525	459	.300	.947	-.358	-.195		-.179	-.137	-.090	.068	.068									
420	.500	.000	.828	.774		.623	.623	.458	.257	.623	460	.600	.000	.752	.825		.900	.944	.986	1.078	1.078									
421	.500	.100	.254	.304		.357	.366	.351	.257	.261	461	.600	.100	.241	.316		.324	.320	.279	.324	.324									
422	.500	.200	.098	.170		.249	.263	.239	.150	.097	462	.600	.200	.070	.198		.184	.191	.178	.251	.251									
423	.500	.300	.027	.114		.162	.174	.153	.062	-.016	463	.600	.300	-.042	.133		.121	.144	.140	.226	.226									
424	.500	.400	.029	.080		.105	.109	.087	-.006	-.106	464	.600	.400	-.114	.079		.071	.092	.114	.214	.214									
425	.500	.600	.034	.047		.057	.052	.029	-.065	-.167	465	.600	.500	-.163	.029		.022	.050	.087	.201	.201									
426	.500	.800	.035	.014		.016	.007	-.020	-.113	-.208	466	.600	.600	-.225	-.031		-.038	.000	.049	.174	.174									
427	.500	.700				-.020	-.034	-.065	-.144	-.279	467	.600	.700	-.270	-.052		-.048	-.013	.046	.164	.164									
428	.500	.800	.015	-.051		-.063	-.083	-.122	-.190	-.345	468	.600	.800	-.269	-.102		-.098	-.054	.010	.128	.128									
429	.500	.900	-.021	-.079		-.100	-.123	-.165	-.222	-.400	469	.600	.900	-.281	-.150		-.146	-.095	-.033	.068	.068									
430	.830	.000	1.037	.923		.733	.668	.573	.518	.564	470	.800	.000	.671	.761		.859	.915	.973	1.077	1.077									
431	.830	.200	.080	.090		.098	.115	.110	.057	.050	471	.800	.200	.017	.125		.104	.131	.137	.246	.246									
432	.830	.400	.058	.063		.036	.023	.008	-.045	-.171	472	.800	.400	-.037	.034		.037	.068	.102	.213	.213									
433	.830	.600	.070	.030		-.014	-.034	-.054	-.098	-.140	473	.800	.600	-.198	-.044		-.041	.000	.059	.188	.188									
434	.830	.800	.050	-.019		-.079	-.105	-.132	-.152	-.262	474	.800	.800	-.219	-.119		-.108	-.058	.012	.138	.138									
$\beta = 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$																							$\beta = 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
401	.110	.647				.741	.760	.779	.625	.435	.345	441	.130	.647			.766	.764	.757	.729	.734	.758								
402	.110	.681				.989	1.013	1.033	.830	.604	.584	442	.130	.682			1.084	1.081	1.062	1.032	1.013	1.012								
403	.110	.698				.962	.970	.949	.812	.605	.537	443	.130	.700			.961	.965	.967	.978	.977	.978								
404	.110	.748				.613	.637	.656	.580	.416	-.155	444	.130	.748			.779	.774	.764	.757	.753	.753								
405	.110	.832				.445	.463	.475	.423	.257	-.285	445	.130	.832			.673	.663	.640	.621	.611	.612								
406	.110	.949										446	.130	.949			.477	.471	.453	.439	.428	.429								
410	.250	.000				.897	.882	.902	.795	.812	.998	450	.300	.000			.924	.940	.972	1.000	1.026	1.064								
411	.250	.200				.383	.388	.393	.286	.122	-.069	451	.300	.200			.363	.359	.354	.319	.312	.312								
412	.250	.400				.336	.341	.339	.252	.087	-.156	452	.300	.400			.321	.322	.331	.337	.348	.391								
413	.250	.600				.633	.637	.628	.543	.370	.236	453	.300	.600			.609	.613	.617	.628	.634	.654								
414	.250	.830										454	.300	.830			.691	.690	.688	.679	.682	.714								
415	.250	.668				.935	.946	.974	.869	.648	.840	455	.300	.668			1.009	1.011	1.006	.993	.997	1.025								
416	.250	.685				.899	.898	.887	.823	.646	.674	456	.300	.685			.882	.890	.901	.917	.927	.930								
417	.250	.738				.648	.653	.638	.538	.331	.457	457	.300	.738			.661	.664	.666	.678	.687	.704								
418	.250	.825				.506	.514	.522	.514	.431	.061	458	.300	.825			.534	.534	.532	.541	.548	.567								
419	.250	.947				.345	.350	.355	.361	.301	-.142	459	.300	.947			.357	.357	.359	.371	.381	.405								
420	.500	.000				.855	.829	.776	.708	.405	.864	460	.600	.000			.867	.894	.942	.991	1.032	1.077								
421	.500	.100				.427	.428	.429	.361	.251	.288	461	.600	.100			.400	.402	.403	.361	.350	.399								
422	.500	.200				.348	.351	.353	.275	.149	.096	462	.600	.200			.301	.304	.308	.291	.295	.348								
423	.500	.300				.308	.310	.308	.234	.098	-.019	463	.600	.300			.276	.281	.293	.299	.307	.358								
424	.500	.400				.313	.315	.306	.239	.105	-.082	464	.600	.400			.288	.295	.308	.323	.337	.378								
425	.500	.500				.350	.351	.340	.282	.161	-.072	465	.600	.500			.331	.338	.351	.368	.385	.419								
426	.500	.600				.334	.332	.322	.283	.182	.039			.600			.315	.324	.341	.365	.385	.424								
427	.500	.700				.088	.079	.068	.056	-.007	-.082	467	.600	.700			.067	.079	.086	.104	.112	.123								
428	.500	.800				-.152	-.161	-.163	-.215	-.313	-.463	468	.600	.800			-.171	-.158	-.132	-.089	-.060	-.025								
429	.500	.900				-.301	-.310	-.316	-.294	-.337	-.450	469																		

TABLE V.- PRESSURE COEFFICIENTS MEASURED ON VERTICAL TAIL AND SPEED BRAKES - Continued

(d) $M = 1.05$ Pressure coefficients C_p on -

Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Upper tail								Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Lower tail													
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$						
			$\beta = -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$																								
401	.110	.647										.130	.647														
402	.110	.681										.130	.682														
403	.110	.698										.130	.700														
404	.110	.748										.130	.748														
405	.110	.832										.130	.832														
406	.110	.949										.130	.949														
410	.250	.000										.300	.000														
411	.250	.200										.300	.200														
412	.250	.400										.300	.400														
413	.250	.600										.300	.600														
414	.250	.630										.300	.633														
415	.250	.668										.300	.668														
416	.250	.685										.300	.686														
417	.250	.738										.300	.738														
418	.250	.825										.300	.825														
419	.250	.947										.300	.947														
420	.500	.000										.600	.000														
421	.500	.200										.600	.200														
422	.500	.400										.600	.400														
423	.500	.600										.600	.600														
424	.500	.630										.600	.633														
425	.500	.668										.600	.668														
426	.500	.685										.600	.686														
427	.500	.700										.600	.700														
428	.500	.800										.600	.800														
429	.500	.900										.600	.900														
430	.830	.000										.800	.000														
431	.830	.200										.800	.200														
432	.830	.400										.800	.400														
433	.830	.600										.800	.600														
434	.830	.800										.800	.800														
$\beta = -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$																											
401	.110	.647										.130	.647														
402	.110	.681										.130	.682														
403	.110	.698										.130	.700														
404	.110	.748										.130	.748														
405	.110	.832										.130	.832														
406	.110	.949										.130	.949														
410	.250	.000										.300	.000														
411	.250	.200										.300	.200														
412	.250	.400										.300	.400														
413	.250	.600										.300	.600														
414	.250	.630										.300	.633														
415	.250	.668										.300	.668														
416	.250	.685										.300	.686														
417	.250	.738										.300	.738														
418	.250	.825										.300	.825														
419	.250	.947										.300	.947														
420	.500	.000										.600	.000														
421	.500	.200										.600	.200														
422	.500	.400										.600	.400														
423	.500	.600										.600	.600														
424	.500	.630										.600	.633														
425	.500	.668										.600	.668														
426	.500	.685										.600	.686														
427	.500	.700										.600	.700														
428	.500	.800										.600	.800														
429	.500	.900										.600	.900														
430	.830	.000										.800	.000														
431	.830	.200										.800	.200														
432	.830	.400										.800	.400														
433	.830	.600										.800	.600														
434	.830	.800										.800	.800														
$\beta = -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$																											
401	.110	.647										.130	.647														
402	.110	.681										.130	.682														
403	.110	.698										.130	.700														
404	.110	.748										.130	.748														
405	.110	.832										.130	.832														
406	.110	.949										.130	.949														
410	.250	.000										.300	.000														
411	.250	.200										.300	.200														
412	.250	.400										.300	.400														
413	.250	.600										.300	.600														
414	.250	.630										.300	.633														
415	.250	.668										.300	.668														
416	.250	.685										.300	.686														
417	.250	.738										.300	.738														
418	.250	.825										.300	.825														
419	.250	.947										.300	.947														
420	.500	.000										.600	.000														
421	.500	.200										.600	.200														
422	.500	.400										.600	.400														
423	.500	.600										.600	.600														
424	.500	.630										.600	.633														
425	.500	.668										.600	.668														
426	.500	.685										.600	.686														
427	.500	.700				</																					

TABLE V.- PRESSURE COEFFICIENTS MEASURED ON VERTICAL TAIL AND SPEED BRAKES - Continued

(d) $M = 1.05$ - ContinuedPressure coefficients C_p on -

Model orifice number	Y b/2	X c	Upper tail								Model orifice number	Y b/2	X c	Lower tail																		
			α = -20°	α = -10°	α = -3°	α = 0°	α = 5°	α = 10°	α = 15°	α = 20°				α = -20°	α = -10°	α = -3°	α = 0°	α = 5°	α = 10°	α = 15°	α = 20°											
			β = 0°; δ _E = 0°; δ _V = 0°; brakes closed																													
			β = 0°; δ _E = 0°; δ _V = 0°; brakes closed																													
401	.110	.647	-.162	-.177		-.117	-.127	-.173	-.206	-.284	441	.130	.647	-.288	-.179		-.132	-.136	-.141	-.123	-.037											
402	.110	.681	.019	-.029		.062	.048	-.023	-.102	-.228	442	.130	.682	-.266	-.075		.022	.011	.024	.134	.229											
403	.110	.698	.028	-.010		.055	.040	-.013	-.075	-.204	443	.130	.700	-.228	.003		.048	.060	.074	.134	.221											
404	.110	.748	-.032	-.070		-.011	-.033	-.102	-.141	-.245	444	.130	.748	-.230	-.062		-.009	.005	.026	.085	.173											
405	.110	.832	-.092	-.130		-.045	-.077	-.162	-.242	-.323	445	.130	.832	-.296	-.104		-.061	-.047	-.016	.046	.125											
406	.110	.949	-.176	-.208		-.113	-.181	-.273	-.386	-.450	446	.130	.949	-.378	-.191		-.131	-.117	-.083	-.022	.029											
410	.250	.000	1.117	1.056		1.041	1.027	.972	.829	.832	450	.300	.000				.964	.990	1.012	1.040	1.077											
411	.250	.200	.097	.100		.202	.222	.188	.180	.232	451	.300	.200	.192	.171		.194	.187	.147	.146	.211											
412	.250	.400	.069	.020		.139	.142	.100	.049	.041	452	.300	.400	.041	.098		.131	.135	.117	.129	.225											
413	.250	.600	.010	-.016		.044	.036	-.006	-.053	-.135	453	.300	.600	-.128	-.004		.036	.042	.041	.070	.159											
414	.250	.830	-.176	-.193		-.141	-.148	-.176	-.217	-.277	454	.300	.830	-.257	-.160		-.129	-.128	-.132	-.109	-.039											
415	.250	.668	.043	-.006		.081	.066	-.007	-.064	-.165	455	.300	.668	-.206	-.080		-.007	.004	.029	.137	.234											
416	.250	.685	.033	-.006		.069	.054	.005	-.069	-.172	456	.300	.686	-.177	-.005		.049	.054	.066	.134	.222											
417	.250	.738	.000	-.040		.010	-.011	-.062	-.109	-.199	457	.300	.738	-.213	-.035		.010	.024	.045	.103	.190											
418	.250	.825	-.048	-.085		-.024	-.050	-.108	-.186	-.262	458	.300	.825	-.256	-.076		-.031	-.018	.013	.076	.165											
419	.250	.947	-.116	-.154		-.121	-.150	-.180	-.280	-.352	459	.300	.947	-.355	-.169		-.094	-.083	-.048	.013	.064											
420	.500	.000	1.096	1.014		.941	.913	.864	.721	.695	460	.600	.000				.955	.992	1.024	1.062	1.106											
421	.500	.100	.183	.138		.214	.235	.215	.220	.251	461	.600	.100	-.057	.184		.201	.198	.156	.163	.228											
422	.500	.200	.077	.063		.174	.193	.168	.154	.198	462	.600	.200	.201	.150		.165	.163	.129	.135	.213											
423	.500	.300	.019	.029		.150	.157	.126	.100	.120	463	.600	.300	.123	.118		.144	.144	.122	.130	.217											
424	.500	.400	.068	.029		.122	.123	.086	.050	.044	464	.600	.400	.041	.079		.104	.104	.086	.111	.208											
425	.500	.500	.054	.019		.091	.085	.044	.005	-.027	465	.600	.500	-.029	.038		.083	.090	.086	.111	.208											
426	.500	.600	.031	-.002		.052	.039	.001	-.029	-.087	466	.600	.600				.034	.040	.046	.087	.185											
427	.500	.700	.010	-.019		.038	.018	-.024	-.068	-.151	467	.600	.700	-.155	-.022		.029	.039	.054	.111	.191											
428	.500	.800	-.017	-.053		-.004	-.031	-.084	-.142	-.198	468	.600	.800	-.210	-.067		-.014	.000	.030	.090	.165											
429	.500	.900	-.048	-.088		-.026	-.056	-.124	-.200	-.295	469	.600	.900	-.264	-.108		-.055	-.037	-.003	.060	.122											
430	.830	.000	1.107	1.004		.911	.871	.804	.748	.646	470	.800	.000				.946	.988	1.028	1.072	1.119											
431	.830	.200	.092	.052		.096	.088	.049	.033	.037	471	.800	.200				.153	.151	.126	.136	.218											
432	.830	.400	.074	.012		.058	.036	-.009	-.026	-.049	472	.800	.400	.034	.060		.104	.112	.100	.119	.213											
433	.830	.600	.054	.016		.008	.004	-.025	-.074	-.121	473	.800	.600	-.073	-.019		.036	.047	.055	.105	.199											
434	.830	.800	.019	-.018							474	.800	.800	-.184	-.074		-.011	.006	.036	.100	.171											
			β = 0°; δ _E = 0°; δ _V = 0°; brakes open											β = 0°; δ _E = 0°; δ _V = 0°; brakes open											β = 0°; δ _E = 0°; δ _V = 0°; brakes open							
401	.110	.647				-.106	-.099	-.096	-.145	-.207	-.256	441	.130	.647				-.109	-.120	-.126	-.142	-.130										
402	.110	.681				.087	.091	.086	.021	-.117	-.213	442	.130	.682				.081	.063	.044	.022	.097										
403	.110	.698				.074	.080	.076	.017	-.107	-.198	443	.130	.700				.086	.076	.080	.071	.102										
404	.110	.748				-.001	.008	.003	-.072	-.159	-.250	444	.130	.748				.018	.009	.013	.012	.048										
405	.110	.832				-.058	-.044	-.057	-.142	-.264	-.343	445	.130	.832				-.036	-.045	-.037	-.037	.010										
406	.110	.949				-.119	-.110	-.142	-.232	-.383	-.481	446	.130	.949				-.119	-.124	-.114	-.101	-.056										
410	.250	.000				1.061	1.047	1.037	.987	.824	.843	450	.300	.000				.977	.985	1.009	1.025	1.051										
411	.250	.200				.302	.310	.332	.289	.272	.291	451	.300	.200				.300	.285	.265	.222	.206										
412	.250	.400				.178	.187	.205	.160	.104	.057	452	.300	.400				.194	.176	.165	.130	.134										
413	.250	.600				.062	.068	.073	.023	-.054	-.153	453	.300	.600				.068	.056	.052	.037	.053										
414	.250	.830				-.119	-.113	-.109	-.143	-.201	-.269	454	.300	.830				-.100	-.111	-.109	-.114	-.092										
415	.250	.668				.065	.078	.078	-.022	-.193	-.270	455	.300	.668				.030	.002	-.035	-.075	.023										
416	.250	.685				.074	.080	.078	-.022	-.090	-.224	456	.300	.686				.079	.068	.066	.053	.079										
417	.250	.738				.027	.014	.011	-.065	-.088	-.217	457	.300	.738				.027	.018	.017	.010	.122										
418	.250	.825				-.040	-.028	-.038	-.111	-.230	-.359	458	.300	.825				-.016	-.026	-.022	-.025	.019										
419	.250	.947				-.136	-.132	-.143	-.184	-.327	-.473	459	.300	.947				-.094	-.097	-.092	-.083	-.044										
420	.500	.000				.981	.955	.911	.860	.785	.783	460	.600	.000				.974	.992	1.028	1.056	1.092										
421	.500	.100				.417	.420	.437	.410	.388	.357	461	.600	.100				.390	.377	.364	.333	.335										
422	.500	.200				.300	.312	.339	.314	.291	.275	462	.600	.200				.272	.250	.239	.201	.201										
423	.500	.300				.226	.238	.265	.237	.206	.190	463	.600	.300				.233	.214	.200	.160	.162										
424	.500	.400				.191	.200	.217	.186	.144	.118	464	.600	.400				.196	.182	.177	.145	.148										
425	.500	.500				.162	.169	.179	.140	.088	.054	465	.600	.500				.163	.151	.151	.129	.141										
426	.500	.600				.132	.136	.138	.099	.036	-.001	466	.600	.600				.114	.105	.107	.093	.117										
427	.500	.700				.101	.106	.103	.061	-.027	-.066	467	.600	.700				.094	.087	.091	.086	.117										
428	.500	.800				.063	.063	.053	.000	-.072	-.131	468	.600	.800																		

TABLE V.- PRESSURE COEFFICIENTS MEASURED ON VERTICAL TAIL AND SPEED BRAKES - Continued

(d) M = 1.03 - Continued

Pressure coefficients C_p on -

Model office number	$\frac{Y}{b/2}$	$\frac{X}{c}$	Upper tail							Model office number	$\frac{Y}{b/2}$	$\frac{X}{c}$	Lower tail								
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$				$\alpha = 20^\circ$	$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta = 0^\circ; \delta_e = 0^\circ; \delta_v = -7.5^\circ; \text{brakes closed}$																		
401	.110	.647								441	.130	.647									
402	.110	.681									.130	.682									
403	.110	.698									.130	.700									
404	.110	.748									.130	.748									
405	.110	.932									.130	.832									
406	.110	.949									.130	.949									
410	.250	.000									.300	.000									
411	.250	.200									.300	.200									
412	.250	.400									.300	.400									
413	.250	.600									.300	.600									
414	.250	.630									.300	.633									
415	.250	.668									.300	.668									
416	.250	.685									.300	.686									
417	.250	.738									.300	.738									
418	.250	.825									.300	.825									
419	.250	.947									.300	.947									
420	.500	.000									.600	.000									
421	.500	.100									.600	.100									
422	.500	.200									.600	.200									
423	.500	.300									.600	.300									
424	.500	.400									.600	.400									
425	.500	.500									.600	.500									
426	.500	.600									.600	.600									
427	.500	.700									.600	.700									
428	.500	.800									.600	.800									
429	.500	.900									.600	.900									
430	.830	.000									.800	.000									
431	.830	.200									.800	.200									
432	.830	.400									.800	.400									
433	.830	.600									.800	.600									
434	.830	.800									.800	.800									
$\beta = 0^\circ; \delta_e = 0^\circ; \delta_v = 7.5^\circ; \text{brakes closed}$																					
401	.110	.647									.130	.647									
402	.110	.681									.130	.682									
403	.110	.698									.130	.700									
404	.110	.748									.130	.748									
405	.110	.932									.130	.832									
406	.110	.949									.130	.949									
410	.250	.000									.300	.000									
411	.250	.200									.300	.200									
412	.250	.400									.300	.400									
413	.250	.600									.300	.600									
414	.250	.630									.300	.633									
415	.250	.668									.300	.668									
416	.250	.685									.300	.686									
417	.250	.738									.300	.738									
418	.250	.825									.300	.825									
419	.250	.947									.300	.947									
420	.500	.000									.600	.000									
421	.500	.100									.600	.100									
422	.500	.200									.600	.200									
423	.500	.300									.600	.300									
424	.500	.400									.600	.400									
425	.500	.500									.600	.500									
426	.500	.600									.600	.600									
427	.500	.700									.600	.700									
428	.500	.800									.600	.800									
429	.500	.900									.600	.900									
430	.830	.000									.800	.000									
431	.830	.200									.800	.200									
432	.830	.400									.800	.400									
433	.830	.600									.800	.600									
434	.830	.800									.800	.800									
$\beta = 0^\circ; \delta_e = 0^\circ; \delta_v = 7.5^\circ; \text{brakes closed}$																					
401	.110	.647									.130	.647									
402	.110	.681									.130	.682									
403	.110	.698									.130	.700									
404	.110	.748									.130	.748									
405	.110	.932									.130	.832									
406	.110	.949									.130	.949									
410	.250	.000									.300	.000									
411	.250	.200									.300	.200									
412	.250	.400									.300	.400									
413	.250	.600									.300	.600									
414	.250	.630									.300	.633									
415	.250	.668									.300	.668									
416	.250	.685									.300	.686									
417	.250	.738									.300	.738									
418	.250	.825									.300	.825									
419	.250	.947									.300	.947									
420	.500	.000									.600	.000									
421	.500	.100									.600	.100									
422	.500	.200									.600	.200									
423	.500	.300									.600	.300									
424	.500	.400									.600	.400									
425	.500	.500									.600	.500									
426	.500	.600									.600	.600									
427	.500	.700									.600	.700									
428	.500	.800									.600	.800									
429	.500	.900									.600	.900									
430	.830	.000									.800	.000									
431	.830	.200									.800	.200									
432	.830	.400									.800	.400									
433	.830	.600									.800	.600									
434	.830	.800									.800	.800									
$\beta = 0^\circ; \delta_e = 0^\circ; \delta_v = 7.5^\circ; \text{brakes closed}$																					
401	.110	.647									.130	.647									
402	.110	.681									.130	.682									
403	.110	.698									.130	.700									
404	.110	.748									.130	.748									
405	.110	.932									.130	.832									
406	.110	.949									.130	.949									
410	.250	.000									.300	.000									
411	.250	.200									.300	.200									
412	.250	.400									.300	.400									
413	.250	.600									.300	.600									
414	.250	.630									.300	.633									
415	.250	.668									.300	.668									
416	.250	.685									.300	.686									
417	.250	.738									.300	.738									
418	.250	.825									.300	.825									
419	.250	.947									.300	.947									
420	.500	.000									.600	.000									
421	.500	.100									.600	.100									
422	.500	.200									.600	.200									
423	.500	.300									.600	.300									
424	.500	.400																			

TABLE V. - PRESSURE COEFFICIENTS MEASURED ON VERTICAL TAIL AND SPEED BRAKES - Continued

(a) $M = 1.03$ - ConcludedPressure coefficients C_p on -

Model orifice number	Y b/2	X c	Upper tail								Model orifice number	Y b/2	X c	Lower tail														
			α = -20°	α = -10°	α = -5°	α = 0°	α = 5°	α = 10°	α = 15°	α = 20°				α = -20°	α = -10°	α = -5°	α = 0°	α = 5°	α = 10°	α = 15°	α = 20°							
			β = 5°; δ _E = 0°; δ _V = 0°; brakes closed																									
401	.110	.647	-.141	-.129		-.071	-.055	-.094	-.226	-.374	441	.130	.647	-.344	-.169		-.102	-.102	-.094	-.056	.021							
402	.110	.681	.088	.043		.118	.109	.012	-.167	-.326	442	.130	.682	-.320	-.011		.009	.067	.134	.208	.287							
403	.110	.698	.094	.052		.107	.101	.001	-.131	-.318	443	.130	.700	-.320	-.011		.082	.102	.137	.203	.280							
404	.110	.748	.052	-.019							444	.130	.748	-.289	-.057		.021	.042	.080	.151	.228							
405	.110	.832	.044	-.082		.002	-.018	-.116	-.269	-.420	445	.130	.832	-.368	-.108		-.023	-.001	.039	.106	.177							
406	.110	.949	-.024	-.143		-.038	-.063	-.189	-.376	-.516	446	.130	.949	-.457	-.176		-.091	-.063	-.022	.024	.072							
β = 5°; δ _E = 0°; δ _V = 0°; brakes closed																												
410	.250	.000	1.085	1.031		.954	.814	.584	.426	.930	450	.300	.000	.913	.655		.989	1.024	1.050	1.076	1.113							
411	.250	.200	.245	.261		.364	.389	.273	.240	.001	451	.300	.200	.094	.288		.315	.298	.268	.253	.333							
412	.250	.400	.075	.119		.213	.240	.180	.081	-.133	452	.300	.400	-.072	.166		.166	.171	.171	.205	.295							
413	.250	.600	.029	.039		.090	.100	.021	-.065	-.271	453	.300	.600	-.226	.014		.066	.073	.089	.145	.225							
414	.250	.630	-.153	-.146							454	.300	.633	-.322	-.149		-.101	-.097	-.088	-.044	.028							
415	.250	.668	.105	.053		.125	.121	.023	-.119	-.295	455	.300	.668	-.285	-.018		-.013	.053	.130	.208	.291							
416	.250	.685	.101	.060		.111	.106	.011	-.103	-.302	456	.300	.686	-.250	.015		.074	.100	.134	.201	.278							
417	.250	.738	.073	.007		.058	.048	-.034	-.136	-.332	457	.300	.738	-.277	-.030		.037	.059	.095	.167	.244							
418	.250	.825	.068	-.041		.024	.005	-.088	-.203	-.375	458	.300	.825	-.308	-.072		.003	.031	.070	.137	.206							
419	.250	.947	-.003	-.107		-.041	-.071	-.166	-.296	-.456	459	.300	.947	-.410	-.148		-.067	-.032	.009	.058	.110							
β = 5°; δ _E = 0°; δ _V = 0°; brakes open																												
420	.500	.000				.877	.826	.648	.742	.897	460	.600	.000	.854	.813		.951	1.004	1.046	1.088	1.127							
421	.500	.100		.342		.416	.429	.387	.368	.326	461	.600	.100	.297	.356		.376	.357	.328	.310	.394							
422	.500	.200		.198		.313	.335	.290	.259	.162	462	.600	.200	.139	.232		.234	.224	.213	.219	.322							
423	.500	.300		.112		.226	.248	.210	.154	.057	463	.600	.300	.030	.168		.163	.174	.173	.203	.298							
424	.500	.400		.080		.169	.191	.145	.072	-.032	464	.600	.400	-.056	.115		.139	.151	.159	.198	.286							
425	.500	.500		.070	.081	.134	.143	.086	.006	-.111	465	.600	.500	-.124	.064		.111	.123	.138	.193	.276							
426	.500	.600		.051	.053	.101	.097	.035	-.041	-.172	466	.600	.600	-.158	.001		.054	.071	.103	.170	.256							
427	.500	.700				.078	.065	.005	-.098	-.263	467	.600	.700	-.200	-.010		.053	.078	.111	.175	.251							
428	.500	.800		.076	-.007	.038	.015	-.059	-.152	-.278	468	.600	.800	-.236	-.061		.008	.042	.085	.149	.220							
429	.500	.900		.066	-.041	.012	-.017	-.105	-.202	-.323	469	.600	.900	-.299	-.108		-.034	.003	.052	.110	.171							
β = 5°; δ _E = 0°; δ _V = 0°; brakes open																												
430	.830	.000	1.098	.997		.789	.710	.591	.551	.635	470	.800	.000	.768	.752		.916	.984	1.036	1.082	1.128							
431	.830	.200	.170	.121		.157	.176	.155	.116	.106	471	.800	.200	.059	.159		.140	.167	.205	.313								
432	.830	.400	.111	.087		.112	.111	.053	-.018	-.068	472	.800	.400	-.087	.071		.124	.136	.149	.197	.286							
433	.830	.600	.090	.063		.085	.064	.003	-.058	-.120	473	.800	.600	-.115	-.021		.043	.073	.112	.181	.267							
434	.830	.800	.106	.018		.027	-.007	-.071	-.137	-.199	474	.800	.800	-.203	-.082		-.002	.038	.084	.154	.229							
β = 5°; δ _E = 0°; δ _V = 0°; brakes open																												
401	.110	.647				.807	.822	.844	.570	.625	.351	441	.130	.647				.832	.828	.818	.782	.817						
402	.110	.681				1.045	1.063	1.081	.673	.837	.630	442	.130	.682				1.133	1.125	1.111	1.081	1.055						
403	.110	.698				1.024	1.026	1.011	.684	.798	.598	443	.130	.700				1.018	1.021	1.028	1.037	1.037						
404	.110	.748				.693	.714	.734	.708	.464	-.059	444	.130	.748				.862	.837	.829	.819	.816						
405	.110	.832				.538	.555	.579	.551	.336	-.205	445	.130	.832				.752	.737	.714	.689	.679						
406	.110	.949										446	.130	.949				.576	.560	.542	.523	.514						
β = 5°; δ _E = 0°; δ _V = 0°; brakes open																												
410	.250	.000				.973	.958	.939	.672	.474	.977							.992	1.004	1.032	1.056	1.082						
411	.250	.200				.463	.475	.486	.374	.266	-.007	450	.300	.000				.458	.442	.419	.378	.358						
412	.250	.400				.421	.429	.430	.363	.178	-.121	451	.300	.200				.414	.410	.405	.394	.410						
413	.250	.600				.707	.710	.704	.507	.538	.309	452	.300	.400				.685	.686	.689	.691	.701						
414	.250	.630				.992	1.004	1.036	.635	.957	.894	453	.300	.600				.764	.759	.754	.728	.736						
415	.250	.668				.962	.958	.952	.635	.857	.745	454	.300	.633				1.070	1.068	1.061	1.034	1.043						
416	.250	.685				.723	.726	.734	.632	.667	.555	455	.300	.668				.946	.951	.965	.982	.991						
417	.250	.738				.590	.596	.611	.509	.562	.265	456	.300	.686				.738	.737	.740	.742	.754						
418	.250	.825				.440	.445	.457	.466	.438	.051	457	.300	.738				.621	.615	.612	.609	.618						
419	.250	.947										458	.300	.825				.456	.451	.452	.454	.465						
420	.500	.000				.928	.894	.835	.670	.729	.893	459	.300	.947				.932	.959	1.007	1.046	1.090						
421	.500	.100				.504	.512	.516	.454	.396	.344							.489	.482	.459	.407	.394						
422	.500	.200				.430	.440	.446	.378	.295	.181	460	.600	.000				.396	.387	.373	.343	.346						
423	.500	.300				.393	.401	.407	.341	.216	.080	461	.600	.100				.372	.370	.368	.350	.364						
424	.500	.400				.400	.403	.399	.336	.200	-.005	462	.600	.200				.383	.384	.389	.382	.402						
425	.500	.500				.437	.439	.430	.353	.257	-.029	463	.600	.300				.423	.426	.434	.435	.456						
426	.500	.600				.423	.419	.414	.317	.310	.170	464	.600	.400				.406	.413	.427	.435	.460						
427	.500	.700				.178	.167	.159	.103	.109	.069	465	.600	.500				.157	.170	.194	.213	.249						
428	.500	.800				-.061	-.072	-.075	-.108	-.127	-.186	466	.600	.600				-.078	-.066	-.040	-.023	.009						
429	.500	.900				-.212	-.225	-.223	-.243	-.280	-.367	467	.600	.700				-.080	-.068	-.040	-.023	.009						
β = 5°; δ _E = 0°; δ _V = 0°; brakes open																												
430	.830	.000				.846	.805	.727	.614	.543	.601	469	.600	.900				.237	.223	.197	-.183	-.154						
431	.830	.200				.327																						

TABLE V.- PRESSURE COEFFICIENTS MEASURED ON VERTICAL TAIL AND SPEED BRAKES - Continued

(e) $M = 1.18$

Pressure coefficients C_p on -

Model orifice number	Upper tail										Model orifice number	Lower tail																			
	$\frac{y}{b/2}$	$\frac{x}{c}$	$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 3^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$		$\frac{y}{b/2}$	$\frac{x}{c}$	$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 3^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$										
			$\beta = -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$											$\beta = -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$																	
401	.110	.647				-.068	-.077	-.033	-.044	-.112	.441	.130	.647				-.066	-.054	-.043	-.010	.036										
402	.110	.681				.031	.032	.023	-.027	-.009	.442	.130	.682				.016	.005	.034	.125	.282										
403	.110	.698				.035	.024	.050	-.007	-.019	.443	.130	.700				.047	.045	.062	.156	.280										
404	.110	.748				-.001	-.006	.013	-.066	-.091	.444	.130	.748				.011	.048	.049	.139	.249										
405	.110	.832				-.047	-.053	.053	-.139	-.177	.445	.130	.832				-.029	-.007	.027	.104	.215										
406	.110	.949				-.091	-.102	-.107	-.183	-.261	.446	.130	.949				-.074	-.052	-.025	.048	.152										
410	.250	.000				1.080	.941	.454	.398	.943	.450	.300	.000				.942	1.002	1.041	1.065	1.086										
411	.250	.000				-.107	-.208	-.385	-.384	-.399	.451	.300	.000				-.092	-.058	.015	.004	.105										
412	.250	.000				-.004	.010	-.005	.077	-.066	.452	.300	.000				.103	.012	.000	.041	.127										
413	.250	.000				.061	.024	.033	.021	.056	.453	.300	.000				.068	.075	.096	.147	.218										
414	.250	.030				-.081	-.116	-.075	-.104	-.101	.454	.300	.633				-.051	-.054	-.042	.001	.078										
415	.250	.668				-.034	.038	.004	-.076	.031	.455	.300	.668				.000	.040	.058	.143	.249										
416	.250	.685				.044	.035	.059	-.021	.037	.456	.300	.685				.037	.033	.145	.159	.258										
417	.250	.738				.025	.004	.004	-.074	.050	.457	.300	.738				.035	.042	.065	.150	.260										
418	.250	.825				-.012	-.030	.029	-.114	-.117	.458	.300	.825				.000	.014	.047	.126	.240										
419	.250	.947				-.066	-.088	-.073	-.138	-.195	.459	.300	.947				-.041	-.027	.007	.078	.183										
420	.500	.000				.997	.915	.818	.818	.833	.460	.600	.000				.934	1.005	1.054	1.092	1.125										
421	.500	.100				-.301	-.391	-.318	-.428	-.443	.461	.600	.100				-.327	-.254	-.219	-.208	-.191										
422	.500	.200				-.140	-.229	-.299	-.362	-.407	.462	.600	.200				-.135	-.071	-.020	.054	.249										
423	.500	.300				-.098	-.081	-.072	-.035	-.384	.463	.600	.300				.016	.009	.058	.112	.177										
424	.500	.400				.088	.129	.093	.148	.116	.464	.600	.400				.156	.004	-.004	.040	.126										
425	.500	.500				.095	.115	.104	.131	.164	.465	.600	.500				.131	.071	.033	.071											
426	.500	.600				.076	.059	.045	.046	.087	.466	.600	.600				.076	.088	.107	.153	.215										
427	.500	.700								.467	.600	.700					.029	.041	.056	.129	.241										
428	.500	.800				.010	.001	-.028	-.067	.063	.468	.600	.800				.015	.026	.044	.126											
429	.500	.900				-.016	-.025	-.051	-.091	-.096	.469	.600	.900				-.012	.003	.032	.103	.217										
430	.830	.000				.969	.888	.810	.783	.774	.470	.800	.000				.957	1.028	1.083	1.128	1.170										
431	.830	.200				-.169	-.267	-.325	-.409	-.390	.471	.800	.200				-.052	.045	.117	.186	.264										
432	.830	.400				.092	.155	.178	.160	.078	.472	.800	.400				.145	-.013	-.001	.050	.130										
433	.830	.600				.065	.068	.059	.052	.071	.473	.800	.600				.079	.087	.112	.156	.228										
434	.830	.800				.017	.023	.010	-.026	-.019	.474	.800	.800				.015	.026	.050	.128	.230										
$\beta = -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$																						$\beta = -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$									
401	.110	.647				-.142	-.031	.627	.511	.386	.676	.721	.647	.130	.647			.231	.487	.659	.659	.700									
402	.110	.681	.018			-.001	.697	.487	.811	.513	.838	.941	.682	.130	.682	-.017	.093	.311	.876	.904	.853	.863									
403	.110	.698	.052			-.019	.665	.847	.603	.859	.870	.639	.443	.130	.700	-.015	.077	.314	.898	.999	.957	.920									
404	.110	.748	.048			-.054	.861	.729	.740	.898	.776	.672	.444	.130	.748	-.070	.023	.369	.770	.878	.904	.899									
405	.110	.832	.005			-.105	.675	.659	.655	.795	.653	.513	.445	.130	.832	-.138	.001	.691	.694	.715	.712	.718									
406	.110	.949	-.079			-.167	.559	.578	.560	.618	.507	.359	.446	.130	.949	-.220	.011	.692	.608	.582	.563	.561									
410	.250	.000	1.231	1.184	1.126	1.082	.855	.463	.386	.943	.450	.300	.000	.734	.601	.091	.940	1.008	1.043	1.064	1.083										
411	.250	.000	.998	-.016	-.073	-.104	-.180	.343	-.394	-.396	.451	.300	.200	-.416	-.610	.898	-.031	-.031	.015	.056	.111										
412	.250	.000	.945	-.108	.068	.185	.198	.116	.132	.024	.452	.300	.400	.101	.099	.200	.263	.278	.273	.369	.456										
413	.250	.000	.200	.081	-.034	.011	.532	.021	.583	.053	.453	.300	.400	.026	.405	.305	.546	.623	.625	.678	.727										
414	.250	.030	-.157	-.133	.088	.537	.377	.711	.640	.431	.455	.300	.633	-.100	-.119	.422	.711	.944	.825	.789	.802										
415	.250	.668	-.057	-.011	.761	.783	.408	.954	.807	.563	.455	.300	.668	-.010	-.032	.871	.944	.825	.789	.802	.891										
416	.250	.685	-.040	-.003	.967	.899	.447	.881	.701	.631	.456	.300	.686	.008	.011	.811	.979	.950	.887	.827	.901										
417	.250	.738	.052	-.029	.845	.761	.512	.793	.621	.697	.457	.300	.738	.052	-.026	.663	.787	.898	.876	.822	.829										
418	.250	.825	.070	-.061	.631	.537	.381	.583	.507	.458	.458	.300	.825	-.111	-.034	.654	.657	.661	.675	.665	.701										
419	.250	.947	-.016	-.110	.514	.537	.452	.580	.507	.454	.459	.300	.947	-.188	-.044	.610	.545	.527	.519	.534	.555										
420	.500	.000	1.198	1.114	1.034	.994	.904	.787	.828	.836	.460	.600	.000	.761	.741	.878	.936	1.012	1.057	1.090	1.120										
421	.500	.100	-.173	-.254	-.260	-.299	-.399	-.347	-.421	-.442	.461	.600	.100	-.568	-.495	-.361	-.318	-.240	-.210	-.218	-.203										
422	.500	.200	.081	-.034	-.091	-.142	-.285	-.294	-.364	-.406	.462	.600	.200	.422	-.317	.018	-.063	-.055	.015	.043	.104										
423	.500	.300	.154	-.075	-.093	-.047	.091	.001	.028	-.133	.463	.600	.300	.074	.133	.242	.262	.019	.062	.125	.365										
424	.500	.400	.091	-.021	.187	.275	.287	.218	.228	.094	.464	.600	.400	.169	.199	.218	.278	.301	.298	.371	.438										
425	.500	.500	.027	-.044	.371	.340	.330	.344	.323	.276	.465	.600	.500	.148	.144	.225	.338	.413	.437	.468	.516										
426	.500	.600	-.015	-.005	.400	.381	.312	.352	.333	.301	.466	.600	.600	.343	1.390	.292	.397	.445	.463	.496	.540										
427	.500	.700	.000	-.061	.026	.631	.537	.452	.580	.507	.458	.300	.700	.000	.201	.099	.167	.209	.208	.307	.388										
428	.500	.800	-.015	-.035	.058	.034	.005	.036	-.038	.001	.468	.600	.800	.000	-.074	.017	-.002	.041	.078	.101	.136										
429	.500	.900	.041	-.055	-.088	-.095	-.111	-.099	-.164	-.127	.469	.600	.900	.000	-.268	.205	-.074	.009	.125	.152	.168										
430	.830	.000	1.201	1.100	1.011	.944	.880	.795	.776	.766	.470	.800	.000	.736	.779	.920	.957	1.034	1.085	1.127	1.168										
431	.830	.200	.194	-.084	-.114	-.170	-.274	-.321	-.400	-.427	.471	.800	.200	-.475	-.268	.205	-.079	.009	.125	.152	.168										
432	.830	.400	.110	.005	.155	.263	.264	.266	.271	.282	.472	.800	.400	.124	.177	.220	.281	.294	.282	.352	.416										
433	.830	.600	.037	-.057	.311	.296	.265	.282	.265	.234	.473	.800	.600	.057	.084	.255	.338	.380	.396	.430	.472										
434	.830	.800	-.048	-.010	.191	.177	.152	.127	.094	.132	.474	.800	.800	.000	.103	.140	.178	.200	.240	.280	.284										

TABLE V.- PRESSURE COEFFICIENTS MEASURED ON VERTICAL TAIL AND SPEED BRAKES - Continued

(e) $M = 1.18$ - ContinuedPressure coefficients C_p on -

Model orifice number	y b/2	x c	Upper tail										Model orifice number	y b/2	x c	Lower tail											
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$	$\alpha = -20^\circ$	$\alpha = -10^\circ$				$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$						
			$\beta = 0^\circ; \delta_E = 0^\circ; \delta_V = 0^\circ; \text{brakes closed}$													$\beta = 0^\circ; \delta_E = 0^\circ; \delta_V = 0^\circ; \text{brakes closed}$											
401	.110	.647	-.011	-.074		-.013	-.003	-.031	-.098	-.151	441	.130	.647	-.144	-.035		-.017	-.022	-.008	.027	.080						
402	.110	.681	.119	-.002		.053	.060	.034	-.085	-.147	442	.130	.682	-.155	-.012		.010	.015	.037	.105	.279						
403	.110	.698	.138	.015		.087	.086	.066	-.069	-.142	443	.130	.700	-.141	.021		.072	.047	.061	.141	.307						
404	.110	.748	.097	-.011		.038	.043	.038	-.097	-.177	444	.130	.748	-.156	.087		.048	.054	.068	.144	.266						
405	.110	.832	.042	-.063		-.014	-.012	-.019	-.133	-.242	445	.130	.832	-.228	-.002		.006	.014	.034	.112	.234						
406	.110	.949	-.044	-.132		-.095	-.085	-.109	-.210	-.304	446	.130	.949	-.280	-.076		-.048	-.036	.015	.065	.179						
$\beta = 0^\circ; \delta_E = 0^\circ; \delta_V = 0^\circ; \text{brakes closed}$																											
410	.250	.000	.008	1.144		1.067	1.011	.988	.920	.928	450	.300	.000				.986	1.054	1.109	1.148	1.189						
411	.250	.200	.257	.164		.166	.250	.199	.051	-.350	451	.300	.200	-.318	.209		.163	.150	.179	.207	.272						
412	.250	.400	.082	.080		.142	.171	.183	.216	.220	452	.300	.400	.207	.175		.162	.149	.183	.222	.288						
413	.250	.600	.121	.028		.101	.112	.078	.016	-.029	453	.300	.600	-.012	.084		.098	.092	.115	.172	.253						
414	.250	.630	-.023	-.089		-.032	-.019	-.036	-.099	-.140	454	.300	.633	-.115	-.020		-.014	-.021	-.005	.034	.091						
415	.250	.668	.145	.005		.065	.072	.059	-.049	-.094	455	.300	.668	-.117	.013		.031	.032	.055	.126	.277						
416	.250	.685	.146	.027		.086	.091	.090	-.031	-.090	456	.300	.686	-.088	.061		.065	.073	.092	.155	.298						
417	.250	.738	.126	.014		.071	.082	.060	-.058	-.137	457	.300	.738	-.129	.069		.067	.062	.073	.155	.280						
418	.250	.825	.087	-.016		.027	.038	-.078	-.078	-.181	458	.300	.825	-.173	.020		.022	.035	.054	.133	.259						
419	.250	.947	.024	-.067		-.021	-.009	-.020	-.116	-.240	459	.300	.947	-.157	-.030		-.024	-.011	.039	.095	.213						
420	.500	.000	1.197	1.102		.906	.898	.839	.815	.819	460	.600	.000	.806	.823		.971	1.049	1.113	1.164	1.213						
421	.500	.100	.335	.231		.090	-.015	-.121	-.244	-.323	461	.600	.100	-.364	-.161		.133	.192	.244	.278	.333						
422	.500	.200	.229	.128		.226	.263	.295	.250	-.048	462	.600	.200	-.247	.247		.158	.088	.129	.191	.259						
423	.500	.300	.172	.099		.150	.186	.233	.283	.303	463	.600	.300	.275	.214		.152	.136	.136	.128	.201						
424	.500	.400	.109	.084		.123	.150	.156	.181	.200	464	.600	.400	.195	.168		.144	.130	.170	.216	.275						
425	.500	.500	.058	.068		.116	.142	.129	.107	.104	465	.600	.500	.116	.134		.132	.121	.153	.203	.280						
426	.500	.600	.102	.041		.110	.118	.104	.039	.019	466	.600	.600				.102	.100	.120	.173	.257						
427	.500	.700	.119	.008		.078	.084	.092	-.006	-.063	467	.600	.700	-.063	.082		.064	.071	.085	.152	.273						
428	.500	.800	.114	.010		.053	.054	.065	-.029	-.106	468	.600	.800	-.110	.052		.044	.053	.069	.149	.263						
429	.500	.900	.085	-.010		.020	.025	.024	-.056	-.131	469	.600	.900	-.148	.005		.010	.025	.054	.127	.245						
430	.830	.000	1.205	1.101		.950	.872	.805	.756	.740	470	.800	.000	.792	.831		.983	1.057	1.128	1.175	1.224						
431	.830	.200	.242	.108		.107	.144	.163	.147	.156	471	.800	.200				.150	.088	.136	.192	.264						
432	.830	.400	.146	.090		.112	.144	.163	.147	.156	472	.800	.400	.187	.165		.141	.119	.160	.210	.268						
433	.830	.600	.078	.060		.112	.112	.103	.062	.047	473	.800	.600	.043	.085		.101	.105	.127	.182	.266						
434	.830	.800	.134	.023		.060	.063	.064	.012	-.030	474	.800	.800	-.069	.037		.041	.054	.077	.153	.267						
$\beta = 0^\circ; \delta_E = 0^\circ; \delta_V = 0^\circ; \text{brakes open}$																											
401	.110	.647				-.003	.007	.015	.002	-.069	-.145	441	.130	.647			.009	.001	-.011	-.015	.016	.062					
402	.110	.681				.068	.080	.087	.086	-.034	-.147	442	.130	.682			.041	.035	.031	.029	.083	.176					
403	.110	.698				.086	.099	.109	.103	-.025	-.141	443	.130	.700			.091	.075	.054	.044	.097	.222					
404	.110	.748				.048	.059	.080	.078	-.056	-.179	444	.130	.748			.085	.071	.059	.046	.100	.219					
405	.110	.832				-.021	-.002	.009	-.014	-.122	-.252	445	.130	.832			.035	.025	.016	.013	.067	.178					
406	.110	.949				-.094	-.069	-.069	-.109	-.221	-.336	446	.130	.949			-.033	-.039	-.042	-.005	.020	.137					
410	.250	.000				1.106	1.069	1.005	.978	.907	.919	450	.300	.000			.947	.993	1.061	1.112	1.150	1.188					
411	.250	.200				.323	.335	.347	.340	.400	.322	451	.300	.200			.309	.303	.290	.283	.284	.324					
412	.250	.400				.210	.227	.245	.230	.233	.148	452	.300	.400			.216	.206	.182	.185	.219	.279					
413	.250	.600				.110	.125	.138	.132	.033	-.092	453	.300	.600			.134	.122	.106	.102	.145	.213					
414	.250	.630				-.007	.003	.006	.008	-.070	-.187	454	.300	.633			.016	.011	.005	.008	.047	.096					
415	.250	.668				-.002	.013	.028	.019	-.064	-.162	455	.300	.668			.014	.010	-.003	-.017	.012	.070					
416	.250	.685				.076	.094	.110	.111	-.036	-.212	456	.300	.686			.091	.082	.066	.048	.105	.184					
417	.250	.738				.058	.079	.116	.101	-.049	-.212	457	.300	.738			.109	.077	.048	.026	.086	.204					
418	.250	.825				.005	.027	.046	.009	-.096	-.277	458	.300	.825			.045	.038	.024	.015	.063	.175					
419	.250	.947				-.051	-.030	-.028	-.058	-.173	-.352	459	.300	.947			-.019	-.025	-.027	.007	.028	.138					
420	.500	.000				1.064	1.022	.937	.841	.820	.818	460	.600	.000			1.001	1.050	1.108	1.153	1.193	1.232					
421	.500	.100				.429	.431	.426	.404	.219	-.061	461	.600	.100			.392	.398	.405	.420	.447	.489					
422	.500	.200				.332	.349	.360	.359	.398	.383	462	.600	.200			.288	.275	.261	.262	.271	.324					
423	.500	.300				.254	.272	.291	.292	.321	.294	463	.600	.300			.235	.222	.195	.193	.195	.258					
424	.500	.400				.206	.224	.245	.239	.240	.208	464	.600	.400			.206	.195	.176	.188	.227	.285					
425	.500	.500				.182	.195	.209	.202	.171	.128	465	.600	.500			.199	.188	.170	.180	.223	.285					
426	.500	.600				.156	.167	.180	.167	.124	.057	466	.600	.600			.171	.165	.147	.143	.168	.264					
427	.500	.700				.115	.127	.144	.160	.081	-.002	467	.600	.700			.132	.122	.115	.111	.158	.242					
428	.500	.800				.096	.120	.136	.134	.051	-.003	468	.600	.800			.117	.104	.082	.083	.144	.244					
429	.500	.900				.071	.085	.093	.089	.024	-.072	469	.600	.900			.075	.068	.066	.092	.132	.243					
430	.830	.000				.991	.952	.877	.819	.783	.743	470	.800	.000			.922	1.024	1.079	1.128	1.172	1.215					
431	.830	.200				.490	.442	.447																			

TABLE V.- PRESSURE COEFFICIENTS MEASURED ON VERTICAL TAIL AND SPEED BRAKES - Continued

(e) $M = 1.18$ - ContinuedPressure coefficients C_p on -

Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Upper tail									Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Lower tail										
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$	$\alpha = -20^\circ$				$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				
			$\beta = 0^\circ; \delta_0 = 0^\circ; \delta_V = -7.5^\circ; \text{brakes closed}$																						
			$\beta = 0^\circ; \delta_0 = 0^\circ; \delta_V = -7.5^\circ; \text{brakes closed}$																						
401	.110	.647			-.021	-.018	-.028	-.072	-.120	-.179	441	.130	.647			-.040	-.034	-.035	-.021	.012	.056				
402	.110	.681			.052	.043	.027	-.021	-.108	-.166	442	.130	.682			-.008	.000	-.004	.029	.128	.238				
403	.110	.698			.083	.084	.059	.002	-.089	-.166	443	.130	.700			.054	.065	.039	.054	.156	.240				
404	.110	.748			-.001	-.007	-.015	-.040	-.146	-.235	444	.130	.748			-.009	.001	.012	.028	.100	.183				
405	.110	.832			-.074	-.076	-.091	-.125	-.217	-.303	445	.130	.832			-.061	-.055	-.041	-.026	.061	.158				
406	.110	.949			-.126	-.129	-.147	-.168	-.272	-.359	446	.130	.949			-.102	-.096	-.077	-.016	.029	.145				
410	.250	.000			1.108	1.072	1.010	.970	.892	.905	450	.300	.000			.942	.989	1.062	1.114	1.151	1.188				
411	.250	.200			-.046	-.056	-.074	-.301	-.385	-.463	451	.300	.200			-.085	-.067	-.035	.008	.046	.087				
412	.250	.400			.002	.030	.047	.069	.128	.122	452	.300	.400			.101	.069	.052	.071	.061	.267				
413	.250	.600			.086	.087	.075	.031	-.019	-.032	453	.300	.600			.062	.073	.077	.099	.190	.265				
414	.250	.830			-.039	-.041	-.046	-.074	-.124	-.155	454	.300	.830			-.036	-.028	-.032	-.030	.067	.108				
415	.250	.668			.033	.028	.041	.015	-.094	-.118	455	.300	.668			-.009	.008	.005	.051	.105	.219				
416	.250	.685			.045	.040	.026	.003	-.080	-.125	456	.300	.686			-.007	.011	.024	.040	.135	.215				
417	.250	.738			-.013	-.022	-.028	-.058	-.153	-.201	457	.300	.738			-.031	-.020	-.007	.002	.094	.186				
418	.250	.825			-.055	-.061	-.081	-.088	-.182	-.241	458	.300	.825			-.066	-.060	-.031	-.015	.085	.188				
419	.250	.947			-.063	-.058	-.063	-.079	-.180	-.278	459	.300	.947			-.086	-.078	-.038	.002	.066	.184				
420	.500	.000			1.052	1.007	.914	.832	.825	.819	460	.600	.000			.914	.958	1.016	1.052	1.081	1.112				
421	.500	.100			-.066	-.070	-.182	-.298	-.416	-.462	461	.600	.100			-.096	-.071	-.037	.002	.022	.049				
422	.500	.200			-.091	-.087	.011	-.137	-.349	-.478	462	.600	.200			-.021	-.112	-.100	-.065	.010	.153				
423	.500	.300			.007	.007	.095	.118	.179	.132	463	.600	.300			.154	.118	.045	.025	.071	.159				
424	.500	.400			.070	.102	.116	.116	.138	.125	464	.600	.400			.128	.115	.085	.097	.086	.245				
425	.500	.500			.081	.089	.091	.072	.050	.053	465	.600	.500			.108	.095	.097	.134	.193	.274				
426	.500	.600			.073	.079	.063	.018	-.026	-.031	466	.600	.600			.046	.066	.075	.097	.159	.227				
427	.500	.700			-.060	-.066	-.082	-.110	-.184	-.185	467	.600	.700			-.060	-.051	-.052	-.034	.040	.133				
428	.500	.800			-.039	-.044	-.057	-.083	-.154	-.214	468	.600	.800			-.053	-.053	-.047	-.036	.049	.143				
429	.500	.900			-.055	-.062	-.083	-.088	-.161	-.226	469	.600	.900			-.070	-.066	-.055	-.015	.054	.156				
430	.830	.000			1.017	.979	.905	.832	.749	.728	470	.800	.000			.920	.967	1.029	1.073	1.112	1.152				
431	.830	.200									471	.800	.200			.155	.008	.028	.085	.142	.209				
432	.830	.400			.067	.086	.116	.130	.128	.151	472	.800	.400			.121	.113	.086	.079	.043	.210				
433	.830	.600			.071	.079	.070	.033	.003	-.006	473	.800	.600			.055	.072	.083	.099	.156	.225				
434	.830	.800			-.013	-.017	-.026	-.040	-.091	-.129	474	.800	.800			-.041	-.038	-.027	-.019	.058	.131				
$\beta = 0^\circ; \delta_0 = 0^\circ; \delta_V = 7.5^\circ; \text{brakes closed}$																									
401	.110	.647			.703	.720	.740	.729	.678	.589	441	.130	.647			.749	.758	.748	.742	.760	.804				
402	.110	.681			.948	.968	.982	.950	.885	.761	442	.130	.682			1.098	1.097	1.063	1.025	.974	1.022				
403	.110	.698			1.073	1.071	1.048	.998	.946	.793	443	.130	.700			1.034	1.067	1.128	1.120	1.082	1.101				
404	.110	.748			.583	.569	.535	.781	.784	.617	444	.130	.748			.859	.877	.904	.914	.926	.937				
405	.110	.832			.733	.756	.712	.634	.640	.425	445	.130	.832			.808	.801	.767	.751	.750	.765				
406	.110	.949			.612	.640	.624	.556	.523	.261	446	.130	.949			.692	.665	.628	.608	.604	.613				
410	.250	.000			1.117	1.076	1.014	.983	.932	.925	450	.300	.000			.948	.994	1.059	1.104	1.142	1.180				
411	.250	.200			.153	.252	.277	.279	.115	-.180	451	.300	.200			.266	.240	.162	.174	.200	.293				
412	.250	.400			.260	.290	.317	.308	.215	.210	452	.300	.400			.333	.333	.318	.341	.409	.481				
413	.250	.600			.692	.699	.706	.689	.643	.533	453	.300	.600			.678	.691	.705	.712	.734	.777				
414	.250	.830			.646	.689	.721	.731	.665	.596	454	.300	.830			.749	.735	.680	.680	.702	.772				
415	.250	.668			.927	.966	.997	1.015	.914	.839	455	.300	.668			1.092	1.066	1.015	.920	1.001	.992				
416	.250	.685			1.068	1.057	1.046	1.016	.949	.841	456	.300	.686			1.065	1.076	1.101	1.059	1.093	1.060				
417	.250	.738			.830	.825	.816	.806	.762	.680	457	.300	.738			.831	.831	.845	.862	.873	.894				
418	.250	.825			.668	.682	.691	.684	.647	.562	458	.300	.825			.722	.709	.692	.691	.702	.733				
419	.250	.947			.541	.557	.576	.554	.561	.445	459	.300	.947			.585	.567	.551	.547	.559	.584				
420	.500	.000			1.013	.968	.898	.835	.809	.812	460	.600	.000			.937	.988	1.061	1.114	1.160	1.204				
421	.500	.100			.137	.111	-.007	-.122	-.240	-.322	461	.600	.100			.200	.149	.196	.237	.269	.324				
422	.500	.200			.319	.240	.273	.312	.281	.113	462	.600	.200			.218	.184	.101	.124	.183	.333				
423	.500	.300			.145	.185	.235	.256	.275	.298	463	.600	.300			.274	.253	.177	.185	.304	.409				
424	.500	.400			.308	.326	.348	.345	.203	.189	464	.600	.400			.340	.345	.335	.353	.411	.479				
425	.500	.500			.438	.439	.444	.434	.372	.217	465	.600	.500			.432	.440	.455	.469	.504	.553				
426	.500	.600			.467	.472	.472	.463	.431	.354	466	.600	.600			.471	.478	.487	.497	.530	.572				
427	.500	.700			.278	.274	.263	.255	.241	.207	467	.600	.700			.276	.281	.297	.317	.355	.405				
428	.500	.800			.061	.056	.041	.032	.030	.000	468	.600	.800			.059	.050	.074	.097	.132	.186				
429	.500	.900			-.079	-.083	-.093	-.106	-.100	-.142	469	.600	.900			-.085	-.087	-.076	-.055	-.022	.030				
430	.830	.000			1.005	.960	.872	.800	.755	.737	470	.800	.000			.962	.995	1.067	1.121	1.169	1.214				
431	.830	.200									471	.800	.200			.222	.182	.107	.131	.185	.354				
432	.830	.400			.301	.327	.337	.336	.271	.173	472	.800	.400			.340	.340	.350	.370	.407	.470				
433	.830	.600			.353	.349	.343	.329	.312	.259	473	.800	.600			.387	.394	.409	.426	.461	.509				
434	.830	.800			.201	.192																			

TABLE V.- PRESSURE COEFFICIENTS MEASURED ON VERTICAL TAIL AND SPEED BRAKES - Concluded

(e) $M = 1.18$ - ConcludedPressure coefficients C_p on -

Model orifice number	$\frac{y}{b/2}$	$\frac{z}{c}$	Upper tail								Model orifice number	$\frac{y}{b/2}$	$\frac{z}{c}$	Lower tail														
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$				$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$							
			$\beta = 5^\circ; \delta_e = 0^\circ; \delta_y = 0^\circ; \text{brakes closed}$																									
			$\beta = 5^\circ; \delta_e = 0^\circ; \delta_y = 0^\circ; \text{brakes closed}$																									
401	.110	.647	.033	-.014		.057	.075	.048	-.066	-.244	441	.130	.647	-.264	-.023		.023	.014	.015	.061	.123							
402	.110	.681	.131	.047		.119	.151	.129	-.037	-.210	442	.130	.682	-.260	.101		.031	.029	.047	.123	.358							
403	.110	.698	.162	.072		.155	.160	.118	-.021	-.216	443	.130	.700	-.280	.101		.068	.064	.082	.175	.357							
404	.110	.748	.118	.042							444	.130	.748	-.276	.062		.077	.075	.104	.186	.314							
405	.110	.832	.061	.005		.036	.037	.008	-.129	-.307	445	.130	.832	-.328	.012		.029	.040	.090	.180	.282							
406	.110	.949	-.026	-.059		-.003	-.020	-.066	-.208	-.401	446	.130	.949	-.378	-.063		-.023	-.002	.066	.163	.219							
410	.250	.000	1.192	1.138		1.034	.875	.447	.381	.931	450	.300	.000	.843	.692		1.046	1.105	1.142	1.177	1.210							
411	.250	.200	.390	.349		.414	.387	.212	.267	.149	451	.300	.200	.231	.297		.385	.363	.347	.362	.410							
412	.250	.400	.216	.183		.282	.303	.235	.224	-.048	452	.300	.400	.014	.226		.239	.223	.217	.260	.360							
413	.250	.600	.171	.092		.169	.182	.138	.057	-.182	453	.300	.600	-.168	.128		.123	.117	.128	.198	.300							
414	.250	.800	.026	-.032							454	.300	.800	.633	.247	.005		.013	.008	.016	.066	.136						
415	.250	.668	.160	.050		.102	.113	.136	-.010	-.211	455	.300	.668	-.240	.054		.042	.043	.062	.140	.358							
416	.250	.685	.169	.081		.135	.153	.124	-.011	-.216	456	.300	.685	-.222	.114		.077	.084	.106	.181	.354							
417	.250	.738	.150	.069		.117	.112	.083	-.022	-.244	457	.300	.738	-.252	.092		.074	.077	.107	.195	.328							
418	.250	.825	.108	.043		.069	.067	.030	-.070	-.279	458	.300	.825	-.290	.045		.042	.058	.106	.199	.311							
419	.250	.947	.048	.006		.024	.011	-.043	-.144	-.341	459	.300	.947	-.325	-.019		.002	.025	.092	.189	.252							
420	.500	.000				.992	.909	.773	.839	.838	460	.600	.000	.840	.852		1.033	1.093	1.141	1.187	1.227							
421	.500	.100	.495	.439		.455	.427	.357	.398	.392	461	.600	.100	.386	.365		.432	.427	.419	.434	.486							
422	.500	.200	.343	.305		.380	.376	.311	.346	.244	462	.600	.200	.246	.278		.304	.294	.285	.308	.387							
423	.500	.300	.249	.210		.296	.304	.264	.271	.137	463	.600	.300	.139	.210		.238	.225	.224	.262	.363							
424	.500	.400	.204	.159		.233	.249	.233	.191	.055	464	.600	.400	.052	.168		.180	.177	.187	.248	.350							
425	.500	.500	.203	.127		.187	.200	.191	.118	-.014	465	.600	.500	-.019	.143		.167	.167	.167	.231	.337							
426	.500	.600	.193	.103		.149	.153	.134	.061	-.143	466	.600	.600	.000	.116		.107	.110	.131	.199	.310							
427	.500	.700				.113	.117	.100	.002	-.143	467	.600	.700	-.148	.103		.078	.087	.105	.184	.333							
428	.500	.800	.141	.061		.077	.076	.052	-.032	-.179	468	.600	.800	-.182	.066		.058	.073	.117	.201	.320							
429	.500	.900	.111	.059		.053	.048	.017	-.058	-.207	469	.600	.900	-.211	.013		.024	.048	.119	.203	.296							
430	.830	.000	1.198	1.100		.930	.844	.756	.737	.743	470	.800	.000	.832	.851		1.010	1.078	1.130	1.178	1.222							
431	.830	.200	.297	.129		.236	.230	.199	.219	.202	471	.800	.200	.189	.179		.217	.212	.231	.275	.372							
432	.830	.400	.237	.129		.134	.132	.131	.081	.057	472	.800	.400	.037	.114		.132	.145	.174	.244	.347							
433	.830	.600	.224	.109		.092	.081	.073	.008	-.051	473	.800	.600	-.068	.101		.081	.096	.126	.203	.323							
434	.830	.800	.182	.075		.068	.059	.037	-.016	-.090	474	.800	.800	-.120	.043		.042	.063	.115	.204	.328							
												$\beta = 5^\circ; \delta_e = 0^\circ; \delta_y = 0^\circ; \text{brakes open}$																
401	.110	.647				.835	.857	.878	.528	.709	.334	441	.130	.647			.899	.878	.859	.825	.869	.909						
402	.110	.681				1.092	1.122	1.095	.593	.901	.571	442	.130	.682			1.225	1.209	1.178	1.138	1.132	1.144						
403	.110	.698				1.138	1.141	1.015	.626	.834	.670	443	.130	.700			1.133	1.145	1.150	1.157	1.150	1.148						
404	.110	.748										444	.130	.748			.945	.941	.932	.929	.931	.941						
405	.110	.832				.784	.808	.832	.817	.543	-.081	445	.130	.832			.862	.841	.815	.796	.795	.806						
406	.110	.949				.654	.672	.710	.696	.418	-.351	446	.130	.949			.710	.690	.668	.650	.649	.657						
410	.250	.000				1.080	1.037	.888	.461	.435	.946	450	.300	.000			.997	1.046	1.104	1.143	1.177	1.210						
411	.250	.200				.423	.441	.418	.233	.298	.157	451	.300	.200			.415	.409	.380	.365	.390	.478						
412	.250	.400				.399	.415	.430	.380	.286	-.009	452	.300	.400			.396	.383	.367	.386	.454	.540						
413	.250	.600				.776	.783	.754	.502	.605	.340	453	.300	.600			.767	.768	.765	.774	.801	.836						
414	.250	.800										454	.300	.800			.831	.810	.787	.761	.805	.867						
415	.250	.668				1.027	1.049	1.136	.596	1.033	.843	455	.300	.668			1.161	1.146	1.127	1.077	1.120	1.146						
416	.250	.685				1.083	1.079	1.048	.622	.905	.737	456	.300	.685			1.073	1.084	1.100	1.114	1.115	1.113						
417	.250	.738				.829	.826	.840	.684	.726	.505	457	.300	.738			.948	.945	.845	.856	.867	.889						
418	.250	.825				.686	.693	.736	.745	.646	.168	458	.300	.825			.741	.725	.717	.718	.733	.760						
419	.250	.947				.561	.564	.595	.616	.538	-.163	459	.300	.947			.588	.582	.578	.580	.595	.620						
420	.500	.000				1.046	.997	.916	.753	.849	.881	460	.600	.000			.980	1.024	1.090	1.140	1.184	1.226						
421	.500	.100				.462	.474	.452	.367	.423	.405	461	.600	.100			.446	.443	.434	.427	.452	.526						
422	.500	.200				.391	.408	.413	.344	.369	.285	462	.600	.200			.345	.332	.311	.303	.352	.465						
423	.500	.300				.347	.357	.376	.361	.304	.170	463	.600	.300			.320	.303	.279	.295	.382	.491						
424	.500	.400				.387	.396	.393	.358	.271	.088	464	.600	.400			.373	.366	.360	.389	.453	.536						
425	.500	.500				.475	.477	.462	.388	.334	.036	465	.600	.500			.475	.472	.473	.493	.539	.599						
426	.500	.600				.495	.485	.484	.362	.298	.205	466	.600	.600			.493	.493	.493	.503	.524	.568						
427	.500	.700				.269	.258	.268	.168	.118	.067	467	.600	.700			.258	.266	.268	.320	.366	.423						
428	.500	.800				.041	.041	.033	.029	.007	-.053	468	.600	.800			.035	.040	.061	.094	.139	.197						
429	.500	.900				-.090	-.103	-.094	-.083	-.129	-.219	469	.600	.900			-.103	-.100	-.084	-.054	-.008	.040						
430	.830	.000				.993	.932	.842	.741	.710	.718	470	.800	.000			.965	1.008	1.077	1.131	1.177	1.220						
431	.830	.200				.265	.266	.276	.249	.237	.230	471	.800	.200			.259	.247	.231	.253	.328	.466						
432	.830	.400				.288	.288	.264	.255	.212	.146	472	.800	.400			.338	.322	.329	.370	.438	.524						

TABLE VI.- PRESSURE COEFFICIENT MEASURED AT THE BASE OF THE UPPER VERTICAL TAIL

(a) $M = 0.60$ Pressure coefficients C_p on -

Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Upper tail							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
600	.150	1.000				-.323	-.333	-.317	-.360	-.414
			$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
600	.150	1.000	-.257	-.309	-.604	-.608	-.600	-.537	-.576	-.600
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
600	.150	1.000	-.239	-.284		-.271	-.291	-.313	-.320	-.390
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
600	.150	1.000			-.298	-.300	-.302	-.318	-.365	-.448
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = -7.5^\circ; \text{brakes closed}$							
600	.150	1.000			-.288	-.287	-.287	-.308	-.359	-.436
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 7.5^\circ; \text{brakes closed}$							
600	.150	1.000			-.560	-.564	-.568	-.523	-.567	-.681
			$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
600	.150	1.000	-.262	-.318		-.345	-.335	-.335	-.392	-.445
			$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
600	.150	1.000			-.625	-.621	-.612	-.536	-.584	-.610

TABLE VI.- PRESSURE COEFFICIENT MEASURED AT THE BASE OF THE UPPER VERTICAL TAIL - Continued

(b) $M = 0.90$ Pressure coefficients C_p on -

Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Upper tail							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
600	.150	1.000				-.365	-.374	-.355	-.409	-.500
			$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
600	.150	1.000	-.331	-.357	-.666	-.659	-.659	-.641	-.690	-.795
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
600	.150	1.000	-.305	-.297		-.293	-.304	-.337	-.372	
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
600	.150	1.000			-.385	-.383	-.388	-.409	-.454	-.568
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = -7.5^\circ; \text{brakes closed}$							
600	.150	1.000			-.355	-.359	-.362	-.375	-.406	-.502
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 7.5^\circ; \text{brakes closed}$							
600	.150	1.000			-.669	-.654	-.670	-.676	-.715	-.791
			$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
600	.150	1.000	-.345	-.356		-.375	-.383	-.371	-.417	-.543
			$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
600	.150	1.000			-.665	-.655	-.651	-.635	-.689	-.815

TABLE VI.- PRESSURE COEFFICIENT MEASURED AT THE BASE OF THE UPPER VERTICAL TAIL - Continued

(c) $M = 0.95$ Pressure coefficients C_p on -

Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Upper tail							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
600	.150	1.000				-.401	-.404	-.378	-.386	-.487
			$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
600	.150	1.000	-.369	-.388	-.744	-.745	-.758	-.704	-.726	-.814
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
600	.150	1.000	-.330	-.324		-.330	-.336	-.349	-.377	-.459
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
600	.150	1.000			-.450	-.449	-.442	-.445	-.471	-.558
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = -7.5^\circ; \text{brakes closed}$							
600	.150	1.000			-.420	-.414	-.399	-.400	-.427	-.509
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 7.5^\circ; \text{brakes closed}$							
600	.150	1.000			-.756	-.746	-.757	-.769	-.746	-.767
			$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
600	.150	1.000	-.378	-.393		-.426	-.422	-.389	-.413	-.531
			$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
600	.150	1.000			-.772	-.774	-.780	-.711	-.743	-.828

TABLE VI.- PRESSURE COEFFICIENT MEASURED AT THE BASE OF THE UPPER VERTICAL TAIL - Continued

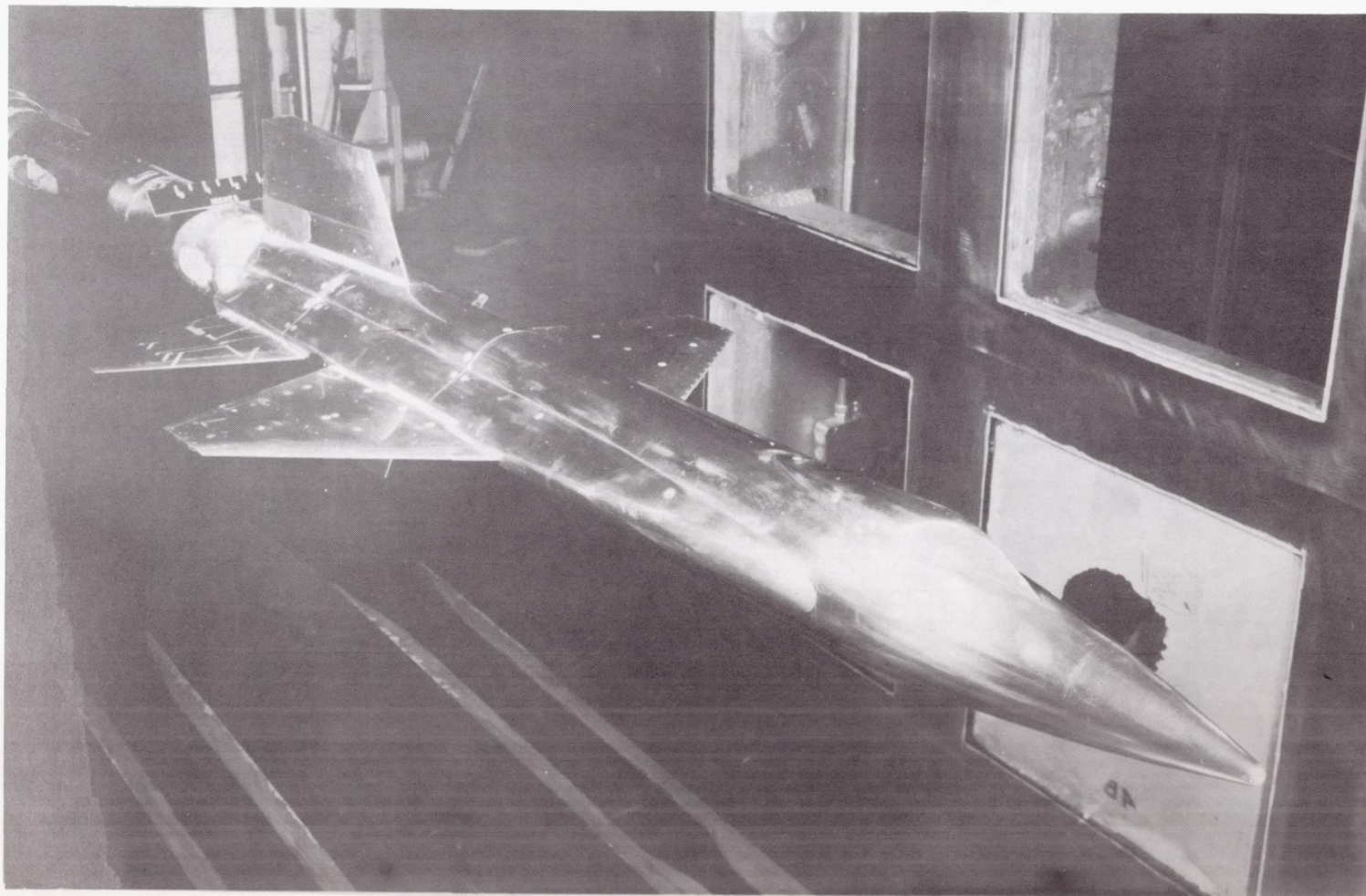
(d) $M = 1.03$ Pressure coefficients C_p on -

Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Upper tail							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -5^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
600	.150	1.000				-.429	-.458	-.515	-.565	-.623
			$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
600	.150	1.000	-.450	-.480	-.726	-.729	-.746	-.788	-.827	-.854
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
600	.150	1.000	-.419	-.439		-.376	-.425	-.489	-.544	-.574
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
600	.150	1.000			-.505	-.498	-.531	-.598	-.658	-.690
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = -7.5^\circ; \text{brakes closed}$							
600	.150	1.000			-.508	-.494	-.525	-.584	-.644	-.681
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 7.5^\circ; \text{brakes closed}$							
600	.150	1.000			-.715	-.719	-.744	-.791	-.812	-.838
			$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
600	.150	1.000	-.465	-.480		-.444	-.484	-.550	-.605	-.672
			$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
600	.150	1.000			-.714	-.734	-.757	-.781	-.828	-.861

TABLE VI.- PRESSURE COEFFICIENT MEASURED AT THE BASE OF THE UPPER VERTICAL TAIL - Concluded

(e) $M = 1.18$ Pressure coefficients C_p on -

Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Upper tail							
			$\alpha = -20^\circ$	$\alpha = -10^\circ$	$\alpha = -3^\circ$	$\alpha = 0^\circ$	$\alpha = 5^\circ$	$\alpha = 10^\circ$	$\alpha = 15^\circ$	$\alpha = 20^\circ$
			$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
600	.150	1.000				-.372	-.342	-.381	-.422	-.457
			$\beta \approx -5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
600	.150	1.000	-.347	-.380	-.573	-.572	-.586	-.591	-.632	-.628
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
600	.150	1.000	-.276	-.341		-.337	-.333	-.365	-.417	-.420
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
600	.150	1.000			-.449	-.442	-.432	-.462	-.504	-.549
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = -7.5^\circ; \text{brakes closed}$							
600	.150	1.000			-.441	-.438	-.430	-.446	-.492	-.531
			$\beta \approx 0^\circ; \delta_e = 0^\circ; \delta_v = 7.5^\circ; \text{brakes closed}$							
600	.150	1.000			-.586	-.574	-.582	-.600	-.613	-.641
			$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes closed}$							
600	.150	1.000	-.348	-.376		-.377	-.356	-.404	-.449	-.481
			$\beta \approx 5^\circ; \delta_e = 0^\circ; \delta_v = 0^\circ; \text{brakes open}$							
600	.150	1.000			-.586	-.586	-.592	-.601	-.611	-.618



L-58-2647

Figure 1.- Photographs of the 0.0667-scale pressure model of the X-15 airplane as tested in the Langley 8-foot transonic pressure tunnel.

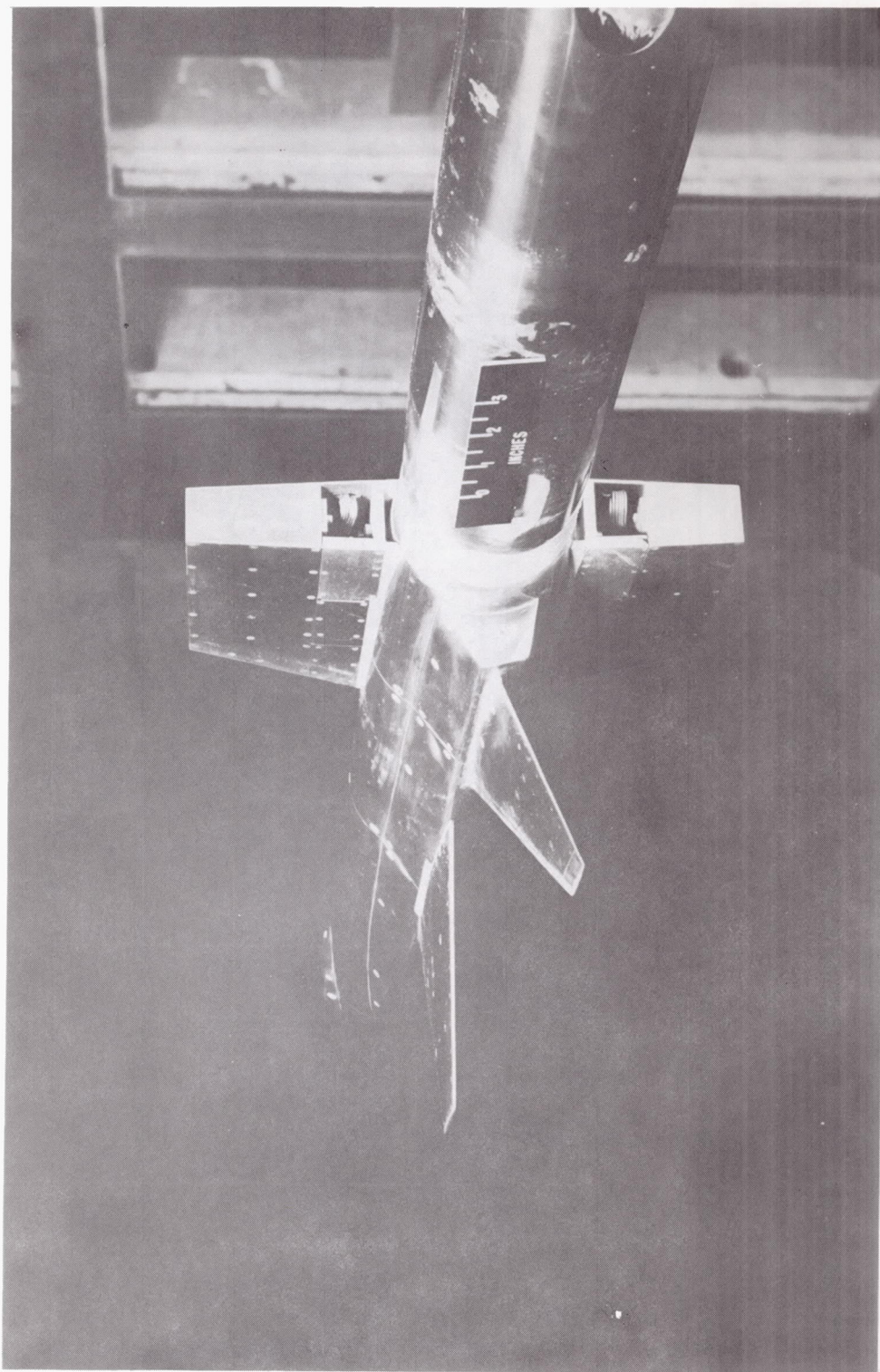


Figure 1.- Concluded.

L-58-2649

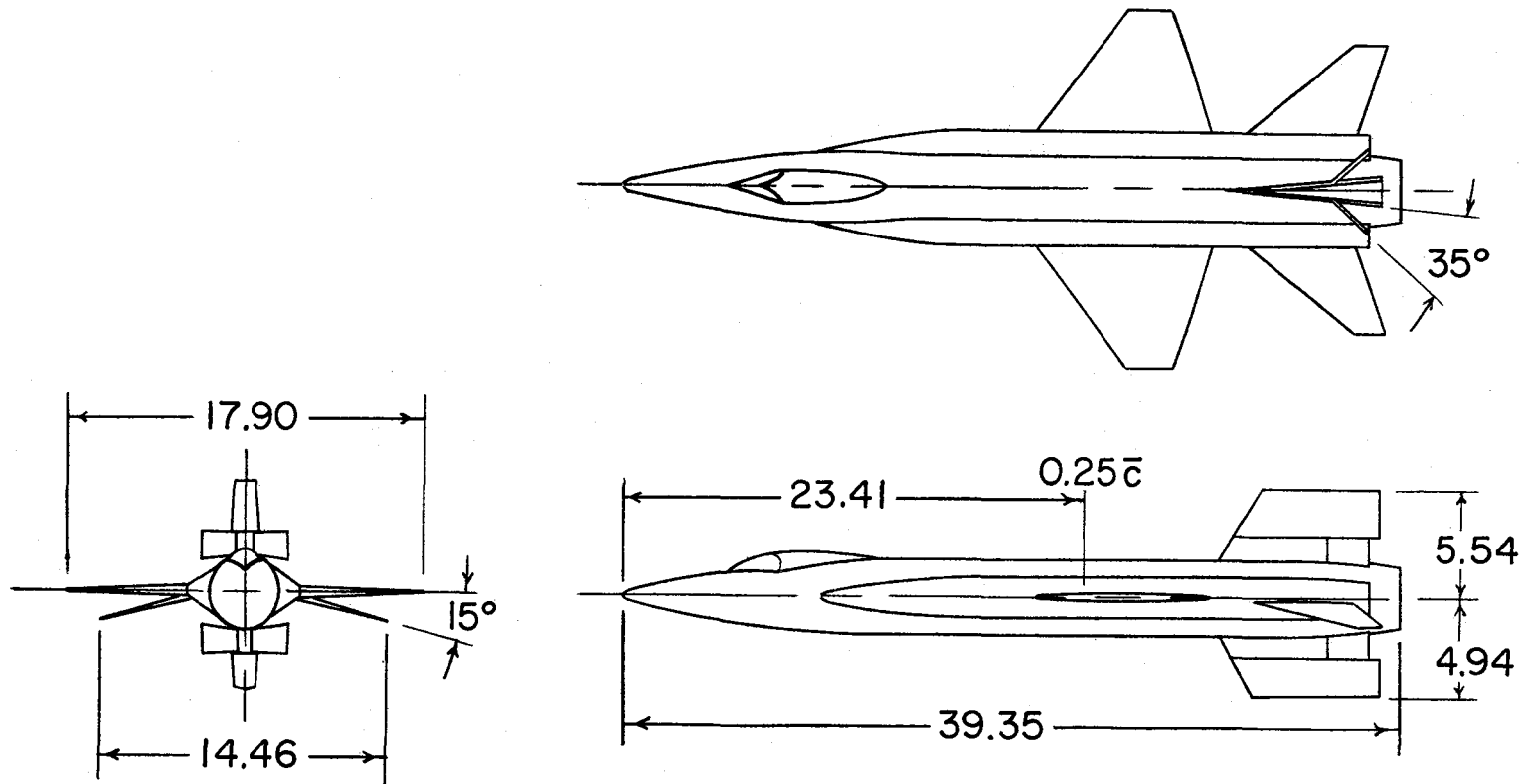
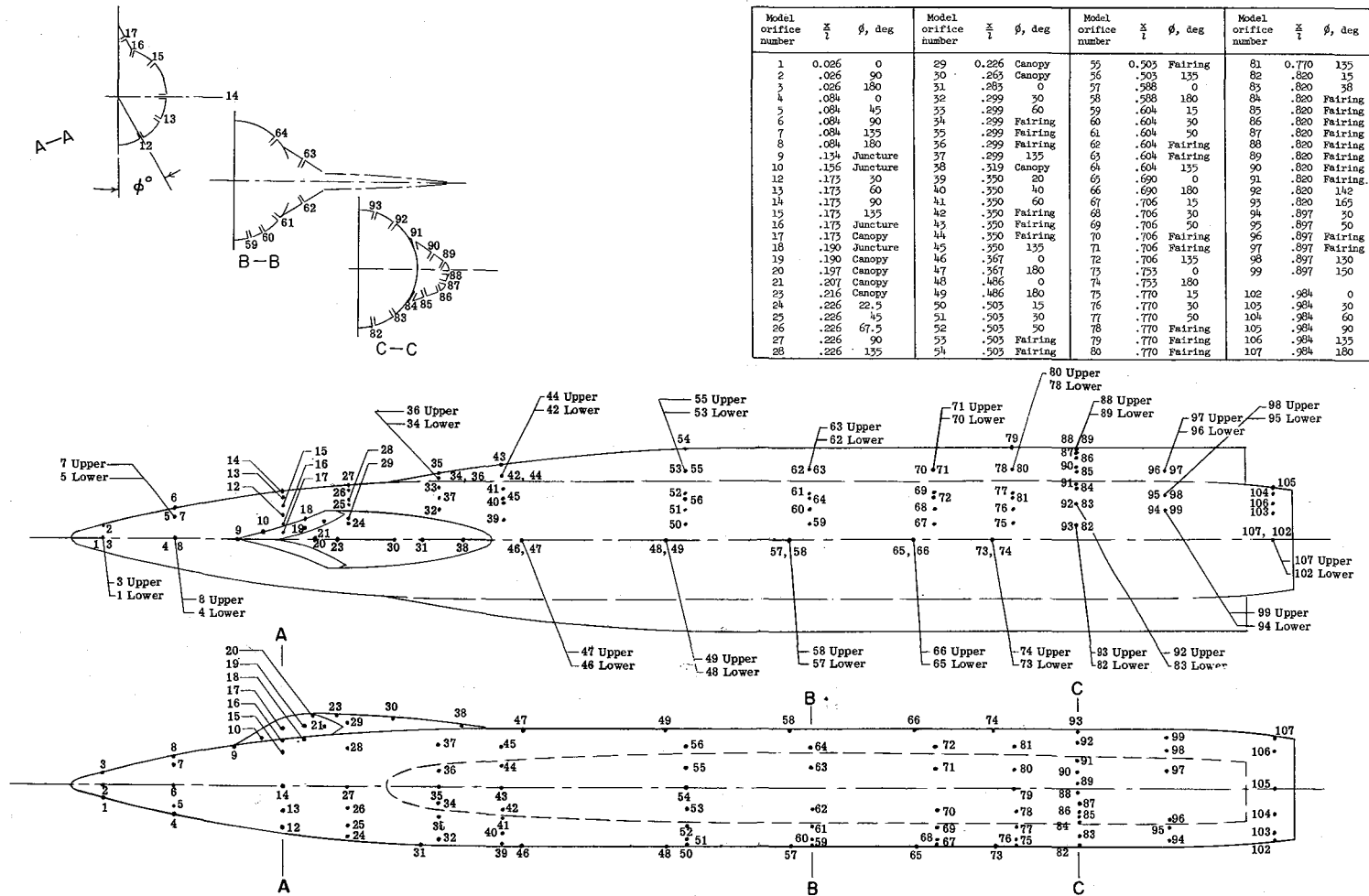


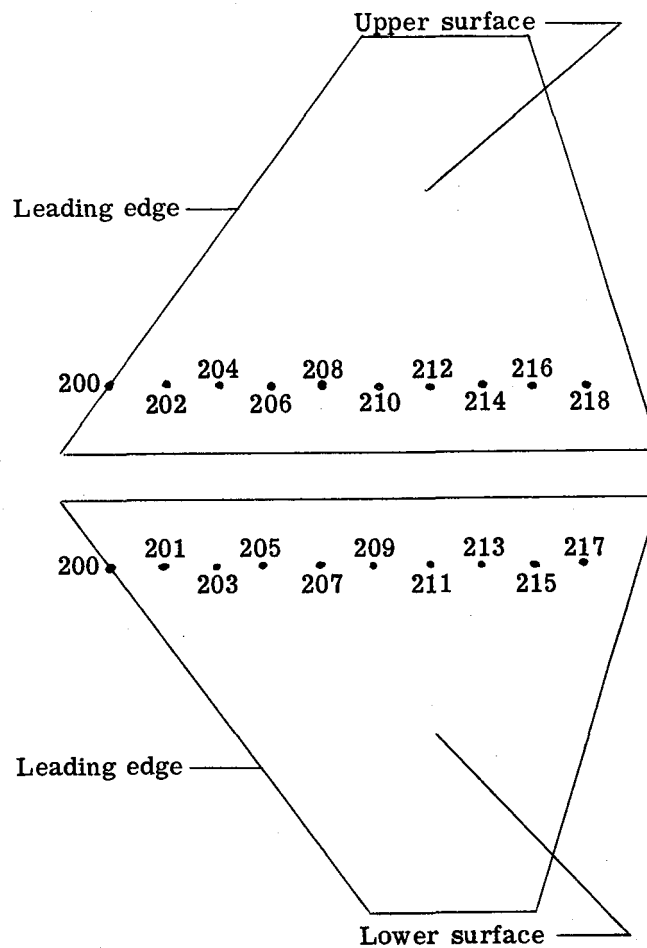
Figure 2.- Drawing of the 0.0667-scale pressure model of the X-15 with speed brakes open. All dimensions are in inches unless otherwise noted.



(a) Orifices on fuselage.

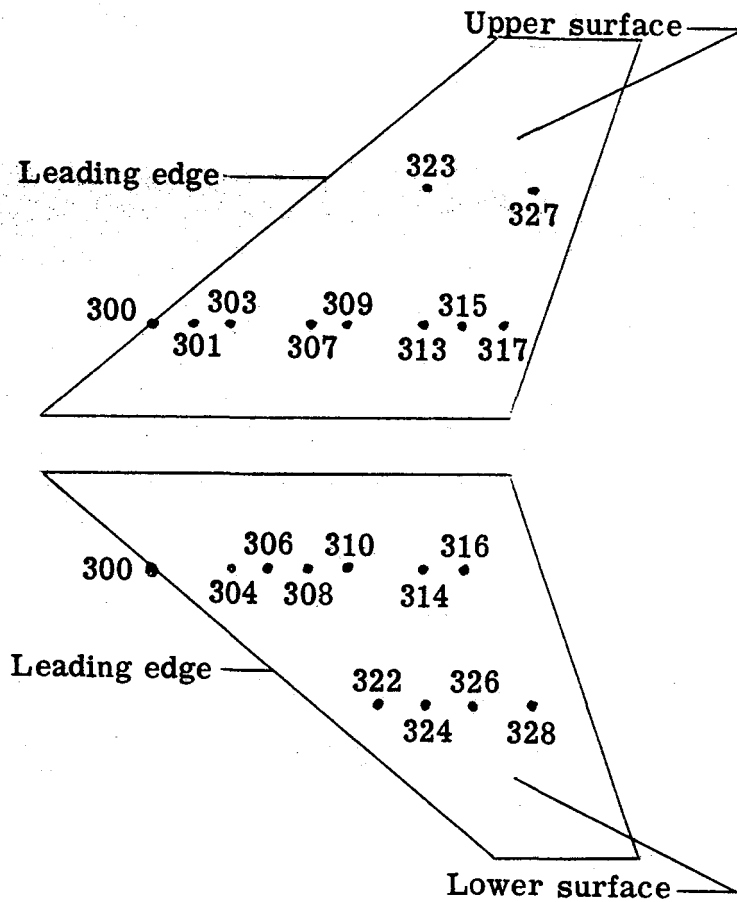
Figure 3.- Location of orifices on model.

Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$
200	0.18 ↓	0
201		.10
202		.10
203		.20
204		.20
205		.30
206		.30
207		.40
208		.40
209		.50
210		.50
211		.60
212		.60
213		.70
214		.70
215		.80
216		.80
217		.90
218		.90



(b) Orifices on right wing.

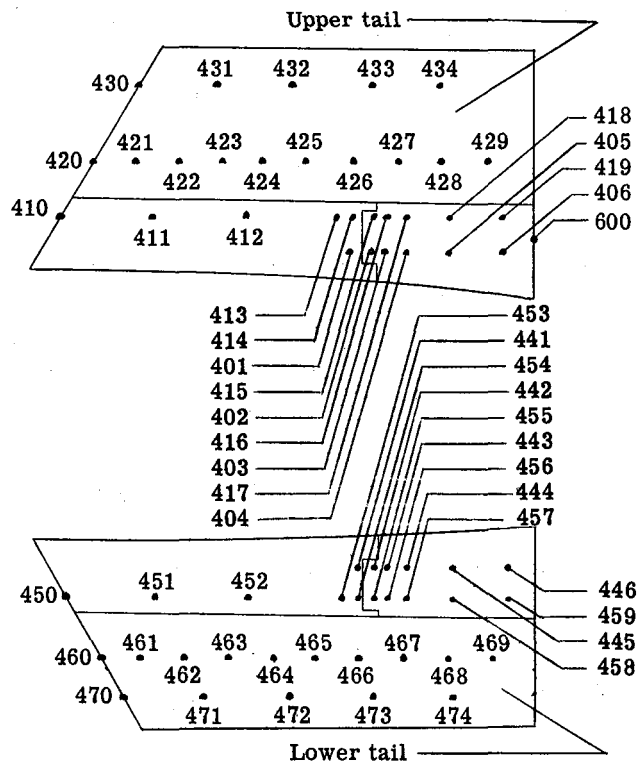
Figure 3.- Continued.



Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$
300	0.26	0
301		.10
303		.20
304		.20
306		.30
307		.40
308		.40
309		.50
310		.50
313		.70
314		.70
315		.80
316		.80
317		.90
322	.62	.20
323		.40
324		.40
326		.60
327		.80
328		.80

(c) Orifices on right horizontal tail.

Figure 3.- Continued.



Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$	Model orifice number	$\frac{y}{b/2}$	$\frac{x}{c}$
401	0.11	0.647	425	0.50	0.500	455	0.30	0.668
402	.11	.681	426	.50	.600	456	.30	.686
403	.11	.698	427	.50	.700	457	.30	.738
404	.11	.748	428	.50	.800	458	.30	.825
405	.11	.832	429	.50	.900	459	.30	.947
406	.11	.949	430	.83	0	460	.60	0
410	.25	0	431	.83	.200	461	.60	.100
411	.25	.200	432	.83	.400	462	.60	.200
412	.25	.400	433	.83	.600	463	.60	.300
413	.25	.600	434	.83	.800	464	.60	.400
414	.25	.630	441	.13	.647	465	.60	.500
415	.25	.668	442	.13	.682	466	.60	.600
416	.25	.685	443	.13	.700	467	.60	.700
417	.25	.738	444	.13	.748	468	.60	.800
418	.25	.825	445	.13	.832	469	.60	.900
419	.25	.947	446	.13	.949	470	.80	0
420	.50	0	450	.30	0	471	.80	.200
421	.50	.100	451	.30	.200	472	.80	.400
422	.50	.200	452	.30	.400	473	.80	.600
423	.50	.300	453	.30	.600	474	.80	.800
424	.50	.400	454	.30	.633	600	.15	1.000

(d) Orifices on right vertical tail including speed brakes.

Figure 3.- Concluded.

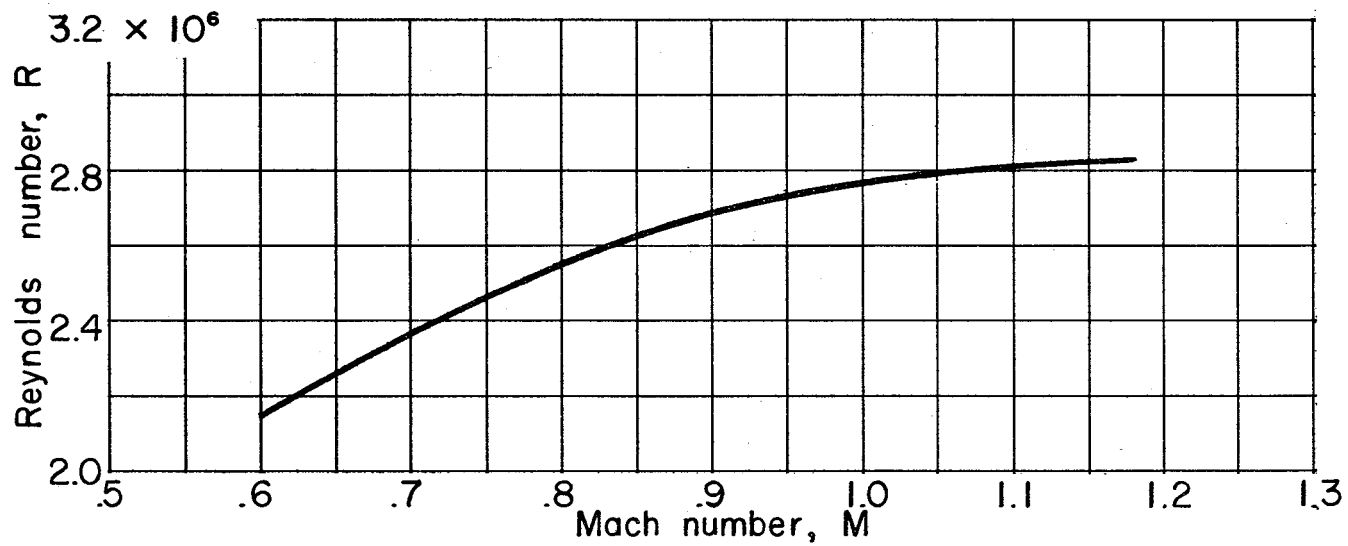


Figure 4.- Variation with Mach number of test Reynolds number based on wing $\bar{c} = 8.22$ inches.